

maint	senanc	e , repai	r.ove	erhaul	_	www.galaxyae	rospace.my   en	quiry@gala:	xyaerospa	ce.my									
CLIENT/OWNER: SERIAL NO. HOURS LDG/CYCLE										WORKSH	EET NO:								
					AIF	RCRAFT		RE	FER W	ORKPACK			WORK/INSP/DESC: 300 FH INSPECTION						
AIRCRAF REGISTF					#1	ENGINE:		RE	FER W	ORKPACK			WORKPA	CK REF:					
BASE/FA					#2	ENGINE:		RE	FER W	ORKPACK			AJL REF NO.:						
DATE IN:			UT: RE	FER ORKPACK						NG / N1	N	= / N2	SHEET: 8 OF						
Reason				ONLINE						Raised b	y and d	ate:	Other req	uirements	s/inform	ation:			
Item						Descr	ription						Technicia	an *	Eng. Cl	RS	Date		
1	MI53-	05 TAIL S	TRUC	TURE AS	SEMBLY (	TAIL CONE	·	ELS UPF	PER ANI	D LOWER .	JOINT)								
	DETA	ILED INSF	PECTIO	ON FOR (	CRACKS (	BOTH INTE	RNAL AND I	EXTERNA	AL SIDE	S)									
	REFERENCE (DMC): 39-A-53-40-00-00A-31AB-A																		
	REMARKS:																		
1	1 MI53-06 TAIL STRUCTURE ASSEMBLY (TAIL CONE JOINT STA 11020)																		
	DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)																		
	DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)  REFERENCE (DMC): 39-A-53-40-00-00A-31AB-A																		
	REMARKS:																		
1	MI53-07 FUSELAGE STRUCTURE ASSEMBLY (FORWARD MGB ROD BACKUP STRUCTURE)																		
	DETAILED INSPECTION FOR CRACKS																		
	FROM	1 31003 To	O 3101	1.		NDED UP TO	(	5											
	2022)					-31AA-A, <mark>TM</mark>													
	REMA	ARKS: TAS	SK HAS	S BEEN E	XTENDEI	O TO 1300 F	H, REFER N	IOTE 11	(39-A-04	1-20-00-00 <i>i</i>	A-028E-P	")							
	1	08 FUSEL BER 3P53			RE ASSE	MBLY (AFT	MGB ROD E	BACKUP	STRUC	TURE AND	FITTING	S PART							
	DETA	ILED INSF	PECTIO	ON FOR (	RACKS														
	P/N 30		)135 O	R THE CF	RUCIFOR	IDED UP TC M FITTING F													
						IED, THE 30 711 OR P/N					FH IF TH	E							
	REFE 2022)		OMC):	39-A-53-1	0-00-00A	-31AA-A, <mark>TM</mark>	II139-548 (TI	MI EXPIR	RED DA	ΓΕ ON 29 S	EPTEME	BER							
	REMA	ARKS: TAS	SK HAS	S BEEN E	XTENDEI	D TO 1300 F	H, REFER N	IOTE 11	(39-A-0	4-20-00-00	A-028E-P	°)							
The	work re	ecorded	above	has bee	n carried	d out in acc	ordance w	ith the r	equiren	nents of th	ne Mala	ysian Civ	il Aviation F	Regulation	for the	time beir	ng in		
						ment is co								-					
						d out in acc						1							
				and in t	nat respe	ect the airc	ratt / equip	ment is	consid	ered tit fo	release	to servi	ce.						
TICK	WHER	E APPLICAE	BLE																
PARTS LABELED & RETURNED	D.D. RAISED	DUPLICATE INSP.	GROUN RUN	D FLIGHT TES	TORQUE CHK.	ADDITIONAL WORKSHEET	MONITORED DEFECT	PLANNING FORECAST			D.D. STATUS	AIRCRAFT LOG BOOK	ENGINE LOG BOOK	PROPELLER LOG BOOK	LOG CARD	OEM/COMP LOG CARD	MOD RECORD BOOK		



ARCAPT TYPE	maint	senance , repair , overhaul		erospace.my   en	quiry@galaxyaero	space.my								
ARCHER TYPE REGISTRATION: REFERENCE COLD SCHOOL  RESIDENCE  DESCRIPTION  REFERENCE COLD SCHOOL  DETAILED INSPECTION FOR CRACKS  DESCRIPTION  DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)  REFERENCE (DMC) 39-AS-34-CO-004-31A-A REFERENCE (DMC	CLIENT/0	DWNER:		SERIAL N	O. HOURS	LDG/	CYCLE		WORKSHEET NO:					
RECERTATION  ASSESSMENT OUT PROBLEM  2 ENGINE  RECERT WORKSPACE.  ALL REFT  BOST  ALL REFT  BOST  ALL REFT  ALL REFT  ALL REFT  BOST  ALL REFT  ALL REFT  ALL REFT  BOST  BO	VIDCD VE	T TVDE:	AIRCRAFT		REFER	WORKPACK			SOUTH INCIDENTIAL					
Reason for raising:    Reason for raising:   Reason for			#1 ENGINE:		REFER	WORKPACK								
Reason for nailing:  Reason for nailing:  Reason for nailing:  Research for nailing:  Resea			#2 ENGINE:		REFER	WORKPACK								
Rem	DATE IN:	REFER OUT: REFER WORKPACK				NG / N1	NF	7 N2	SHEET:	8	OF			
1 Mis3-06 FAIL STRUCTURE ASSEMBLY (TAIL CONE JOINT STA 11020)  DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)  REFERENCE (CMC): 39-A-53-40-00-JOA-31AB-A REMARKS:  1 Mis3-06 STRUCTURE ASSEMBLY (FORWARD MGB ROD BACKUP STRUCTURE)  DETAILED INSPECTION FOR CRACKS  REFERENCE (CMC): 39-A-53-10-00-JOA-31AB-A THI 199-548 NEW ISSUE EXPRIV SEPTEMBER 20TH 2022  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT SN FROM 31-00 TO 31011.  REMARKS: NOT APPLICABLE BY ARCRAFT SN AND EXTENDED TO 1300 FH INSPECTION  1 Mis3-06 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 395330A11251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PA 953500A130 CF THE CRUCIFORM FITTING RETTOMODIFICATION PAY 305306P93711 ARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3750 FH ARE READING. THE 300 FM INTERVAL IS REDUCED TO 150 FH IF THE RETOMODIFICATION PAY 305306P3321 IS NOT APPLIED.  REFERENCE (CMC): 39-A-53-10-00-DO-31AA-A TIM 139-548 NEW ISSUE EXPRIV SEPTEMBER 201H 2022  REMARKS:  □ The work recorded above has been carried out in accordance with the requirements of the Misignysian Civil Avisition Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  □ The work recorded above has been carried out in accordance with the requirements of the Misignysian Civil Avisition Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  □ The work recorded above has been carried out in accordance with the requirements of the Linguistic for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.	Reason	for raising:				Raised b	y and da	ate:	Other requ	uirements/	information:			
DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)  REFERENCE (DMC): 38 A 63-40-00-00A-31AB-A REMARKS:  1 MIS-307 FUSELAGE STRUCTURE ASSEMBLY (FORWARD MOB ROD BACKUP STRUCTURE)  DETAILED INSPECTION FOR CRACKS  REFERENCE (DMC): 39-A-33-10-00-00A-31AB-A TIME 133-S40 RIVINISULE EXPIREY SERFEMBER 281H 2022  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT SN FROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300 FH INTERVAL IS REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300 FH INTERVAL IS REMARKS: NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PN 306500X001/35 OR THE CRUCIPOWN FITTING RETROMODIFICATION PN 305300F38/711 ARE INSTALLED ON THE KOTORCASAT:  NOTE: WHEN THE 3750 FH ARE REAGHED. THE 30F FH INTERVAL IS REDUCED TO 130 FH IF THE RETROM CONTICATION NO 305300F38/71 OR PN 305300F38/711 OR PN 305300F38/711 OR PN 305300F38/711 ARE REFERENCE, (DMC): 39-A-31-10-00-00A-31AA-A TIME 1305-44 NEW ISSUE EXPIRE SEPTEMBER 281H 2022  REMARKS:  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 TICK / WHERE AREA CABLE  1 TICK	Item		Desc	ription					Technicia	n * E	Eng. CRS	Date		
REFERENCE (DMC): 39-A-53-40-00-00-31AB-A REMARKS:  1 MIS-307 FUSELACE STRUCTURE ASSEMBLY (PORWARD MGB ROD BACKUP STRUCTURE)  DETAILED INSPECTION FOR CRACKS  REFERENCE (DMC): 39-A-53-40-00-00-31AB-A (TMI IS9-SE IN EWI ISSUE EXPIRY SEPTEMBER 29TH 2022)  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT SIN FROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300FH INSPECTION  1 MIS3-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBERS 3PS300A1125)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PIN 365300X00135 GR THE CRUCTORM FITTING RETROMODIFICATION PIN 365300P30711 ARE INSTALLED ON THE ROTROCKAPT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETROMODIFICATION PIN 365300P30711 ARE INSTALLED ON THE ROTROCKAPT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETROMODIFICATION PIN 365300P3071 ARE INSTALLED ON THE ROTROCKAPT.  TOTIC WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETROMODIFICATION PIN 365300P3071 OR PIN 365300P30711 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31A-A-A  TIMI 130 FH NEW INSUE EXPIRY SEPTEMBER 20TH 2022  REMARKS:  1 THE work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to	1	MI53-06 TAIL STRUCTURE ASSEM	IBLY (TAIL CONE	JOINT STA	11020)									
REMARKS:  1 MIGS-OF FUSELAGE STRUCTURE ASSEMBLY (FORWARD MGB ROD BACKUP STRUCTURE)  DETAILED INSPECTION FOR CRACKS  REFERENCE (DMC): 39-A 63-10-00-00A-31AA-A  TIMI 135-648 NEW ISSUE EXPRY SEPTEMBER 29TH 2022  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT SN FROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300 FH INSPCTION  1 MIGS-OB FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART  NUMBER 3PS303A 125:1)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PIN 3GS30XX00135 OR THE CRUCIFORM FITTING RETROMODIFICATION PIN 3GS30BP30711 ARE  INSTALLED ON THE ROTORCRAFT  NOTE: WHEN THE 3750 FH ARE REACHED. THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE  REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A  TIMI 139-548 NEW ISSUE EXPRY SEPTEMBER 29TH 2022  REMARKS:  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 TICK WHERE APPLICABLE  1 DICK WHERE APPLICABLE  2 MS		DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)												
1 MIGS-07 FUSELAGE STRUCTURE ASSEMBLY (FORWARD MGB ROD BACKUP STRUCTURE)  DETAILED INSPECTION FOR CRACKS  REFERENCE (DMC): 39-A-53-10-00-00-31A-A.A  TMI 139-S8 NEW ISSUE EXPIRY SEPTEMBER, 29TH 2022  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT S.N  FROM 31003 TO 3101-1.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300/FH INSPCTION  1 MIGS-06 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 9F9330A1251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP 10 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PIN 36530X00135 OR THE CRUCIFORM FITTING RETROMODIFICATION PIN 363306936711 ARE INSTALLED ON THE ROTORCHAFT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION PIN 36330693071 OR PIN 365300939211 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A  TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER, 29TH 2022  REMARKS:  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 THE work recorded above has been carried out in accordance with the requirements of the Interval of the In														
DETAILED INSPECTION FOR CRACKS  REFERENCE (DMC): 39-A-S3-10-00-00A-31AA-A  TM 159-249 NEW I SSUE EXPIRY SEPTEMBER 23TH 2022  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT S/N FROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT S/N AND EXTENDED TO 1300 FH INSPECTION  1 MIS3-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB RDD BACKUP STRUCTURE AND FITTING PART NUMBER 395330A11251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PAY 305300X0315 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 305306998711 ARE INSTALLED ON THE ROTORCAFAT:  NOTE: WHEN THE 3750 FH ARE REACHED. THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETROMODIFICATION P/N 305306938711 IS NOT APPLIED.  REFERENCE (DMC): 39-A-S3-10-00-00A-31AA-A  TMI 139-348 NEW ISSUE EXPIRY SEPTEMBER 23TH 2022  REMARKS:  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 The work recorded above has been carried out in accordance with the requirements of the force in the time being in force and in that respect the aircraft / equipment is considered fit for release to service.														
REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A TMI 139-348 NEW ISSUE EXPIRY SEPTEMBER 29TH 2022  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT SN RROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300FH INSPCTION   1 MISS-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 3P5300A1251)  DETAILED INSPCTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PAN 3053000X015 OR THE CRUCIFORM FITTING RETROMODIFICATION PIN 305306P36711 ARE INSTALLED ON THE ROTORCRAFT:  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION PIN 365306P36711 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER 29TH 2022  REMARKS:  1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 THO work recorded above has been carried out in accordance with the requirements of the force and in that respect the aircraft / equipment is considered fit for release to service.	1 MI53-07 FUSELAGE STRUCTURE ASSEMBLY (FORWARD MGB ROD BACKUP STRUCTURE)													
NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT S/N FROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300/FH INSPCTION   1 MIS3-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 3/P5330A11251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PIN 3053000/03/30 OR THE CRUCIFORM FITTING RETROMODIFICATION PIN 305300P30211 IARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3/750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION PIN 305300P3021 TO R PIN 305300P30211 IS NOT APPLIED.  REFERENCE (DMC): 39-A 53-10-00-00A-31AA-A TIMI 139-548 NEW ISSUE EXPIRY SEPTEMBER 20TH 2022  REMARKS:   1 The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  1 The work recorded above has been carried out in accordance with the requirements of the 1 Tick ▼ WHERE APPLICABLE  1 TICK ▼ WHERE APPLICABLE  1 DID 1000000 TO 100000 TO 100000000000000000		DETAILED INSPECTION FOR CRACKS												
FROM 31003 TO 31011.  REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300FH INSPCTION    MI53-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 3P5330411251)   DETAILED INSPECTION FOR CRACKS   NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY P/N 365300X00135 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 3653086936711 ARE INSTALLED ON THE GROTORCHAFT     NOTE: WHEN THE 3750 FH ARR REACHED. THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 365306936711 OR P/N 365306936711 IS NOT APPLIED.    REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A														
Miss-op Fuselage structure assembly (AFT Mgb ROD Backup Structure and Fitting Part Number sps330A11251)   Detailed inspection for cracks														
NUMBER 3P5330A11251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PN 3G5300X00135 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 3G5306P36711 ARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 3G5306P39211 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31A-A  TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER 29TH 2022  REMARKS:   The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the Toto the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  **MESTATE**  **ROTATION FOR THE FUND FILE OF THE STATE OF		REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300FH INSPCTION												
NUMBER 3P5330A11251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PN 3G5300X00135 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 3G5306P36711 ARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 3G5306P39211 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31A-A  TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER 29TH 2022  REMARKS:   The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the Toto the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  **MESTATE**  **ROTATION FOR THE FUND FILE OF THE STATE OF					DP									
DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY P/N 3G530000135 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 3G5306P36711 ARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 3G5306P36711 OR P/N 3G5306P39211 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A  I'M 139-S48 NEW ISSUE EXPIRY SEPTEMBER.29TH 2022  REMARKS:  The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  MASS  D. DIPHICAE GROUND RIGHTIES TORUE ACCIONAL MONITORIS PLANING DARY SAULS D. ARCAPT BNORLED PROFILES LOG CRANCOM MODERCARE	1		ASSEMBLY (AFT	MGB ROD E	BACKUP STR	JCTURE AND	FITTING	PART						
NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY PN 3G5300X00135 OR THE CRUCIPORM PITTING RETROMODIFICATION P/N 3G5306P36711 ARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 3G5306P36711 OR P/N 3G5306P39211 IS NOT APPLIED.  REFERENCE (DMC): 339-A-53-10-00-00A-31AA-A  TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER;29TH 2022  REMARKS:   "The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requiremen		,	CKS	۹4 ٔ										
NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 3G5306P392711 OR P/N 3G5306P39211 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A  TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER,29TH 2022  REMARKS:  The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK WHERE APPLICABLE  PARTS  DD  DUPLICATE GROUND FIGHTES TOROUT ADDITIONAL MONITORID PLANING DIARY SIAIUS DD ARCRAFT ENGREDG PROPELLER LOG OFMICOMAP MODIFICIONE  MONITORING DIARRES IN THE BUSINESS DESCRIPTION OF THE BUSINESS DD ARCRAFT ENGREDG PROPELLER LOG OFMICOMAP MODIFICIONE  TOROUTE WHERE APPLICABLE  PARTS  DD  DUPLICATE GROUND FIGHTEST TOROUT ADDITIONAL MONITORID PLANING DIARY SIAIUS DD ARCRAFT ENGREDG PROPELLER LOG OFMICOMAP MODIFICIONE  MODIFICATION OF THE STATE OF THE		NOTE: THE 300 FH INTERVAL IS E P/N 3G5300X00135 OR THE CRUC	EXTENDED UP TO	•										
REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER,29TH 2022  REMARKS:   "The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  "The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK WHERE APPLICABLE  PARTS  DD. DUPLICATE (ROUND PLIGHTIES) TORQUE ADDITIONAL MONITORED PLANNING DUARY STATUS DD. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MODRECORD							H IF THE	≣						
*The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.    *The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.    *The work recorded above has been carried out in accordance with the requirements of the		RETRO MODIFICATION P/N 3G530	JOP 307 II OR P/N	1 365306739	211 15 NOT A	PPLIED.								
*The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  *The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK *WHERE APPLICABLE  PARTS DD DUPLICATE GROUND FLIGHTIEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS DD AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD				MBER,29TH	2022									
force and in that respect the aircraft / equipment is considered fit for release to service.   The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTIEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD		REMARKS:												
force and in that respect the aircraft / equipment is considered fit for release to service.   The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTIEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD														
force and in that respect the aircraft / equipment is considered fit for release to service.   The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTIEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD														
force and in that respect the aircraft / equipment is considered fit for release to service.   The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTTEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD														
force and in that respect the aircraft / equipment is considered fit for release to service.   The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTIEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD														
The work recorded above has been carried out in accordance with the requirements of the for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.  ☐ TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTIEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD							ne Malay	rsian Civil	Aviation R	egulation	for the time I	being in		
TICK ✓ WHERE APPLICABLE  PARTS D.D. DUPLICATE GROUND FLIGHTTEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD	☐ *The	work recorded above has been c	arried out in ac	cordance w	ith the requi	ements of th		to service	a.					
PARTS D.D. DUPLICATE GROUND FLIGHT TEST TORQUE ADDITIONAL MONITORED PLANNING DIARY STATUS D.D. AIRCRAFT ENGINE LOG PROPELLER LOG OEM/COMP MOD RECORD		-	. sopeot are unc	oquip		.35.56 11.101		307 910						



maint	enance , repair , overhaul	www.galaxyae	erospace.my   enquir	y@galaxyaeros	pace.my							
CLIENT/C	DWNER:		SERIAL NO. HOURS LDG/CYCLE					WORKSHEET NO:				
		AIRCRAFT		REFER V	VORKPACK			WORK/INSP/DESC: **1300 FH INSPECTION				
AIRCRAF REGISTR		#1 ENGINE:		REFER V	ER WORKPACK			WORKPACK REF:				
BASE/FA		#2 ENGINE:		REFER V	VORKPACK			AJL REF N	10.:			
	REFER OUT: REFER WORKPACK				NG / N1	NF	7 N2	SHEET:	1	OF		
	for raising:				Raised by	/ and da	ate:	Other req	uirement	ts/informat	on:	
Item		Desc	ription					Technicia	ın ,	* Eng. CRS		Date
1	MI53-07 FUSELAGE STRUCTURE		•	OD BACKUF	STRUCTUR	E)						Date
	DETAILED INSPECTION FOR CRA	CKS										
	REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A  (TMI 139-548 NEW ISSUE EXPIRY SEPTEMBER,29TH 2022)											
	NOTE: THE 300 FH INTERVAL IS E FROM 31003 TO 31011.	۷										
	FROM 31003 TO 31011.  REMARKS:											
1	1 MI53-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 3P5330A11251)											
	DETAILED INSPECTION FOR CRA	CKS		" &	-							
	NOTE: THE 300 FH INTERVAL IS E P/N 3G5300X00135 OR THE CRUC INSTALLED ON THE ROTORCRAF	IFORM FITTING										
	NOTE: WHEN THE 3750 FH ARE R RETRO MODIFICATION P/N 3G530					H IF THE	≣					
	REFERENCE (DMC): 39-A-53-10-00 TMI 139-548 NEW ISSUE		MBER 29TH 201	22								
	11011 103 040 142 17 10002		WDLN,20111202	<b></b>								
	REMARKS:	()										
1	MI53-09 FUSELAGE STRUCTURE	ASSEMBLY (ANT	TITORQUE BEA	M BACKUP	STRUCTURE	)						
	DETAILED INSPECTION FOR CRAI	CKS										
	REFERENCE (DMC): 39-A-53-10-00	)-00A-31AB-A										
	REMARKS:											
force The	work recorded above has been cand in that respect the aircraft / ework recorded above has been cane time being in force and in that	equipment is co arried out in acc	ensidered fit fo cordance with	or release to the require	service. ements of the	e			egulatio	n for the tir	ne being	g in
	WHERE APPLICABLE	, , , , , ,	- darle con									
			1101						05			
PARTS LABELED &		RQUE ADDITIONAL CHK. WORKSHEET		PLANNING DI FORECAST UP	ARY STATUS DATE UPDATE	D.D. STATUS	AIRCRAFT LOG BOOK	ENGINE LOG BOOK	PROPELLER LOG BOOK		EM/COMP LOG CARD	MOD RECORD BOOK



	senance , repair , overhaul		34 7226   Fax: +60 erospace.my   end		erospace.	my								
CLIENT/0	DWNER:		SERIAL NO	D. HOUI	RS	LDG/0	CYCLE		WORKSHEET NO:					
AIDODAE	T.T./DE	AIRCRAFT		REFE	R WOR	RKPACK			WORK/INSP/DESC: 300 FH INSPECTION					
AIRCRAF REGISTF		#1 ENGINE:		REFE	R WOR	RKPACK			WORKPACK REF:					
BASE/FA		#2 ENGINE:		REFE	R WOR	WORKPACK			AJL REF NO.:					
DATE IN:	REFER OUT: REFER WORKPACK					NG / N1	NF	/ N2	SHEET:	8	OF			
Reason 1	for raising: NSPECTION (AF+ENG) TO BE CARR							ate:	Other req	uirements	/inform	ation:		
Item		Desc	cription						Technicia	an *	Eng. CF	RS	Date	
1	MI53-06 TAIL STRUCTURE ASSEM	MBLY (TAIL CON	E JOINT STA	11020)										
	DETAILED INSPECTION FOR CRACKS (BOTH INTERNAL AND EXTERNAL SIDES)													
	REFERENCE (DMC): 39-A-53-40-00													
	REMARKS:													
1	MI53-07 FUSELAGE STRUCTURE													
	DETAILED INSPECTION FOR CRA	CKS					•	0	2					
	REFERENCE (DMC): 39-A-53-10-00 : TMI NO. 139-548, NEW I		EPTEMBER 2	9TH 2022			4	X						
	NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT S/N FROM 31003 TO 31011.													
	REMARKS:													
				3										
1	MI53-08 FUSELAGE STRUCTURE NUMBER 3P5330A11251)	ASSEMBLY (AFT	F MGB ROD B	ACKUP ST	ructu	JRE AND I	FITTING	PART						
	DETAILED INSPECTION FOR CRA	cks												
	NOTE: THE 300 FH INTERVAL IS E P/N 3G5300X00135 OR THE CRUC INSTALLED ON THE ROTORCRAF	FORM FITTING												
	NOTE: WHEN THE 3750 FH ARE R RETRO MODIFICATION P/N 3G530						H IF THE	Ī						
	REFERENCE (DMC): 39-A-53-10-00		MBER 20TH	2022										
	REMARKS:	- LA INTOLITE												
	work recorded above has been c and in that respect the aircraft /						e Malay	sian Civi	Aviation R	Regulation	for the	time bein	ıg in	
☐ *The	work recorded above has been cone time being in force and in that	arried out in ac	cordance wi	th the req	quireme	nts of the		to service	 ce.					
	WHERE APPLICABLE													
PARTS LABELED &	D.D. DUPLICATE GROUND FLIGHTTEST TO	DRQUE ADDITIONAL CHK. WORKSHEET	MONITORED DEFECT	PLANNING FORECAST	DIARY UPDATE	STATUS UPDATE	D.D. STATUS	AIRCRAFT LOG BOOK	ENGINE LOG BOOK	PROPELLER LOG BOOK	LOG CARD	OEM/COMP LOG CARD	MOD RECORD BOOK	
RETURNED		· · · · · · · · · · · · · · · · · · ·	52.20		U1 W111 L	U. Delle		1 -50 5000	1 2001	1 -50 500%	0.00	-50 O IIID	1	



maint	senance , repair , overhaul		erospace.my   en	quiry@galaxyaero:	space.my								
CLIENT/C	DWNER:		SERIAL NO	O. HOURS	LDG/	CYCLE		WORKSHEET NO:					
AIRCRAF	T TYPE: AW139	AIRCRAFT		REFER	WORKPACK			WORK/INSP/DESC: **1300 FH INSPECTION					
REGISTR		#1 ENGINE:		REFER	WORKPACK			WORKPAC					
BASE/FA		#2 ENGINE:		REFER	WORKPACK				AJL REF NO.: SHEET: 1 OF				
	REFER OUT: REFER WORKPACK				NG / N1	NF	7 N2	011221.	'	OF			
1300 FH	for raising: SCHEDULED INSPECTION TO BE CA MO.AMP.AW139 ISSUE 1 REV 0 07J.		W		Raised b	y and da	ate:	Other requ	uirements	s/information			
Item		Desc	ription					Technicia	ın *	Eng. CRS	Date		
1	MI53-07 FUSELAGE STRUCTURE	ASSEMBLY (FOR	RWARD MGB	ROD BACKU	P STRUCTUR	E)							
	DETAILED INSPECTION FOR CRAC	CKS											
	REFERENCE (DMC): 39-A-53-10-00 : TMI NO. 139-548, NEW IS		EPTEMBER 2	9TH 2022									
NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH FOR ALL HELICOPTERS EXCEPT S/N FROM 31003 TO 31011.													
	REMARKS: NOT APPLICABLE BY AIRCRAFT SN AND EXTENDED TO 1300FH												
1	MI53-08 FUSELAGE STRUCTURE A NUMBER 3P5330A11251)												
	DETAILED INSPECTION FOR CRACKS												
	NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY P/N 3G5300X00135 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 3G5306P36711 ARE INSTALLED ON THE ROTORCRAFT.												
	NOTE: WHEN THE 3750 FH ARE RI RETRO MODIFICATION P/N 3G530					H IF THE	E						
	REFERENCE (DMC): 39-A-53-10-00 TMI 139-548 NEW ISSUE	_	MBER,29TH	2022)									
	REMARKS:	00											
1	MI53-09 FUSELAGE STRUCTURE	ASSEMBLY (ANT	TTORQUE BE	EAM BACKUP	STRUCTURE	<del>.</del> )							
	DETAILED INSPECTION FOR CRAC	CKS											
	REFERENCE (DMC): 39-A-53-10-00	-00A-31AB-A											
	REMARKS:												
*The	work recorded above has been as	erried out in con	oordanaa	ith the requir	oments of th	e Mala	reign Civil	Aviation	earlotica	for the time	heing in		
force The	work recorded above has been ca and in that respect the aircraft / e work recorded above has been ca	equipment is co arried out in acc	onsidered fit cordance w	for release t	o service. ements of th	e			egulation	i ioi tile time			
	ne time being in force and in that i	respect the airc	raft / equip	ment is cons	idered fit for	release	to servic	e.					
TICK	WHERE APPLICABLE												
PARTS LABELED &		RQUE ADDITIONAL HK. WORKSHEET	MONITORED DEFECT		IARY STATUS PDATE UPDATE	D.D. STATUS	AIRCRAFT LOG BOOK	ENGINE LOG BOOK	PROPELLER LOG BOOK		/COMP MOD RECOR CARD BOOK		



maint	tenance , repair , overhaul	www.galaxyae	erospace.my   enqui	ry@galaxyaeros	space.my								
CLIENT/0	OWNER:		SERIAL NO.	L NO. HOURS LDG/CYCLE					WORKSHEET NO:				
		AIRCRAFT		REFER	WORKPACK			WORK/INSP/DESC: 300 FH INSPECTION					
AIRCRAF		#1 ENGINE:		REFER	WORKPACK			WORKPACK REF:					
REGISTR BASE/FA		#2 ENGINE:		REFER	WORKPACK			AJL REF	10.:				
	REFER OUT: REFER WORKPACK				NG / N1	NF	: / N2	SHEET:	8	OF			
	for raising:				Raised b	y and da	ate:	Other req	uirement	ts/informat	tion:		
Item		Desc	ription					Technicia	ın ,	* Eng. CRS	8	Date	
1	MI53-05 TAIL STRUCTURE ASSEM		<u>'</u>	S UPPER A	ND LOWER J	IOINT)							
						,							
	DETAILED INSPECTION FOR CRA	ACKS (BOTH INTE	RNAL AND EX	TERNAL SI	DES)								
	REFERENCE (DMC): 39-A-53-40-0	0-00A-31AB-A											
	REMARKS:												
1	MI53-06 TAIL STRUCTURE ASSEM	MBLY (TAIL CONE	JOINT STA 11	020)				0)					
	DETAILED INSPECTION FOR CRA	ACKS (BOTH INTE	RNAL AND EX	TERNAL SI	DES)		<b>\</b> '-						
	REFERENCE (DMC): 39-A-53-40-0												
	REMARKS:												
1	MI53-07 FUSELAGE STRUCTURE ASSEMBLY (FORWARD MGB ROD BACKUP STRUCTURE)												
	DETAILED INSPECTION FOR CRACKS												
	NOTE : THE 300 FH INTERVAL IS	EXTENDED UP T	O 1300 FH FO	R ALL HELIC	COPTERS EX	CEPT S/N	N						
	FROM 31003 TO 31011.			2									
	REFERENCE (DMC): 39-A-53-10-0 2022)	0-00A-31AA-A, <mark>TN</mark>	/II139-548 (TMI	EXPIRED D	ATE ON 29 S	EPTEMB	SER)						
	REMARKS: TASK HAS BEEN EXT	ENDED TO 1300 F	H, REFER NO	TE 11 (39-A	-04-20-00-00 <i>P</i>	A-028E-P)	)						
		200											
	•	)											
	*The work recorded above has been carried out in accordance with the requirements of the Malaysian Civil Aviation Regulation for the time being in force and in that respect the aircraft / equipment is considered fit for release to service.												
_	work recorded above has been of					10							
	he time being in force and in that						to service	e.					
	WHERE APPLICABLE												
PARTS LABELED &		ORQUE ADDITIONAL CHK. WORKSHEET			PDATE UPDATE	D.D. STATUS	AIRCRAFT LOG BOOK	ENGINE LOG BOOK	PROPELLER LOG BOOK		OEM/COMP LOG CARD	MOD RECORD BOOK	



	senance , repair , overhaul		Fax: +603 7 rospace.my   enquir		ace.my							
CLIENT/0	DWNER:		SERIAL NO.	HOURS	LDG/C	YCLE	WORKSH	WORKSHEET NO: WORK/INSP/DESC: 300 FH INSPECTION WORKPACK REF:				
AIDODAE	T TVDE.	AIRCRAFT		REFER W	ORKPACK		WORK/IN					
AIRCRAF REGISTR		#1 ENGINE:		REFER W	ORKPACK		WORKPA					
BASE/FA		#2 ENGINE:		REFER W	ORKPACK		AJL REF	NO.:				
DATE IN:	REFER OUT: REFER WORKPACK				NG / N1	NF / N	2 SHEET:	9	OF			
Reason f	or raising:				Raised by	and date:	Other rec	uirements,	/information:			
Item		Desc	ription				Technicia	an * I	Eng. CRS	Date		
1	MI53-08 FUSELAGE STRUCTURE ASSEMBLY (AFT MGB ROD BACKUP STRUCTURE AND FITTING PART NUMBER 3P5330A11251)  DETAILED INSPECTION FOR CRACKS  NOTE: THE 300 FH INTERVAL IS EXTENDED UP TO 1300 FH IF THE FUSELAGE STRUCTURE ASSEMBLY P/N 3G5300X00135 OR THE CRUCIFORM FITTING RETROMODIFICATION P/N 3G5306P36711 ARE INSTALLED ON THE ROTORCRAFT.  NOTE: WHEN THE 3750 FH ARE REACHED, THE 300 FH INTERVAL IS REDUCED TO 150 FH IF THE RETRO MODIFICATION P/N 3G5306P36711 OR P/N 3G5306P39211 IS NOT APPLIED.  REFERENCE (DMC): 39-A-53-10-00-00A-31AA-A, TMI139-548 (TMI EXPIRED DATE ON 29 SEPTEMBER 2022)  REMARKS: TASK HAS BEEN EXTENDED TO 1300 FH, REFER NOTE 11 (39-A-04-20-00-00A-028E-P)											
1	FO OO TAIL ACCEMBLY											
	53-38 TAIL ASSEMBLY  DO A DI (HAMMER TAPPING CHECK) FOR SIGNS OF DEBONDING  REFERENCE (DMC): 39-A-53-40-00-00A-31AJ-A  REMARKS:											
1	DO AN OC FOR PLAY (NO REMOVAL AND QUANTITATIVE MEASUREMENT REQUIRED). IF UNUSUAL PLAY IS FELT, REMOVE ROD ASSEMBLY AND PERFORM A DI FOR CONDITION, DAMAGE AND PLAY (RADIAL PLAY CHECK ONLY) (MAXIMUM PLAY ALLOWED 0.3 MM - 0.0118 IN)  REFERENCE (DMC): 39-A-55-11-00-00A-320A-A; 39-A-55-11-02-00A-31AA-B; 39-A-55-11-03-00A-31AA-B REMARKS:											
1	62-03 MAIN ROTOR BLADE TIP AN	ID MASS BALANO	CE WEIGHT PO	CKET COVE	R AREA							
	DO A GVI FOR DEBONDING AND I	DAMAGE (BLADE	REMOVAL NO	T REQUIRED	))							
	REFERENCE (DMC): 39-A-62-11-00	0-00B-310A-A										
	REMARKS:											
force *The for th	work recorded above has been c and in that respect the aircraft / e work recorded above has been c ne time being in force and in that	equipment is co arried out in acc respect the airc	ensidered fit for cordance with	r release to the requirer	service. ments of the	)		Regulation	for the time b	eing in		
PARTS LARFLED &	D.D. DUPLICATE GROUND FLIGHTTEST TO	DRQUE ADDITIONAL WORKSHEET	MONITORED PI	LANNING DIAR	RY STATUS LIPDATE	D.D. AII	RCRAFT ENGINE LOG	PROPELLER LOG BOOK	LOG OEM/COI	MP MOD RECORD		



	tenance, repair, overhaul		4 7226   Fax: +603 erospace.my   enqu		pace.my							
CLIENT/0	OWNER:		SERIAL NO.	HOURS	LDG/C	YCLE	W	WORKSHEET NO:				
AIRCRAF	T TYPE:	AIRCRAFT		REFER W	VORKPACK			WORK/INSP/DESC: **1300 FH INSPECTIO				
REGISTE		#1 ENGINE:		REFER W	VORKPACK			WORKPACK REF:				
BASE/FA		#2 ENGINE:		REFER W	VORKPACK			JL REF N				
DATE IN:	REFER OUT: REFER WORKPACK				NG / N1	NF/N	N2 SI	HEET:	1	OF		
Reason	for raising:				Raised by	and date	: 0	ther requ	uirements/	/informatio	n:	
Item		Desc	ription				Т	echnicia	n * E	Eng. CRS		Date
1	MI53-07 FUSELAGE STRUCTURE	ASSEMBLY (FOR	RWARD MGB R	OD BACKUP	STRUCTURE	≣)						
	DETAILED INSPECTION FOR CRA	ACKS										
	REFERENCE (DMC): 39-A-53-10-0 : TMI NO. 139-548, NEW I		EPTEMBER 29	TH 2022								
	NOTE : THE 300 FH INTERVAL IS FROM 31003 TO 31011.											
	REMARKS:			0								
1	MI53-08 FUSELAGE STRUCTURE NUMBER 3P5330A11251)	TTI, GP	RT				+					
	DETAILED INSPECTION FOR CRA											
	NOTE: THE 300 FH INTERVAL IS E P/N 3G5300X00135 OR THE CRUC INSTALLED ON THE ROTORCRAF	IBLY										
	NOTE: WHEN THE 3750 FH ARE R RETRO MODIFICATION P/N 3G530	,				H IF THE						
	REFERENCE (DMC): 39-A-53-10-0 TMI 139-548 NEW ISSUE		M⊾	)22								
	REMARKS:	00	7.									
1	MI53-09 FUSELAGE STRUCTURE	ASSEMELY (ANT	TTORQUE BEA	M BACKUP	STRUCTURE)							
	DETAILED INSPECTION FOR CRA	ACKS										
	REFERENCE (DMC): 39-A-53-10-0	0-00A-31AB-A										
	REMARKS:											
force	work recorded above has been ce and in that respect the aircraft /	equipment is co	onsidered fit fo	or release to	service.		an Civil Av	viation Re	egulation	for the time	e being	in
	work recorded above has been on the time being in force and in that						service.					
TICK	✓ WHERE APPLICABLE	ı		ı						,		
PARTS LARFLED &	D.D. DUPLICATE GROUND FLIGHTTEST TO	ORQUE ADDITIONAL WORKSHEET	MONITORED	PLANNING DIA		D.D. A	AIRCRAFT E	NGINE LOG	PROPELLER LOG ROOK	LOG OEN	M/COMP G CARD	MOD RECORI