

SAFETY NOTICE

S.N. N° DATE REV.

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AW 189 CRASH CARD

GROUND EMERGENCY AND RESCUE OPERATIONS



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<u>INTRODUCTION</u>

The purpose and objective of this card is to support ground personnel to respond to accident or incident crash-rescue operations on AW189 within their capability and training and to be able to rescue survivors of a crash in a safe, efficient manner. This crash card is not intended to cover every contingency which may arise, nor does the card detail every safety emergency ingress and egress practice. Specialized basic aircraft firefighting training should be sought to supplement the information contained herein.

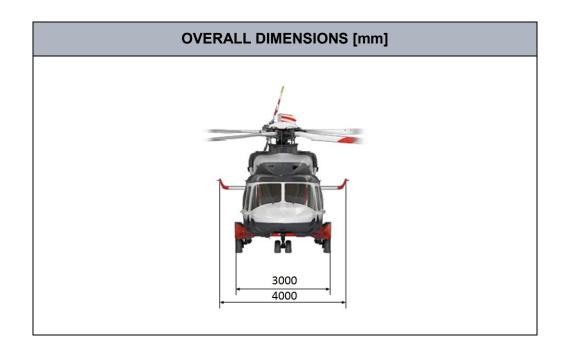
It is intended that the AW189 configuration herein covered is generic and it is under the Operators' responsibility to estimate the applicability on their current configurations.

Moreover, this document will not be updated to be personalized for each operator in terms of both language and configuration used.

GENERAL INFORMATION

WEIGHT	[kg]
Empty Weight	5120-6120
Max Take off Weight	8600

OCCUPANCY	[pax]
Max Crew (Cockpit)	2
Max Passengers (Cabin)	Max 16/19



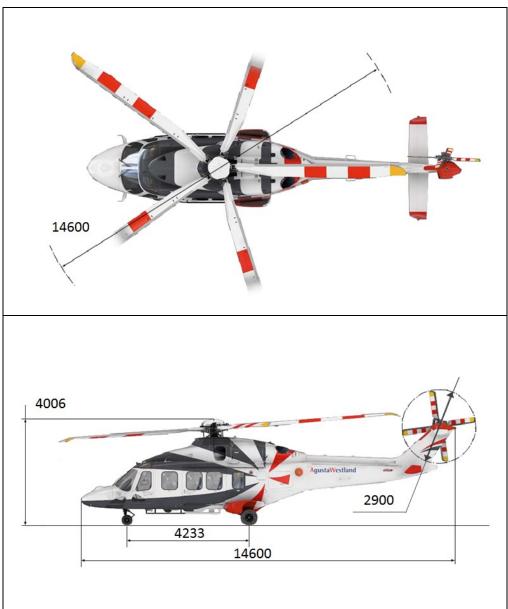
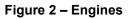


Figure 1 – Overall Dimensions [mm]

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7 (W 165 OI WOIT OF WILD STOOM	D EMEROLITOT
POWERPLANTS	#
GE CT7-2E1 or Safran Aneto-1K	2
APU Microturbo eAPU60H (between the engines)	1
APU Microturbo eAPU60H (between the engines) 1	



OIL	Capacity [I]
Engine	6.4 (Safran) 6.9 (GE)
APU	2.8
MGB	27
TGB	1.22
IGB	1.87
Hydraulic system – max nominal pressure 207 bar	

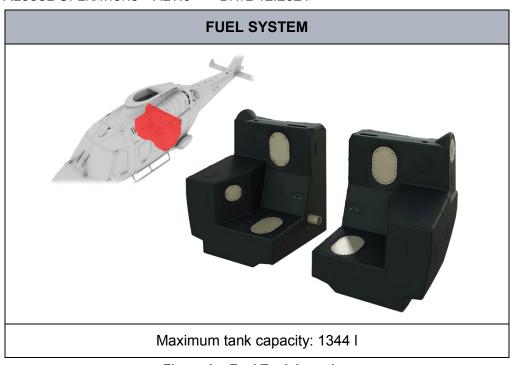


Figure 3 – Fuel Tank Location

AUXILIARY FUEL TANK INSTALLATION (KIT - STANDARD CONFIGURATION) Auxiliary Central Tank (capacity 510 I) Two Sub Floor Tanks (RH and LH – total capacity 270 I)

Figure 4 – Auxiliary Fuel Tank Location – Standard Configuration Page 8 of 39

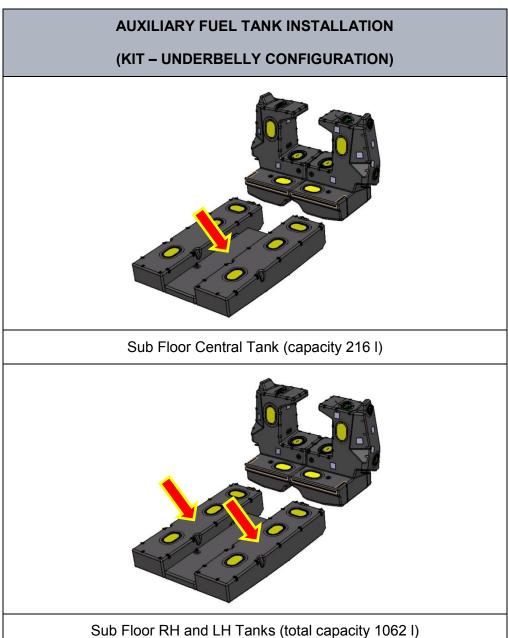


Figure 5 – Auxiliary Fuel Tank Location – Underbelly Configuration

STRUCTURAL MATERIALS

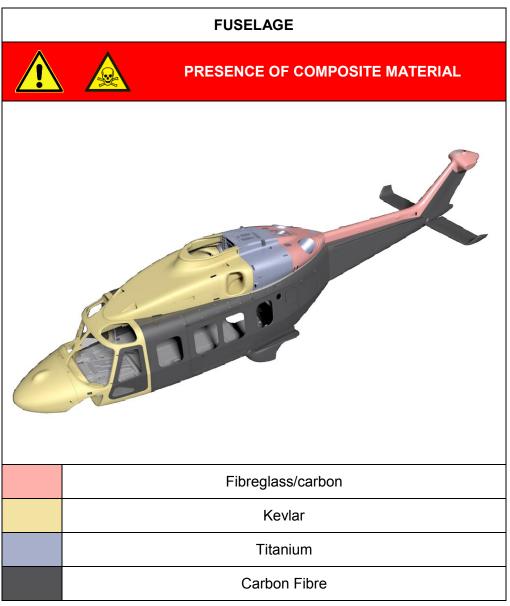


Figure 6 – Structural Materials – Fuselage

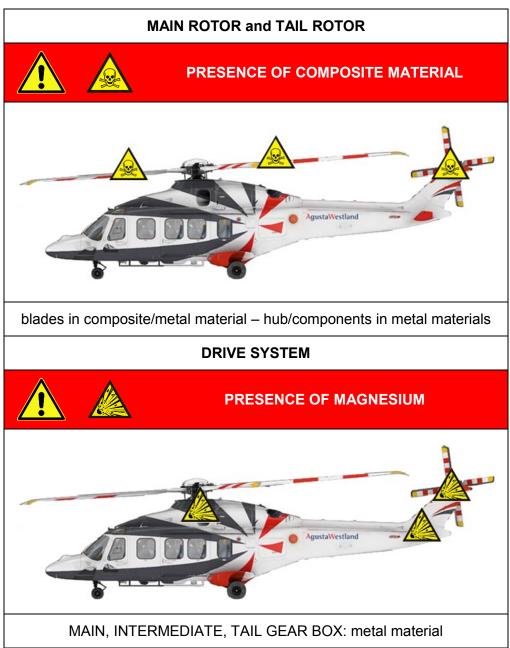


Figure 7 – Structural Materials

HAZARDS

HAZARDS - APPROACH CONES

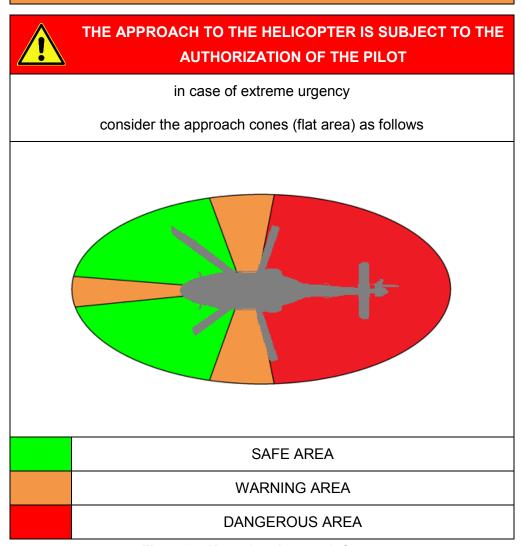


Figure 8 – Hazards – Approach Cones

HAZARD - FRAGMENTATION IN CASE OF WHEEL FIRE DEBRIS FROM RIMS AND TIRES MAY POSE A FRAGMENATION HAZARD PRESSURIZED WHEELS (~5.9 BAR) MLG Right Wheel NLG approach path MLG Left Wheel **APPROACH PATH** DANGEROUS AREA

Figure 9 - Hazards - Fragmentation

HAZARDS – PRESSURIZED RECIPIENTS



CHECK THE INTEGRITY OF THE AREAS ADJACENT TO THE PRESSURIZED RECIPIENTS AND APPROACH AS APPROPRIATE

AND ALL ROADIL AD ALL ROLLING		
SYSTEM (number of bottles)	REF IN FIGURE	Pressure [bar] @21 C
Floating (2)	1	>344
Liferaft (2)	2	>270
Engine Fire Extinguisher (2)	3	>24 (43 aneto)
APU Fire Extinguisher (1)	4	>24
Wheels (3)	-	~5.9
Cabin Fire Extinguisher (1 up to 3)	Inside cockpit (next	9
Oxygen Bottles (1 up to 3)	to pilot's seat) and cabin, depending on configurations	200



Figure 10 – Hazards – Pressure Recipients

HAZARDS - EXPLOSIVE CHARGES

PRESENCE OF EXPLOSIVE CHARGES APPROACH AS APPROPRIATE		
ACTIVATION (number of charges) REF IN FIGURE		FIGURE
Rescue Hoist (1)	1	right side
Cargo Hook (1)	2	bottom side
Engine Fire Extinguisher (2)	3	-
APU Fire Extinguisher (1)	3	In between



Figure 11 – Hazards – Explosive Charges

HAZARD - BATTERIES



BATTERIES MAY EXPLODE WITH FIRE



POSSIBLE RELEASE OF TOXIC GASES

BATTERY	REF IN FIGURE	
Main (1)	1	Nose Avionics Bay
Emergency Locator Transmitter, ELT (1) / ADELT (1)	2	Tail / Rear LH side of the fuselage
ADI STBY Emergency (1)	3	RH side of Nose Avionics Bay
Emergency Lights System Battery Pack (1)	4	Nose Avionics Bay
Helicopter Emergency Egress Lighting System Battery packs (6)	(5)	Inside Doors under window, under the two small windows of LH side of HC – depending on configuration
Flight Data Recorder (1)	6	Rear Avionic Bay
Life Raft ELT (2), Sea Light (2), Torch (4)	7	Sponsons



Figure 12 – Hazards – Batteries Location

HAZARDS – HOT SURFACES

PRESENCE OF HOT SURFACES	
SURFACE	REF IN FIGURE
Pitots (2, see Figure 16)	1
Engine exhausts (2)	2
APU exhaust (1)	3
Air intake (2)	4
Latches, handles, metallic components (in case of fire)	access doors/panels

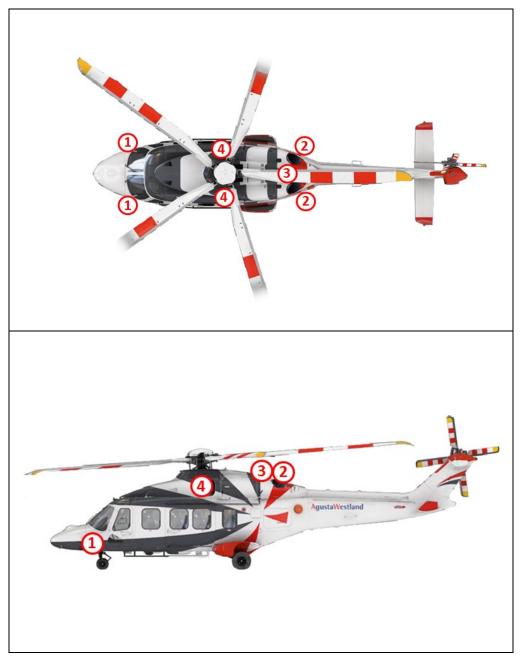


Figure 13 – Hazards – Hot Surfaces

HAZARD - FUEL AND LIQUIDS IN HYDRAULIC SYSTEM

PRESENCE OF FLAMMABLE LIQUIDS MAX NOMINAL PRESSURE 207 BAR IN HYDRAULIC SYSTEM

FUEL TANK



max tank capacity: 1344 I (see Figure 3)

AUXILIARY FUEL TANK - KIT STANDARD CONFIGURATION



Auxiliary Central Tank (capacity 510 I, see Figure 4)

Sub Floor Tanks (total capacity 270 I, see Figure 4)

AUXILIARY FUEL TANK - KIT UNDERBELLY CONFIGURATION



Sub Floor Central Tank (capacity 216 I, see Figure 5)

Sub Floor Tanks (total capacity 1062 I, see Figure 5)

Figure 14 – Hazards – Fuel Location

HAZARDS – APU HOT SURFACES POSSIBLE RELEASE OF TOXIC GASES

Figure 15 – Hazards – APU

HAZARDS - PITOTS

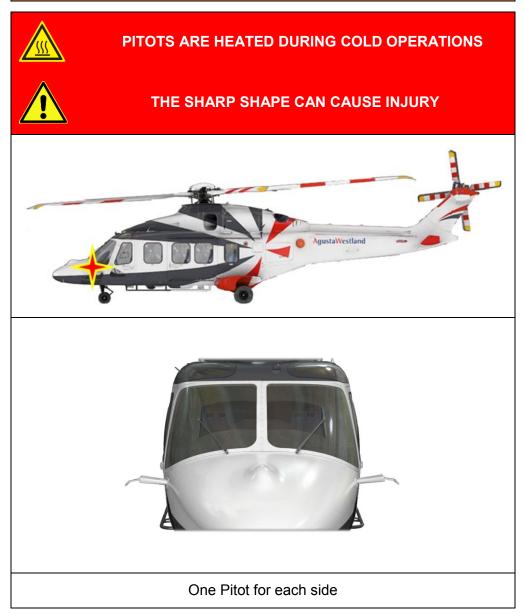


Figure 16 - Hazards - Pitots

HAZARDS - EMERGENCY FLOATATION SYSTEM (EFS)

THE FRONT AND BACK BALOONS MAY INFLATE SUDDENLY THE INFLATION BOTTLES INSTALLED RIGHT AFTER THE **SPONSONS HAVE A PRESSURE ABOVE 344 BAR**

Figure 17 – Hazards – Emergency Floatation System

Two Baloons for each side. See Figure 43 for the de-activation procedure

HAZARDS – EMERGENCY LIFE RAFTS



Figure 18 – Hazards – Emergency Life Rafts

HAZARDS - EMERGENCY LOCATION TRANSMITTER (ELT)

THE ELT BEACON MAY DEPLOY SUDDENLY One ELT Beacon for the Helicopter

Figure 19 – Hazards – Emergency Location Transmitter

HAZARDS - IGNITER BOXES



Figure 20 - Hazards - Igniter Boxes

SAFETY INFORMATION: GROUND STAFF (OUTSIDE

THE HELICOPTER)



IT IS RECOMMENDED TO APPROACH PERSONNEL NOT
ADEQUATELY TRAINED ON GENERAL RISKS OR
HELICOPTER EMERGENCY MEASURES



PERSONNEL IS REQUIRED TO WEAR THE APPROPRIATE
PERSONAL PROTECTIVE EQUIPMENT



USE CAUTION WHEN APPROACHING THE HELICOPTER,
CHECKING THE STRUCTURE INTEGRITY



THE APPROACH OF THE HELICOPTER IS NOT ALLOWED IN CASE OF ANY POSSIBILE RECOVERY OR IGNITION OF FLAMES. VERIFY THE ABSENCE OF SPILLS OF FLUIDS AND FUEL



IN THE EVENT OF SMOKE, FLAMES, SPARKS FIRE
FIGHTING TRAINED PERSONNEL ONLY IS ALLOWED TO
OPERATE



IT MIGHT BE NECESSARY TO WEAR THE SELF-CONTAINED BREATHING APPARATUS



POSSIBLE PRESENCE OF STATIC ELECTRICITY ON THE HELICOPTER



ELECTRICALLY GROUND THE HELICOPTER IF POSSIBLE

ACCESS INTO THE HELICOPTER

AgustaWestland

NORMAL ACCESS DOORS

USE NORMAL ACCESS

IF THE NORMAL ACCESS CANNOT BE USED, ACT ON THE EMERGENCY ACCESS DOORS

Pilot and Copilot Doors (LH/RH) - Passenger Doors (RH/LH)

EMERGENCY ACCESS DOORS



Pilot and Copilot Emergency Windows (LH/RH)

Passenger Emergency Windows (4 RH/ 4 LH) (*)

(*) STC configurations may differ from that reported in this document

Figure 21 – Normal and Emergency Access Doors Scheme

NORMAL ACCESS - OPEN THE PILOT/CO-PILOT DOOR

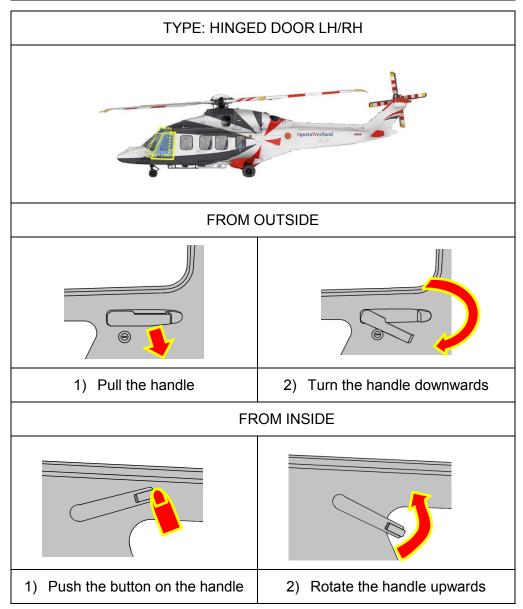


Figure 22 – Pilot/Co-Pilot Door - Opening Procedure

EMERGENCY ACCESS - OPEN THE PILOT/CO-PILOT EMERGENCY EXITS

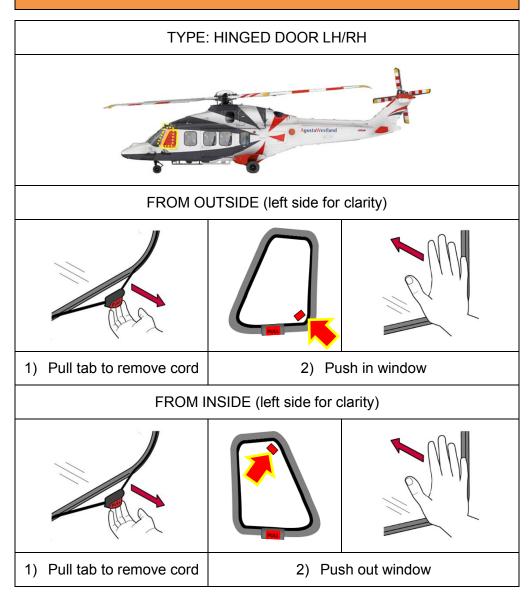
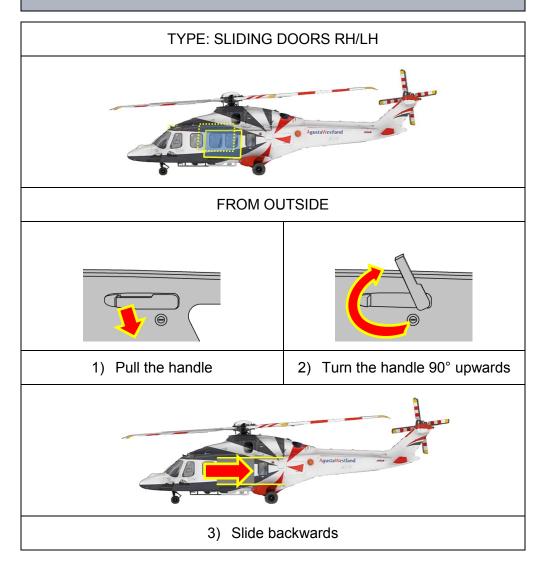


Figure 23 – Pilot/Co-Pilot Emergency Exits - Opening Procedure

NORMAL ACCESS - OPEN THE PASSENGER DOOR



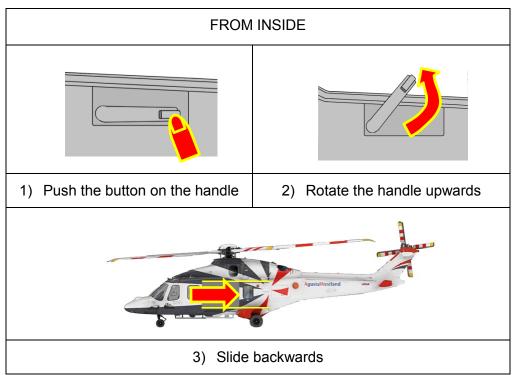
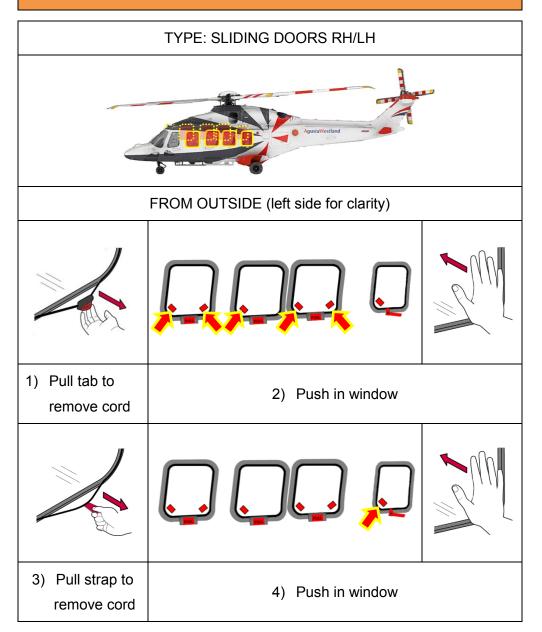


Figure 24 – Sliding Passenger Door - Opening Procedure

EMERGENCY ACCESS - OPEN THE PASSENGER EMERGENCY EXITS



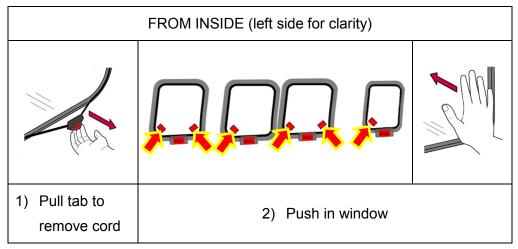


Figure 25 – Sliding Passenger Emergency Exits - Opening Procedure

NORMAL ACCESS - OPEN THE PASSENGER DOOR

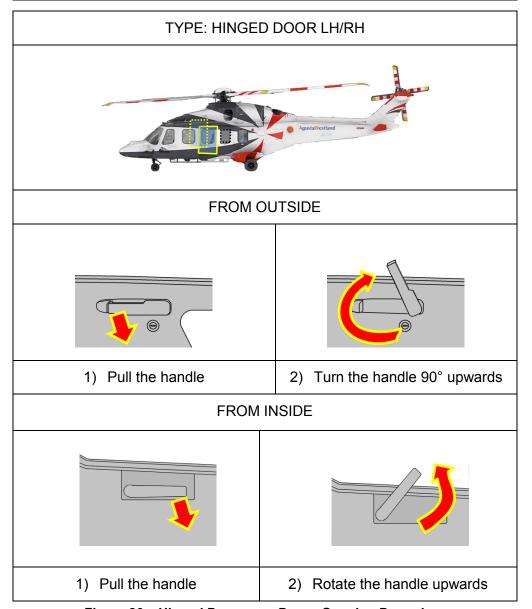


Figure 26 – Hinged Passenger Door - Opening Procedure

EMERGENCY ACCESS - OPEN THE PASSENGER EMERGENCY EXITS

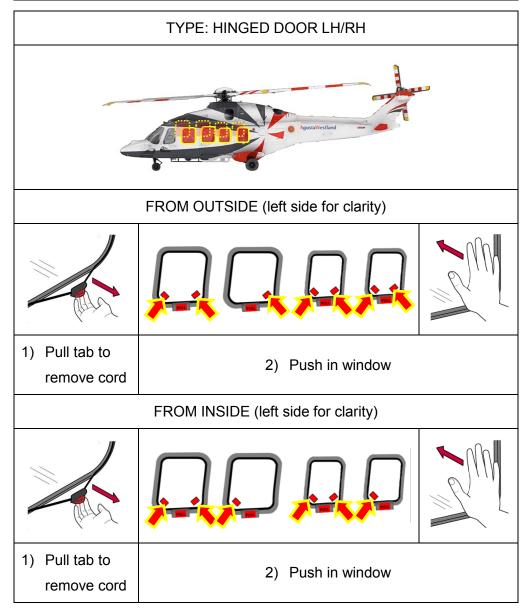
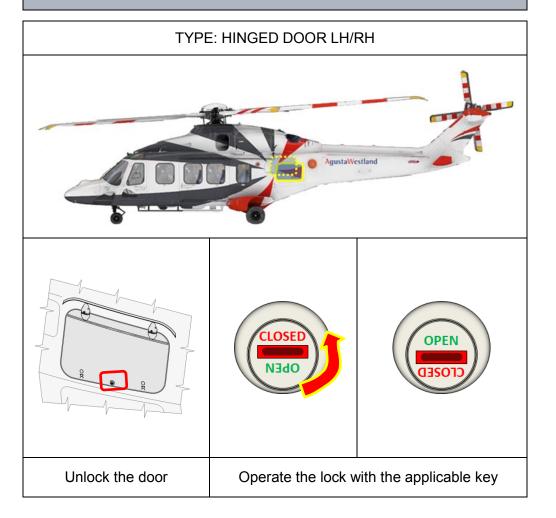


Figure 27 – Hinged Passenger Emergency Exits - Opening Procedure

OPEN THE BAGGAGE DOOR



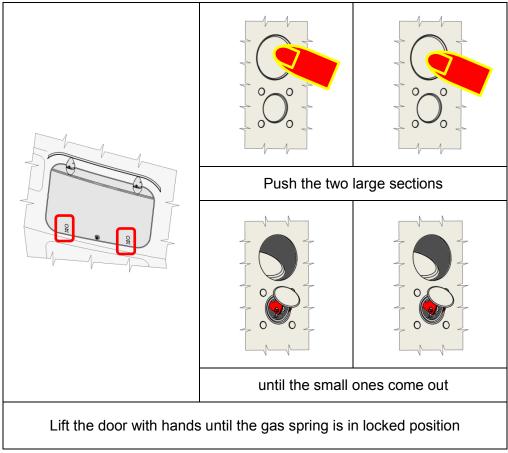
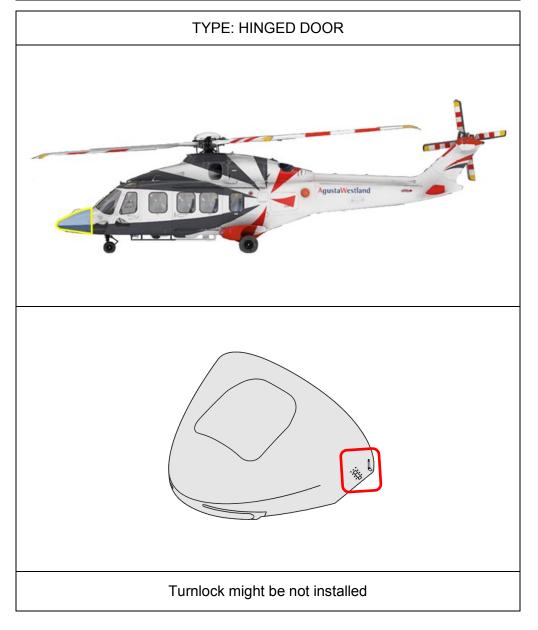


Figure 28 – Baggage Door - Opening Procedure

OPEN THE NOSE RADOME DOOR



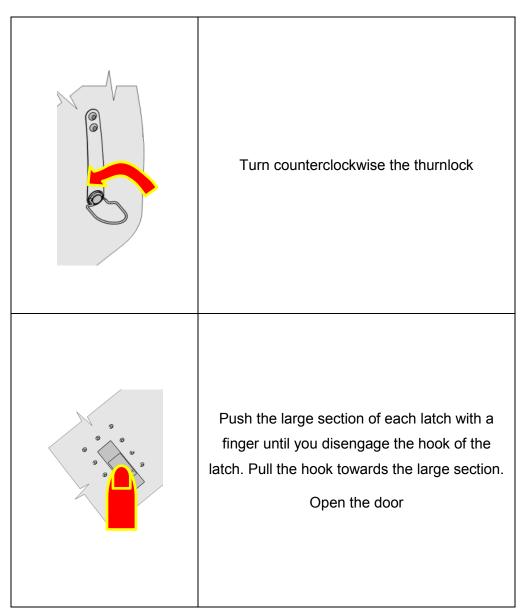
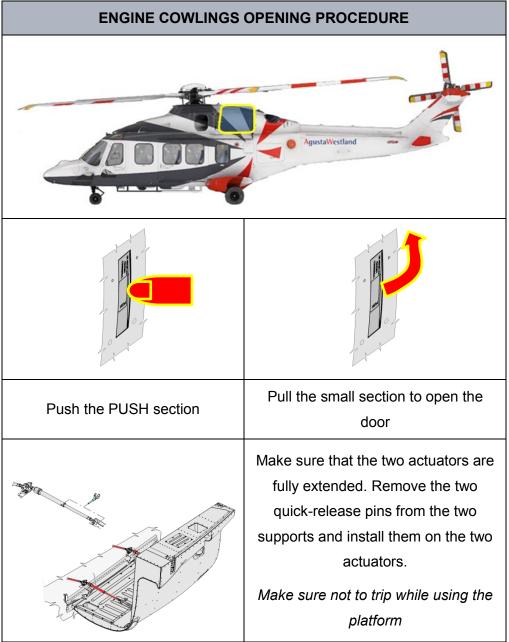


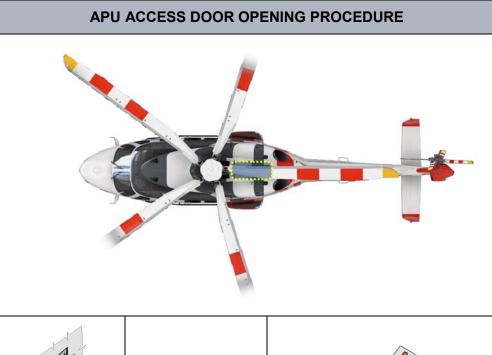
Figure 29 - Nose Radome Door - Opening Procedure

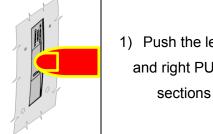
ENGINE AND APU ACCESS



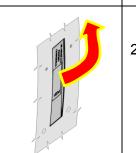
IGNITER BOXES - VOLTAGE HIGH AMPERAGE OUTPUT
BEFORE SERVICING DISCONNECT INPUT CURRENT
BEFORE OPERATING CONNECT OUTPUT LEADS AND
IGNITER PLUG



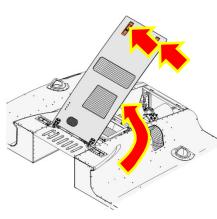




1) Push the left and right PUSH



2) Pull the left and right small sections to open the door



3) Lift the door with hands until the locking rod is in the locked position.

Figure 30 - Engine Cowlings Opening Procedure

Figure 31 - APU Cowling Opening Procedure

FIREFIGHTING RECOMMENDATIONS

FIRE IN THE BAGGAGE COMPARTMENT



IF SMOKE/FIRE IS CONFIRMED USE THE EXTINGUISHER
OTHERWISE OPEN THE BAGGAGE DOOR DIRECTLY

See Figure 28 for the baggage door opening procedure

RECOMMENDED FIRE FIGHTING AGENTS

engine fires	HALON or dry chemicals
exhaust nozzle fire	HALON or dry chemicals
APU fire	HALON or dry chemicals
fuel fire	dry chemical for leaking fuel and foam on ground spill area
wheel and brake fires	water fog or dry powder
	approach landing gear as per Figure 9 strand upwind of fire to avoid hydraulic fluid fumes
electrical fires	HALON or dry chemicals
cockpit and cabin area fires	HALON 1211 or dry powder
baggage compartment fire	HALON
aft equipment compartment fire	HALON

Figure 32 – Recommended Fire Fighting Agents

SAFETY INFORMATION: GROUND STAFF (INSIDE THE

HELICOPTER)

THE FOLLOWING PROCEDURES MUST BE CARRIED OUT



- 1) IN CASE OF EMERGENCY ON GROUND
- 2) ONLY IF PILOTS ARE INCAPACITATED
- 3) STRICTLY IN ORDER OF PRESENTATION



USE CAUTION WHEN MOVING INSIDE THE HELICOPTER,
CHECKING THE STRUCTURE INTEGRITY. SIGNS COULD
INCLUDE BUT ARE NOT LIMITED TO, DEFORMITY OF
STRUCTURE, FLAME IMPINGEMENT OR UNEVEN
SURFACES



IN CASE OF CHOCKS AVAILABILITY

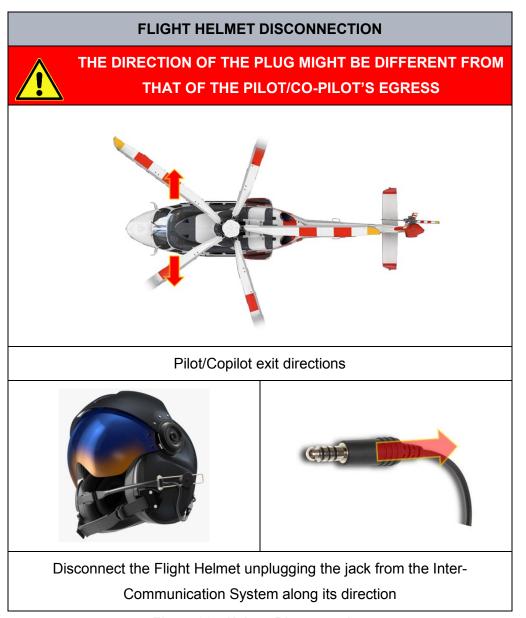
ACCESS THEIR NEED

AND LOCK THE WHEELS



WHEN ENTERING THE COCKPIT AREA, BE CAREFUL NOT
TO MOVE THE CYCLIC AND COLLECTIVE CONTROLS
BEFORE SHUTTING DOWN THE ENGINES

PERSONNEL RESCUE



SAFETY BELTS Pilot/copilot seat belt Passengers' seat belt Rotate the gear of the buckle in ANY direction to unfasten the belt

Figure 34 – Opening of the Safety Belt

EMERGENCY APU AND ENGINE SHUTDOWN



PERFORM THIS PROCEDURE TO FAST SHUTDOWN THE ENGINE AND ALLOW QUICKLY THE OTHER RESCUE OPERATIONS

IN CASE THERE IS NO FIRE OR NO MAJOR EMERGENCY

SKIP STEPS FROM 1 TO 4



TO PRESERVE THE INTEGRITY OF THE SYSTEMS

AND PERFORM:

- STEPS FROM 5 TO 7
- NEXT OPERATIONS

IN CASE OF MAJOR EMERGENCY OR FIRE



PERFORM:

- STEPS FROM 1 TO 4
- NEXT OPERATIONS

STEP 1 - APU FUEL SUPPLY INTERRUPTION - MAJOR EMERGENCY/FIRE



APU Control Panel is located in the Interseat Console





Raise the guard switch

Press the FIRE/ARM push button to shutdown the APU

Figure 35 - APU Shutdown - Major Emergency/Fire

STEP 2 - ENGINE FUEL SUPPLY INTERRUPTION - MAJOR

EMERGENCY/FIRE

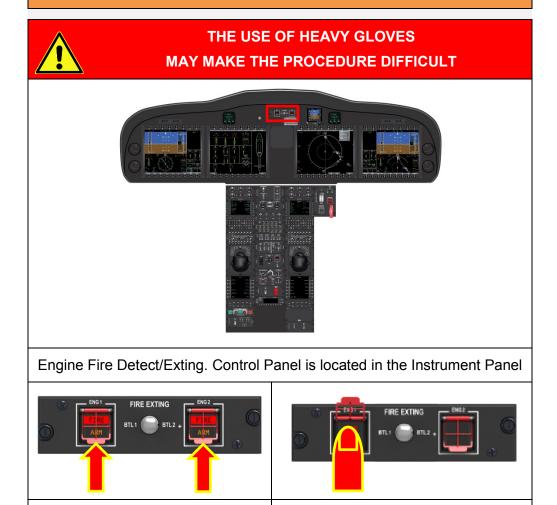


Figure 36 – Engine Fuel Supply Interruption – Major Emergency/Fire

Press the FIRE/ARM push button of

both the engines to shut off the valve

STEP 3 - APU FIRE EXTINGUISHING - MAJOR EMERGENCY/FIRE



Figure 37 – APU Fire Extinguishing – Major Emergency/Fire

Raise the guard switches of both

the engines (see ENG 1 in figure)

STEP 4 - ENGINE FIRE EXTINGUISHING - MAJOR EMERGENCY/FIRE

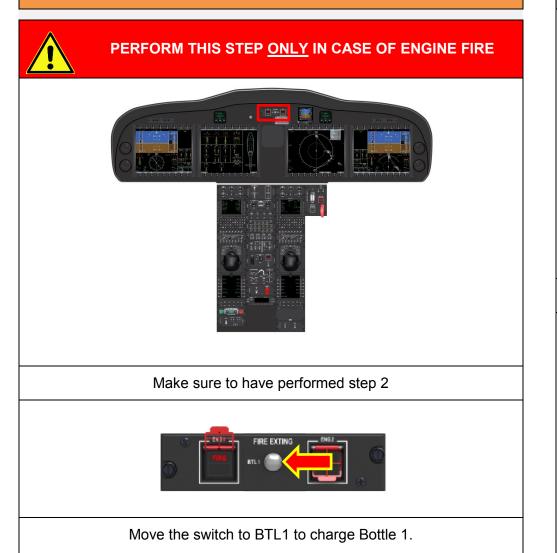


Figure 38 – Engine Fire Extinguishing – Major Emergency/Fire

If necessary do the same with BTL2



The Engine Control Panel is located in the Interseat Console



Push and rotate counterclockwise to switch off APU

Figure 39 - Normal APU Shutdown - Minor Emergency

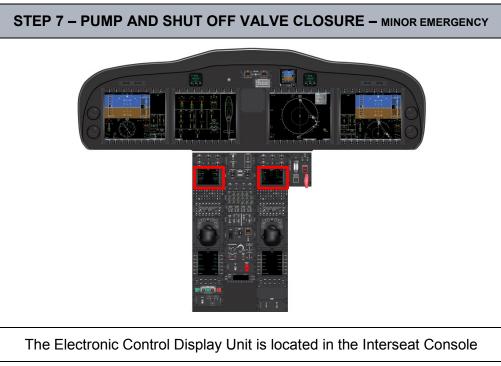
STEP 6 - ENGINE SHUTDOWN - MINOR EMERGENCY

The Engine Control Panel is located in the Interseat Console



Push and rotate counterclockwise to switch off each engine

Figure 40 – Normal Engine Shutdown – Minor Emergency
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Set the Pump 1 &2 to Off and the Shut Off Valve 1&2 to Close

Figure 41 – Normal Pump and Shut Off Valve Closure – Minor Emergency

NEXT OPERATIONS

ROTOR BRAKE THE PROCEDURE DOESN'T WORK WITH ENGINES STILL ON The Rotor Brake lever is on the Overhead Console

EMERGENCY FLOATATION SYSTEM (EFS)

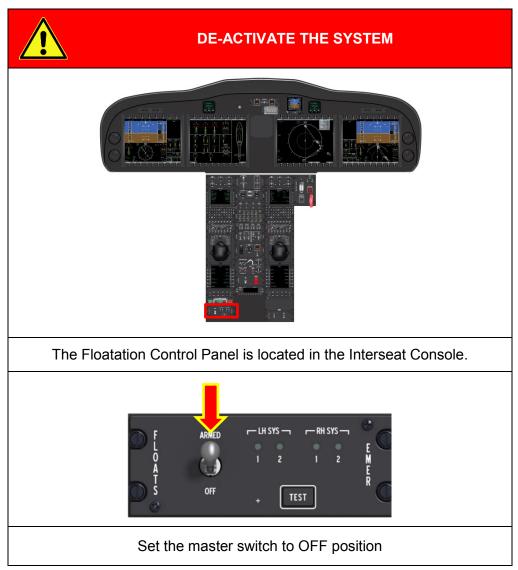


Figure 43 – Floatation System – De-activation

Figure 42 - Rotor Brake

Move the lever from OFF to the BRAKE position and pump within the

indicated range if necessary

EMERGENCY LOCATOR TRASMITTER (ELT)

DE-ACTIVATE THE SYSTEM AND INFORM THE AIR
TRAFFIC CONTROL ABOUT THE AIRCRAFT EVENT AND
LOCATION





The System Control Panel is on the Instrument Panel

Set the switch to ARM

Figure 44 – ELT System – De-activation

INTERRUPTION OF THE POWER SUPPLY FROM THE CONTROL PANEL



Electrical Power Generation and Distribution System, EPGDS, location





Move the Battery Master lever to OFF

Figure 45 – Interruption of the Power Supply from the Control Panel

MANUAL DISCONNECTION OF THE BATTERIES



PERFORM THIS PROCEDURE ONLY WHEN THE ENGINES ARE SWITCHED OFF AND THE ROTORS ARE STOPPED



IN CASE OF NEED TO RECONNECT THE BATTERIES

CONTACT LHD AI&P TEAM – RISK OF CVFDR DATA LOSS

See Figure 29 for the nose radome door opening procedure



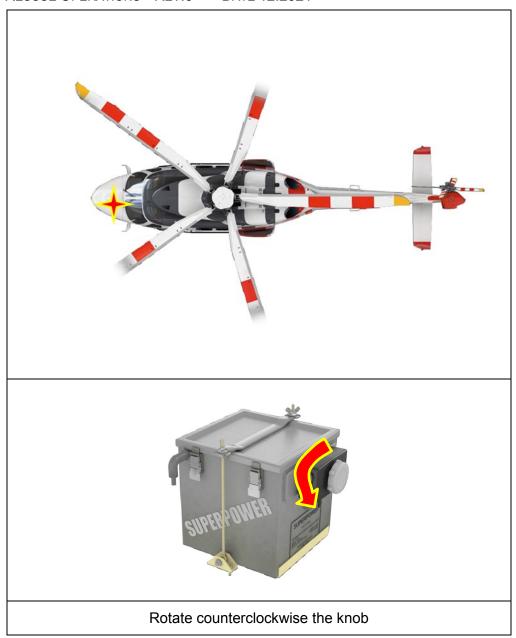


Figure 46 – Battery Location and Disconnection Procedure



Figure 47 – Adjustment of the Pilot Seats



Figure 48 – Removal of the Pilot Seats