A	Alf	RWORTHINESS DIRECTI No F-2004-090	VE	Distribution:	Issue date: June 23, 2004	Page : <b>1/2</b>
Direction générale de l'aviation civile France	This Airworthiness Directive is published by the DGAC:  On behalf of EASA, the Primary Airworthiness Authority for the affected product.  as the Registration Airworthiness Authority for the affected aircraft				Translation of « Consigne de Navigabilité » of same number. In case of difficul ty, reference should be made to the French issue.	
GSAC publication	No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.					
Corresponding foreign Airworthiness Directive(s):			Airworthiness Directive(s) replaced:			
Not applicable			None			
Person in charge of airworthiness: EUROCOPTER			Type(s): AS 355 helicopters			
Type certificate(s) No. 168			1			
TCDS No 168						
ATA chapter:		Subject:				
29 Hydraulic system - Hyd		raulic	cut off fund	ction		

## 1. **EFFECTIVITY**:

AS 355 E helicopters, pre-MOD 073263.

## 2. REASON:

The aim of this Airworthiness Directive (AD) is to eliminate the possibility of a load unbalance in the flight controls, due to residual pressure in the system, after cutting off the hydraulic assistance, following a hydraulic failure, in particular in low outside temperature conditions.

## 3. MANDATORY ACTION AND COMPLIANCE TIME:

The following measure is rendered mandatory from the effective date of this AD:

- At the latest by December 31, 2004, embody the modifications to the electrical wiring, in compliance with the instructions specified in § 2.A., 2.B. and 2.C. of referenced EUROCOPTER AS 355 Alert Service Bulletin No. 29.00.04.

#### 4. REFERENCE PUBLICATION:

EUROCOPTER AS 355 Alert Service Bulletin No. 29.00.04 (Any subsequent approved revision to this ASB is acceptable).

#### 5. **EFFECTIVE DATE**:

July 03, 2004.



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## 6. REMARK:

For any questions concerning the technical content of the requirements in this AD, please contact:

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## 7. APPROVAL:

This AD is approved under EASA reference No 2004-6366 dated June 15, 2004.