

Aircraft Registration:	9M-HFA	Serial No:	1487	Date:	30/01/2023	Page No:	1
Aircraft Hours:	1911.40	Landings:	3352.00	Engine 1:	1911.40	Engine 2:	0.00

Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
<b>SBs</b>						
Types	SB Number	Subject	Last Done	Next Due	Interval	Limitation Notes
Repetitive SBs	319 73 4839 VERSION A	CHECK OF THE PREFERRED INJECTOR PIPE ASSY	-	-	0	MYCOPTER SERVICE BULLETIN STATUS ITEM #1 PAGE 1
Repetitive SBs	SB NO. EC120-05-014, REVISION 2	COLLECTIVE PITCH CONTROL - PERIODIC CHECK OF THE DRIVE TUBE/CONTROL PINION BONDED CONNECTION LOCATED ON THE LH/RH TWIST GRIPS	31/05/2022	19/07/2024	26	REFER TIC-EC120B-21-0015.  APPLICABLE. AIRCRAFT FITTED WITH LH AND RH TWIST GRIPS. REFER WORK REPORT NO. 354/2008.  COMPLIANCE: INITIAL - WITHIN 13 MONTHS FOLLOWING RECEIPT OF THIS SERVICE BULLETIN. REPEAT - EVERY 26 MONTHS. NOTE: THIS SB IS ONLY CANCELLED UPON COMPLIANCE OF SB EC120-76-010 (TIC NO. TIC-EC120B-21-0014)
Repetitive SBs	No. EC120-05A023	LIGHTING AND ANCILLARIES CONTROL UNIT (LACU) INSPECTION OF THE LACU	-	31/03/2023	0	REFER TIC-EC120B-21-0074, DATED 19.07.2021.  TO BE COMPLIED WITHIN 600 FH OR 24 MONTHS, WHICHEVER OCCURS FIRST. REPETITIVE INSPECTION AT EVERY 1320 FH.  REF: AD 2021-0168
Repetitive SBs	No. EC120-05A023	LIGHTING AND ANCILLARIES CONTROL UNIT (LACU) INSPECTION OF THE LACU	0.00	2122.40	1320	REFER TIC-EC120B-21-0074, DATED 19.07.2021.  TO BE COMPLIED WITHIN 600 FH OR 24 MONTHS, WHICHEVER OCCURS FIRST. REPETITIVE INSPECTION AT EVERY 1320 FH.  REF: AD 2021-0168
Repetitive SBs	EASB 05A020R2 (EASA AD 2021-0069) 1000FH	EASB 05A020R2 (EASA AD 2021-0069)	0.00	1960.80	1000	
Repetitive SBs	SIN No. 2740-S-25	CHECK OF THE KANNAD EMERGENCY LOCATOR TRANSMITTER ATTACHMENT STRAPS	31/05/2022	30/06/2022	1	LAST DONE @1910.10 FH ON 24.03.2022 REF: 9M-HFA-2720

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Repetitive SBs	EASB 05A020R2 (EASA AD 2021-0069)	TIME LIMITS - MAINTENANCE CHECKS - TAIL ROTOR HUB - CHECK OF THE TAIL ROTOR HUB BODY - INSPECTION OF THE HUB BODY	1906.40	1921.40	15	SUPERSEDES AD 2019-0272R1  LAST DONE @1906.40 FH ON 15.03.2022 REF: WP 9M-HFA-2649
Repetitive SBs	ASB 05A019 R2 PARA 3.B.2.c (EASA AD 2023-0036)	PROTECTIVE MEASURE TIME LIMIT - MAINTENANCE CHECKS - MAIN ROTOR HUB - CHECK OF THE SCISSORS BOLTS	15/03/2022	22/03/2022	7	SUPERSEDES BY AD 2019-0139  LAST DONE @1906.40 FH ON 15.03.2022 REF: WP 9M-HFA-2649
Repetitive SBs	ASB 05A019 R2 PARA 3.B.2.c (EASA AD 2023-0036)	PROTECTIVE MEASURE TIME LIMIT - MAINTENANCE CHECKS - MAIN ROTOR HUB - CHECK OF THE SCISSORS BOLTS	1906.40	1921.40	15	SUPERSEDES BY AD 2019-0139  LAST DONE @1906.40 FH ON 15.03.2022 REF: WP 9M-HFA-2649
Repetitive SBs	EASB 05A012 R1 (EASA AD 2010-0026-E)	TIME LIMIT - MAINTENANCE - CHECK FOR CRACK IIN THE MAIN ROTOR HUB	1906.40	0.00	15	LAST DONE @1906.40 FH ON 15.03.2022 REF: WP 9M-HFA-2649 INTEGRATED INTO ALS REV 003 2022.07.18
Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
Standard SBs	EC120-62-021, REV 0	MAIN ROTOR INSTALLATION OF STEEL AMPEP BUSHES ON THE EQUIPPED SCISSORS LINKS	-	-	-/OPTIONAL	
<b>Airframe SBs</b>						
Standard SBs	SB NO. EC120-00-002, REV. 2	INTRODUCTION TO ALERT SERVICE BULLETINS, SERVICE BULLETIN AND TECHNICAL DIRECTIVES UPDATED FOR HARMONIZATION	-	-	-/ FOR INFO ONLY	FOR INFO ONLY. REFER TIC-EC120B-22-0007
Standard SBs	EASB NO. EC120B-05A020, REVISION 2	TIME LIMITS - MAINTENANCE CHECKS - TAIL ROTOR HUB CHECK OF THE TAIL ROTOR HUB BODY	-	-	-/TO BE COMPLIED	TIC-EC120B-21-0002
Standard SBs	ASB NO. EC120B-63A024,REV. 0	MAIN ROTOR DRIVE - MAIN GEAR BOX (MGB) MAIN MODULE REPLACEMENT OF THE LOWER HOUSING S/N M1274 MANUFACTURER P/N (MP/N) C632A2117101 OF THE MGB	-	-	-/TO BE COMPLIED	TIC-EC120B-22-0016
Standard SBs	SB NO. EC120-52-021, REVISION 0	PRODUCT IMPROVEMENT - DOORS - CARGO DOOR AND REAR ACCESS DOOR - NEW EC120 DOOR LOCK KEY	-	-	-/TO BE COMPLIED	TIC-EC120B-21-0078  APPLICABILITY: APPLICABLE TO THIS AIRCRAFT DUE TO AIRCRAFT IS EQUIPPED WITH FITTED LOCK AND PANELS
Standard SBs	SB NO. EC120-31-010, REV. 0	OBSOLESCENCE MANAGEMENT - INDICATING AND RECORDING SYSTEMS - INTERCHANGEABILITY OF THE VEHICLE ENGINE MANAGEMENT DISPLAY (VEMD)	-	-	-/TO BE COMPLIED	TIC-EC120B-21-0146  APPLICABILITY: APPLICABLE DUE TO AIRCRAFT FITTED WITH AFFECTED VEMD.

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Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
Standard SBs	SB NO. EC120-32-009, REV. 1	LANDING GEAR- NEW BINDING CLAMPS OF THE LANDING GEAR FAIRINGS	-	-	-/TO BE COMPLIED	TIC-EC120B-22-0001  APPLICABILITY: APPLICABLE TO THIS AIRCRAFT SERIAL NUMBER
Standard SBs	SB NO. EC120-00-002, REV. 2	GENERAL - HARMONIZATION - INTRODUCTION TO ALERT SERVICE BULLETINS, SERVICE BULLETIN AND TECHNICAL DIRECTIVES UPDATED FOR HARMONIZATION	-	-	-/NOT APPLICABLE	TIC-EC120B-22-0007  APPLICABILITY: APPLICABLE TO THIS AIRCRAFT TYPE.
Standard SBs	SB NO. EC120-63-023, REV. 0	OBSOLESCENCE MANAGEMENT - MAIN ROTOR DRIVE - ROTOR BRAKE REPLACEMENT OF THE ROTOR BRAKE PAD	-	-	-/OPEN	TIC-EC120B-22-0012
Standard SBs	SB NO. EC120-24-015, REV. 2	ELECTRICAL POWER - ELECTRICAL MASTERBOX- SEPARATION OF THE DIRECT BATTERY DISTRIBUTION FROM THE ELECTRICAL MASTER BOX	-	-	-/TO BE COMPLIED	TIC-EC120B-22-0014  APPLICABILITY: APPLICABLE TO THIS AIRCRAFT DUE TO SERIAL NUMBER AFFECTED.
Standard SBs	SB NO. EC120-26-002, REV. 0	PROTECTIVE MEASURE - FIRE PROTECTION - HAND FIRE EXTINGUISHER (HFE) - CHECK OF THE ABSENCE OF DISCHARGE OF THE HFE	-	-	-/NOT APPLICABLE	TIC-EC120B-22-0045  NOT APPLICABLE DUE TO AFFECTED ITEM NOT FITTED TO AIRCRAFT.
Standard SBs	ASB NO. EC120-32A014,REV. 0	LANDING GEAR INSPECTION FOR COROSION OF THE LANDING GEAR	-	23/03/2023	-/TO BE COMPLIED	TIC-EC120B-22-0005
Standard SBs	ASB NO. 05A006, REVISION 1	TAIL ROTOR DRIVE SHAFT - DAMPER FRICTION RING ON THE REAR DRIVE SHAFT	-	-	-/CANCELLED	TIC-EC120B-21-0083
Standard SBs	SB NO. EC120-24-017, REVISION 0	STARTER GENERATOR - INTRODUCTION OF A CALENDAR LIMIT FOR THE TIME BETWEEN OVERHAUL (TBO) OF THE STARTER GENERATOR	-	-	-/COMPLIED WITH	TIC-EC120B-21-0013
Standard SBs	SB NO. EC120-76-010, REVISION 0	INE CONTROLS - REPLACEMENT OF THE TWIST GRIP DRIVE TUBE	-	-	-/OPTIONAL	REFER TIC-EC120B-21-0014.  COMPLY WITH THIS SB EC120-76-010 R0 WILL OBSOLETE SB EC120-05-014, REVISION 2 COMPLIANCE AS SENTENCE IN TIC-EC120B-21-0015.
Standard SBs	SB 32-002	SKID SHOES REPLACE	-	-	-/COMPLIED WITH	REF: WP-1786-001
Standard SBs	AD NO. 2021-0079 // ASB NO. EC120-64A003, REVISION 0	TAIL ROTOR INSPECTION OF THE DRAIN HOLES ALIGNMENT OF THE TAIL ROTOR BLADES (TRB)	1910.10	-	-/COMPLIED WITH	HFA 2739
Standard SBs	AD NO. 2021-0079 // ASB NO. EC120-64A003, REVISION 0	TAIL ROTOR INSPECTION OF THE DRAIN HOLES ALIGNMENT OF THE TAIL ROTOR BLADES (TRB)	31/05/2022	-	-/COMPLIED WITH	HFA 2739

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Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
Standard SBs	SB NO. EC120-53-018, REVISION 0	CORRECTIVE MEASURE - REAR FUSELAGE - REPAIR AT THE TAIL BOOM VHF ATTACHMENT	-	-	-/OPTIONAL	<p>REFER: TIC-EC120B-20-044</p> <p>REMARKS:</p> <p>APPLICABLE TO ALL EC120 HELICOPTER</p> <p>COMPLIANCE: HELICOPTERS/INSTALLED EQUIPMENT OR PARTS: THE WORK MUST BE PERFORMED ON THE HELICOPTER BY THE OPERATOR. COMPLY WITH PARAGRAPH 3. OF THIS SERVICE BULLETIN.</p> <p>NON-INSTALLED EQUIPMENT OR PARTS: MODIFY STOCKS AS REQUIRED AND AS PER PARAGRAPHS 3.B.2.B. TO 3.B.2.E.</p> <p>REASON: THE PURPOSE OF THIS SERVICE BULLETIN IS TO REPAIR THE TAIL BOOM DUE TO CORROSION AND/OR CRACKS AT THE VHF (VERY HIGH FREQUENCY) ANTENNA ATTACHMENTS.</p>

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Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
Standard SBs	ASB NO. EC120-53A017, REVISION 1	PROTECTIVE MEASURE - FUSELAGE - REAR FUSELAGE CORROSION ON THE TAIL BOOM	-	-	-/Complied With	<p>REFER: TIC-EC120B-20-045</p> <p>REMARKS:</p> <p>APPLICABLE TO ALL EC120 HELICOPTERS.</p> <p>COMPLIANCE: HELICOPTERS/INSTALLED EQUIPMENT OR PARTS: THE WORK MUST BE PERFORMED ON THE HELICOPTER BY THE OPERATOR. COMPLY WITH PARAGRAPH 3. OF THIS SERVICE BULLETIN WITHIN 6 MONTHS FROM RECEIPT OF REVISION 0 OF THIS ALERT SERVICE BULLETIN ISSUED ON THE DATE INDICATED IN THE PAGE FOOTER.</p> <p>NON-INSTALLED EQUIPMENT OR PARTS: BEFORE INSTALLING THE TAIL BOOM IDENTIFIED IN PARAGRAPH 1.A.2 OF THIS SERVICE BULLETIN. CHECK STOCKS AND MODIFY WHEN APPLICABLE AS PER 3.B.2.</p> <p>REASON: CORROSION WAS FOUND ON THE EXTERNAL TAIL BOOM SKIN, UNDER THE VHF ANTENNA. THIS GALVANIC TYPE CORROSION IS DUE TO: - A CONTACT BETWEEN THE ALUMINIUM SKIN AND THE STAINLESS-STEEL ANTENNA ATTACHMENT INSERTS. - THE PROTECTION WHICH IS NOT SUFFICIENT</p> <p>CORROSION PROPAGATION CANNOT BE DETECTED WITHOUT REMOVING THE VHF ANTENNA AND REQUIRES THOROUGH INSPECTION. IF THIS CORROSION IS NOT</p>

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Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
						DETECTED, THE TAIL BOOM CAN UNDERGO TO BUCKLING DURING SEVERE LANDING. CONSEQUENTLY, AIRBUS HELICOPTERS MAKES IT MANDATORY TO CHECK THE PREVIOUSLY SPECIFIED AREAS FOR CORROSION.
Standard SBs	SB NO. EC120-88-022, REVISION 0	ELECTRICAL HARNESSSES - REAR FUSELAGE - IMPROVEMENT OF THE HARNESS FOR THE CHIP DETECTOR OF THE TAIL GEAR BOX (TGB)	-	-	-/OPTIONAL	<p>REFER: TIC-EC1208-20-040</p> <p>REMARKS:</p> <p>APPLICABLE TO ALL EC120 SERIAL NUMBERS.</p> <p>AIRBUS HELICOPTERS RECOMMENDS COMPLIANCE WITH PARAGRAPH 3. OF SB EC120B-88-022 DURING ONE OF THE NEXT MAINTENANCE INSPECTIONS AS PER OPERATIONAL CONSTRAINTS.</p> <p>DURING A FLIGHT, THE PILOT REPORTED APPEARANCE OF SEVERAL FLASHES AND FOLLOWED BY FIXED LIGHTING OF GB CHIP ON THE WARNING PANEL. THE INVESTIGATION SHOWED THAT THERE WERE NO PARTICLES ON THE TGB CHIP DETECTOR, BUT CHAFING WAS DETECTED ON THE WIRES OF THE CHIP DETECTOR HARNESS.</p>

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Standard SBs	SB NO. EC120-63-021, REVISION 0	PRODUCT IMPROVEMENT - MAIN ROTOR DRIVE - MODIFICATION OF THE GUIDING OF THE ROTOR BRAKE PADS	-	-	-/OPTIONAL	<p>REFER: TIC-EC1208-20-035</p> <p>REMARKS:</p> <p>APPLICABILITY: TO REQUEST PHYSICAL INSPECTION FOR EQUIPMENT APPLICABILITY OF PN: C635A1201056 AND C635C1013102 ON EC120B SN 1487 AND 1275.</p> <p>COMPLIANCE: OPTIONAL UPON CUSTOMER REQUEST.</p> <p>REASON: THE PURPOSE OF THIS SERVICE BULLETIN IS TO IMPROVE THE GUIDING OF THE ROTOR BRAKE PADS. THIS IMPROVEMENT CONSISTS IN INCREASING THE GUIDING LENGTH FOR THE BRAKE PADS AND MODIFYING THE SPRING TO PROVIDE A FULLY AXIAL ELASTIC RETURN FORCE ON THE PADS.</p>

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Standard SBs	EASB NO. EC120-05A022, REVISION 0	PROTECTIVE MEASURE -TIME LIMITS - MAINTENANCE CHECKS - LANDING GEAR - CHECK FOR CRACKS ON THE LANDING GEAR	-	-	-/NOT APPLICABLE	<p>REFER: TIC-EC120B-20-037</p> <p>REMARKS:</p> <p>NOT APPLICABLE DUE TO BOTH AIRCRAFT SERIAL NUMBER ARE NOT AFFECTED BY THIS EASB.</p> <p>COMPLIANCE:</p> <p>1.UPON RECEIPT OF THIS EASB (20/10/2020), INSERT THE PAGE IN APPENDIX 4. OF THIS EASB IN THE FLIGHT MANUAL</p> <p>2.COMPLY WITH PARA 3.B.2.A. EACH BFF UNTIL THE FIRST COMPLIANCE WITH PARA 3.B.2.B. OF THIS ASB</p> <p>3.WITHIN 3M FROM RECEIPT OF REVISION 0 OF THIS ASB ISSUED ON 20/10/2020, COMPLY WITH PARA 3 (EXCEPT PARA 3.B.2.A.) OF THIS ASB</p> <p>AFTER COMPLIANCE WITH PARA 3.B.2.B. OF THIS ASB, REMOVE THE APPENDED PAGE PREVIOUSLY ADDED IN 1. FROM THE FLIGHT MANUAL.</p> <p>4.THEN, COMPLY WITH PARA 3 (EXCEPT PARAGRAPH 3.B.2.A.) EVERY 13 AUTOROTATION LANDINGS</p> <p>REASON:</p> <p>AFTER REPLACEMENT OF A LANDING GEAR DUE TO SINKING, INSPECTION WAS CARRIED OUT ON THE DISCARDED SKID TUBE WHICH RESULT IN FINDING OF A CRACK INSIDE THE SKID TUBE, UNDER THE REAR CROSSBAR FITTING.</p> <p>AFTER INVESTIGATION BY AH, IT WAS FOUND THAT THIS CRACK CAN INITIATE IF THE HELICOPTER PERFORMS A LARGE NUMBER OF AUTOROTATION LANDINGS AND WOULD LEAD TO ROLL-OVER OF THE</p>



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						HELICOPTER AFTER A HARD LANDING. THEREFORE, AH MAKES COMPLIANCE WITH THIS ASB MANDATORY.
Standard SBs	ASB No. EC120-25A030, REVISION 2	PROTECTIVE MEASURE - EQUIPMENT AND FURNISHINGS - EMERGENCY FLOTATION SYSTEM INSPECTION OF THE FLOTATION BALLOON SUPPLY HOSES	-	-	-/NOT APPLICABLE	REFER: TIC-EC120B-20-027  REMARKS: NOT APPLICABLE DUE TO SN: 1275 AND 1487 DID NOT EQUIPPED WITH AFFECTED PN.  REASON: DURING MAINTENANCE OPERATIONS ON THE EMERGENCY FLOATATION SYSTEM OF AN AS365 HELICOPTER, THE THIRD COMPARTMENT OF THE LH FRONT FLOTATION BALLOON DID NOT INFLATE. AFTER ANALYSIS, THE HOSE THAT SUPPLIES THE GAS FLOW FROM THE CYLINDER ASSEMBLY TO THE RELATED FLOTATION BALLOON COMPARTMENT WAS FOUND CLOGGED.
Standard SBs	SB EC120-88-001	ELECTRICAL HARNESS - DIRECT CURRENT POWER SUPPLY SYSTEM. CHECK OF THE 5VB HARNESS	-	-	-/OPTIONAL	OPTIONAL-APPLICABLE. COMPLY UPON CUSTOMER REQUEST.  MYCOPTER SERVICE BULLETIN COMPLIANCE SUMMARY - AIRFRAME ITEM #23 PAGE 1
Standard SBs	SB EC120-05-018	TIME LIMITS - MAINTENANCE CHECKS - MAIN ROTOR CHECKING THE SKIN ON MAIN ROTOR BLADES	-	-	-/NOT APPLICABLE	NOT APPLICABLE. SERIAL NUMBER NOT AFFECTED  MYCOPTER SERVICE BULLETIN COMPLIANCE SUMMARY - AIRFRAME PAGE. 1 OF 1
Standard SBs	ASB 25A030	PROTECTIVE MEASURE - EQUIPMENT AND FURNISHINGS - EMERGENCY FLOTATION SYSTEM - INSPECTION OF THE FLOTATION BALLOON SUPPLY HOSES	-	-	-/NOT APPLICABLE	NOT APPLICABLE. AIRCRAFT NOT QUIPPED WITH FLOATS.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #22 PAGE 3

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Standard SBs	ASB 62A020	PROTECTIVE MEASURE - TIME LIMITS - MAINTENANCE CHECKS - MAIN ROTOR HUB - INSPECTION OF ROTATING AND NON-ROTATING SCISSORS BOLTS	-	-	-/COMPLIED WITH	COMPLIED AS PER ASB EC120B-62A020 REFER HFA 007/008  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #18 PAGE 3
Standard SBs	ASB 04A008	CORRECTIVE MEASURE - LIMITATIONS - TAIL ROTOR HUB - INTRODUCTION OF A NEW PART NUMBER AND RE-IDENTIFICATION OF THE EQUIPPED BLADES OF THE TAIL ROTOR HUB	-	-	-/NOT APPLICABLE	NOT APPLICABLE DUE TO T/R BLADE S/N NOT AFFECTED.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #17 PAGE 3
Standard SBs	ASB 52A018	DOORS - REINFORCEMENT OF THE DOOR LOCKING / UNLOCKING MECHANISM	-	-	-/COMPLIED WITH	DECLARED ON 16.02.2017 ACCOMPLISHED BY HELIDAX. REFER WORK ORDER CIFH/26  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #16 PAGE 2
Standard SBs	ASB 31A008	INDICATING AND RECORDING SYSTEMS - REPLACEMENT OF PUDH-BUTTON FLOAT ARM AND SMS UNIT	-	-	-/NOT APPLICABLE	NOT APPLICABLE DUE TO NOT INSTALLED ON AIRCRAFT  MYCOPTER MODBOOK SB STATUS - AIRCRAFT ITEM #15 PAGE 2
Standard SBs	EASB 04A007	LIMITATIONS - EMERGENCY FLOATATION GEAR CONTROL ON THE LIGHTING AND ANCILLARY CONTROL UNIT (LACU)	-	-	-/NOT APPLICABLE	NOT APPLICABLE DUE TO NO EMERGENCY FLOATATION GEAR INSTALLED.  MYCOPTER MOD BOOK SB STATUS - AIRFRAME ITEM #14 PAGE 2
Standard SBs	ASB 53A015	FUSLEAGE - TAIL BOOM - CORROSION ON THE TAIL BOOM	-	-	-/COMPLIED WITH	DECLARED ON 17.11.2014. ACCOMPLISHED BY SWISS HELICOPTER  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #13 PAGE 2

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Standard SBs	ASB 52A017	DOORS - CREW DOORS - CHECK OF THE INSTALLATION OF THE CREW DOOR ACTUATORS	-	-	-/NOT APPLICABLE	NOT AVAILABLE DUE TO COMPONENT NOT INSTALLED ON AIRCRAFT.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #12 PAGE 2
Standard SBs	ASB 52A014	DOORS - SLIDING DOOR - ATTACHMENT OF THE SLIDING DOOR STAR SUPPORT	-	-	-/NOT APPLICABLE	DECLARED ON 05.02.2015 @ 1783:24 FH  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #11 PAGE 2
Standard SBs	ASB 25A026	EQUIPMENT AND FURNISHINGS - EMERGENCY FLOATATION GEAR - PROTECTOR INSTALLATION ON THE REAR BRACKET AND THE SUPPLY COUPLINGS	-	-	-/NOT APPLICABLE	NOT APPLICABLE DUE TO AIRCRAFT NOT EQUIPPED WITH FLOATS.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #10 PAGE 2
Standard SBs	EASB 24A012	ELECTRICAL POWER - EMERGENCY SWITCH (EMER SW0 WIRING	-	-	-/NOT APPLICABLE	NOT APPLICABLE BY AIRCRAFT SERIAL NUMBER  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #9 PAGE 2
Standard SBs	ASB 05A011	TIME LIMITS - MAINTENANCE CHECKS - EMERGENCY FLOATATION GEARS; PERIODIC CHECK OF THE FLOATS WITH INTERVALS IN HOURS	-	-	-/NOT APPLICABLE	DECLARED ON 06.07.2010 BY HELI TECHNIQUE  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #7 PAGE 1
Standard SBs	ASB 04A004	LIMITATIONS - SIREN LOAD RELEASE UNITS	-	-	-/NOT APPLICABLE	NOT APPLICABLE. AIRCRAFT NOT EQUIPPED WITH HOOK  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #6 PAGE 1
Standard SBs	ASB 76A005	ENGINE CONTROLS - REPLACEMENT OF THE TWIST GRIP DRIVE TUBE	-	-	-/NOT APPLICABLE	DECLARED ON 30.10.2008 BT IX AIR.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #5 PAGE 1

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Standard SBs	ASB 76A006	ENGINE CONTROLS - CHECK OF THE TWIST GRIP DRIVE TUBE AND CONTROL PINION ATTACHMENT	-	-	-/NOT APPLICABLE	DECLARED ON 30.10.2008 BY IX AIR.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #4 PAGE 1
Standard SBs	ASB 25A023	EQUIPMENT AND FURNISHINGS - ELECTRICAL BONDING OF ENERGY ABSORBING SEATS	-	-	-/NOT APPLICABLE	DECLARED ON 15.02.2008 BY IX AIR  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #3 PAGE 1
Standard SBs	EASB 05A008 R2	EQUIPMENT AND FURNISHINGS - CRACK IN THE BELT BUCKLE	-	-	-/COMPLIED WITH	APPLIED ON 31.08.2007 BY HELICOPTERS ITALIA. REFER AIRCRAFT LOGBOOK WORK ORDER 336/2007  MYCPTER MODBOOK SB STATUS - AIRFRAME ITEM #2 PAGE 1
Standard SBs	ASB 05A006	TIME LIMITS - MAINTENANCE CHECKS - TAIL ROTOR DRIVE SHAFT: DAMPER FRICTION RING ON THE REAR DRIVE SHAFT	-	-	-/COMPLIED WITH	DECLARED ON 25.05.2009. ACCOMPLISHED BY IX AIR.  MYCOPTER MODBOOK SB STATUS - AIRFRAME ITEM #1 PAGE 1  ASB CANCELLED
Standard SBs	319 71 4089 VERSION A	AIR INTAKE GRIDS. ELIMINATION OF BRAZING AND SIMPLIFICATION OF THE DESIGN (MOD TF89)	-	-	-/OPTIONAL	OPTIONAL DUE TO NON MANDATORY SB. MYCOPTER SB STATUS ITEM #8 PAGE 2
Standard SBs	319 79 4831 VERSION D	LUBRICATION UNIT INSPECTION OF JET BONDING ON EJECTOR ASSEMBLY	-	-	-/SUPERSEDED	SUPERSEDED BY MANDATORY SB NO 319 79 4835 MYCOPTER SB STATUS ITEM #6 PAGE 2
Standard SBs	319 73 4837 VERSION B	UPDATING OF THE ENGINE LOG BOOK TO EXTEND THE TBO OF THE ARRIUS 2F PUMP AND METERING VALVE ASSEMBLY, ADJUSTED	-	-	-/FOR INFO ONLY	MYCOPTER SB STATUS ITEM #4 PAGE 2
Standard SBs	A319 79 4841 VERSION A	REPLACEMENT OF THE MAGNETIC HEADS OF THE ELECTRICAL MAGNETIC PLUG ASSEMBLIES	-	-	-/NOT APPLICABLE	NOT APPLICABLE BY ENGINE MODIFICATION STATUS. MYCOPTER SB STATUS ITEM #3 PAGE 1
Standard SBs	A319 79 4840 VERSION A	INSPECTION OF THE MAGNETIC HEADS OF THE ELECTRICAL MAGNETIC PLUG ASSEMBLIES	-	-	-/NOT APPLICABLE	NOT APPLICABLE BY ENGINE MODIFICATION STATUS. MYCOPTER SB STATUS ITEM #2 PAGE 1

Aircraft Registration:	9M-HFA	Serial No:	1487	Date:	30/01/2023	Page No:	13
Aircraft Hours:	1911.40	Landings:	3352.00	Engine 1:	1911.40	Engine 2:	0.00

Types	SB Number	Subject	Last Done	Next Due	Interval/Status	Limitation Notes
Standard SBs	319 72 4088 VERSION A	GAS GENERATOR MODULE (M02). CHANGE OF REAR BEARING DUCT MATERIAL	-	12/10/2020	-/OPTIONAL	MODIFICATION TF88 TO BE COMPLIED AT THE NEXT ENGINE MAJOR MAINTENANCE / INSPECTION (ENGINE HSI).  MYCOPTER MODIFICATION RECORD BOOK SERVICE BULLETIN COMPLIANCE - ENGINE ITEM #4 PAGE 2
<b>Engine SBs</b>						
Standard SBs	SB NO. 319 73 4850, VERSION B	PREVENTIVE FLAME-OUT CHECK VIA THE PREFERENCE INJECTOR VALVE TEST	-	-	-/NOT APPLICABLE	TIC-EC120B-22-0038  APPLICABILITY: NOT APPLICABLE TO THIS ENGINE DUE TO THE PREFERENCE INJECTOR LAST REPAIRED BY SA ERMETO HAS LOGGED MORE THAN 100 FLIGHT HOURS.
Standard SBs	319 73 4077 VERSION F	ADJUSTED FCU ASSEMBLY ADDITION AN AUXILIARY DELTA P VALVE	-	-	-/COMPLIED WITH	APPLICABLE. FCU INSTALLED HAD BEEN COMPLIED WITH TF77. FCU S/N: 332M
Standard SBs	SB NO. 319 72 2846, VERSION B	ARRIUS 2F SERVICE BULLETIN CREATION OF CONFIGURATION APPLICABLE TO ARRIUS 2 B1, B1A, B2, F, K1 AND K2 TURBOSHAFT ENGINES DURING OVERHAUL	-	-	-/ FOR INFO ONLY	REFER TIC-EC120B-21-0021.  COMPLIANCE: APPLICABLE, DURING GENERAL OVERHAUL
Standard SBs	SB NO. 319 77 4097, VERSION B	ENGINE INDICATING - PYROMETRIC HARNESS - CHANGE IN THE MATERIAL OF THE THERMOCOUPLE PROBES AND ROUTING MODIFICATION. MODIFICATION Tf 97.	0.00	2004.80	-/TO BE COMPLIED	REFER TIC-EC120B-21-0022.  TO COMPLY WITH SB WHEN THE PYROMETRIC HARNESS REPLACED AT SMI 400 FH INSPECTION.