

SUBJECT: TIME LIMITS
Inspection of the Tail Rotor Blade Assembly

REMARKS:

- This Service Bulletin forms the subject of the 30th AS 350 Airworthiness Directive.
- Revision 1 forming the subject of the 32nd AS 350 Airworthiness Directive is intended:
 - . to schedule the inspection detailed in paragraph 1C (1) every 10 flying hours.
 - . to allow 10 hours to elapse after detection of the sound of broken fibers prior to performing inspection with removal as per in paragraph 1C (2).
 - Revision 2 forming the subject of the 37th AS 350 Airworthiness Directive is intended:
 - . to extend the check without removal from 10 to 30 hours.
 - Revision 3 of this Service Bulletin, which forms the subject of the 37th AS 350 Airworthiness Directive Revision 1:
 - . modifies the check interval from 300 hours to 400 flying hours.
 - Revision 4 of this Service Bulletin, which forms the subject of the 37th AS 350 Airworthiness Directive Revision 2:
 - . includes the new tail rotor blade assembly part numbers in paragraph 1A (only page 1 is concerned).
 - Revision 4 index "a" of this Service Bulletin, which does not modify the terms of the 37th AS 350 Airworthiness Directive Revision 2, corrects the Tail Rotor Blade assembly Part Numbers in paragraph 1A.
 - Revision 5 of this Service Bulletin forms the subject of Revision 3 of the 37th AS 350 Airworthiness Directive, except for paragraph 1D(3), which is classified "Recommended" by the aircraft manufacturer, and:
 - . reflects the extension of the type "T" inspection interval from 400 to 500 hours.
 - . specifies the compliance conditions in paragraph 1D.
 - . classifies paragraph 1D (3) "Recommended".
 - Revision 6 of this Service Bulletin informs that this Service Bulletin is no longer applicable following the integration of its contents in the documentation:
 - . for B2 and B3 helicopters versions:
 - . Airworthiness Limitations Section (ALS),
 - . Aircraft Maintenance Manual (AMM),
 - . Master Servicing Manual (MSM).
 - . for other versions:
 - . Airworthiness Limitations Section (ALS),
 - . Maintenance Manual (MET),
 - . Master Servicing Manual (MSM).

Corresponds to modification: Not applicable

OFFICE OF APPROVAL
D.G.A.C. Approved : December 1982

Approval by Direction Générale de l'Aviation Civile (D.G.A.C) is limited to helicopters subject to Airworthiness Certificate .

following the integration of its contents in the documentation
This Service Bulletin is no longer applicable

1 - PLANNING INFORMATION

A- APPLICABILITY
AS 350 Helicopters, All Versions.

Tail rotor blade assembly P/N	350A08-1011 350A12-0020 350A12-0030 355A12-0031 355A12-0040	}	All dash numbers
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B- REASON
To check the tail rotor blade assembly for condition.

C- DESCRIPTION
Not applicable.

D- COMPLIANCE
Not applicable.

Export Control:
US Export Control - No US content. This Item does not contain any U.S. origin ITAR or EAR content.
FR Export Control - Not Listed. This Item is not listed against the EC regulations in the EU/FR

SERVICE

BULLETIN

E- APPROVAL

The technical content of this document is approved under the authority of the Design Organization Approval ref. EASA. 21J.700.

The technical content of this document is approved under the prerogatives of the recognition of design capability ref. EMAR21J-015-DGA for French Government helicopters.

The technical content of this document is approved by Airbus Helicopters Airworthiness Department for export military versions.