

GALAXY AEROSPACE (M) SDN. BHD. [1040262-D]

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WORKSHEET

CLIENT/OWNER: ROYAL MALAYSIA WORKSHEET NO: 14633-001 SERIAL NO. HOURS LDG/CYCLE POLICE WORK/INSP/DESC: 50 FH INSPECTION AIRCRAFT 31726 REFER WORKPACK AIRCRAFT TYPE: AW139 WORKPACK REF: 9M-PMB-14633 PCE-KB1830 #1 ENGINE: REFER WORKPACK REGISTRATION: 9M-PMB AJL REF NO.: REFER WORKPACK REFER WORKPACK #2 ENGINE: PCE-KB1832 BASE/FACILITY: WMSA SHEET: 16 OF 19 DATE IN: REFER NG/N1 NF/N2 Reason for raising: Raised by and date: Other requirements/information: SCHEDULED INSPECTIONS PERFORM I.A.W AIRCRAFT MAINTENANCE PROGRAM RMPAOF/ CAMO; Hanis NIL CAMO/AMP/AW139 ISSUE 3 REV. 0 DATED 03 FEB 2021, IETP - 42ND ISSUE DATE 2022-12-01 17/02/2023 UPDATED: 2023-01-27, EMM PTC6C-67C 30.2 REVISION DATE 2023-02-07 OR LATER APPROVED REVISIONS Technician * Eng. CRS Date Description Item 62-48 MAIN ROTOR TENSION LINK 69 DO A DI FOR PRESENCE OF CRACKS OF THE DROOP STOP SUPPORT. REFERENCE (DMC): 39-A-62-22-00-00A-31AK-A REMARKS: 40. SATU-8107 63-35 ROTOR BRAKE COVER 70 DO A GVI FOR CONDITION AND INTEGRITY OF ATTACHMENTS REFERENCE (DMC): 39-A-63-50-00-00A-310A-A REMARKS: C/O- SATI3-64-38 TAIL ROTOR INSTALLATION DO A GVI OF TAIL ROTOR INSTALLATION COMPONENTS REFERENCE (DMC): 39-A-64-21-00-00A-310A-A REMARK: 40. SATTS -71-15 LEFT AND RIGHT INSULATED TUBES (PART NUMBERS 3G7160V02951 AND 3G7160V03051) 72 (ENGINE INLET PARTICLE SEPARATOR SYSTEM) DO A FC. FAULT FIND TASK TO VERIFY SEALING ABILITY UNDER PRESSURE REFERENCE (DMC): 39-A-75-51-00-00A-340A-K REMARKS: NOT APPLICABLE DUE TO NOT FITTED 71-17 MAIN PANEL (2 OFF) (ENGINE INLET PARTICLE SEPARATOR SYSTEM) 73 DO A DI FOR INTEGRITY OF ATTACHMENT TO STRUCTURE AND ABSENCE OF CRACKS REFERENCE (DMC): 39-A-75-51-00-00A-31AB-K REMARKS: NOT APPLICABLE DUE TO NOT FITTED *Certifies that the work specified above, except as otherwise specified, was carried out in accordance with CAA Malaysia Requirements and in respect to that work the aircraft / aircraft component is considered ready for release to service. Am 0/2016/02 *Certifies that the work specified above, except as otherwise specified, was carried out in accordance with and in respect to that work the aircraft / aircraft component is considered ready for release to service. ☐ TICK ✓ WHERE APPLICABLE