



CLIENT/OWNER: YTL POWER GENERATION SDN BHD	SERIAL NO. 41358	HOURS	LDG/CYCLE	WORKSHEET NO: 13938-001
AIRCRAFT TYPE: AW139	AIRCRAFT	REFER WORKPACK		WORK/INSP/DESC: 50 FH INSPECTION
REGISTRATION: 9M-YTL	#1 ENGINE: PCE-KB1443	REFER WORKPACK		WORKPACK REF: 9M-YTL-13938
BASE/FACILITY: WMSA	#2 ENGINE: PCE-KB1457	REFER WORKPACK		AJL REF NO.: REFER WORKPACK
DATE IN: REFER WORKPACK OUT: REFER WORKPACK		NG / N1	NF / N2	SHEET: 14 OF 17

Reason for raising:
SCHEDULED INSPECTION TO BE CARRIED OUT I.A.W. AMP NO.YTLPG/CAMO/AMP/AW139 ISSUE 1
REV. 2,AW139 IETP - 41ST ISSUE 2022-06-28 UPDATED 2022-11-16 TO THE ALL LATEST
ISSUE/REVISION/UPDATES. 50 FH INSPECTION (AIRFRAME AND ENGINES)

Raised by and date:
CAMO; Thavamalar
21/12/2022

Other requirements/information:
NIL

Item	Description	Technician	* Eng. CRS	Date
70	63-35 ROTOR BRAKE COVER REFERENCE (DMC): 39-A-63-50-00-00A-310A-A DO A GVI FOR CONDITION AND INTEGRITY OF ATTACHMENTS REMARKS: CARRIED OUT FOUND SATISFACTORY	<i>[Signature]</i> 811	<i>[Signature]</i> 	15/05/23
71	64-38 TAIL ROTOR INSTALLATION DO A GVI OF TAIL ROTOR INSTALLATION COMPONENTS DMC REFERENCE: 39-A-64-21-00-00A-310A-A REMARK: CARRIED OUT FOUND SATISFACTORY	<i>[Signature]</i> 811	<i>[Signature]</i> 	15/05/2023
72	71-15 LEFT AND RIGHT INSULATED TUBES (PART NUMBERS 3G7160V02951 AND 3G7160V03051) (ENGINE INLET PARTICLE SEPARATOR SYSTEM) DO A FC. FAULT FIND TASK TO VERIFY SEALING ABILITY UNDER PRESSURE REFERENCE (DMC): 39-A-75-51-00-00A-340A-K REMARKS: NOT APPLICABLE DUE COMPONENT NOT INSTALLED.			/
73	71-17 MAIN PANEL (2 OFF) (ENGINE INLET PARTICLE SEPARATOR SYSTEM) REFERENCE (DMC): 39-A-75-51-00-00A-31AB-K DO A DI FOR INTEGRITY OF ATTACHMENT TO STRUCTURE AND ABSENCE OF CRACKS REMARKS: NOT APPLICABLE DUE TO NOT FITTED			/
74	CM71-03 ENGINE POWER AVAILABLE TREND MONITORING (4-DISPLAY HELICOPTER ONLY) REFERENCE (DMC): N/A PERFORM AND RECORD ON THE SUITABLE PROVIDED FORMS (SEE PT6C-67C OR PT6C-67C1 ENGINE MAINTENANCE MANUAL, WHICHEVER APPLICABLE) THE POWER ASSURANCE CHECK TREND OF BOTH THE ENGINES NOTE: TASK TO BE PERFORMED BY FOLLOWING THE INSTRUCTIONS CONTAINED IN THE LAST ISSUE OF P&W PT6C-67C OR PT6C-67C1 ENGINE MAINTENANCE MANUAL, WHICHEVER APPLICABLE. REMARKS: REFER TO ATTACHMENT FOR RESULT OF PAC CARRIED OUT. FOUND SATISFACTORY	<i>[Signature]</i> 8106	<i>[Signature]</i> 	17/05/23

Certifies that the work specified above, except as otherwise specified, was carried out in accordance with CAA Malaysia Requirements and in respect to that work
the aircraft / aircraft component is considered ready for release to service. AMO/2016/02

Certifies that the work specified above, except as otherwise specified, was carried out in accordance with _____
and in respect to that work the aircraft / aircraft component is considered ready for release to service.

TICK ✓ WHERE APPLICABLE