

UPDATE ON AIR VEHICLE MAINTENANCE PLANNING INFORMATION (AMPI)

CHAPTER 4: AIRWORTHINESS LIMITATION

INTRODUCTION

IF NOT DIFFERENTLY SPECIFIED, THE RETIREMENT LIVES ARE BASED ON THE FOLLOWING ASSUMPTIONS:

- FOR EXTERNAL HOIST OPERATIONS: 4 LIFTS PER FLYING HOUR, CALCULATED ON AT LEAST 100 FH BASIS.

RETIREMENT LIVES – AFFECTED PARTS

REF	PART	PART NUMBER	RETIREMENT LIFE	APPLICABILITY
RL012A	MAIN ROTOR BLADE ASSEMBLY	3G6210A00131	20000 FH or 50200 landings whichever comes first (Table 4) (Note 4) (Para 4 - External Hoist Operation)	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-JPM, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT FITTED ON A/C
RL016	MAIN ROTOR SCISSORS ATTACHMENT FLANGE ASSEMBLY	3G6220A00632	41700	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-JPM, 9M-YPG, 9M-YTL & 9M-BOD DUE TO COMPONENT NOT FITTED ON A/C. APPLICABLE TO 9M-BOC & 9M-BOD DUE TO COMPONENT FITTED ON A/C
RL019	UPPER SCISSOR LEVER ASSEMBLY	3G6230A00832	14280	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-JPM, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT NOT FITTED ON A/C.
RL056	SINGLE HOIST MOUNT	3G2591V02551	100100 lifts (Note 16) (Note 17)	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-JPM, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT NOT FITTED ON A/C

RL057	DOUBLE HOIST MOUNT	3G2591V02451	100800 lifts (Note 16) (Note 17)	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-JPM, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT NOT FITTED ON A/C
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TABLE 2: RETIREMENT LIVES – LIST OF AFFECTED PARTS

NOTES:

16. IF THE NUMBER OF LIFTS IS NOT KNOWN THROUGHOUT THE ENTIRE SERVICE LIFE OF THE HOIST MOUNT, THE ACCRUED NUMBER OF LIFTS SHALL BE CALCULATED MULTIPLYING THE HOIST MOUNT FLIGHT HOURS BY A FACTOR OF 1.65.
17. IF BOTH THE NUMBER OF LIFTS AND THE SERVICE LIFE ARE NOT KNOWN, THE HOIST MOUNT RETIREMENT LIFE SHALL BE REDUCED BY 10000 LIFTS.

EXTERNAL HOIST OPERATION

REF	PART	PART NUMBER	LIFE PENALTY FACTOR	APPLICABILITY
RL012	MAIN ROTOR BLADE ASSEMBLY	3G6210A00131	0.1	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT FITTED ON A/C
RL022	MGB SHAFT, MAIN ROTOR	3G6320A01852	0.1	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-JPM, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT NOT FITTED ON A/C

TABLE 5: EXTERNAL HOIST LIFT – LIFE PENALTY FACTOR

NOTE:

2. DELETED

MANDATORY INSPECTION – GENERAL

REF	PART	TASK	INTERVAL	REFERENCE (DMC)	APPLICABILITY
MI64-02	TAIL ROTOR ELASTOMERIC SPHERICAL BEARING (PART NUMBER 3G6420V00153 AND 3G6420V00154)	DETAILED INSPECTION FOR CRACKS IN THE ELASTOMER	100	39-A-64-11-00-00A-31AB-A	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD.
MI64-03	DELETED				

TABLE 2: MANDATORY INSPECTION – LIST OF AFFECTED PARTS

CERTIFICATION MAINTENANCE REQUIREMENT

REF	PART	TASK	INTERVAL	REFERENCE (DMC)	APPLICABILITY
CM71-03	ENGINE POWER AVAILABLE TREND MONITORING (4- DISPLAY HELICOPTER ONLY)	PERFORM AND RECORD ON THE SUITABLE PROVIDED FORMS (SEE PT6C-67C OR PT6C-67C1 ENGINE MAINTENANCE MANUAL, WHICHEVER APPLICABLE) THE POWER ASSURANCE CHECK TREND OF BOTH THE ENGINES	50	NOTE 3	UPDATE ON NEW ENGINE MODEL

TABLE 2: CERTIFICATION MAINTENANCE REQUIREMENTS

NOTE:

- TASK TO BE PERFORMED BY FOLLOWING THE INSTRUCTIONS CONTAINED IN THE LAST ISSUE OF P&W PT6C-67C OR PT6C-67C1 ENGINE MAINTENANCE MANUAL, WHICHEVER APPLICABLE.

CHAPTER 5: SCHEDULED/UNSHEDULED MAINTENANCE

SCHEDULE / UNSCHEDULED MAINTENANCE

1.2 CONTINUED AIRWORTHINESS

THE MAINTENANCE REQUIREMENTS IDENTIFIED IN THIS CHAPTER, TOGETHER WITH THOSE IN THE FOLLOWING LIST, CONSTITUTE THE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS FOR THE HELICOPTER:

- AIRWORTHINESS LIMITATIONS AS IN CHAPTER 04 OF THIS PUBLICATION
- HELICOPTER PRE-FLIGHT CHECKS IDENTIFIED IN SECTION 2 OF THE ROTORCRAFT FLIGHT MANUAL
- PT6C-67C AND PT6C-67C1 ENGINE SCHEDULED MAINTENANCE REQUIREMENTS

PERMITTED INSPECTION /CHECK INTERVAL TOLERANCE- TOLERANCE RULE

UNLESS SPECIFIED DIFFERENTLY, THE TOLERANCES FOR THE SCHEDULED INSPECTIONS/CHECKS ARE AS FOLLOWS:

- HOURLY INTERVAL - TEN PERCENT (10 %) OR 50 HOURS MAXIMUM WHICHEVER IS LESS
- CALENDAR INTERVAL - TEN PERCENT (10 %) OR 30 DAYS MAXIMUM WHICHEVER IS LESS
- LANDINGS INTERVAL - TEN PERCENT (10 %) OR 100 LANDINGS MAXIMUM WHICHEVER IS LESS.

WHEN AN INSPECTION/CHECK IS POSTPONED WITH RESPECT TO THE PRESCRIBED SCHEDULE, ALWAYS WITHIN THE MAXIMUM ALLOWABLE TOLERANCE, SUBSEQUENT INTERVALS WILL BE COMPUTED AS PER THE ORIGINAL SCHEDULE AND RELATED TOLERANCE. EXAMPLES:

- TASK LIMIT: 100 FH. MAXIMUM TOLERANCE: 10 FH. IF THE INSPECTION IS CONDUCTED AT 105 FH, THE SUBSEQUENT ONE MUST BE PERFORMED AS PER ORIGINAL SCHEDULE AT 200 FH (+ 10 FH)
- TASK LIMIT: 24 MONTHS. MAXIMUM TOLERANCE: 30 DAYS. IF THE INSPECTION IS CONDUCTED AT 25 MONTHS, THE SUBSEQUENT ONE MUST BE PERFORMED AS PER ORIGINAL SCHEDULE AT 48 MONTHS (+ 30 DAYS).
- TASK LIMIT: 2000 LANDINGS. MAXIMUM TOLERANCE: 100 LANDINGS. IF THE INSPECTION IS CONDUCTED AT 2100 LANDINGS, THE SUBSEQUENT ONE MUST BE PERFORMED AS PER ORIGINAL SCHEDULE AT 2000 LANDINGS (+ 100 LANDINGS).

WHEN AN INSPECTION/CHECK IS ANTICIPATED WITH RESPECT TO THE PRESCRIBED SCHEDULE, SUBSEQUENT INTERVALS WILL BE CONDUCTED, ACCORDING TO THE SCHEDULE, STARTING FROM THE ACTUAL TIME (HOURS OR CALENDAR DATE) THE INSPECTION/CHECK WAS PERFORMED. EXAMPLES:

- TASK LIMIT: 100 FH. MAXIMUM TOLERANCE: 10 FH. IF THE INSPECTION IS CONDUCTED AT 85 FH, THE SUBSEQUENT ONE MUST BE PERFORMED 185 FH (+ 10 FH)
- TASK LIMIT: 24 MONTHS. MAXIMUM TOLERANCE: 30 DAYS. IF THE INSPECTION IS CONDUCTED AT 23 MONTHS, THE SUBSEQUENT ONE MUST BE PERFORMED AT 47 MONTHS (+ 30 DAYS).
- TASK LIMIT: 2000 LANDINGS. MAXIMUM TOLERANCE: 100 LANDINGS. IF THE INSPECTION IS CONDUCTED AT 1900 LANDINGS, THE SUBSEQUENT ONE MUST BE PERFORMED AS PER ORIGINAL SCHEDULE AT 3900 LANDINGS (+ 100 LANDINGS).

COMPONENT OVERHAUL SCHEDULE – GENERAL

REF	COMPONENT	PART NUMBER	OVERHUAL INTERVAL	APPLICABILITY
CO62-03	MAIN ROTOR DAMPER	3G6220V02051	3000	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
CO71-02	DELETED			NIL

TABLE 2: LIST OF COMPONENTS

DISCARD TIME SCHEDULE

REF	COMPONENT	PART NUMBER	OVERHUAL INTERVAL	APPLICABILITY
DT25-19	BATTERY PACK (PORTABLE LOCATOR BEACON)	A0696Y 305579A	5 YEARS (NOTE 10)	
DT31-06	QAR BATTERY	D51640-0001 (NOTE 17)	10 YEARS	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
DT71-04	SHUT OFF VALVE (2 OFF)	3G7160V01451	10000FH	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
DT56-03	COCKPIT DOOR EMERGENCY EXIT WINDOW GASKET, RETAINERS AND FILLER WEDGES	3G5630A01451 3G5630A01551 3G5630A01651 3G5630A01751 3G5630A03251 3G5630A01851	4 YEARS (NOTE 20)	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT

DT56-04	CABIN EMERGENCY EXIT WINDOW GASKET, RETAINERS AND FILLER WEDGES	3G5630A02451 3G5630A02551 3G5630A02651 3G5630A02951 3G5630A01951 3G5630A02051 3G5630A02151 3G5630A02251 3G5630A02351	4 YEARS (NOTE 20)	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
DT56-05	EMERGENCY EXIT WINDOW EXTERNAL PULL TAB (CLIP AND HANDLE)	3G5630A01351 3G5630A01251	4 YEARS (NOTE 20)	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT

TABLE 2: LIST OF COMPONENTS

NOTE:

25. THE DISCARD TIME IS INTENDED FROM THE MANUFACTURING DATE OR ACCORDING TO THE EXPIRATION DATE REPORTED ON THE LOG CARD OF THE COMPONENT, WHICHEVER COMES FIRST.

MAINTENANCE TASK - GENERAL

NO	ITEM	TASK	CONDITION/LIMIT STANDARD	CONDITION/LIMIT PHASED	CONDITION/LIMIT PROGRESSIVE	REFERENCE	APPLICABILITY
02-03	MAIN LANDING GEAR WHEEL (GOODRICH LANDING GEAT INSTALLATION)	DO A FC TO MEASURE THE TYRE INFLATION (1034 BAR IF THE HELICOPTER IS ON THE JACKS AND 1075 BAR IF THE HELICOPTER IS ON GROUND). IF THE PRESSURE IS NOT WITHIN LIMITS PERFORM INFLATION PROCEDURE (SEE APPLICABLE AMP DMC)	50	BWKL/50	BWKL/50	N/A	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT

07-05	MAIN LANDING GEAR WHEEL (GOODRICH LANDING GEAT INSTALLATION)	DO A FC TO MEASURE THE TYRE INFLATION (1034 BAR IF THE HELICOPTER IS ON THE JACKS AND 1075 BAR IF THE HELICOPTER IS ON GROUND). IF THE PRESSURE IS NOT WITHIN LIMITS PERFORM INFLATION PROCEDURE (SEE APPLICABLE AMP DMC)	50	BWKL/50	BWKL/50	N/A	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
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TABLE 2: LIST OF REQUIREMENTS FOR GENERAL VISUAL CHECKS

NO	ITEM	TASK	CONDITION/LIMIT STANDARD	CONDITION/LIMIT PHASED	CONDITION/LIMIT PROGRESSIVE	REFERENCE	APPLICABILITY
22-02	COLLECTIVE LVDT	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF BUNDLES AND CONDITION, SAFETY AND SECURITY OF CONNECTORS	1 YEAR	1 YEAR	1 YEAR	39-A-76-11-00-00A-310A-A	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD
22-03	TRIM SERVO ACTUATORS	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF BUNDLES AND CONDITION, SAFETY AND SECURITY OF CONNECTORS	2 YEARS	2 YEARS	2 YEARS	39-A-22-12-00-00A-310A-A	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD
24-03	POWER DISTRIBUTION PANELS	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF BUNDLES AND CONDITION, SAFETY AND SECURITY OF CONNECTORS	2 YEARS	2 YEARS	2 YEARS	39-A-24-61-00-00A-310A-A	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD DUE TO COMPONENT FITTED ON A/C
25-74	SEAT BELTS (ZODIAC AEROSPACE)	DO A GVI OF THE SEATBELT FABRIC FOR DAMAGE AND CONDITION, INCLUDING THE RESTRAINT SYSTEM ATTACHMENTS AND ALL THREADED PARTS	1 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON	1 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON	1 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON	39-J-25-21-00-01A-310A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-

			THE HELICOPTER [52][67]	THE HELICOPTER [52][67]	THE HELICOPTER [52][67]		BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
25-75	SEAT BUCKLE RESTRAINT SYSTEM (ZODIAC AEROSPACE)	DO AN OC FOR PROPER FUNCTIONING	1 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON THE HELICOPTER [52][67]	1 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON THE HELICOPTER [52][67]	1 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON THE HELICOPTER [52][67]	39-J-25-21-01-00A-320A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
25-77	SEAT ABSORBERS (ZODIAC AEROSPACE)	DO A GVI OF THE FUSES SPACERS AND THE ABSORBERS PASSENGER DEVICES FOR CONDITION, CORROSION AND DAMAGE	2 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON THE HELICOPTER [42][67]	2 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON THE HELICOPTER [42][67]	2 YEAR / PRIOR THE INSTALLATION OF THE NEW COMPONENT ON THE HELICOPTER [42][67]	39-J-25-21-00-03A-310A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
25-78	SEAT ATTACHEMETNS (ZODIAC AEROSPACE)	DO A GVI OF THE SEAT ATTACHMENTS TO THE RAILS FOR WEAR, CORROSION AND CRACKS	2 YEARS [67]	2 YEARS [67]	2 YEARS [67]	39-J-25-21-00-03A-310A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT
25-98	ADELT WATER ACTIVATION SWITCH	DO AN OPERATIONAL CHECK OF THE ADELT WATER ACTIVATED SWITCH FOR CORRECT FUNCTIONING	3 YEARS [68]	3 YEARS [68]	3 YEARS [68]	39-A-25-64-03-00A-320A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT

							MODEL NOT FITTED ON AIRCRAFT
91-04	PITCH AND ROLL ACTUATORS	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF BUNDLES AND CONDITION, SAFETY AND SECURITY OF CONNECTORS	2 YEARS	2 YEARS	2 YEARS	39-A-91-10-00-00D-310A-A	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD

TABLE 3: LIST OF REQUIREMENTS FOR SCHEDULED / UNSCHEDULED MAINTENANCE CHECKS

NOTE:

20. TASK TO BE PERFORMED BY FOLLOWING THE INSTRUCTIONS CONTAINED IN THE LAST ISSUE OF THE P&W PT6C-67C AND PT6C-67C1 ENGINE MAINTENANCE MANUAL.

67. FOR INTERVAL TRANSITION, REFER FIGURE 3.

68. THIS CHECK IS APPLICABLE ONLY TO ADELTA WATER ACTIVATION SWITCH MODE STRIKE 3 AND SUBSEQUENT. TO IDENTIFY THE COMPONENT MOD STRIKE, REFER TO MOD RECORD LABEL INSTALLED ON THE SWITCH WHICH REPORTS THE MODIFICATION APPLIED TO THE COMPONENT.

NO	ITEM	TASK	CONDITION/LIMIT STANDARD	CONDITION/LIMIT PHASED	CONDITION/LIMIT PROGRESSIVE	REFERENCE	APPLICABILITY
25-07	RESCUE HOIST CABLE CUTTER ELECTRICAL CONNECTIONS	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF THE BUNDLES. CONNECTORS FOR CONDITIONN SAFETY AND SECURITY.	1 YEARS [2]	1 YEARS [2]	1 YEARS [2]	39-A-25-91-01-00C-310A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT

TABLE 4: LIST OF REQUIREMENTS – RESCUE HOIST SYSTEM (BREEZE)

NOTE:

2. FOR INTERNAL TRANSITION, REFER TO FIGURE 3.

NO	ITEM	TASK	CONDITION/LIMIT STANDARD	CONDITION/LIMIT PHASED	CONDITION/LIMIT PROGRESSIVE	REFERENCE	APPLICABILITY
25-27	RESCUE HOIST CABLE CUTTER ELECTRICAL CONNECTIONS	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF THE BUNDLES. CONNECTORS FOR CONDITIONN SAFETY AND SECURITY.	1 YEARS [5]	1 YEARS [5]	1 YEARS [5]	39-A-25-94-01-00C-310A-K	NOT APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-YPG, 9M-YTL,9M-BOC & 9M-BOD DUE TO COMPONENT P/N NOT FITTED ON AIRCRAFT

TABLE 5: LIST OF REQUIREMENTS – DOUBLE RESCUE HOIST SYSTEM (GOODRICH)

NOTE:

5. FOR INTERNAL TRANSITION, REFER TO FIGURE 3.

NO	ITEM	TASK	CONDITION/LIMIT STANDARD	CONDITION/LIMIT PHASED	CONDITION/LIMIT PROGRESSIVE	REFERENCE	APPLICABILITY
25-34	RESCUE HOIST CABLE CUTTER ELECTRICAL CONNECTIONS	DO A GVI TO DETECT CORROSION OR MECHANICAL DAMAGE OF THE BUNDLES. CONNECTORS FOR CONDITIONN SAFETY AND SECURITY.	1 YEARS [5]	1 YEARS [5]	1 YEARS [5]	39-A-25-96-01-00C-310A-K	APPLICABLE TO 9M-PMA, 9M-PMB, 9M-PMC, 9M-PMD, 9M-PME, 9M-PMF, 9M-BOC & 9M-BOD DUE TO COMPONENT P/N FITTED ON AIRCRAFT. NOT APPLICABLE TO 9M-YPG & 9M-YTL DUE TO COMPONENT NOT FITTED ON AIRCRAFT.

TABLE 6: LIST OF REQUIREMENTS – SINGLE RESCUE HOIST SYSTEM (GOODRICH)

NOTE:

4. FOR INTERNAL TRANSITION, REFER TO FIGURE 3.

NO	EVENT	REFERENCE (DMC)	INITIALS
1-23	DUST DEVIL ATMOSPHERIC EVENT	39-A-00-70-00-23A-28AA-A	-

TABLE 2: CONDITIONAL INSPECTIONS REQUIREMENTS