

No	Description	Current (Revision 010: January 21 2021)	Previous (Revision 009: Sept 16 2021)	Remarks
1	3.1 Section INTRO - USE OF THE MSM Paragraph 5.2.2 modified.	Calendar Time Calendar time can be expressed: in years (Y), in months (M), in days (D), from the date written on the label of the equipment item (ED: Expiry Date). 5.2.1 Airframe initial aging date The helicopter airframe initial aging date to be taken into account is the date marked on the helicopter identification plate. 5.2.2 Initial aging date for a component Unless specified differently, the start date (T0) to be taken into account for the aging of components and dynamic assemblies is the date of first installation or first depreservation to be recorded on the Log Card for: new components or dynamic assemblies (with OTL or TBO) not new components or dynamic assemblies (with TBO) after accomplishment of an overhaul In this case, the MSM Tasks do not contain any specification. The storage time before installation is therefore not to be taken into account on the condition that storage was in accordance with the applicable documentation, except for the following specific cases. For some components, the aging date to be taken into account is the date of manufacture marked on the Log Card (FM). In this case, TSM (Time Since Manufacture) is specified in each task concerned in the MSM. Specific cases:	The initial aging date for a component The initial aging date for a component and a dynamic component to be taken into account is generally the date of first installation or first destorage, to be recorded on the equipment log card. In this case, there are no other specifications in the MSM tasks. The storage time before installation is therefore not to be taken into account on the condition that storage was in accordance with the applicable documentation. For some components, the aging date to be taken into account is the date of manufacture marked on the Log Card (FM). In this case, TSM (Time Since Manufacture) is specified in each task concerned in the MSM. Particular cases: Components or dynamic components are sometimes subject to a limit since installation and a limit since manufacture. In this case, both limits followed by TSI (Time Since Installation) and TSM (Time Since Manufacture) are indicated, as per the following example:	For info only
2	3.2 Section SB INCORPORATED - SB INCORPORATED Modified (added)	05-035 R1 Reduction of the TBO of the deployment control 25-66 unit of the CPI 503 beacon (HR SMITH) 26-004 R0 FIRE PROTECTION - Obsolescence of cockpit 26-22 2021.01.22 fire extinguisher 63A013 R1 MAIN ROTOR DRIVE - Tail rotor drive flange 63-24 2021.01.22 Installation of rear output stop		For info only
3	3.3 Section 05.10.00 25-69 OPT - LIFE RAFTS 25/69/06/000/000/005 Life raft MP/N modified 12 Y	Life raft CMM 25.64.47 245431-0 (704A42891043) OTL 12 Y TSM 0 245431-1 (704A42891050) 245434-1 (704A42891051) Tropical and damp atmosphere Type SR10. You can apply for an extension by returning the assembly in question to the manufacturer for examination.	25/69/06/000/000/000005 Life raft CMM 25.64.47 245431-0 (704A42691043) OTL 12 Y TSM 0 245431-1 (704A42691061) 245434-1 (704A42691060) Tropical and damp atmosphere Type SR10. You can apply for an extension by returning the assembly in question to the manufacturer for examination.	For info only. No life raft installed.



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4	3.3 Section 05.10.00 25-69 OPT - LIFE RAFTS 25/69/06/000/000/015 Life raft MP/N modified 15 Y	25/69/06/000/0000015 Life raft CMM 25.64.47	Life raft CMM 25.64.47 245431-0 (704A42891043) OTL 15 Y TSM 0 245431-1 (704A42891081) 245431-1 (704A42891081) Type SR10. You can apply for an extension by returning the assembly in question to the manufacturer for examination.	For info only. No life raft installed.
5	3.3 Section 05.10.00 45-11 OPT - CVFDR 45/11/01/000/000/030 CVFDR recorder Description modified MP/N modified OC	45/11/01/000/000/030 CVFDR recorder 980-6021-032 (704A45381025) OC 980-6021-066 (704A45381029) HONEYWELL AR204C / AR602C. Fitted with the DUKANE DK100 or DK120 or DATASONIC ELP 362D emergency locator transmitter.	45/11/01/000/000/030 CVFDR recorder 980-6021-032 (704A45381025) OC 980-6021-066 (-) HONEYWELL AR204C. Fitted with the DUKANE DK100 or DK120 or DATASONIC ELP 362D emergency locator transmitter.	For info only. 9M-SAS's CV/FDR (AR-602C) 980-6021-032
6	Section 05.20.00 28-20 FUEL DISTRIBUTION 28/20/00/211/000/010 Transfer pump Description modified Documentation modified 2400 FH	28/20/00/211/000/010 Transfer pump CMM 28.19.03 2400 FH 240 FH 28-19-03-220-007-A01 28-19-03-220-008-A01 Check the brushes and the armature.	28/20/00/211/000/010 Transfer pump <i>CMM 28.19.03</i> 2400 FH 240 FH Check the brushes and the commutator.	For info only.
7	Section 05.20.00 31-33 VIDEO RECORDING 31/33/00/520/211/100 Camera cockpit Task deleted 1 Y		31/33/00/520/211//100 Camera cockpit AMM 31.33.00.725 1 Y 36 D Vision 1000. Camera data recovery.	To update in Aeronet.



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9	Section 05.20.00 45-11 OPT – CVFDR 45/11/01/723/000/000 Immersion probe Description modified 600 FH // 2 Y Section 05.20.00 52-30 Cargo door - 52/30/00/520/221/000 Compartment and battery door locks New task integrated	45/11/01/723/000/000 Immersion probe AMM 45.11.01.723 600 FH 60 FH GRT 05 // 2 Y 73 D HONEYWELL AR204C / AR602C. Functional test. 52/30/00/520/221/000 Compartment and battery door locks AMM 52.00.00.221 600 FH 60 FH GRT 06 // 2 Y 73 D POST MOD 0752C68 Check.	45/11/01/723/000/000 Immersion probe AMM 45.11.01.723 600 FH 60 FH GRT 05 // 2 Y 73 D HONEYWELL AR204C. Functional test.	The 9M- SAS was not a POST MOD 0763C81
10	600 FH // 2 Y Section 05.20.00 63-24 TAIL ROTOR DRIVE POWER TAKE- OFF 63/24/01/221/000/005 MGB tail drive-shaft flange Periodicity modified 300 FH	63/24/01/221/000/005 MGB tail drive-shaft flange AMM 63.24.01.221 300 FH 30 FH 63.24.01.221.001 63.24.01.221.003 POST MOD 0763C81 Check axial and radial play of the flange.	63/24/01/221/000/005 MGB tail drive-shaft flange AMM 63.24.01.221 PO 300 FH 30 FH 63.24.01.221.001 63.24.01.221.003 POST MOD 0763C81 Check axial and radial play of the flange.	The 9M- SAS was a POST MOD 0763C81
11	Section 05.50.00 64-10 ROTOR BLADES - 64/10/00/221/001/032 Tail rotor blade Description modified Documentation modified	64/10/00/221/001/032 Tail rotor blade AMM 64.10.00.221A 64.10.00.221.003 64.10.00.221.004 365A12-0060-01 (-) Inspection. With blade removal. With TRH fairing removal. With TRH fairing removal. Note 1: If there is separation of blade shielding in zone A2 and A5: inspection every 100 FH or 1 Y, as long as the replacement criteria are not met (without blade removal). Note 2: If there is separation of blade shielding in zone C and E: inspection every 50 FH, 100 FH, 150 FH and 300 FH (without blade removal). If no change is noticed, resume the regular inspection cycle.	Tail rotor blade AMM 64.10.00.221A 64.10.00.221A 64.10.00.221.003 365A12-0060-01 (-) Inspection. With blade removal. With TRH fairing removal. Note 1: If bonding separation except in areas A2-A5, check every 50, 100, 150 and 300 FH (without blade removal). If no changes are noted, resume the standard inspection cycle. Note 2: If bonding separation in areas A2-A5 and/or cracks in all the areas of the shielding, check every 100 FH or 1 Y until the replacement criteria are reached (without blade removal).	This PN is not applicable since the Tail Rotor Blade of 9M-SAS PN is 365A12- 0070-01



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	Section 05.50.00 64/10/00/221/001/042 Tail rotor blade Description modified Documentation modified	64/10/00/221/001/042 Tail rotor blade AMM 64.10.00.221B 64.10.00.221.003 64.10.00.221.004 365A12-0070-00 (-) 365A12-0070-01 (-) Visual inspection. With blade removal. With TRH fairing removal. Note 1: If there is separation of blade shielding in zone A2 and A5: inspection every 100 FH or 1 Y, as long as the replacement criteria are not met (without blade removal). Note 2: If there is separation of blade shielding in zone C and E: inspection every 50 FH, 100 FH, 150 FH and 300 FH (without blade removal). If no change is noticed, resume the regular inspection cycle.	64/10/00/221/001/042 Tail rotor blade AMM 64.10.00.221B 64.10.00.221.003 365A12-0070-00 (-) 365A12-0070-01 (-) Visual inspection. With blade removal. With TRH fairing removal. Note 1: If bonding separation except in areas A2-A5, check every 50, 100, 150 and 300 FH (without blade removal). Should be no further deterioration, resume normal inspection cycle. Note 2: If bonding separation in areas A2-A5 and/or cracks in all the areas of the shielding, check every 100 FH or 1 Y until the replacement criteria are reached (without blade removal).	The Tail Rotor Blade of 9M-SAS PN is 365A12- 0070-01. To update the inspection hours.