



AIROD

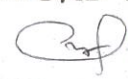

DISCREPANCY REPORT

AIRCRAFT TYPE: A109E REG. NO: PM-B0B DR. NO: 1502 / B0B
 SOURCE REFERENCE: BTs
 DATE RAISED: 2/3/15 CHECK TYPE: _____ WORK ORDER (JCN): LB201502B

DISCREPANCY

A/F HRS	2938.70	NO.1 ENG	2938.70	(TSN/TSO)	NO.2 ENG	622.00	(TSN/TSO)
---------	---------	----------	---------	-----------	----------	--------	-----------

BT109EP-141 REV A DTD 27.02.15 (MANDATORY) - TAIL ROTOR PITCH CONTROL LINK ASSY P/N 10-0130-05-117 TO BE COMPLIED WITH ON PT 1 ONLY (BEFORE NEXT FLIGHT).


RAISED BY: Spkesh Amish SIGN & STAMP:   DATE: 2/3/15

DISPOSITION
TO CARRY OUT BT109EP-141 REV A DTD 27.02.15 T/R PITCH CTRL LINK ASSY P/N 10-0130-05-117 TO COMPLY PT 1 ONLY.

SIGN: WK STAMP:  DATE: 2/3/15

MHP:	MHA:	CUST. APP. SIGNATURE:	MATERIAL COST:
------	------	-----------------------	----------------

CORRECTIVE ACTION

MECH	LAE/AH & DATE
<u>BT109EP-141 REV A DTD 27.02.15 ON T/R PCL ASSY P/N 10-0130-05-117 TO COMPLY PT 1 ONLY. FOUND SATIS.</u>	<u>WK</u>  <u>2/3/15</u>

COMPONENT/ PART CHANGE

PART NO.	DESCRIPTION	S/NO. ON	S/NO. OFF	QTY.	BATCH NO.
<u>N/A</u>					

DUPLICATE / RII / CRITICAL TASK
N/A N/A
 First Insp. Sign / Stamp / Date _____ Second Insp. Sign / Stamp / Date _____

The work recorded above has been carried out in accordance with the requirements of the MCA R for the time being in force and in that respect, the aircraft / equipment is considered fit for release to service.

BOLLETTINO TECNICO

The technical content of this document is approved
under the authority of DOA nr EASA.21J.005.

N° **109EP-141**

DATE February 27, 2015

REV. A

Compliance with
this Bollettino is:

MANDATORY

SUBJECT: TAIL ROTOR PITCH CONTROL LINK ASSY P/N 109-0130-05-117.

REASON: Perform an inspection of the subject link assy for excessive friction of the spherical bearing.

HELICOPTERS AFFECTED:

PART I[^] and PART II[^]:

All Agusta A109E helicopters that install the tail rotor pitch control link assy P/N 109-0130-05-117 which have accumulated less than 100 (one hundred) FH since their overhaul and all tail rotor pitch control link assy P/N 109-0130-05-117 which have accumulated less than 100 (one hundred) FH since their overhaul (e.g. spherical bearing replacement) and present in stock.

NOTE

If it is not possible to determine if or when the link has been overhauled, the link is affected by this Bollettino Tecnico.

COMPLIANCE:

PART I[^]:

Before the next flight.

PART II[^]:

Within and not later than 5 (five) flight hours from the date of issuance of this Bollettino Tecnico.

An appropriate entry should be made in the aircraft log book upon accomplishment.
If ownership of aircraft has changed, please, forward to new owner.
