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AgustaWestland Products

SERVICE BULLETIN

ALERT

N° 139-686

DATE: November 8, 2021 **REV.:** /

TITLE

ATA 53 – TAIL GEAR BOX FITTING INSPECTION

REVISION LOG

First Issue

An appropriate entry should be made in the aircraft log book upon accomplishment. If ownership of aircraft has changed, please, forward to new owner.



1. PLANNING INFORMATION

A. EFFECTIVITY

All AB/AW139 helicopters installing TGB fittings P/N 3G5351A01151 or P/N 3G5351A01152 with more than 3000 FH.

<u>NOTE</u>

During the installation of a new/repaired tail boom assembly, verify the applicability of this SB on the TGB fitting installed on the tail boom.

B. COMPLIANCE

<u>Part I:</u>

Within and not later than 25 (twenty-five) FH after the issue date of this Service Bulletin. Part II:

Immediately after compliance with Part I, if evidence of cracks is found during the inspection.

C. CONCURRENT REQUIREMENTS

N.A.

D. REASON

This Service Bulletin is issued in order to provide the necessary instruction on how to perform a detailed inspection of the TGB fitting.

E. DESCRIPTION

This Service Bulletin prescribes a detailed inspection of the TGB fitting in order to detect potential cracks. If cracks are found, the Service Bulletin prescribes the immediate replacement of the component.

Part I prescribes the detailed inspection of RH and FWD parts of the TGB fitting with the mandatory use of a borescope.

Part II prescribes the TGB fitting replacement in case of findings found during the application of the Part I.



F. APPROVAL

The technical content of this Service Bulletin is approved under the authority of DOA nr. EASA.21.J.005. For helicopters registered under other Aviation Authorities, before applying the Service Bulletin, applicable Aviation Authority approval must be checked within Leonardo Helicopters customer portal.

EASA states mandatory compliance with inspections, modifications or technical directives and related time of compliance by means of relevant Airworthiness Directives. If an aircraft listed in the effectivity embodies a modification or repair not LHD certified and affecting the content of this Service Bulletin, it is responsibility of the Owner/Operator to obtain a formal approval by Aviation Authority having jurisdiction on the aircraft, for any adaptation necessary before incorporation of the present Service Bulletin.

G. MANPOWER

To comply with this Service Bulletin, the following Maintenance-Man-Hours (MMH) are deemed necessary:

<u>Part I:</u> approximately four (4) MMH;

Part II: approximately thirty-six (36) MMH.

Maintenance-Man-Hours are based on hands-on time and can change with personnel and facilities available.

H. WEIGHT AND BALANCE

N.A.

I. REFERENCES

1) PUBLICATIONS

Following Data Modules refer to AMP:

DATA I	MODULE	DESCRIPTION	<u>PART</u>		
DM01	39-A-00-20-00-00A-120A-A	Helicopter on ground for a safe maintenance.	I, II		
DM02	39-A-52-44-01-00A-520A-A	Access panels - Remove procedure.	II		
DM03	39-A-52-44-01-00A-720A-A	Access panels - Install procedure.	П		
Following Data Modules refer to ASRP:					
DATA I	MODULE	DESCRIPTION	<u>PART</u>		
DM04	39-A-53-52-07-00A-920A-D	Tail gearbox fitting - Replacement	П		



2) ACRONYMS & ABBREVIATIONS

- AMDI Aircraft Material Data Information
- AMP Aircraft Maintenance Publication
- AMPI Aircraft Maintenance Planning Information
- ASRP Aircraft Structural Repair Publication
- DM Data Module
- DOA Design Organization Approval
- EASA European Aviation Safety Agency
- FH Flight Hours
- FWD Forward
- LHD Leonardo Helicopters
- MMH Maintenance Man Hours
- P/N Part Number
- S/N Serial Number
- TGB Tail Gear Box

3) ANNEX

N.A.

J. PUBLICATIONS AFFECTED

N.A.

K. SOFTWARE ACCOMPLISHMENT SUMMARY

N.A.



2. MATERIAL INFORMATION

A. REQUIRED MATERIALS

1) PARTS

<u>PART I</u>

N.A.

<u>PART II</u>

#	P/N	ALTERNATIVE P/N	DESCRIPTION	Q.TY	LVL	NOTE	LOG P/N
1	3G5351A01152		TGB fitting	1		(1)	-

Refer also to IPD for the spares materials required to comply with the AMP DMs referenced in the accomplishment instructions.

2) CONSUMABLES

The following consumable materials, or equivalent, are necessary to accomplish this Service Bulletin:

#	SPEC./LHD CODE NUMBER	DESCRIPTION	Q.TY	NOTE	PART
2	CCC-C-440, Class I	Cheesecloth (C028)	AR		I
3	TT-N-95-B Code No. 531055030	Aliphatic Naphtha (C059)	AR		I
4	P-D-680, Type II or MIL-PRF-680B, Type II	Cleaning solvent (C010)	AR		I

Refer also to AMDI for the consumable materials required to comply with the AMP DM referenced in the accomplishment instructions.

3) LOGISTIC MATRIX

N.A.

NOTE

(1) For helicopters installing the fitting P/N 3G5351A01151 it is possible also to install P/N 3G5351A01151 as a valid alternative to P/N 3G5351A01152.

B. SPECIAL TOOLS

The following special tools, or equivalent, are necessary to accomplish this Service Bulletin:

#	P/N	DESCRIPTION	Q.TY	NOTE	PART
5	RMGE-DS-06-2010-LH or approved alternative	Platform left (GG-01-00)	1	(B1)	I, II
6	RMGE-SL-06-2010-RH or approved alternative	Platform right (GG-02-00)	1	(B1)	I, II



#	P/N	DESCRIPTION	Q.TY	NOTE	PART
7	Commercial	Borescope	1	(B1)	I
8	Commercial	Flash light	1	(B1)	I
9	3G5305G12331	Tool TGB/IGB fitting inst	1	(B2) (B3)	П

Refer also to ITEP for the special tools required to comply with the AMP DM referenced in the accomplishment instructions.

SPECIAL TOOLS NOTES

(B1) Item to be procured as a local supply.

- (B2) Tool P/N 3G5305G12331 is required to perform Part II and may be supplied as part of tool P/N 3G5350G00831.
- (B3) Please contact Leonardo Helicopters Division order administration to request the tools supply on loan. As soon as the present Service Bulletin is implemented the tools supplied on loan shall be promptly returned to Leonardo Helicopters Division.

C. INDUSTRY SUPPORT INFORMATION

In ref. to PART II application:

As reported on PART I step 3, only if Product Support Engineering confirms the replacement, please Issue relevant MMIR form to your Warranty Administration Dpt. Please note that "Product Support Engineering's approvals" is mandatory to evaluate your request, otherwise MMIR could be rejected.

Owners/Operators who comply with the instructions of this Service Bulletin no later than the applicable date in the "Compliance" section will be eligible to receive required materials on free of charge basis, except for Consumable Materials and Special Tools. NOTE: Customers who fail to comply with the instructions in this Service Bulletin before

the compliance date are not eligible for the aforementioned special policy.

Please Issue relevant MMIR form to your Warranty Administration Dpt

3. ACCOMPLISHMENT INSTRUCTIONS

GENERAL NOTES

- a) Place an identification tag on all components that are re-usable, including the attaching hardware removed to gain access to the modification area and adequately protect them until their later re-use.
- b) Exposed thread surface and nut must be protected using a layer of tectyl according to MIL-C-16173 grade I.

<u>PART I</u>

- 1. In accordance with AMP DM 39-A-00-20-00-00A-120A-A, prepare the helicopter on ground for a safe maintenance. Disconnect the battery, all electrical power sources and/or the external power supply.
- 2. Put the platform (GG-01-00), or an approved alternative, adjacent to the left side of the fuselage.
- 3. Put the platform (GG-02-00), or an approved alternative, adjacent to the right side of the fuselage
- 4. In accordance with AMP DM 39-A-52-44-01-00A-520A-A, remove the access panels 360AL, 360AT and 360CT to get access to the right side of the fin.

<u>NOTE</u>

With reference to the following step, the use of borescope is mandatory to comply with the required inspection.

- 5. With reference to Figures 1 and 2 perform, with the use of a borescope, a detailed inspection for cracks of the RH and FWD sides of the TGB fitting (area delimitated by circles in Figure 1) in accordance to the following procedure:
 - 5.1 Properly illuminate the area to inspect (if necessary use a flash light).
 - 5.2 Clean the area to inspect with the Cheesecloth (C028) and the Aliphatic naphtha (C059) or Cleaning solvent (C010).
 - 5.3 With reference to Figure 1 View C and View D, accurately examine the outer surfaces of the forward and right side of TGB fitting body above the horizontal row of fastener holes (area delimitated by circles in figure), making sure there are not cracks or damage.
 - 5.4 Accurately examine the area around the vertical and horizontal row of fastener



holes (refer also to Figure 2 "View looking forward side of TGB fitting") of the inner forward side of the tail gearbox attachment-fitting.

- 5.5 Accurately examine the area around the vertical and horizontal row of fastener holes (refer also to Figure 2 "View looking right side of TGB fitting") of the inner right side of the tail gearbox attachment-fitting.
- 6. In case of any finding resulting from the inspection required by step 5, take a picture of the crack and of the involved area and contact Product Support Engineering (engineering.support.lhd@leonardocompany.com) to send photos of the cracks. Record for compliance with Part I of this Service Bulletin on the helicopter logbook and immediately proceed with Part II.
- 7. In accordance with AMP DM 39-A-52-44-01-00A-720A-A, reinstall the access panels 360AL, 360AT and 360CT.
- 8. Remove the platforms from the fuselage sides.
- 9. Return the helicopter to flight configuration and record for compliance with Part I of this Service Bulletin on the helicopter logbook.
- 10. Send the attached compliance form to the following mail box:

engineering.support.lhd@leonardocompany.com

As an alternative, gain access to My Communications section on Leonardo WebPortal and compile the "Service Bulletin Application Communication".



<u>PART II</u>

1. In accordance with AMP DM 39-A-00-20-00-00A-120A-A, prepare the helicopter on ground for a safe maintenance. Disconnect the battery, all electrical power sources and/or the external power supply.

<u>NOTE</u>

With reference to step 2, apply ASRP DM 39-A-53-52-07-00A-920A-D also to install P/N 3G5351A01151, if required.

- 2. In accordance with ASRP DM 39-A-53-52-07-00A-920A-D replace the damaged TGB fitting with a new one.
- 3. In accordance with AMP DM 39-A-52-44-01-00A-720A-A, install all the access panels previously removed.
- 4. Remove the platforms from the fuselage sides.
- 5. Return the helicopter to flight configuration and record for compliance with Part II of this Service Bulletin on the helicopter logbook.
- 6. Send the attached compliance form to the following mail box:

engineering.support.lhd@leonardocompany.com

As an alternative, gain access to My Communications section on Leonardo WebPortal and compile the "Service Bulletin Application Communication".



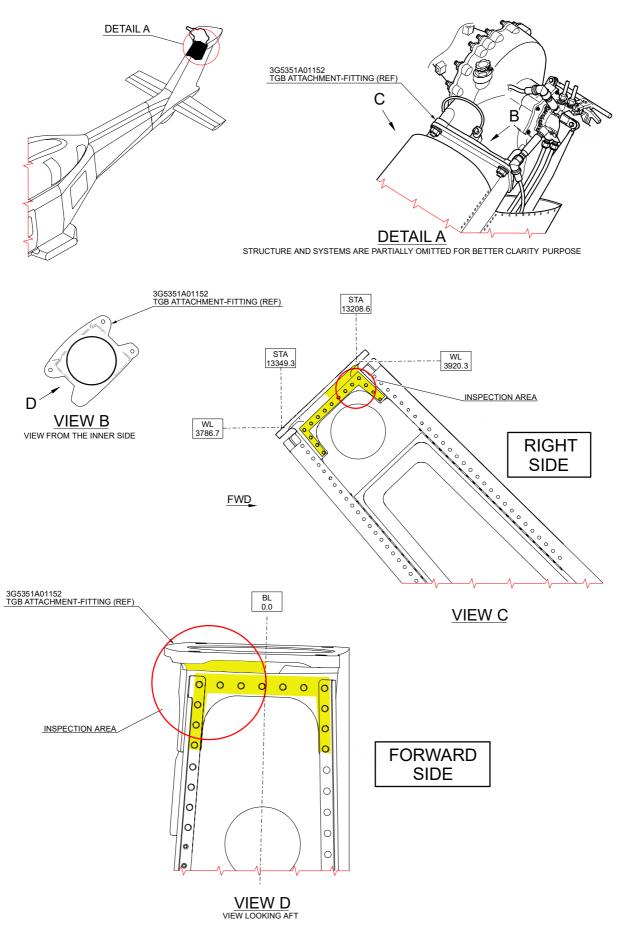
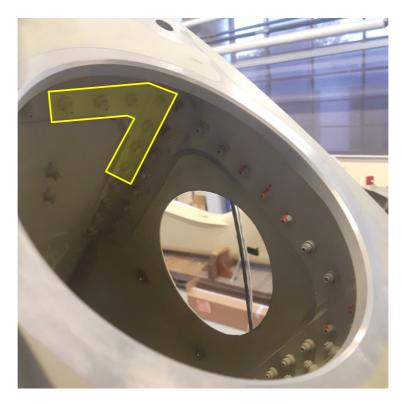
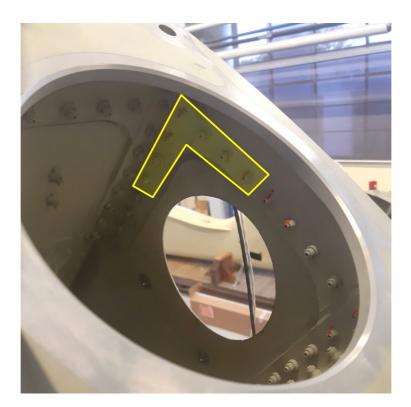


Figure 1





VIEW LOOKING FORWARD SIDE OF TGB FITTING (FOR REFERENCE ONLY)



VIEW LOOKING RIGHT SIDE OF TGB FITTING (FOR REFERENCE ONLY)

S.B. N°139-686 ALERT DATE: November 8, 2021 REVISION: / Figure 2





Please send to the following address: LEONARDO S.p.A. CUSTOMER SUPPORT & SERVICES - ITALY		SERVICE BULLETIN COMPLIANCE FORM Date:				Date:
		Number:				
PRODUCT SUPPORT ENGINEE Via Giovanni Agusta, 520	RING & LICENSES DEPT.					
21017 Cascina Costa di Samara Tel.: +39 0331 225036 Fax: +39	ate (VA) - ITALY 0331 225988	Revision:				
Customer Name and Addre	ess:			Telephone:		
				Fax:		
				B.T. Compli	ance Date:	
Helicopter Model	S/N		Total N	umber	Total Hours	T.S.O.
Remarks:						
Information:	Information:					

We request your cooperation in filling this form, in order to keep out statistical data relevant to aircraft configuration up-to-date. The form should be filled in all its parts and sent to the above address or you can communicate the application also via Technical Bulletin Application Communication Section placed in Leonardo AW Customer Portal - MyCommunications Area. We thank you beforehand for the information given.