## CT7-2E TEMPORARY REVISION - 05-0017 MM 05-11-03, SERVICE LIFE LIMITS 004 GAS GENERATOR TURBINE ROTOR PARTS - SERVICE LIFE LIMITS

Dated: 10/03/2023

SUBJECT:GAS GENERATOR TURBINE ROTOR PARTS - SERVICE LIFE LIMITSREASON:This Temporary Revision (TR) adds the data from TR 05-0017 which changes the Life<br/>Limits for Stage 2 Turbine Aft Cooling Plate PN 5166T27P01.FILING INSTRUCTIONS:Put this TR in the CT7-2E Maintenance Manual, GEK 112043-02, replacing 05-11-03,<br/>SERVICE LIFE LIMITS 004, dated May 31/2017. Record this TR in the Record of<br/>Temporary Revisions. Update your media via the GE Aviation Web portal.

## GE Designated: -CONFIDENTIAL-

The information contained in this document is GE proprietary information and is disclosed in confidence. It is the property of GE and shall not be used, disclosed to others or reproduced without the express written consent of GE, including, but without limitation, it is not to be used in the creation, manufacture, development, or derivation of any repairs, modifications, spare parts, designs, or configuration changes or to obtain FAA or any other government or regulatory approval to do so. If consent is given for reproduction in whole or in part, this notice and the notice set forth on each page of this document shall appear in any such reproduction in whole or part.

This technical data is considered subject to the Export Administration Regulations (EAR) pursuant to 15 CFR Parts 730-774. Transfer of this data by any means to a Non-U.S. Person, whether in the United States or abroad, without the proper U.S. Government authorization (e.g., License, exemption, NLR, etc.), is strictly prohibited.

Copyright (2023) General Electric Company, U.S.A.

## \* \* \* FOR CT7-2E1

2.

TASK 05-11-03-200-807

1. <u>General Information.</u>

This section contains the FAA approved service life limits for the gas generator turbine rotor parts. The remaining service life (in cycles) is to be determined by subtracting the accumulated total flight cycles from the life limit cycles stated for each part listed in paragraph 2. Refer to 05-11-00, SERVICE LIFE LIMITS 004 for the definition and calculation of cycle count. The operator is responsible to be sure that the cycle limited parts are not used beyond their cycle limit. Service Life Limits for Gas Generator Turbine Rotor Parts.

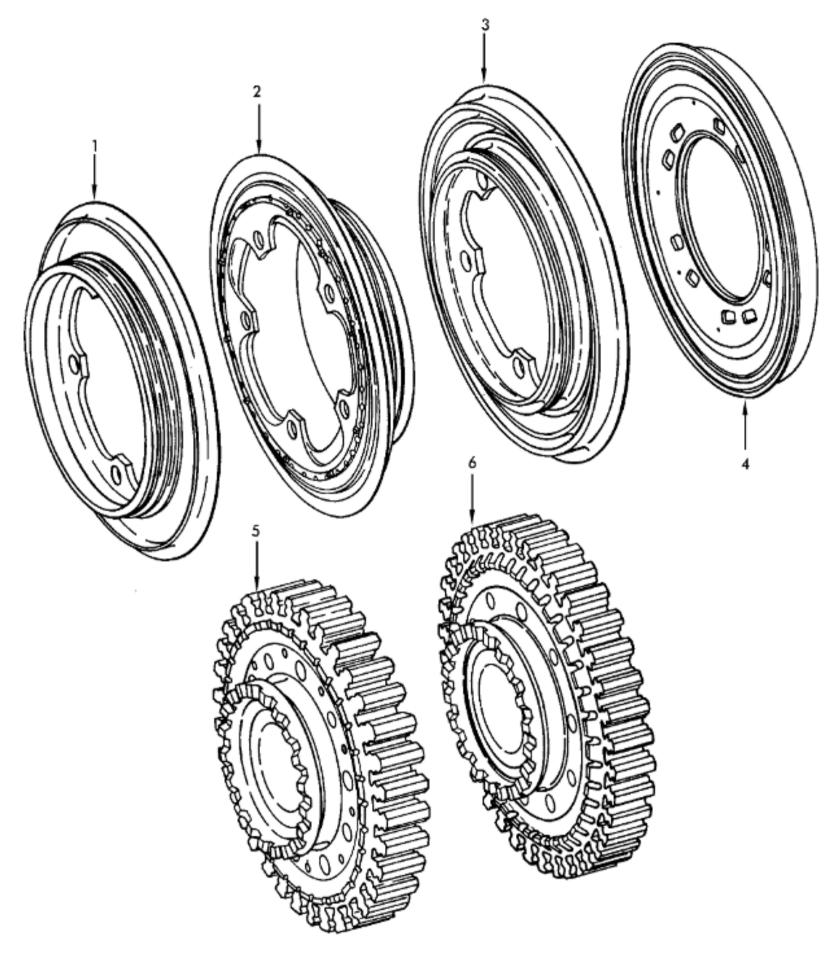
## Refer to Figure 1.

These life limits have been established assuming CT7-2E1 mission profiles representative of an AW189 Helicopter Application and an aircraft transmission limit of 355.5 lb-ft for All Engine Operation.

Figure Item Number	Nomenclature	Part Number	Life Limits (Cycles Since New)
1	Stage 1 Turbine Forward Cooling Plate	4185T23P01	7,500
2	Stage 1 Turbine Aft Cooling Plate	4185T24P01	7,500
3	Stage 2 Turbine Forward Cooling Plate	4106T80P03	7,500
4	Stage 2 Turbine Aft Cooling Plate	5166T27P01	3,100
5	Stage 1 Turbine Disk	4185T13P01	7,500
б	Stage 2 Turbine Disk	4185T14P01	6,800

\* \* \* FOR CT7-2E1

Date Printed: 2023/10/04



1179286-00

Figure 1 Gas Generator Turbine Rotor Parts (Typical) - Service Life Limits

GE Designated: - CONFIDENTIAL Subject to the restrictions on the media

GE PROPRIETARY INFORMATION - Not to be used, disclosed to others or reproduced without the express written consent of GE. Technical data is considered ITAR and/or EAR controlled; transfer of this data to a Non-US Person, without USG authorization, is strictly prohibited.