

DUPLICATE

RETURNED

ADS TIONAL

WORKSHEET

GALAXY AEROSPACE (M) SDN. BHD. [1040262-D]

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CLIENT/OWNER: ROYAL MALAYSIA SERIAL NO. HOURS LDG/CYCLE WORKSHEET NO: 15364-001 POLICE WORK/INSP/DESC: 50 FH INSPECTION AIRCRAFT 31807 REFER WORKPACK AIRCRAFT TYPE: AW139 REGISTRATION: 9M-PMA #1 FNGINE WORKPACK REF: PCF-KR2069 REFER WORKPACK 9M-PMA-15364 BASE/FACILITY: WBKK #2 ENGINE: PCE-KB2067 AJL REF NO.: REFER WORKPACK REFER WORKPACK DATE IN: REFER WORKPACE OUT: REFER SHEET 14 OF 16 NG/N1 NF/N2 Reason for raising: SCHEDULED INSPECTIONS PERFORM I.A.W AIRCRAFT MAINTENANCE PROGRAM Raised by and date: Other requirements/information: RMPAOF/CAMO/AMP/AW139 ISSUE 3 REV. 0 DATED 03 FEB 2021, IETP - 42ND ISSUE DATE CAMO; Hanis NII 2022-12 -01 UPDATED: 2023-02-28, EMM PTC6C-67C 30.2 REVISION DATE 2023-02-07 OR LATER 20/04/2023 APPROVED REVISIONS. DUE: 1697:10 FH Item Description Technician * Eng. CRS Date 71 64-38 TAIL ROTOR INSTALLATION DO A GVI OF TAIL ROTOR INSTALLATION COMPONENTS DMC REFERENCE: 39-A-64-21-00-00A-310A-A 18/5/23 REMARK: C/O SATIS 72 71-15 LEFT AND RIGHT INSULATED TUBES (PART NUMBERS 3G7160V02951 AND 3G7160V03051) (ENGINE INLET PARTICLE SEPARATOR SYSTEM) REFERENCE (DMC): 39-A-75-51-00-00A-340A-K DO A FC. FAULT FIND TASK TO VERIFY SEALING ABILITY UNDER PRESSURE REMARKS: NOT APPLICABLE DUE TO NOT FITTED 73 71-17 MAIN PANEL (2 OFF) (ENGINE INLET PARTICLE SEPARATOR SYSTEM) REFERENCE (DMC): 39-A-75-51-00-00A-31AB-K DO A DI FOR INTEGRITY OF ATTACHMENT TO STRUCTURE AND ABSENCE OF CRACKS REMARKS: NOT APPLICABLE DUE TO NOT FITTED M004 74 CM71-03 ENGINE POWER AVAILABLE TREND MONITORING (4-DISPLAY HELICOPTER ONLY) REFERENCE (DMC): N/A PERFORM AND RECORD ON THE SUITABLE PROVIDED FORMS (SEE PT6C-67C OR PT6C-67C1 ENGINE MAINTENANCE MANUAL, WHICHEVER APPLICABLE) THE POWER ASSURANCE CHECK TREND OF **BOTH THE ENGINES** 18/5/23 NOTE: TASK TO BE PERFORMED BY FOLLOWING THE INSTRUCTIONS CONTAINED IN THE LAST ISSUE 8407 OF P&W PT6C-67C OR PT6C-67C1 ENGINE MAINTENANCE MANUAL, WHICHEVER APPLICABLE. CROSS REFER 9M-PMA-15701 REMARKS: (10 75 MI62-01 MAIN ROTOR BLADE ASSEMBLY REFERENCE (DMC): 39-A-62-11-00-00A-310A-A GENERAL VISUAL INSPECTION FOR DAMAGE NOTE: ROTOR STARTING OR STOPPING WITH WIND SPEED ABOVE 33 KTS. FOR EACH ROTOR START OR STOP PERFORMED WITH WIND SPEED ABOVE 33 KTS THE ADDITIONAL INSPECTIONS REPORTED 1815/23 IN TABLE 6 MUST BE MANDATORILY CARRIED OUT. REMARKS: CLO SAUS *Certifies that the work specified above, except as otherwise specified, was carried out in accordance with CAA Malaysia Requirements and in respect to that work the aircraft / aircraft component is considered ready for release to service. AMO/2016/02 *Certifies that the work specified above, except as otherwise specified, was carried out in accordance with and in respect to that work the aircraft / aircraft component is considered ready for release to service. TICK ✓ WHERE APPLICABLE

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Mão RECORD