	AIRWORTHINESS DIRECTIVE No F-2004-033		Distribution:	Issue date:	Page :
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Direction générale de l'aviation civile France	This Air	worthiness Directive is published by the DG on behalf of EASA, the Primary Airworthine affected product. as the Registration Airworthiness Authority aircraft.	ess Authority for the	Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.	
GSAC publication	No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.				
Corresponding foreign Airworthiness Directive(s):			Airworthiness Directive(s) replaced:		
Not applicable			None		
Person in charge of airworthiness: EUROCOPTER			Type(s): AS 350 helicopters		
Type certificate(s) No. 84					
TCDS No 157					
ATA chapter:	Subject:				
28		Fuel - Fuel bleed lever			

1. EFFECTIVITY:

AS 350 helicopter versions B, BA, BB, B1, B2, B3 and D, pre-MOD 073239.

2. <u>REASON</u>:

This Airworthiness Directive (AD) is issued following some cases of loss of the fuel bleed lever in flight, on AS 350 helicopters which could have led to a collision with the tail rotor.

3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

The following measures are rendered mandatory from the effective date of this AD:

- **3.1.** No later than within 100 flying hours or 6 months (whichever limit is reached first), remove and modify the fuel bleed lever in compliance with the instructions described in paragraphs 2.B and 2.C of referenced EUROCOPTER AS 350 Alert Service Bulletin (ASB) No. 28.00.16.
- **3.2.** Before installation on an aircraft of a fuel bleed lever held as spares, comply with the instructions described in paragraphs 2.B and 2.C of the referenced ASB.

4. <u>REFERENCE PUBLICATION</u>:

EUROCOPTER AS 350 Alert Service Bulletin No. 28.00.16 (Any subsequent approved revision of this ASB is acceptable).

5. EFFECTIVE DATE:

Upon receipt from March 17, 2004.



6. <u>REMARK</u>:

For any questions concerning the technical content of the requirements in this AD, please contact:

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7. <u>APPROVAL</u>:

This AD is approved under EASA reference No 2004-2105 dated March 09, 2004.