FEDERAL AVIATION ADMINISTRATION AIRWORTHINESS DIRECTIVES

SMALL AIRCRAFT, ROTORCRAFT, GLIDERS BALLOONS, AIRSHIPS, AND UAS

BIWEEKLY 2023-25

11/20/2023 - 12/03/2023



Federal Aviation Administration Continued Operational Safety Policy Section, AIR-141 P.O. Box 25082 Oklahoma City, OK 73125-0460

| | | SMALL AIRCRAFT | |
|--------------------------|-------------------------|---|--|
| AD No. | Information | Manufacturer | Applicability |
| | Information Key: E- Emo | ergency; COR - Correction; R - Replaces, A- Affects | |
| Biweekly 2023-01 | | | |
| 2022-26-01 | | GE Aviation Czech s.r.o. | M601D-11,M601E-11,M601E-11A,M601E- 11AS,M601E-11S,M601F,H75-100,H75- 200,H80,H80-100,H80-200,H85-100,H85-200 |
| 2022-27-03 | | Leonardo S.p.a. | AB139,AW139 |
| 2022-27-08 | | Bell Textron Canada Limited | 407 |
| Biweekly 2023-02 | | | |
| 2022-27-09 | | Airbus Helicopters | EC130T2 |
| 2023-01-02 | | Leonardo S.p.a. | A109,A109A,A109A II,A109C,A109E,A109K2,A109S,AW109SP |
| Biweekly 2023-03 | | | |
| 2023-01-07 | | GE Aviation Czech s.r.o. | H75-100,H75-200,H80,H80-100,H80- 200,H85-100,H85-200 |
| 2023-01-11 | | Safran Helicopter Engines S.A. | Makila IA, Makila IAI |
| 2023-01-12 | D 2022 01 00 | Sarran Helicopter Engines S.A. | Arriel IC, Arriel ICI, Arriel IC2 |
| 2023-02-03 2023-02-04 | R 2022-01-09 | Mooney International Corporation | Stemme S 10- v 1, Stemme S 12 M20C, M20D, M20E, M20F, M20G |
| | | | |
| Biweekly 2023-04 | | | |
| 2023-01-04 | | Airbus Helicopters | AS350B,AS350BA,AS350B1,AS350B2,AS35 0B3,AS350D,AS355E,AS355F,AS355F1,AS3 55F2 AS355N AS355NP |
| 2023-01-07 | | GE Aviation Czech s.r.o. | H75-100,H75-200,H80,H80-100,H80- |
| 2023-01-08 | | Continental Aerospace Technologies GmbH | 200,H85-100,H85-200 TAE 125-02-99,TAE 125-02-114 |
| 2023-01-10 | | GE Aviation Czech s.r.o. | M601E-11,M601E-11A,M601E- |
| 2023-02-12 | | Continental Aerospace Technologies Inc. | GTSIO-520-C,GTSIO-520-D,GTSIO-520- |

E,GTSIO-520-F,GTSIO-520-H,GTSIO-520-K,GTSIO-520-L,GTSIO-520-M,GTSIO-520-N,IO-470-A,IO-470-C,IO-470-D,IO-470-E,IO-470-F,IO-470-G,IO-470-H,IO-470-J,IO-470-K,IO-470-L,IO-470-LO,IO-470-M,IO-470-N,IO-470-P,IO-470-R,IO-470-S,IO-470-T,IO-470-U,IO-470-V,IO-470-VO,IO-520-A,IO-520-B,IO-520-BA,IO-520-BB,IO-520-C,IO-520-CB,IO-520-D,IO-520-E,IO-520-F,IO-520-J,IO-520-K,IO-520-L,IO-520-M,IO-520-MB,IO-520-N,IO-520-NB,IO-520-P,IO-550-A,IO-550-B,IO-550-C,IO-550-D,IO-550-E,IO-550-F,IO-550-G,IO-550-L,IO-550-N,IO-550-P,IO-550-R,IOF-550-B,IOF-550-C,IOF-550-D,IOF-550-E,IOF-550-F,IOF-550-L,IOF-550-P,IOF-550-R,LIO-470-A,LIO-520-P,LTSIO-520-AE,O-470-A,O-470-E,O-470-G,O-470-G-CI,O-470-H,O-470-J,O-470-K,O-470-K-CI,O-470-L,O-470-L-CI,O-470-M,O-470-M-CI,O-470-N,O-470-P,O-470-R,O-470-S,O-470-T,O-470-U,TSIO-470-B,TSIO-470-С,

| | 1 | SMALL AIRCRAFT | |
|------------------|------------------------|--|---|
| AD No. | Information | Manufacturer | Applicability |
| | Information Key: E- Em | ergency; COR - Correction; R - Replaces, A- Affects | |
| | | | TSIO-470-D,TSIO-520-A,TSIO-520- AE,TSIO-520-AF,TSIO-520-B,TSIO-520- BB,TSIO-520-BE,TSIO-520-C,TSIO-520- CE,TSIO-520-D,TSIO-520-DB,TSIO-520- E,TSIO-520-B,TSIO-520-JB,TSIO-520- H,TSIO-520-J,TSIO-520-JB,TSIO-520- K,TSIO-520-KB,TSIO-520-L,TSIO-520- LB,TSIO-520-M,TSIO-520-N,TSIO-520- NB,TSIO-520-U,TSIO-520-R,TSIO-520- T,TSIO-520-U,TSIO-520-R,TSIO-520- VB,TSIO-520-U,TSIO-520-UB,TSIO-520- VB,TSIO-520-WB,TSIO-550-A,TSIO-550- B,TSIO-550-C,TSIO-550-E,TSIO-550- J,TSIOF-550-K,TSIOL-550-A,TSIOL-550-C |
| 2023-03-01 | | Airbus Helicopters Deutschland GmbH | BO-105A,BO-105C,BO-105S,BO-105LS A- 1,BO-105LS A-3,MBB-BK 117 A-1,MBB- BK 117 A-3,MBB-BK 117 A-4,MBB-BK 117 B-1,MBB-BK 117 B-2,MBB-BK 117 C- 1,MBB-BK 117 C-2,MBB-BK 117 D-2 |
| 2023-03-10 | | Schempp-Hirth Flugzeugbau GmbH | Duo-Discus, Duo Discus T |
| Biweekly 2023-05 | | | |
| 2023-01-07 | | GE Aviation Czech s.r.o. | H75-100,H75-200,H80,H80-100,H80- |
| 2023-02-17 | | Textron Aviation Inc. | 200,H85-100,H85-200 210N,210R,P210N,P210R,T210N,T210R,17 177A,177B,177RG,F177RG |
| 2023-03-02 | | Pratt & Whitney Canada Corp. | PT6E-67XP |
| 2023-03-03 | | Leonardo S.p.a. | AB139,AW139 |
| 2023-03-12 | R 2004-04-09 | Pratt & Whitney Canada Corp. | JT15D-1,JT15D-1A,JT15D-1B |
| 2023-03-13 | | Airbus Helicopters | AS355E,AS355F,AS355F1,AS355F2,AS355 |
| 2023-04-08 | | Continental Aerospace Technologies, Inc. (Continental®) | N GTSIO-520-C,GTSIO-520-D,GTSIO-520- H,GTSIO-520-K,GTSIO-520-L,GTSIO-520- M,GTSIO-520-N,GTSIO-520-S,IO-360-A,IO- 360-AB,IO-360-AF,IO-360-C,IO-360-CB,IO- 360-D,IO-360-DB,IO-360-E,IO-360-ES,IO- 360-G,IO-360-GB,IO-360-H,IO-360-HB,IO- |

360-J,IO-360-JB,IO-360-K,IO-360-KB,IO-470-D,IO-470-E,IO-470-G,IO-470-H,IO-470-J,IO-470-K,IO-470-L,IO-470-M,IO-470-N,IO-470-P,IO-470-R,IO-470-S,IO-470-T,IO-470-U,IO-470-V,IO-470-VO,IO-520-A,IO-520-B,IO-520-BA,IO-520-BB,IO-520-C,IO-520-CB,IO-520-D,IO-520-E,IO-520-F,IO-520-J,IO-520-K,IO-520-L,IO-520-M,IO-520-MB,IO-550-A,IO-550-B,IO-550-C,IO-550-D,IO-550-E,IO-550-F,IO-550-G,IO-550-L,IO-550-N,IO-550-P,IO-550-R,LTSIO-360-E,LTSIO-360-EB,LTSIO-360-KB,LTSIO-360-RB,LTSIO-520-AE,O-470-A,O-470-B,O-470-E,O-470-G,O-470-H,O-470-J,O-470-K,O-470-L,O-470-M,O-470-N,O-470-R,O-470-S,O-470-T,O-470-U,TSIO-360-A,TSIO-360-AB,TSIO-360-B,TSIO-360-BB,TSIO-360-C,TSIO-360-CB,TSIO-360-D,TSIO-360-DB,

.

AD No. Information Manufacturer Applicability Information Key: E- Emergency; COR - Correction; R - Replaces, A- Affects TSIO-360-E.TSIO-360-EB.TSIO-360-G,TSIO-360-GB,TSIO-360-H,TSIO-360-HB,TSIO-360-JB,TSIO-360-KB,TSIO-360-LB,TSIO-360-MB,TSIO-360-RB,TSIO-360-SB,TSIO-520-A,TSIO-520-AE,TSIO-520-AF, TSIO-520-B, TSIO-520-BB, TSIO-520-BE, TSIO-520-C, TSIO-520-CE, TSIO-520-D,TSIO-520-DB,TSIO-520-E,TSIO-520-EB,TSIO-520-G,TSIO-520-H,TSIO-520-J,TSIO-520-JB,TSIO-520-K,TSIO-520-KB,TSIO-520-L,TSIO-520-LB,TSIO-520-M,TSIO-520-NB,TSIO-520-P,TSIO-520-R,TSIO-520-T,TSIO-520-UB,TSIO-520-VB,TSIO-520-WB,TSIO-550-A,TSIO-550-B,TSIO-550-C,TSIO-550-E,TSIO-550-G,TSIO-550-K,TSIO-550-N,TSIOF-550-K,TSIOL-550-A,TSIOL-550-B,TSIOL-550-C Biweekly 2023-06 2023-03-14 Schempp-Hirth Flugzeugbau GmbH Duo-Discus, Duo Discus T 2023-03-22 R 2015-09-04 R1 DG Flugzeugbau GmbH,Schempp-Hirth DG-1000T, Duo Discus T Flugzeugbau GmbH 2023-04-20 Cirrus Design Corporation SF50 Biweekly 2023-07 2023-05-03 R 2022-14-14 Alexander Schleicher GmbH & Co. ASW -15, ASW-15B Segelflugzeugbau 2023-05-09 Airbus Helicopters Deutschland GmbH EC135P3,EC135T3,MBB-BK 117 D-2,MBB-BK 117 D-3 2023-05-16 R 2023-04-08 Continental Aerospace Technologies Inc. GTSIO-520-C,GTSIO-520-D,GTSIO-520-H,GTSIO-520-K,GTSIO-520-L,GTSIO-520-M,GTSIO-520-N,GTSIO-520-S,IO-360-A,IO-360-AB,IO-360-AF,IO-360-C,IO-360-CB,IO-360-D.IO-360-DB.IO-360-E.IO-360-ES.IO-

360-G.IO-360-GB.IO-360-H.IO-360-HB.IO-360-J,IO-360-JB,IO-360-K,IO-360-KB,IO-470-A,IO-470-C,IO-470-D,IO-470-E,IO-470-F.IO-470-G.IO-470-H.IO-470-J.IO-470-K.IO-470-L,IO-470-LO,IO-470-M,IO-470-N,IO-470-P,IO-470-R,IO-470-S,IO-470-T,IO-470-U,IO-470-V,IO-470-VO,IO-520-A,IO-520-B,IO-520-BA,IO-520-BB,IO-520-C,IO-520-CB,IO-520-D,IO-520-E,IO-520-F,IO-520-J,IO-520-K,IO-520-L,IO-520-M,IO-520-MB,IO-550-A,IO-550-B,IO-550-C,IO-550-D,IO-550-E,IO-550-F,IO-550-G,IO-550-L,IO-550-N,IO-550-P,IO-550-R,LTSIO-360-E,LTSIO-360-EB,LTSIO-360-KB,LTSIO-360-RB,LTSIO-520-AE,O-470-A,O-470-B,O-470-E,O-470-G,O-470-H,O-470-J,O-470-K,O-470-L,O-470-M,O-470-N,O-470-R,O-470-S,O-470-T,O-470-U,TSIO-360-A,TSIO-360-AB,TSIO-360-B,TSIO-360-BB,TSIO-360-C,TSIO-360-CB,TSIO-360-D,TSIO-360-DB,TSIO-360-E,TSIO-360-EB,TSIO-360-F,TSIO-360-FB,TSIO-360-G,TSIO-360-

SMALL AIRCRAFT

| | Information | SMALL AIKCKAFT | Applicability |
|------------------|------------------------|---|--|
| AD NO. | Information | Manufacturer | Аррисавину |
| | Information Key: E- Em | hergency; COR - Correction; R - Replaces, A- Affects | |
| | | | GB,TSIO-360-H,TSIO-360-HB,TSIO-360- JB,TSIO-360-KB,TSIO-360-LB,TSIO-360- MB,TSIO-360-RB,TSIO-360-SB,TSIO-520- A,TSIO-520-AE,TSIO-520-AF,TSIO-520- B,TSIO-520-BB,TSIO-520-BE,TSIO-520- C,TSIO-520-E,TSIO-520-D,TSIO-520- DB,TSIO-520-E,TSIO-520-L,TSIO-520- G,TSIO-520-H,TSIO-520-L,TSIO-520- JB,TSIO-520-LB,TSIO-520-KB,TSIO-520- L,TSIO-520-LB,TSIO-520-M,TSIO-520- NB,TSIO-520-P,TSIO-520-R,TSIO-520- T,TSIO-520-UB,TSIO-520-R,TSIO-520- WB,TSIO-550-A,TSIO-550-B,TSIO-550- C,TSIO-550-R,TSIO-550-G,TSIO-550- K,TSIO-550-N,TSIOF-550-K,TSIOL-550- A,TSIOL-550-B,TSIOL-550-C |
| 2023-06-11 | | Viking Air Limited | DHC-2 Mk.I |
| Biweekly 2023-08 | | | |
| 2023-07-51 | Е | Leonardo S.p.a. | AB139,AW139 |
| | | | |
| Biweekly 2023-09 | | | |
| 2023-06-05 | | Bell Textron Canada Limited | 206A,206A-1 (OH-58A),206B,206B- |
| 2023-07-08 | | Pilatus Aircraft Ltd. | PC-12/47E |
| Biweekly 2023-10 | | | |
| 2023-06-14 | | Pratt & Whitney Canada Corp. | PW308A,PW308C |
| 2023-07-03 | | Leonardo S.p.a. | AB412,AB412 EP |
| Biweekly 2023-11 | | | |
| 2023-08-06 | A 2020-20-08 | Airbus Helicopters | AS332C,AS332C1,AS332L,AS332L1,AS332 |
| 2023-08-07 | | Allied Ag Cat Productions Inc. | L2,EC225LP G-164A,G-164B |
| Biweekly 2023-12 | | | |
| 2023-09-07 | R 2022–02–01 | Sikorsky Aircraft Corporation | S-92A |
| 2023-09-12 | | Pilatus Aircraft Ltd. | PC-12,PC-12/45,PC-12/47,PC-12/47E |
| 2023-10-02 | R 2021–23–12 | The Boeing Company,Airbus SAS,Bombardier Inc.,Embraer S.A.,Gulfstream Aerospace Corporation,Gulfstream Aerospace LP,Textr Aviation Inc.,Pilatus Aircraft Limited,Fokker Services B.V.,Saab AB Support and Services,De Havilland Aircraft of Canada Limited,Airbus Canada Limited Partnership,ATR - GIE Avions de Transport Régional,MHI RJ Aviation ULC,BAE Systems (Operations) Limited,Lockheed Martin Corporation,Lockheed Martin Aeronautics Company,Viking Air | 18,23,35,36,50,58,60,65,70,76,77,95,99,100,1 11,120,140,150,152,170,172,175,177,180,182, 185,188,190,195,200,206,207,208,210,300,31 on4,320,321,335,336,337,340,382,390,400,401,4 r 02,404,406,408,411,414,421,425,441,500,501, 510,525,550,551,552,560,650,680,700,750,19 00,2000,4000,1049-54,1049B-55 (Navy R7V- 1),1049C-55,1049D-55,1049E-55,1049F-55 (USAF C-121C),1049H-82,1049G-82,1125 Westwind Astra,1329-23A,1329-23E,1329- 25,1329- 23D,150A,150B,150C,150D,150E,150F,150G ,150H,150J,150K,150M,150L, |

| 170A,170B,172A,172B,172C,172D,172E,172 G,172F (USAF T-41A),172H (USAF T- 41A),172I,172K,172L,172M,172N,172P,172Q 172R,172RG,172S,175A,175B,175C,177A,1 177B 180A 180B 180C 180D 180E 180E 180C |
|--|
| 170A,170B,172A,172B,172C,172D,172E,172 G,172F (USAF T-41A),172H (USAF T- 41A),172I,172K,172L,172M,172N,172P,172Q 172R,172RG,172S,175A,175B,175C,177A,1 77B 180C 180D 180C 180D 180E 180C |
| J. J. 1807, 1805, 1802, 1807, |
| |

A,A18D,A200 (C-12A),A200 (C-

SMALL AIDCDAFT

| AD No. | Information | Manufacturer | Applicability |
|--------|-------------------------|---|---|
| | Information Key: E- Eme | rgency; COR - Correction; R - Replaces, A- Affe | ects |
| | | | 12C),A200C (UC-12B),A200CT (C- |
| | | | 12D),A200CT (C-12F),A200CT (FWC- |
| | | | 12D),A200CT (RC-12D),A200CT (RC- |
| | | | 12G),A200CT (RC-12H),A200CT (RC- |
| | | | 12K),A200CT (RC-12P),A200CT (RC- |
| | | | 12Q),A23,A23-19,A23A,A23- |
| | | | 24,A24,A24K,A300 B2-1A,A300 B2- 1C A 200 B2 202 A 200 B2K 2C A 200 B4 |
| | | | 2C \(\Delta\) 300 B2-203,A300 B2K-3C,A300 B4- |
| | | | 601 A 300 B4-603 A 300 B4-605R A 300 B4- |
| | | | 620,A300 B4-622,A300 B4-622R,A300 C4- |
| | | | 605R Variant F,A300 F4-605R,A300 F4- |
| | | | 622R,A310-203,A310-204,A310-221,A310- |
| | | | 222,A310-304,A310-322,A310-324,A310- |
| | | | 325,A-314,A318-111,A318-112,A318- |
| | | | 121,A318-122,A319-111,A319-112,A319- |
| | | | 113,A319-114,A319-115,A319-131,A319- |
| | | | 132,A319-133,A319-151N,A319-153N,A319 171N A220 211 A220 212 A220 214 A220 |
| | | | 1/111,A320-211,A320-212,A320-214,A320- 216 A 320-231 A 320-232 A 320-233 A 320 |
| | | | 251N A320-252N A320-253N A320- |
| | | | 271N.A320-272N.A320-273N.A321- |
| | | | 111,A321-112,A321-131,A321-211,A321- |
| | | | 212,A321-213,A321-231,A321-232,A321- |
| | | | 251N,A321-251NX,A321-252N,A321- |
| | | | 252NX,A321-253N,A321-253NX,A321- |
| | | | 271N,A321-271NX,A321-272N,A321- |
| | | | 272NX,A330-201,A330-202,A330-203,A330 |
| | | | 225,A550-225F,A550-245,A550-245F,A550- 301 |
| | | | 301,A330-302,A330-303,A330-321,A330- |
| | | | 343.A330-841.A330-941.A340-211.A340- |
| | | | 212,A340-213,A340-311,A340-312,A340- |
| | | | 313,A340-541,A340-642,A35,A350- |
| | | | 941,A350-1041,A36,A36TC,A380-841,A380 |
| | | | 842,A380-861,A45 (Military T-34A; B- |
| | | | 45),A56TC,A65,A65-8200,A75 (Army PT- |
| | | | 13A; -13B; -13C),A/5JI (Army P1- 18) A75L2 A75L200 A75N1 (Army DT 17) |
| | | | 10,A/5L5,A/5L500,A/5N1 (Anny F1-17, - $17\Delta \cdot Navy N2S_{1} \cdot -4) \Delta 99 \Delta 99 \Delta Army \Delta T_{-}$ |
| | | | 11.Astra SPX.AT-6 (SNJ-2).AT-6A (SNJ- |
| | | | 3),AT-6B,AT-6C (SNJ-4),AT-6D (SNJ-5),AT |
| | | | 6F (SNJ-6),ATR42-200,ATR42-300,ATR42- |
| | | | 320,ATR42-500,ATR72-101,ATR72- |
| | | | 102,ATR72-201,ATR72-202,ATR72- |
| | | | 211,ATR72-212,ATR72-212A,Avro 146- |
| | | | RJ/0A, Avro 146-RJ85A, Avro 146- |
| | | | KJ100A,B100,B19,B200,B200C,B200C (U- 12E) B200C (C-12D) B200C (UC |
| | | | 12F) B200C (UC- 12F) B200C (UC- |
| | | | 121),B200C (CC- 12M),B200CGT.B200CT.B200GT.B200T B2 |
| | | | 3,B24R,B300,B300C,B300C (MC- |
| | | | 12W),B300C (UC- |
| | | | 12W),B35,B36TC,B50,B60,B75 (Navy N2S- |
| | | | 5),B95,B95A,B99,BAC 1-11 400 Series,BAC |
| | | | 1-11 200 Series, BAe 146-100A, BAe 146- |
| | | | 200A,BAe 146-300A,BAe.125 Series |
| | | | 800A,BAe.125 Series 800A (C-29A),BAe.12 |
| | | | Series 800A |

| | | SMALL AIRCRAF | T |
|--------|--------------------------|--|---|
| AD No. | Information | Manufacturer | Applicability |
| | Information Key: E- Emer | rgency; COR - Correction; R - Replaces, A- Aff | Pects |
| | | | (U-125).BAe.125 Series 800B.BAe.125 Series |
| | | | 1000A.BAe.125 Series 1000B.BC-1A.BD- |
| | | | 500-1A10.BD-500-1A11.BD-700-1A10.BD- |
| | | | 700-1A11.BD-700-2A12.Beagle B.121 Series |
| | | | 1,Beagle B.121 Series 2,Beagle B.121 Series |
| | | | 3,BH.125 Series 400A,BH.125 Series |
| | | | 600A,C23,C24R,C35,C-45G,C-45H,C50,C54- |
| | | | DC,C54A-DC,C54B-DC,C54D-DC,C54G- |
| | | | DC,C54E- |
| | | | DC,C90,C90A,C90GT,C90GTi,C99,CL-215- |
| | | | 1A10,CL-215-6B11 (CL-215T Variant),CL- |
| | | | 215-6B11 (CL-415 Variant), CL-44J, CL-600- |
| | | | 1A11 (CL-600),CL-600-2A12 (CL-601),CL- |
| | | | 600-2B16 (CL-601-3A), CL-600-2B16 (CL- |
| | | | 601-3R),CL-600-2B16 (CL-604),CL-600- |
| | | | 2B19 (Regional Jet Series 100),CL-600-2B19 |
| | | | (Regional Jet Series 440),CL-600-2C10 |
| | | | (Regional Jet Series 700),CL-600-2C10 |
| | | | (Regional Jet Series 701),CL-600-2C10 |
| | | | (Regional Jet Series 702),CL-600-2C11 |
| | | | (Regional Jet Series 550),CL-600-2D15 |
| | | | (Regional Jet Series 705),CL-600-2D24 |
| | | | (Regional Jet Series 900),CL-600-2E25 |
| | | | (Regional Jet Series |
| | | | 1000),D18C,D18S,D35,D45 (Military 1- |
| | | | 34B),D50,D50A,D50B,D50C,D50E,D50E- |
| | | | 5990,D55,D55A,D75N1 (Army P1- |
| | | | 27),D95A,DC-10-10,DC-10-10F,DC-10- 15 DC 10 20 DC 10 20E (VC 10A VDC |
| | | | 10) DC 10 40E DC 10 40 DC 24 |
| | | | S1C3C DC3A S1CC DC3A S4C4C DC3A |
| | | | SC3G DC3A-SCG DC3C-R-1830.00C DC3C- |
| | | | \$1C3G DC3C-\$C3G DC3C-\$4C4G DC3D-P- |
| | | | 1830-90C DC3_G102 DC3_G102 Δ DC3_ |
| | | | G103A DC3-G202A DC-4 DC-6B DC- |
| | | | 7B DC-7C DC-6 DC-6A DC-7 DC-8-11 DC- |
| | | | 8-12.DC-8-21.DC-8-31.DC-8-32.DC-8- |
| | | | 33.DC-8-41.DC-8-42.DC-8-43.DC-8-51.DC- |
| | | | 8-52.DC-8-53.DC-8-55.DC-8-61.DC-8- |
| | | | 61F.DC-8-62.DC-8-62F.DC-8-63.DC-8- |
| | | | 63F,DC-8-71,DC-8-71F,DC-8-72,DC-8- |
| | | | 72F,DC-8-73,DC-8-73F,DC-8F-54.DC-8F- |
| | | | 55.DC-9-11.DC-9-12.DC-9-13.DC-9-14.DC- |

Mk.I,DHC-2 Mk.II,DHC-2 Mk.III,DHC-4,DHC-4A,DHC-6-1,DHC-6-100,DHC-6-200,DHC-6-300,DHC-6-400,DHC-7-1,DHC-7-100,DHC-7-101,DHC-7-102,DHC-7-103,E17B (Army

3A/RA,DH.125 Series 400A,DHC-2

9-15,DC-9-15F,DC-9-21,DC-9-31,DC-9-32,DC-9-32 (VC-9C),DC-9-32F,DC-9-32F (C-9A),DC-9-32F (C-9B),DC-9-33F,DC-9-34,DC-9-34F,DC-9-41,DC-9-51,DC-9-81 (MD-81), DC-9-82 (MD-82), DC-9-83 (MD-83),DC-9-87 (MD-87),DH.125 Series 1A,DH.125 Series 1A-522,DH.125 Series 1A/R-522,DH.125 Series 1A/S-522,DH.125 Series 3A,DH.125 Series 3A/R,DH.125 Series

| AD No. | Information | Manufacturer | Applicability |
|--------|-------------|--------------|---------------|
| | | | |

Information Key: E- Emergency; COR - Correction; R - Replaces, A- Affects

UC-43D),E17L,E18S,E18S-9700,E310H,E310J,E33,E33A,E33C,E35,E50, E55,E55A,E75 (Army PT-13D; Navy N2S-5; PT-13D/N2S-5),E75N1 (Army PT-13D; Navy N2S-5; PT-13D/N2S-5),E90,E95,EMB-110P1,EMB-110P2,EMB-120,EMB-120FC,EMB-120QC,EMB-120RT,EMB-120ER,EMB-135,EMB-135BJ (Legacy 600),EMB-135BJ (Legacy 650),EMB-135BJ,EMB-135ER,EMB-135KE,EMB-135KL,EMB-135LR,EMB-145EP,EMB-145ER,EMB-145LR,EMB-145MP,EMB-145MR,EMB-145XR,EMB-500,EMB-505,EMB-545,EMB-550,ERJ 170-100 LR,ERJ 170-100 SE,ERJ 170-100 STD,ERJ 170-100 SU,ERJ 170-200 LL,ERJ 170-200 LR,ERJ 170-200 STD,ERJ 170-200 SU,ERJ 190-100 ECJ,ERJ 190-100 IGW,ERJ 190-100 LR,ERJ 190-100 STD,ERJ 190-200 IGW,ERJ 190-200 LR, ERJ 190-200 STD, ERJ 190-300, ERJ 190-400,F150F,F150G,F150H,F150J,F150K,F150 L,F150M,F152,F172D,F172E,F172F,F172G,F 172H,F172K,F172L,F172M,F172N,F172P,F1 7D (UC-43C),F27 Mark 050,F27 Mark 100,F27 Mark 200,F27 Mark 300,F27 Mark 400,F27 Mark 500,F27 Mark 600,F27 Mark 700,F28 Mark 0070,F28 Mark 0100,F28 Mark 1000,F28 Mark 2000,F28 Mark 3000,F28 Mark 4000.F33.F33A.F33C.F337E.F337F.F337G.F 337H,F35,F50,FA150K,FA150L,FA150M,FA 152, Falcon 7X, Falcon 900EX, FALCON 2000.FALCON 2000EX.Falcon 10.Fan Jet Falcon.Fan Jet Falcon Series C.Fan Jet Falcon Series D,Fan Jet Falcon Series E,Fan Jet Falcon Series F, Fan Jet Falcon Series G,FP172D,FR172E,FR172F,FR172G,FR172H ,FR172J,FR172K,FRA150L,FRA150M,FT337 E,FT337F,FT337GP,FT337HP,G-1159,G-1159A,G-1159B,G18S,G33,G50,G58,G-IV,GIV-X,Gulfstream 100,Gulfstream 200, GV, GV-SP, GVI, GVII-G500, GVII-G600,H18,H35,H50,H90,Hawker 750,Hawker 800, Hawker 800 (U-125A), Hawker 800XP,Hawker 850XP,Hawker 900XP,Hawker 1000,HS 748 Series 2A,HS 748 Series 2B,HS.125 Series 1B,HS.125 Series 1B-522,HS.125 Series 1B/R-522,HS.125 Series 1B/S-522,HS.125 Series 3B,HS.125 Series 3B/R,HS.125 Series 3B/RA,HS.125 Series 3B/RB,HS.125 Series 3B/RC,HS.125 Series 400A,HS.125 Series 400B,HS.125 Series 400B/1,HS.125 Series 401B,HS.125 Series 403A(C),HS.125 Series 403B,HS.125 Series 600A,HS.125 Series 600B,HS.125 Series 600B/1,HS.125 Series

СЛЛАТТ

| | | SMALL AIKCKAF | l |
|--------|------------------------|--|--|
| AD No. | Information | Manufacturer | Applicability |
| | Information Key: E- Em | ergency; COR - Correction; R - Replaces, A- Affe | cts |
| | | | 600B/2.HS.125 Series 600B/3.HS.125 Series |
| | | | 700A,HS.125 Series 700B,HS.125 Series |
| | | | F3B,HS.125 Series F3B/RA,HS.125 Series |
| | | | F400B,HS.125 Series F403B,HS.125 Series |
| | | | F600B,IB75A,J35,JRB-6,K35,L-1011-385- |
| | | | 1,L-1011-385-1-14,L-1011-385-1-15,L-1011- |
| | | | 385-3,LC40-550FG,LC40-550G,LC41- |
| | | | 550FG,M19A,M357B,M35,MD-10-10F,MD- 10.20E MD, 11 MD, 11E MD, 88 MD, 00 |
| | | | 10-30F,MD-11,MD-11F,MD-88,MD-90- 30 MU 300 10 MU 300 Mysters Falcon 20 |
| | | | C5 Mystere-Falcon 20 - D5 Mystere-Falcon 20 |
| | | | - E5.Mystere-Falcon 20 - F5.Mystere-Falcon |
| | | | 50,Mystere-Falcon 200,Mystere-Falcon |
| | | | 900,N35,Navy R6D-1,Navy R6D- |
| | | | 1Z,P172D,P206,P206A,P206B,P206C,P206D, |
| | | | P206E,P210N,P210R,P337H,P35,R172E,R17 |
| | | | 2F,R172G,R172H,R172J,R172K,R182,R4D- |
| | | | 8,K4D-8Z,KC- 451 919 A 919D 925 9 A 19 A 9 A 19D 9 A |
| | | | 307B SA-307B-1 SAAB 340B 340A (SAAB |
| | | | SF340A).SAAB 2000.SC-7 Skyvan Series |
| | | | 2,SC-7 Skyvan Series 3,SD17S,SD3-30,SD3- |
| | | | 60,SD3-60 SHERPA,SD3-SHERPA,Super |
| | | | DC- |
| | | | 3,T182,T182T,T188C,T206H,T207,T207A,T2 |
| | | | 10F,1210G,1210H,1210J,1210K,1210L,1210 M T210N T210P T240 T210P T210P T210P |
| | | | M,1210N,1210K,1240,1510P,1510Q,1510K, T337B T337C T337D T337E T337E T337G T |
| | | | 337H T337H-SP T-6G TC-45G TC-45H TC- |
| | | | 45J,TP206A,TP206B,TP206C,TP206D,TP206 |
| | | | E,TR182,TU206A,TU206B,TU206C,TU206D |
| | | | ,TU206E,TU206F,TU206G,U206,U206A,U20 |
| | | | 6B,U206C,U206D,U206E,U206F,U206G,UC- |
| | | | 45J,USAF C-118A,V35,V35A,V35B,12- D 140A 140 46 1640A |
| | | | D,140A,149-40,1049A- 98 177RG 184 105A 202.R 208R 247 D |
| | | | (Army C-73).300-50A-01 (USAF C- |
| | | | 141A),3TM,402-2,45 (Military YT-34),720B |
| | | | Series,80-A1,99A (FACH),A60,ATP,B18S |
| | | | (Army F-2),B75N1 (Navy N2S-3),B90,BD- |
| | | | 100-1A10 (Challenger 300),C18S,CL- |
| | | | 44D4,D17A (Army UC-43F),D17R (Army |
| | | | UC-45A), D1/5, DHC-5, Electra 10- E E177RG E90 EP182 G |
| | | | 159.G178.G35.G36.Galaxy.Gulfstream |
| | | | G150,Gulfstream G280,HU-16D,J50,Jetstream |
| | | | Model 4101,LC42-550FG,NA-260,Navy |
| | | | SNB-1,O-47B,PC-24,S- |
| | | | 307,S550,SE17B,SF17D,SNJ-7,Super |
| | | | Universal,T303,T-34C,TR-1 |
| | | | |
| | | | |

2023-10-05 2023-11-03 R 2023–07–51

Leonardo S.p.a. Honda Aircraft Company LLC AB139,AW139 HA-420

Biweekly 2023-13

2023-09-09

Aerostar Aircraft Corporation, B-N Group Ltd.,Commander Aircraft

PA-60-600 (Aerostar 600), PA-60-601 (Aerostar 601),PA-60-601P (Aerostar

| Applicability |
|--|
| |
| 01P),PA-60-602P (Aerostar 602P),PA-60- 00P (Aerostar 700P),BN-2,BN-2A,BN-2A, BN-2A-8,BN-2A- ,112TC,112TCA,114TC,SR22,SR22T,LTSI)-360-E,LTSIO-360-EB,LTSIO-360- B,LTSIO-360-RB,TSIO-360-FB,TSIO-360- B,TSIO-360-LB,TSIO-360-FB,TSIO-360- B,TSIO-360-LB,TSIO-360-FB,TSIO-360- B,TSIO-360-SB,TSIO-520-BE,TSIO-520- ,TSIO-520-LB,TSIO-520-B,TSIO-550- ,TSIO-550-K,TSIO-550-G,TSIO-550- ,TSIO-550-K,TSIO-550-B,TSIO-550- ,TSIO-550-K,TSIO-550-B,IO-550- ,TSIO-550-K,TSIO-550-B,IO-550- ,TSIO-550-L,TSIO-550-B,IO-550-E,IO-550- ,TSIO-550-L,TSIO-550-B,IO-550-E,IO-550- ,IP2012 Traveller,TB 21,DA 40,F-28C,F- 8C-2,F-28C-2R,F-28F,F-28F- ,280C,280F,280FX,500,H-295 (USAF 110D),H-395 (USAF L-28A or U-10B),4500- 00,4500-300 Series II,IO-540-AA1A5,IO- 40-AG1A5,IO-540-S1A5,TIO-540- LE2A,TIO-540-AH1A,LTIO-540-J2BD,TO- 60-C1A6D,TO-360-E1A6D,ITO-540-J2BD,M- -210TC,IO-540- 4X1,M20J,M20K,M20M,M20TN,M20V,PA- 3,PA-23-160,PA-23-235,PA-23-250,PA-23- 50 (Navy UO-1),PA-E23-250,PA-24- 50,PA-24-260,PA-24-400,PA-28-201T,PA- 8R-201T,PA-28RT-201T,PA-30,PA-31,PA- 1-325,PA-31-350,PA-31P,PA-31P-350,PA- 2-260,PA-32R-300,PA-32R-301T,PA-34- 00,PA-34-200T,PA-34-200T,PA-33P,A-44- 80T,PA-46-310P,PA-46-350P,Lake LA- ,Lake LA-4A,Lake LA-4-200,Lake 250,47G- B,47G-3B-1,47G-3B-2,47G-3B-2A,UH- 21,UH-12L4,EA 400-500,35-33,35-A33,35- 333,5C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,35- 333,C-33,55- 355,P35,N35,N35,P35,S35,V35,V35A,V35B,3 ,A36,A36TC,B36TC,D55,E55,56TC,A56TC, 8,658,60,A60,B60,95,95- 55,B95,B95A,D95A,E95,185,185A,185B,18 C,185D,185E,A185E,A185F,A188,A188A,A 88B,T182,T182,T182,T182,T1845,71882,71862,C06,P206A, 206,P206B,P206C,P206D,P206E,T206H,T72 06A,T7206B,T1206C,T1206B,T1206E,T12 64,T1206B,T1206C,T1206B,T1206E,T12 64,T1206B,T1206C,T1206B,T1206E,T12 64,T1206B,T1206C,T1206B,T1206E,T12 65,N9210N,T2100T,T210K,T210T,T210K,T 10L,T210M,T210N,T240,T303,310,310B,31 C,310D,310E,310F,310G,310H,310I,310I,310 C,310D,310E,310F, |
| |

| SMALL AIRCRAFT | | | | |
|------------------|-------------------------|---|---|--|
| AD No. | Information | Manufacturer | Applicability | |
| | Information Key: E- Emo | ergency; COR - Correction; R - Replaces, A- Affects | | |
| | | | 1,321,335,340,340A,LC40-550FG,LC41- 550FG,LC42- 550FG,FT337E,FT337F,FT337GP,FT337HP, P337H,T337B,T337C,T337D,T337E,T337F,T 337G,T337H,T337H- SP,401,401A,401B,402,402A,402B,402C,404, 411,411A,414,414A,421,421A,421B,421C,A5 00,500-A,500-B,500-S,500-U,560-A,560- E,685,P.68C-TC,P.68TC Observer,EA-400 | |
| 2023-11-05 | R 2021-10-28 | Pilatus Aircraft Ltd. | PC-24 | |
| 2023-11-12 | | DAHER AEROSPACE | TBM 700 | |
| Biweekly 2023-14 | | | | |
| 2023-11-07 | R 2021-23-13 | Airbus Helicopters, Airbus Helicopters Deutschland GmbH, Air Space Design and Manufacturing LLC, Bell Textron Canada Limited, Bell Textron Inc., Brantly Internationa Inc., Centerpointe Aerospace Inc., Columbia Helicopters Inc., The Enstrom Helicopter Corporation, Erickson Air-Crane Incorporated DBA Erickson Air-Crane, Erickson Incorporated DBA Erickson Air- Crane, Hélicoptères Guimbal, Siam Hiller Holdings Inc., Kaman Aerospace Corporation, Leonardo S. p. a., MD Helicopters Inc., PZL-Swidnik S. A., Robinson Helicopter Company, Schweizer RSG LLC, Scotts-Bell 47 Inc, Sikorsky Aircraft Corporation | 47,206,210,212,222,230,234,280,305,369,407, 412,427,429,430,480,505,1100,107- II,204B,205A,205A-1,205B,206A,206A- 11,206L,206L-1,206L-3,206B- 1,206L,206L-1,206L-3,206L- 4,222B,222U,269A,269A-1,269B,269C,269C- 1,269D,280C,280F,280FX,369A,369D,369E,3 69F,369FF,369H,369HE,369HM,369HS,412C F,412EP,47B,47B3,47D,47D1,47E,47G,47G- 2,47G-2A,47G-2A-1,47G-3,47G-3B,47G-3B- 1,47G-3B-2,47G-3B-2A,47G-4A,47G-4A,47G- 5,47G-5A,47H-1,47J,47J-2,47J- 2A,47K,480B,500N,600N,A109,A109A,A109 A II,A109C,A109E,A109K2,A109S,A119,AB13 9,AB412,AB412 EP,AS332C,AS332C1,AS332L,AS332L1,AS3 32L2,AS350B,AS350B1,AS350B2,AS350B3, AS350BA,AS350C,AS350D1,AS350B2,AS350B3, AS350BA,AS350C,AS350D1,AS350D1,AS35 5E,AS355F,AS355F1,AS355F2,AS355N,AS3 55NP,AS-365N2,AS- 365N3,AW109SP,AW119 MKII,AW139,AW169,AW189,B-2,B-2A,B- 2B,BO-105A,BO-105C,BO-105LS A-1,BO- 105LS A-3,BO-105S,CABRI G2,CH- 47D,CH-54A,EC 155B,EC120B,EC130B4,EC130T2,EC155B1, EC225LP,F-28,F-28A,F-28C,F-28C-2,F-28C- 2R,F-28F,F-28F-R,FH-1100,K-190A,K- 240,K-600,MBB-BK 117 A-1,MBB-BK 117 A-3,MBB-BK 117 A-4,MBB-BK 117 B- 1,MBB-BK 117 B-2,MBB-BK 117 C-1,MBB- BK 117 C-2,MBB-BK 117 D-2,MBB-BK 117 D-3,MD900,OH-13E,OH-13H,PZL W- 3A,R22,R22 ALPHA,R22 BETA,R22 MARINER,R44,R44 II,R66,S-51,S-52,S-55,S- 55B,S-55C,S-58A,S-58B,S-58BT,S-58C,S- 58D,S-58DT,S-58E,S-58ET,S-58F,S-58FT,S- 58G,S-58H,S-58HT,S-58J,S-58J,S-58J,S-58D,S-58D,S-58B, | |

| AD No. | Information | Manufacturer | Applicability |
|------------------|-------------------------|--|--|
| | Information Key: E- Eme | ergency; COR - Correction; R - Replaces, A- Affects | |
| | | | Astazou,SA 318C-Alouette Astazou,SA.315B Alouette III,SA.316B Alouette III,SA.316C Alouette III,SA.319B Alouette III,SA330J,SA341G,SA342J,SA-365C,SA- 365C1,SA-365C2,SA-365N,SA-365N1,SA- 366G1,SE 313B-Alouette II,SE 3130-Alouette II,SE.3160 Alouette III,TH-1L,TH-28,UH- 12,UH-12A,UH-12B,UH-12C,UH-12E,UH- 12E-L,UH-12L,UH-12L4,UH-1E,UH-1L,K- 225 |
| 2023-13-51 | Е | Airbus Helicopters | SA341G,SA342J |
| Biweekly 2023-15 | | | |
| 2023-12-04 | | Pilatus Aircraft Ltd. | PC-24 |
| 2023-12-17 | R 2022-19-03 | Pilatus Aircraft Ltd. | PC-12,PC-12/47,PC-12/45,PC-12/47E |
| 2023-12-24 | | GE Aviation Czech s.r.o. | M601E-11AS,M601E-11S,H75-100,H80- |
| 2023-12-26 | R 2021–24–04 | Bell Textron Canada Limited | 100,H85-100 505 |
| 2023-13-14 | 2023-01-12 | Safran Helicopter Engines S.A. | Arriel 1C,Arriel 1C1,Arriel 1C2,Arriel 1K1 |
| Biweekly 2023-16 | | | |
| 2023-13-08 | R 2021–05–03 | Airbus Helicopters | EC225LP |
| 2023-13-51 | R 2022-19-08 | Airbus Helicopters | SA341G,SA342J |
| Biweekly 2023-17 | | | |
| 2023-14-06 | | Airbus Helicopters | EC120B,EC130B4,EC130T2 |
| 2023-14-07 | | Airbus Helicopters | EC155B1 |
| 2023-15-03 | | Safran Helicopter Engines S.A. | Arrius 2B2 |
| 2023-15-07 | | Air Tractor Inc. | AT-802,AT-802A |
| Biweekly 2023-18 | | | |
| 2023-15-06 | | Pilatus Aircraft Ltd. | PC-24 |
| 2023-16-04 | | Piaggio Aviation S.p.A. | P-180 |
| 2023-17-51 | Ε | Bell Textron Canada Limited | 407 |
| Biweekly 2023-19 | | | |
| 2023-17-05 | | Schempp-Hirth Flugzeugbau GmbH | Ventus-2a, Ventus-2b |
| 2023-17-09 | R 2022-13-03 | Cameron Balloons Ltd.,Aerostar International,Ballonbau Worner GmbH,B Kubicek spol s.r.o.,Eagle Balloons Corp., Aerosports Ltd.,Lindstrand Balloons Ltd.,McGrath Michael D. | Fuel Cylinders alony JR |
| 2023-17-13 | | BRP-Rotax GmbH & Co KG | 912 F2,912 F3,912 F4,912 iSc2 Sport,912 iSc3 Sport,912 S2,912 S3,912 S4,914 F2,914 F3,914 F4,AMT-200 (Super Ximango),HK 36 R SUPER DIMONA,HK 36 TC,HK 36 TS,DA20-A1,DV 20 Katana,Sky Arrow 650 TC,SF 25C |
| 2023-17-51 | | Bell Textron Canada Limited | 407 |

| AD No. | Information | Manufacturer | Applicability | |
|---|-------------|--------------|---------------|--|
| Information Key: E- Emergency; COR - Correction; R - Replaces, A- Affects | | | | |

Biweekly 2023-20

| 2023-16-06 | | Hamilton Sundstrand Corporation |
|------------|--------------|---|
| 2023-17-04 | R 2022-04-04 | Continental Aerospace Technologies Inc. |
| | | (Continental) |

C-145,C-125-1,C-125-2,GO-300-F,GO-300-E,GO-300-D,GO-300-C,GO-300-B,GO-300-A,IO-360,IO-360-A,IO-360-A1A,IO-360-A1A1,IO-360-A1A2,IO-360-A1B,IO-360-A1B6.IO-360-A1B6D.IO-360-A1C.IO-360-A1D,IO-360-A1D6,IO-360-A1D6D,IO-360-A2A,IO-360-A2A1,IO-360-A2A2,IO-360-A2B,IO-360-A2C,IO-360-A3A1,IO-360-A3A2,IO-360-A3B6,IO-360-A3B6D,IO-360-A3D6D,IO-360-A4A1,IO-360-A4A2,IO-360-A5A1,IO-360-A5A2,IO-360-A6A1,IO-360-A6A2,IO-360-AB,IO-360-AF,IO-360-B,IO-360-B1A,IO-360-B1A1,IO-360-B1A2,IO-360-B1B,IO-360-B1C,IO-360-B1D,IO-360-B1E,IO-360-B1F,IO-360-B1F6,IO-360-B1G6,IO-360-B2A1,IO-360-B2A2,IO-360-B2E,IO-360-B2F,IO-360-B2F6,IO-360-B3A1,IO-360-B4,IO-360-B3A2,IO-360-B4A,IO-360-B4A1,IO-360-B4A2,IO-360-B5A1,IO-360-B5A2,IO-360-B6A1,IO-360-B6A2, IO-360-C, IO-360-C1A, IO-360-C1A1, IO-360-C1A2, IO-360-C1B, IO-360-C1C,IO-360-C1C6,IO-360-C1D6,IO-360-C1E6,IO-360-C1E6D,IO-360-C1F,IO-360-C1G6,IO-360-C2A1,IO-360-C2A2,IO-360-C3A1,IO-360-C3A2,IO-360-C4A1,IO-360-C4A2,IO-360-C5A1,IO-360-C5A2,IO-360-C6A1,IO-360-C6A2,IO-360-CB,IO-360-D,IO-360-D1A,IO-360-D1A1,IO-360-D1A2,IO-360-D2A1,IO-360-D2A2,IO-360-D3A1,IO-360-D3A2.IO-360-D4A1.IO-360-D4A2.IO-360-D4A?1,IO-360-D5A1,IO-360-D5A2,IO-360-D6A1,IO-360-D6A2,IO-360-DB,IO-360-E,IO-360-E1A,IO-360-E1A1,IO-360-E1A2.IO-360-E2A1.IO-360-E2A2.IO-360-E3A1,IO-360-E3A2,IO-360-E4A1,IO-360-E4A2, IO-360-E5A1, IO-360-E5A2, IO-360-E6A1,IO-360-E6A2,IO-360-ES,IO-360-F1A,IO-360-G,IO-360-GB,IO-360-H,IO-360-HB,IO-360-J,IO-360-J1A6D,IO-360-J1AD,IO-360-JB,IO-360-K,IO-360-K2A,IO-360-KB,IO-360-L2A,IO-360-M1A,IO-360-M1B,IO-360-N1A,IO-360-P1A,IO-470-A,IO-470-C,IO-470-D,IO-470-E,IO-470-F,IO-470-G,IO-470-H,IO-470-J,IO-470-K,IO-470-L,IO-470-LO,IO-470-M,IO-470-N,IO-470-P,IO-470-R,IO-470-S,IO-470-T,IO-470-U,IO-470-V,IO-470-VO,IO-520-A,IO-520-B,IO-520-BA,IO-520-BB,IO-520-C,IO-520-CB,IO-520-D,IO-520-E,IO-520-F,IO-520-J,IO-520-K,IO-520-L,IO-520-M,IO-520-MB,IO-520-N,IO-520-NB,IO-520-P,LIO-520-P,IO-550-A,IO-550-B,IO-550-C,IO-550-D,IO-550-E,IO-550-F.

14SF-17,14SF-19

| AD No. | Information | Manufacturer | Applicability |
|------------------|------------------------|---|---|
| L | Information Key: E- Er | nergency; COR - Correction; R - Replaces, A- Affect | ts |
| | | | IO-550-G,IO-550-L,IO-550-N,IO-550-P,IO- 550-R,O-300-A,O-300-B,O-300-C,O-300- D,O-300-E,O-470-2,O-470-4,O-470-11,O- 470-11-CI,O-470-11B,O-470-11B-CI,O-470- 13,O-470-13A,O-470-15,O-470-A,O-470- B,O-470-B-CI,O-470-E,O-470-G,O-470-G- CI,O-470-H,O-470-J,O-470-K,O-470-K-CI,O- 470-L,O-470-L-CI,O-470-R,O-470-S,O-470-T,O- 470-U,TSIO-360-A,TSIO-360-AB,TSIO-360- B,TSIO-360-B,TSIO-360-C,TSIO-360- CB,TSIO-360-B,TSIO-360-CB,TSIO-360- E,TSIO-360-B,TSIO-360-B,TSIO-360- FB,TSIO-360-C,TSIO-360-B,TSIO-360- KB,TSIO-360-LB,TSIO-360-B,TSIO-360- KB,TSIO-360-LB,TSIO-360-AB,TSIO-360- KB,TSIO-520-A,TSIO-360-RB,TSIO-360- NB,TSIO-520-B,TSIO-360-RB,TSIO-360- SB,TSIO-520-B,TSIO-520-AE,TSIO-520- BE,TSIO-520-C,TSIO-520-CE,TSIO-520- BE,TSIO-520-JB,TSIO-520-K,TSIO-520- KB,TSIO-520-L,TSIO-520-K,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-NB,TSIO-520- KB,TSIO-520-R,TSIO-520-VB,TSIO-520- WB |
| 2023-17-10 | | Vulcanair S.p.A. | Vulcanair V1.0 |
| 2023-18-03 | | Viking Air Limited | DHC-3 |
| 2023-19-04 | | Aircraft Industries a.s. | L 410 UVP-E20,L 410 UVP-E20 CARGO,L- 420 |
| Biweekly 2023-21 | | | |
| 2023-17-07 | | Leonardo S.p.a. | A119,AW119 MKII |
| 2023-19-06 | R 64-09-03 | Viking Air Limited | DHC-2 Mk.II,DHC-2 Mk.I,DHC-2 Mk.III |
| 2023-20-51 | Ε | Airbus Helicopters | AS332C,AS332C1,AS332L,AS332L1,AS332 L2,SA330J |
| Biweekly 2023-22 | | | |
| 2023-20-07 | | Epic Aircraft LLC | E1000 |
| 2023-20-51 | | Airbus Helicopters | AS332C,AS332C1,AS332L,AS332L1,AS332 L2,SA330J |
| Biweekly 2023-23 | | | |
| No ADs | | | |
| Biweekly 2023-24 | | | |
| 2023-20-03 | | Austro Engine GmbH | E4,E4P |
| 2023-21-04 | | Embraer S.A. | EMB-505 |
| 2023-21-06 | | Embraer S.A. | EMB-505 |
| 2023-23-01 | R 2022-01-05 | Airbus Helicopters | EC130T2 |

| AD No. | Information | Manufacturer | Applicability | | | |
|---|--------------|--------------------------------|---------------------------|--|--|--|
| Information Key: E- Emergency; COR - Correction; R - Replaces, A- Affects | | | | | | |
| Biweekly 2023-25 | | | | | | |
| 2023-22-03 | R 2021–08–02 | Safran Helicopter Engines S.A. | Arriel 2D,Arriel 2E | | | |
| 2023-22-11 | | Embraer S.A. | EMB-505 | | | |
| 2023-22-14 | | Airbus Helicopters | SA-365C1,SA-365C2,SA-365N | | | |
| 2023-24-51 | E | Hélicoptères Guimbal | CABRI G2 | | | |

PART 39-AIRWORTHINESS DIRECTIVES

The authority citation for part 39 continues to read as follows:

[Amended]

The FAA amends §39.13 by:

Removing Airworthiness Directive 2021-08-02, Amendment 39-21496 (, May 17, 2021); and

Adding the following new airworthiness directive:

2023–22–03Safran Helicopter Engines, S.A. (Type Certificate Previously Held by Turbomeca, S.A.): Amendment 39–22587; Docket No. FAA–2022–1311; Project Identifier MCAI–2022–00624–E.

(a) Effective Date

This airworthiness directive (AD) is effective December 26, 2023.

(b) Affected ADs

This AD replaces AD 2021-08-02, Amendment 39-21496 (, May 17, 2021).

(c) Applicability

This AD applies to Safran Helicopter Engines, S.A. (type certificate previously held by Turbomeca, S.A.) Model Arriel 2D and Arriel 2E engines.

(d) Subject

Joint Aircraft Service Component (JASC) Code 7250, Turbine section.

(e) Unsafe Condition

This AD was prompted by the manufacturer revising the airworthiness limitations section (ALS) of the existing engine maintenance manual (EMM) to introduce new or more restrictive tasks and limitations for certain life-limited parts. The FAA is issuing this AD to prevent failure of life-limited parts. The unsafe condition, if not addressed, could result in uncontained release of a critical part, damage to the engine, and damage to the helicopter.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

(1) Within 90 days after the effective date of this AD, revise the ALS of the existing EMM or instructions for continued airworthiness and the existing approved maintenance or inspection program, as applicable, by incorporating the actions specified in paragraph (1) of European Union Aviation Safety Agency (EASA) AD 2022–0083, dated May 11, 2022 (EASA AD 2022–0083).

(2) The owner/operator (pilot) holding at least a private pilot certificate may perform the action required by paragraph (g)(1) of this AD for your engine and must enter compliance with the applicable paragraphs of this AD into the engine maintenance records in accordance with and . The record must be maintained as required by , , or .

(h) Provisions for Alternative Actions and Intervals

After the actions required by paragraph (g) of this AD have been done, no alternative actions and associated thresholds and intervals, including life limits, are allowed unless they are approved as specified in the provisions of the "Ref Publication" section of EASA AD 2022–0083.

(i) Alternative Methods of Compliance (AMOCs)

The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in . In accordance with , send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (j) of this AD and email to: .

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(j) Additional Information

For more information about this AD, contact Kevin Clark, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (781) 238–7088; email: .

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under and .

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2022–0083, dated May 11, 2022.

(ii) [Reserved]

(3) For EASA AD 2022–0083, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: . You may find this material on the EASA website at *ad.easa.europa.eu*.

(4) You may view this service information at FAA, Airworthiness Products Section, Operational Safety Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit or email .

Issued on October 26, 2023.

Caitlin Locke,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[Filed 11–17–23; 8:45 am]

BILLING CODE 4910–13–P

PART 39-AIRWORTHINESS DIRECTIVES

The authority citation for part 39 continues to read as follows:

[Amended]

The FAA amends §39.13 by adding the following new airworthiness directive:

2023–22–11Embraer S.A.: Amendment 39–22595; Docket No. FAA–2023–1504; Project Identifier MCAI–2023–00473–A.

(a) Effective Date

This airworthiness directive (AD) is effective December 26, 2023.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Embraer S.A. Model EMB–505 airplanes, as identified in Agência Nacional de Aviação Civil (ANAC) AD 2023–02–01R1, effective March 14, 2023 (ANAC AD 2023–02–01R1), certificated in any category.

(d) Subject

Joint Aircraft System Component (JASC) Code 2215, Autopilot Main Servo.

(e) Unsafe Condition

This AD was prompted by an occurrence of corrosion on the clutch retaining bolt of the aileron autopilot servo mount. The FAA is issuing this AD to address the corrosion in the clutch retaining bolt of the aileron autopilot servo mount. The unsafe condition, if not addressed, could result in failure of the clutch retaining bolt of the aileron autopilot servo mount, which could disengage the clutch from the drive pin and jam the aileron controls, resulting in reduced controllability of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, ANAC AD 2023–02–01R1.

(h) Exceptions to ANAC AD 2023-02-01R1

(1) Where ANAC AD 2023–02–01R1 refers to February 6, 2023, the effective date of ANAC AD 2023–02–01, this AD requires using the effective date of this AD.

(2) Where ANAC AD 2023–02–01R1 requires replacing a part with a new part, for the purposes of this AD, "new" means zero flight hours.

(3) Where the "NOTE" to Table 01 in ANAC AD 2023–02–01R1 specifies "If the airplane operation age and /or the flight hours criteria change before the SB accomplishment, the most restrictive criteria must be obeyed," this AD requires complying with the most restrictive criteria for each applicability range (in months and flight hours) in Table 01 of ANAC AD 2023–02–01R1.

(4) This AD does not adopt paragraph (d) of ANAC AD 2023–02–01R1.

(i) No Reporting Requirement

Although the service information referenced in ANAC AD 2023–02–01R1 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

(j) Alternative Methods of Compliance (AMOCs)

The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in . In accordance with , send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (k) of this AD or email to: . If mailing information, also submit information by email. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local Flight Standards District Office.

(k) Additional Information

For more information about this AD, contact Jim Rutherford, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (816) 329–4165; email: .

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under and .

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Agência Nacional de Aviação Civil (ANAC) AD 2023-02-01R1, effective March 14, 2023.

(ii) [Reserved]

(3) For ANAC AD 2023–02–01R1, contact ANAC, Continuing Airworthiness Technical Branch (GTAC), Rua Doutor Orlando Feirabend Filho, 230-Centro Empresarial Aquarius-Torre B-Andares 14 a 18, Parque Residencial Aquarius, CEP 12.246–190-São José dos Campos-SP, Brazil; phone: 55 (12) 3203–6600; email: ; website: *anac.gov.br/en/*. You may find this material on the ANAC website at *sistemas.anac.gov.br/certificacao/DA/DAE.asp*.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit or email .

Issued on October 30, 2023.

Victor Wicklund,

Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[Filed 11–17–23; 8:45 am]

BILLING CODE 4910-13-P

PART 39-AIRWORTHINESS DIRECTIVES

The authority citation for part 39 continues to read as follows:

[Amended]

The FAA amends §39.13 by adding the following new airworthiness directive:

2023–22–14Airbus Helicopters: Amendment 39–22598; Docket No. FAA–2023–1720; Project Identifier MCAI–2023–00003–R.

(a) Effective Date

This airworthiness directive (AD) is effective December 26, 2023.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Airbus Helicopters Model SA–365C1, SA–365C2, and SA–365N helicopters, certificated in any category.

(d) Subject

Joint Aircraft Service Component (JASC) Code: 6520, Tail rotor gearbox.

(e) Unsafe Condition

This AD was prompted by reports of damaged control rod dual bearings (dual bearings) installed on the tail rotor gearbox (TGB). The FAA is issuing this AD to inspect for particles in the TGB magnetic plug. The unsafe condition, if not addressed, could result in loss of yaw control and subsequent loss of control of the helicopter.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2023–0001, dated January 4, 2023 (EASA AD 2023–0001).

(h) Exceptions to EASA AD 2023–0001

(1) Where EASA AD 2023–0001 requires compliance in terms of flight hours, this AD requires using hours time-in-service.

(2) Where EASA AD 2023–0001 refers to the effective dates specified in paragraphs (h)(2)(i) and (ii) of this AD, this AD requires using the effective date of this AD.

(i) March 21, 2022 (the effective date of EASA AD 2022–0038, dated March 7, 2022).

(ii) The effective date of EASA AD 2023–0001.

(3) Where EASA AD 2023–0001 defines Groups, for Group 2, replace the text "SA 365 N helicopters with an affected part installed that has accumulated 500 flight hours (FH) or more since first installation on a helicopter," with "SA–365N helicopters with an affected part installed that has accumulated 500 or more total hours time-in-service on the affected part or the total hours time-in-service on the affected part cannot be determined."

(4) Where the service information referenced in EASA AD 2023–0001 permits a pilot to perform a check of the magnetic plug, this AD requires that action be performed by a person authorized under .

(5) Where Note 1 of EASA AD 2023–0001 specifies, "Helicopters that were under close monitoring on March 21 2022 (the effective date of EASA AD 2022–0038) must continue the close monitoring procedure up to the first inspection accomplished in accordance with the instructions of ASB 1;" for this AD, replace that text with, "Helicopters that are under close monitoring as of the effective date of this AD must continue close monitoring until the first instance of the requirements in paragraph (1) of EASA AD 2023–0001 are completed."

(6) Where EASA AD 2023–0001 requires replacing the TGB and the service information referenced in EASA AD 2023–0001 specifies replacing the TGB, for this AD, before further flight, remove the TGB from service and replace it with an airworthy part, or repair the TGB in accordance with a method approved by the Manager, Europe Middle East & Africa Section, International Validation Branch, FAA; EASA; or Airbus Helicopters' Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(7) Where paragraphs (5) and (6) of EASA AD 2023–0001 require replacing an affected part, as defined in EASA AD 2023–0001, with a serviceable part, as defined in EASA AD 2023–0001; for this AD, remove the affected part, as defined in EASA AD 2023–0001, from service and replace it with a serviceable part, as defined in EASA AD 2023–0001.

(8) Where paragraph (5) of EASA AD 2023–0001 does not specify a compliance time; for this AD, the compliance time for those actions is before further flight.

(9) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies to do the actions identified in paragraphs (h)(9)(i) and (ii) of this AD, this AD does not include those requirements.

(i) Comply with paragraph 2.D., except this AD requires reporting information, including the information in Appendix 4. of the service information, in accordance with paragraph (h)(18) of this AD.

(ii) Send parts and particles to Airbus Helicopters.

(10) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies replacing the chip detector or conical housing chip detector if there is an anomaly; for this AD, an anomaly may be indicated by the magnetic component of the TGB chip detector or the conical housing chip detector

not being magnetized. If there is an anomaly, this AD requires before further flight, removing from service the TGB chip detector or the conical housing chip detector, as applicable to your model helicopter.

(11) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies making sure that the chip detector or conical housing chip detector is in good condition; for this AD, good condition for the chip detector is indicated when there are no signs of wear on the locking systems (including wear on the bayonets and slotted tubes). If there are any signs of wear on the locking systems, this AD requires, before further flight, removing the TGB chip detector is magnetized. If the conical housing chip detector is not magnetized, this AD requires, before further flight, removing the further flight, removing the conical housing chip detector is magnetized. If the conical housing chip detector from service.

(12) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies replacing the O-rings if necessary; this AD requires, before further flight, removing any affected O-ring from service and replacing it with an airworthy O-ring.

(13) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies removing an affected TGB, returning it to an approved workshop, including sending all the particles found in the affected part; this AD requires, before further flight, removing an affected TGB from service and replacing it with an airworthy part, or repairing the TGB in accordance with a method approved by the Manager, Europe Middle East & Africa Section, International Validation Branch, FAA; EASA; or Airbus Helicopters' DOA. If approved by the DOA, the approval must include the DOA-authorized signature. You are not required to send the particles found in the TGB to Airbus Helicopters or send an affected TGB to an approved workshop.

(14) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies to use tooling, this AD allows the use of equivalent tooling.

(15) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies discarding certain parts, this AD requires removing those parts from service.

(16) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies performing a metallurgical analysis of particles if there is a doubt concerning the type, size, or classification of any collected particle, this AD requires, before further flight, performing a metallurgical analysis if the type, size, or classification of any collected particle cannot be determined.

(17) Where the service information (including any work card) referenced in EASA AD 2023–0001 specifies if there is any doubt remaining (pertaining to particle classification) after performing a metallurgical analysis, contact Airbus, this AD requires, if the type, size, or classification of any collected particle cannot be determined after performing a metallurgical analysis, before further flight, removing an affected TGB from service and replacing it with an airworthy part, or repairing the TGB in accordance with a method approved by the Manager, Europe Middle East & Africa Section, International Validation Branch, FAA; EASA; or Airbus Helicopters' DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(18) Where the service information referenced in EASA AD 2023–0001 requires reporting inspection results, including Appendix 4.A., to Airbus Helicopters, if any M50 particles are found, this AD requires reporting those inspection results along with a detailed description of any information and findings, and if possible, provide photos, at the applicable time in paragraph (h)(18)(i) or (ii) of this AD.

(i) If the inspection was done on or after the effective date of this AD: Submit the report within 10 days after accomplishing the metallurgical analysis.

(ii) If the inspection was done before the effective date of this AD: Submit the report within 10 days after the effective date of this AD.

(19) This AD does not adopt the "Remarks" section of EASA AD 2023-0001.

(i) Special Flight Permits

Special flight permits may be issued in accordance with and , provided no passengers are onboard.

(j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in . In accordance with , send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD. Information may be emailed to: .

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(k) Related Information

For more information about this AD, contact Kevin Kung, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone (781) 238–7244; email.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under and .

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2023–0001, dated January 4, 2023.

(ii) [Reserved]

(3) For EASA AD 2023–0001, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ; internet *easa.europa.eu*. You may find the EASA material on the EASA website *ad.easa.europa.eu*.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Parkway, Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit or email .

Issued on October 30, 2023.

Victor Wicklund,

Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[Filed 11–17–23; 8:45 am]

BILLING CODE 4910–13–P



EMERGENCY AIRWORTHINESS DIRECTIVE www.faa.gov/aircraft/safety/alerts/

DATE: November 21, 2023 AD #: 2023-24-51

Emergency Airworthiness Directive (AD) 2023-24-51 is sent to owners and operators of Hélicoptères Guimbal Model Cabri G2 helicopters.

Background

This emergency AD was prompted by reports of a crack in the pilot cyclic stick base. The European Union Aviation Safety Agency (EASA), which is the aviation authority for the European Union, has issued EASA Emergency AD 2023-0204-E, dated November 20, 2023 (EASA AD 2023-0204-E), to correct an unsafe condition on Hélicoptères Guimbal Model Cabri G2 helicopters. EASA AD 2023-0204-E states that further investigation determined that the root cause of the cracks is fatigue, primarily related to induced loads on the cyclic stick during pre-flight (free play) checks. Accordingly, EASA AD 2023-0204-E requires repetitively inspecting certain part-numbered pilot and co-pilot cyclic stick bases and, depending on the results, corrective action. EASA AD 2023-0204-E also prohibits installing those pilot and co-pilot cyclic stick bases unless its requirements are met.

This emergency AD is intended to detect a cracked pilot or co-pilot cyclic stick base. This condition, if not addressed, could result in failure of the pilot or co-pilot cyclic stick base and subsequent loss of control of the helicopter.

Related Service Information

The FAA reviewed EASA AD 2023-0204-E, which requires repetitively inspecting pilot cyclic stick base part number (P/N) G41-42-801 and co-pilot cyclic stick base P/Ns G41-43-801 and G41-43-802 for a crack. Depending on the results, EASA AD 2023-0204-E requires contacting HG [Hélicoptères Guimbal] for approved instructions to replace a cracked cyclic stick base and accomplishing those instructions accordingly. EASA AD 2023-0204-E also allows removing the dual control (co-pilot cyclic stick) instead of replacing a cracked co-pilot cyclic stick base. Lastly, EASA AD 2023-0204-E prohibits installing a specified pilot or co-pilot cyclic stick base unless it is a new (never installed before) part or, before installation, has passed its required inspection.

The FAA also reviewed Guimbal Mandatory Service Bulletin SB 23-006, Revision B, dated November 14, 2023 (SB 23-006B) which specifies an initial and repetitive inspections of both the pilot and copilot cyclic bases for cracks. SB 23-006B specifies doing the inspection using a flashlight and in case of doubt, performing a dye-penetrant inspection. If there is a crack on the pilot's side, SB 23-006B specifies grounding the helicopter and contacting HG [Hélicoptères Guimbal]; if there is a crack on the copilot's side, SB 23-006B specifies removing the dual controls and contacting HG.

FAA's Determination

These products have been approved by the aviation authority of another country and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in its emergency AD

described above. The FAA is issuing this emergency AD after determining that the unsafe condition described previously is likely to exist or develop on other products of the same type design.

Emergency AD Requirements

This emergency AD requires accomplishing the actions specified in EASA AD 2023-0204-E, described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this emergency AD and except as discussed under "Differences Between this Emergency AD and the EASA Emergency AD."

Explanation of Required Compliance Information

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA developed a process to use some civil aviation authority (CAA) ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has been coordinating this process with manufacturers and CAAs. As a result, EASA AD 2023-0204-E is incorporated by reference in this FAA emergency AD. This emergency AD, therefore, requires compliance with EASA AD 2023-0204-E in its entirety through that incorporation, except for any differences identified as exceptions in the regulatory text of this emergency AD. Using common terms that are the same as the heading of a particular section in EASA AD 2023-0204-E does not mean that operators need comply only with that section. For example, where the emergency AD requirement is not limited to the section titled "Required Action(s) and Compliance Time(s)" in EASA AD 2023-0204-E.

Differences Between this Emergency AD and the EASA Emergency AD

The service information referenced in EASA AD 2023-0204-E specifies performing a dyepenetrant inspection in case of a doubt regarding if there is a crack, whereas this emergency AD does not require that action. If there is cracked pilot or co-pilot cyclic stick base, EASA AD 2023-0204-E requires contacting HG [Hélicoptères Guimbal] for approved instructions to replace it with a serviceable part and accomplishing those instructions accordingly and the service information referenced in EASA AD 2023-0204-E specifies contacting HG [Hélicoptères Guimbal] or removing the dual controls and contacting HG [Hélicoptères Guimbal], whereas this emergency AD requires removing the cracked cyclic stick base from service and replacing it with a serviceable cyclic stick base in accordance with a method approved by the FAA, EASA, or Hélicoptères Guimbal EASA Design Organization Approval.

Justification for Immediate Adoption and Determination of the Effective Date

Section 553(b)(3)(B) of the Administrative Procedure Act (APA) (5 U.S.C. 551 *et seq.*) authorizes agencies to dispense with notice and comment procedures for rules when the agency, for "good cause," finds that those procedures are "impracticable, unnecessary, or contrary to the public interest." Under this section, an agency, upon finding good cause, may issue a final rule without providing notice and seeking comment prior to issuance. Further, section 553(d) of the APA authorizes agencies to make rules effective in less than thirty days, upon a finding of good cause.

An unsafe condition exists that requires the immediate adoption of this emergency AD to all known U.S. owners and operators of these helicopters. The FAA has found that the risk to the flying public justifies foregoing notice and comment prior to adoption of this rule because the affected component is part of an assembly that is critical to the control of a helicopter. As the FAA also has no information pertaining to the quantity of cracked components that may currently exist in the U.S. fleet or how quickly the condition may propagate to failure, the actions required by this emergency

AD must be accomplished before further flight for certain helicopters. Accordingly, notice and opportunity for prior public comment are impracticable and contrary to the public interest pursuant to 5 U.S.C. 553(b)(3)(B).

In addition, the FAA finds that good cause exists pursuant to 5 U.S.C. 553(d) for making this amendment effective in less than 30 days, for the same reasons the FAA found good cause to forego notice and comment.

Authority for this Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Presentation of the Actual Emergency Airworthiness Directive

The FAA is issuing this emergency airworthiness directive under 49 U.S.C. 106(g), 40113, and 44701 according to the authority delegated to me by the Administrator.

2023-24-51 Hélicoptères Guimbal: Project Identifier MCAI-2023-01201-R.

(a) Effective Date

This emergency airworthiness directive (AD) is effective upon receipt.

(b) Affected ADs

None.

(c) Applicability

This emergency AD applies to Hélicoptères Guimbal Model Cabri G2 helicopters, certificated in any category.

(d) Subject

Joint Aircraft Service Component (JASC) Code: 6710, Main Rotor Control.

(e) Unsafe Condition

This emergency AD was prompted by reports of a crack in the pilot cyclic stick base. The FAA is issuing this emergency AD to detect a cracked pilot or co-pilot cyclic stick base. The unsafe condition, if not addressed, could result in failure of the pilot or co-pilot cyclic stick base and subsequent loss of control of the helicopter.

(f) Compliance

Comply with this emergency AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this emergency AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) Emergency AD 2023-0204-E, dated November 20, 2023 (EASA AD 2023-0204-E).

(h) Exceptions to EASA AD 2023-0204-E

(1) Where EASA AD 2023-0204-E defines "the SB," this emergency AD requires using Guimbal Mandatory Service Bulletin SB 23-006, Revision B, dated November 14, 2023.

(2) Where EASA AD 2023-0204-E refers to its effective date, this emergency AD requires using the effective date of this emergency AD.

(3) Where EASA AD 2023-0204-E requires compliance in terms of flight hours, this emergency AD requires using hours time-in-service.

(4) Where Table 1 in EASA AD 2023-0204-E states, "Compliance Time after the Effetive Date," for this emergency AD, replace that text with, "Compliance Time after the Effective Date."

(5) Where Note (1) of EASA AD 2023-0204-E states, "For the initial inspection, a single ferry flight without passengers is allowed to a maintenance location, where the actions required by this AD can be accomplished," for this emergency AD, replace that text with, "For the initial inspection, a single special flight permit may be issued in accordance with 14 CFR 21.197 and 21.199 to a maintenance location where the actions required by this emergency AD can be accomplished, provided there are no passengers onboard."

(6) Where the service information referenced in EASA AD 2023-0204-E states performing a dye-penetrant inspection, this emergency AD does not require that action.

(7) Instead of complying with paragraphs (2) and (3) of EASA AD 2023-0204-E and paragraph d) of the service information referenced in EASA AD 2023-0204-E, for this emergency AD, comply with the following: "As a result of an inspection required by paragraph (1) of EASA AD 2023-0204-E, if there is a crack, before further flight, remove the affected part, as defined in EASA AD 2023-0204-E, from service and replace it with a serviceable part, as defined in EASA AD 2023-0204-E, in accordance with a method approved by the Manager, International Validation Branch, FAA; or EASA; or Hélicoptères Guimbal EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature."

(8) This emergency AD does not adopt the "Remarks" section of EASA AD 2023-0204-E.

(i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this emergency AD, if requested using the procedures found in § 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (j)(1) of this emergency AD or email to: 9-AVS-AIR-730-AMOC@faa.gov. If mailing information, also submit information by email.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(j) Additional Information

(1) For more information about this emergency AD, contact Dan McCully, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone (404) 474-5548; email william.mccully@faa.gov.

(2) For Guimbal service information identified in this emergency AD, contact Hélicoptères Guimbal, 1070, rue du Lieutenant Parayre, Aérodrome d'Aix-en-Provence, 13290 Les Milles, France; phone 33-04-42-39-10-88; email support@guimbal.com; or at guimbal.com. You may also view this referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222-5110.

(3) The subject of this emergency AD is addressed in EASA Emergency AD 2023-0204-E, dated November 20, 2023. For this EASA material, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; Internet easa.europa.eu. You may find this EASA material on the EASA website at ad.easa.europa.eu. You may also view this EASA material at the FAA address identified in paragraph (j)(2) of this emergency AD.

Issued on November 21, 2023.

Caitlin Locke, Director, Compliance & Airworthiness Division, Aircraft Certification Service.