

HELICOPTERS

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# EMERGENCY ALERT SERVICE BULLETIN

## **PROTECTIVE MEASURE**

TIME LIMITS - MAINTENANCE CHECKS - Landing gear Check for cracks on the landing gear ATA 32

For the attention o	f F		ARLE
HELICOPTER CONCERNED	NUMBER	Civil	Version(s)
EC120	05A022	В	ED AF
Revision No.		101	Date of issue
Revision 0			2020-10-20
Revision 1		FIN D.	2020-11-05
	Revision 2		2024-07-03

## Summary:

After the replacement of a label of gear due to sinking, the customer performed an inspection of the discarded skid tube.

During the Capacity a crack was found inside the skid tube, under the rear crossbar fitting.

Airbust H have the thus renders mandatory the inspection of the landing gear of EC120 helicopters to make sure that the still a state no cracks in the skid tube under the rear crossbar fitting.

## Reason for last Revision:

Revision 2 of this ALERT SERVICE BULLETIN informs that this ALERT SERVICE BULLETIN is no longer applicable following:

- Integration in Master Servicing Manual (MSM) of the Operating Time Limit (OTL) of 1450 AutoRotational Landings (ARL)
- Publishing of SB EC120-32-11-0001 allowing the retrofit of landing gears with reinforced pad with Operating Time Limit (OTL) of 5900 AutoRotational Landings (ARL).

## Compliance:

Compliance with this ALERT SERVICE BULLETIN is mandatory.

S/N: 1482, 1568, 1573, 1596, 1600, 1601, 1605, 1606, 1609, 1610, 1611, 1612, 1613, 1615, 1616, 1617, 1620, 1622, 1623, 1624, 1625, 1626, 1628, 1630, 1631, 1635, 1637, 1638, 1641, 1643, 1645, 1646, 1647, 1648, 1649, 1650

#### Export Control:

US Export Control - No US content. This Item does not contain any U.S. origin ITAR or EAR content. FR Export Control - Not Listed. This Item is not listed against the EC regulations in the EU/FR

HELICOPTERS

## 1. PLANNING INFORMATION

### **1.A. EFFECTIVITY**

#### 1.A.1. Helicopters/installed equipment or parts

EC120 helicopters, version B, which have performed more than 1450 autorotation landings since new and have a serial number listed on page 1.

#### 1.A.2. Non-installed equipment or parts

Skids which have performed more than 1450 autorotation landings.

## <u>NOTE 1</u>

All the possible configurations of landing gear delivered by Airbus Helicopters are concerned. As a reminder, the existing landing gear references are:

- C321A2101053,
- C321A2101054,
- C321A2501051, - C321A2601053.

## **1.B. ASSOCIATED REQUIREMENTS**

Not applicable.

## 1.C. REASON

Revision 0:

After the replacement of a landing gear due to sinking, the customer performed an inspection of the discarded skid tube.

During the inspection, a crack was found inside the skid tube, under the rear crossbar fitting. After investigation by Airbus Helicopters, it was found that this crack can initiate if the helicopter performs a large number of autorotation landings.

This crack can lead to the roll-over of the helicopter after a hard landing.

Therefore, Airbus Helicopters makes compliance with this ALERT SERVICE BULLETIN mandatory.

Revision 1:

The purpose of revision 1 of this ALERT SERVICE BULLETIN is to specify that the number of autorotation landings is specific to each landing skid.

Revision 0 2020-10-20 Revision 2 2024-07-03 Page 2/6 *This document is available on the Internet:* <u>www.airbushelicopters.com/techpub/</u> Export Control: US\_EC\_NoUScontent - FR\_EC\_NotListed

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#### Revision 2:

Revision 2 of this ALERT SERVICE BULLETIN informs that this ALERT SERVICE BULLETIN is no longer applicable following:

- Integration in Master Servicing Manual (MSM) of the Operating Time Limit (OTL) of 1450 AutoRotational Landings (ARL)
- Publishing of SB EC120-32-11-0001 allowing the retrofit of landing gears with reinforced pad with Operating Time Limit (OTL) of 5900 AutoRotational Landings (ARL).

Revision 2 of this ALERT SERVICE BULLETIN has no effect on the compliance with former revisions of this ALERT SERVICE BULLETIN.

## 1.D. DESCRIPTION

Not applicable.

#### **1.E. COMPLIANCE**

#### 1.E.1. Compliance at H/C manufacturer level

Not applicable.

#### 1.E.2. Compliance in service

Not applicable.

#### 1.F. APPROVAL

The technical content of this document is approved under the authority of the Design Organization Approval ref. EASA. 21J.700.

#### 1.G. MANPOWER

Not applicable

#### 1.H. WEIGHT AND BALANCE

Not applicable.

## 1.I. POWER CONSUMPTION

Not applicable.

## 1.J. SOFTWARE UPGRADES/UPDATES

Not applicable.

HELICOPTERS

#### 1.K. REFERENCES

Not applicable

## **1.L. OTHER AFFECTED PUBLICATIONS**

Not applicable

## 1.M. PART INTERCHANGEABILITY OR MIXABILITY

Not applicable.

HELICOPTERS

## 2 EQUIPMENT OR PARTS INFORMATION

2.A. EQUIPMENT OR PARTS: PRICE - AVAILABILITY - PROCUREMENT

Not applicable.

### 2.B. LOGISTIC INFORMATION

Not applicable.

## 2.C. EQUIPMENT OR PARTS REQUIRED PER HELICOPTER/COMPONENT

Not applicable.

## 2.D. EQUIPMENT OR PARTS TO BE RETURNED

Not applicable.

HELICOPTERS

## **3** ACCOMPLISHMENT INSTRUCTIONS

## 3.A. GENERAL

Not applicable.

## 3.B. WORK STEPS

Not applicable.

## 3.C. RECORD OF COMPLIANCE

Not applicable.

## 3.D. OPERATING AND MAINTENANCE INSTRUCTIONS

Not applicable.