BUREAU VERITAS

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

inspections and/or modifications described below are mandatory. No person may operate a product to which this Alrworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive

Translation of 'Consigne de Navigabilité'

Réf.:

91-138-043(B)

In case of any difficulty, reference should be made to the French original issue.

AS 355 AEROSPATIALE helicopter

Pitch control lever expansion pin - Tail rotor

This Airworthiness Directive is applicable to AEROSPATIALE helicopters AS 355 E, F, F1, F2 and N.

On an AS 350 helicopter, the failure of one lug of the pitch control lever hinge yoke on the TGB, due to incorrect assembly, has rendered the following measures mandatory upon receipt of the present directive:

- 1/ Within the next 10 flying hours, then every 10 flying hours until paragraph 2 of this Directive has been applied, visually check the pitch control rod support yoke for cracks as per paragraph B of AEROSPATIALE telex service N° 01.34. Analyse results of check according to the telex service indications.
- 2/ Within the next 50 flying hours, check that the expansion pin is correctly installed as per paragraph C of AEROSPATIALE telex service N° 01.34. Analyse results of checks according to the telex service indications.

Ref.: AEROSPATIALE telex service Nº 01.34

EFFECTIVE DATE: UPON RECEIPT

July 1, 1991

AS 355 AEROSPATIALE helicopter

91-138-043(B)