A/C TYPE	EC 155	В	
A/C DECN	9M-SAS	SIN	6583



EUROCOPTER SOUTH EAST ASIA PTE LTD

48 Loyang Way Singapore 508740 Company Registration No.: 197702516C

NO:	20343	FORM NO: M/003 4
JOB NO.		me making
PAGE	OF	

TYPE OF CHECK _____

DEFECT AND RECTIFICATION WORKSHEET/CERTIFICATE OF RELEASE TO SERVICE

REPORTED	DESECTANORY REQUIRED	ACTION TAKEN	COMPONE	NT CHANGES	G.I.N.NO. MA	N MEGH	LIC/ APP NO.
M BY DATE	DEFECT/WORK REQUIRED	ACTION TAKEN	S/N OFF	S/N ON	REPORT NO.	IRS MECH	DATE DATE
KAILIN	COMPLY WITH TLX INFO -	Not-d					
0/20/02	7 TFS NO. 00000359-						W.
	INFORMATION CONCERNING PHE						0
	ACCIDENT INVOLVING AN AS 365	N					Wall W
	HELICOPTER IN THE IRISH SEA					-	p 190
) KAILIN	COMPLY WITH SL NO. 2497/06	Noted.					1
22/02/07	· ·						
	TEST BENCH SHEET						1/2/0
							ASS
GABRIE	L TO CARRY OUT ASB 76AOD	ASB 76A001 C/O. FOUND					
	MODIFICATION OF "OFF/10	SATIS.					
	FLT" SELECTOR LIFTING	RHS SWITCH CAUARD, PN: 001723		_	Mo 574/06	1,	
	INTERLOCKS ON 12 ALPA	4 LUS SWACY CHUARD, PLAN: 061722	-	-	Me575/06	14	
	PANSI	365764 LIMITER PIN, PIN: 365A61-180	7-01 -	_	Mo768/06	U/26/62	107
	(FEF. AD EV-2006-0256	(RI)					/ XE !
						. 700	1
AND IN		T AS OTHERWISE SPECIFIED WAS CARRIED OUT IN A FT/AIRCRAFT COMPONENT IS CONSIDERED READY EASA PART-145: EASA.145.0056 MANUFACTURER'S PUBLICATIO	FOR RELEAS			EQUIREME	NŢS

TELEX INFO - EUROCOPTER - TELEX INFO - EUROCOPTER - TELEX INFO - EUROCOPTER



T.F.S. No. 00000359 du 05/02/2007 EUROCOPTER - MARIGNANE - TLX 42506F

T	Ε	L	E	X	I	N	F	0	R	M	Α	T	10	N
-							_							

AIRCRAFT: AS 365 <u>Civil version(s)</u>: N, N1, N2, N3

Military version(s): F, Fs, Fi, K

AIRCRAFT: AS 565 Military version(s): AA, MA, MB,

SA, SB, UB

AIRCRAFT: AS 366 <u>Civil version(s)</u>: G1

Military version(s): GA

AIRCRAFT: SA 365 Civil version(s): C, C1, C2, C3

AIRCRAFT: EC 155 Civil version(s): B, B1

ATA: 00

<u>SUBJECT</u>: Information concerning the accident involving an AS 365 N helicopter in the Irish Sea

Dear Customer,

Through the various media, you have probably learnt of an accident involving an AS 365 N helicopter in the Irish Sea, at the end of December 2006.

The helicopter plunged into the sea during an offshore personnel shuttle flight between two rigs.

The salvage of the helicopter was delayed by the bad weather conditions.

Special Bulletin S1/2007, issued by Air Accident Investigation Branch on January 30, 2007 and available from their Internet site, specifies that the result of the analysis of the flight recorders and the investigations conducted on the helicopter indicate no signs of a technical malfunction prior to the impact.

Further checks are in progress as part of a standard accident investigation procedure.



SERVICE LETTER

OPERATOR SUPPORT 40220 Tarnos – France Tel. (33) (0) 5 59 74 40 00 Telex 570 042 Fax (33) (0) 5 59 64 74 98

Technical Support Department Fax (33) (0) 5 59 74 45 34

JSV/JMG/MD

General Service Letter No. 2497/06

Subject: All engines.

Engine log book. Engine test bench sheet

Bordes, December 14, 2006

Dear Sir or Madam,

Starting from today and for a period of approximately six months, the TURBOMECA Group will be using two qualified performance-calculation software programs to determine whether engines are compliant during bench testing.

These two software programs use the "engine test bench sheet" in two different formats.

It is therefore normal to find one of these two formats in engine log books.

Although these two "engine test bench sheets" are slightly different in terms of form, in terms of content, the engine acceptance criteria are strictly identical and compliant with the TURBOMECA reference documents.

Please contact us if you require further information or assistance.

J-S. VIGNES
Technical Support Department





EUROCOPTER DIRECTION TECHNIQUE SUPPORT 13725 MARIGNANE CEDEX FRANCE

CIVIL VERSION(S):

B, B1

ALERT SERVICE BULLETIN

No. 76A001

SUBJECT: ENGINE CONTROLS

Modification of "OFF/IDL/FLT" selector lifting interlocks on 12 ALPHA panel

Corresponds to MOD 0739C36

LIST OF APPROVED REVISIONS	REVISION No. 0 APPROVED			
Not applicable	Date: May 4, 2006			
(4 L)	200			



1. PLANNING INFORMATION

1.A. EFFECTIVITY

1.A.1. Helicopters

Helicopters pre-MOD 0739C36.

NOTE

Refer to the aircraft Certificate of Conformity (or RIC -AMS Individual Inspection Log Book), the Equipment Log Cards (FM) or the aircraft Log Book to identify the actual modification status of the aircraft.

1.A.2. Component(s) affected

Selector lifting interlocks P/N 800KU01B6AA or 800KU01D6AA and 801KU01C6AA or 801KU01E6AA and guard switch P/N 365A61-1807-00 on 12 ALPHA panel assembly.

1.B. ASSOCIATED REQUIREMENTS

Not applicable.

1.C. REASON

To secure engine control selectors, in order to prevent engine control selector flipping from "FLT" to "IDLE".

This Alert Service Bulletin will form the subject of an EASA Airworthiness Directive.

1.D. DESCRIPTION

In flight, further to a shock on canopy, unlocking of one of the 2 engine control selectors occurred and selector flipped to IDLE. This resulted in slowing the concerned engine to idle. This was made possible by wear of selector control lever locking pins.

This Alert Service Bulletin consists of installing new selector lifting interlocks with locking detents, in order to prevent untimely flipping of engine control selector to IDLE.

Therefore, EUROCOPTER renders compliance with this ALERT SERVICE BULLETIN mandatory.



1.E. COMPLIANCE

Eurocopter renders compliance with this Alert Service Bulletin mandatory.

- 1.E.1. At the works
- 1.E.1.a. On aircraft

Before delivery, comply with paragraph 2.B. of this Alert Service Bulletin.

1.E.1.b. On spares

Not applicable.

- 1.E.2. Retrofit action
- 1.E.2.a. On aircraft: by operator

Comply with paragraph 2.B. of this Alert Service Bulletin, at the latest within 550 flying hours or 1 year, whichever occurs first, as from receipt of this Alert Service Bulletin (published at the date mentioned in page footer).

1.E.2.b. On spares: by operator

Installation of selector lifting interlocks listed in paragraph 1.A.2. is prohibited within twelve months as from receipt of this Alert Service Bulletin.

1.F. APPROVAL

Approval is limited to civil version helicopters subject to an Airworthiness Certificate.

1.F.1. Approval of modifications

The information or instructions relate to MOD 0739C36 which was approved on November 23, 2005.

1.F.2. Approval of Alert Service Bulletin

The technical information contained in this Alert Service Bulletin No. 76A001 Revision 0 was approved on May 4, 2006 under the authority of EASA Design Organization Approval No. 21J.056.

1.G. MANPOWER

- Qualification: 1 electrician technician.
- Time: 1 hour approximately.



1.H. WEIGHT & BALANCE

- Weight: Not applicable.
- Moment: Not applicable.

1.I. EFFECT ON ELECTRICAL LOADS

Not applicable.

1.J. SOFTWARE MODIFICATION EMBODIMENT STATE

Not applicable.

1.K. REFERENCES

Refer to:

Aircraft Maintenance System(AMM) Task 24.00.00.911.

Standard Practices (MTC) Work Card: 20.07.03.408.

Flight Manual (FLM).

1.L. OTHER DOCUMENTS CONCERNED

Illustrated Parts Catalog (IPC).

1.M. INTERCHANGEABILITY AND MIXABILITY OF PARTS

1.M.1.Interchangeability

Selector lifting interlocks are mechanically but not functionally interchangeable.

1.M.2.Mixability

Selector lifting interlocks are not mixable.



2. ACCOMPLISHMENT INSTRUCTIONS

2.A. GENERAL

- Comply with instructions on electrical generation as per Task 24.00.00.911 (AMM).
- Check aircraft cleanliness and appearance after an inspection or repair as per Work Card 20.07.03.408 (MTC).

2.B. OPERATIONAL PROCEDURE

2.B.1. Preliminary steps

- Shut down aircraft electrical power system.
- Isolate engine control systems on 12 ALPHA panel.
- 2.B.2. Replacement selector lifting interlocks and switch guard on 12 ALPHA panel assembly (Figure 1)
 - Remove switch guard (a). Scrap.
 - Remove selector lifting interlocks (b) and (c) by means of tool (4).
 - Scrap selector lifting interlocks (b) and (c).
 - Install selector lifting interlocks (2) and (3) by means of tool (4).
 - Install switch guard (1).

2.B.3. Functional tests

Not applicable.

2.B.4. Final steps

- Check aircraft cleanliness and appearance as per Work Card 20.07.03.408 (MTC).
- Restore aircraft to flight condition.

2.B.5. Spares

Refer to paragraph 1.E.2.b.

2.C. IDENTIFICATION

 Record compliance with this Alert Service Bulletin No. 76A001 Revision 0 and embodiment of MOD 0739C36 in the aircraft documents.

2.D. OPERATING AND MAINTENANCE INSTRUCTIONS

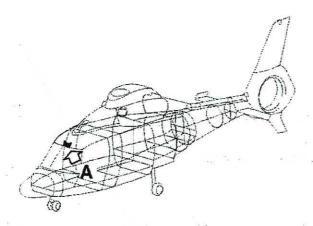
2.D.1. Operating instruction(s)

As per Flight Manual (FLM).

2.D.2. Maintenance instruction(s)

Not applicable.







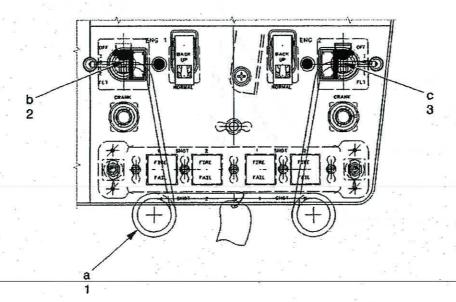


Figure 1



MATERIAL INFORMATION 3.

3.A. MATERIAL: COST - AVAILABILITY

Kits and components will be delivered free of charge upon operator's order until March 31, 2007.

3.B. INFORMATION CONCERNING INDUSTRIAL SUPPORT

Not applicable

3.C. MATERIAL REQUIRED FOR EACH AIRCRAFT, ENGINE/COMPONENTS

3.C.1. Kit or component(s) to be ordered

New P/N (in MPN)	Qty	It.	Key Word	Former P/N	Disposition Disposition
365A61-1807-01	1	1	Guard switch	365A61-1807-00	Scrap
001723	1	2	Selector lifting interlock, LH	800KU01B6AA or 800KU01D6AA	Scrap
001722	1	3	Selector lifting interlock, RH	801KU01C6AA or 801KU01E6AA	Scrap

3.C.2. Material to be ordered separately

Not applicable.

3.C.3. Ingredients to be ordered separately

Not applicable.

3.C.4. Material delivered by the customer

Not applicable.

3.C.5. Tooling

New P/N (in MPN)	Qty	lt.	Key Word	Former P/N	Instructions Disposition
OUT131	4	. 4	Tool for slotted nut removal		-

ERRATUM EC 155

ALERT SERVICE BULLETIN No. 76A001

An error has been found in the subject Alert Service Bulletin No.76A001 Revision 0:

Page 8/10 paragraph: 3.C.1. Kit or component(s) to be ordered

PLEASE READ:

001722 1 2 Switch guard Left side 001723 1 3 Switch guard Right side

INSTEAD OF:

001723 1 2 Selector lifting interlock, LH 001722 1 3 Selector lifting interlock, RH

The date on the replacement page is underlined and remains unchanged.

The page issued previously must be replaced with the page enclosed.

May 10, 2007

J. MARTINEZ

SB KIT Manager



3. MATERIAL INFORMATION

3.A. MATERIAL: COST - AVAILABILITY

Kits and components will be delivered free of charge upon operator's order until March 31, 2007.

3.B. INFORMATION CONCERNING INDUSTRIAL SUPPORT

Not applicable

3.C. MATERIAL REQUIRED FOR EACH AIRCRAFT, ENGINE/COMPONENTS

3.C.1. Kit or component(s) to be ordered

New P/N (in MPN)	Qty	It.	Key Word	Former P/N	Instructions Disposition
365A61-1807-01	1 -	1	Guard switch	365A61-1807-00	Scrap
001722	1.	2	Switch guard Left side	800KU01B6AA or 800KU01D6AA	Scrap
001723	1	3	Switch guard Right side	801KU01C6AA or 801KU01E6AA	Scrap

3.C.2. Material to be ordered separately

Not applicable.

3.C.3. Ingredients to be ordered separately

Not applicable.

3.C.4. Material delivered by the customer

Not applicable.

3.C.5. Tooling

New P/N (in MPN)	Qty	It.	Key Word	Former P/N	Instructions Disposition
OUT131	1	4	Tool for slotted nut removal	e • g	at a



3.D. MATERIAL REQUIRED FOR EACH SPARE PART

3.D.1. Kit to be ordered

As per paragraph 3.C.1.

3.E. RE-IDENTIFIED PARTS

Not applicable.

3.F. TOOLING: COST - AVAILABILITY

For any information, contact the Customer Support Sales Department.

3.G. PROCUREMENT CONDITIONS

3.G.1. Material

Order the required quantity

from

EUROCOPTER
Etablissement de Marignane
Direction VENTES Service Client
S.V.
13725 MARIGNANE CEDEX
FRANCE

NOTE 1

On the purchase order, please specify the mode of transport, the destination and the serial numbers of the aircraft to be modified.

NOTE 2

For ALERT SERVICE BULLETINS, order by: Telex: HELICOP 410 969F Fax: +33 (0)4.42.85.99.96.



3.G.2. Ingredients

The ingredients listed in this Alert Service Bulletin and in the reference documents can be ordered from INTERTURBINE:

Website:

http://www.itlogistics.de +49.41.91.809.300

Telephone:

AOG:

+49.41.91.809.444

APPENDIX

Not applicable.

EASA	AIRW	AIRWORTHINESS DIRECTIVE									
	AD No: 2006 - 0256 R1 Date: 16 October 2006	Date: 16 October 2006									
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.											
Type Approval	Holder's Name:	Type/Model designation(s):									
EUROCOPTER		EC155 Helicopoters									
TCDS Number: F	rance No. 159										
Foreign AD: Not	applicable.										
Supersedes: Not	applicable.										
ATA 76	Engine Controls - E 12 ALPHA	ngine Control Selector Switch Guard on Panel									
Manufacturer:	EUROCOPTER (formerly	EUROCOPTER FRANCE).									
Applicability:	EUROCOPTER EC155 MOD 0739C36.	B and B1 helicopters before embodiment of									
Reason:	the two engine control sel	ive (AD) is issued following a case of unlocking of one of lectors, in flight, further to a shock on the canopy.									
	engine to idle.	The selector flipped to the IDLE position. This resulted in slowing the concerned engine to idle.									
	I are the second the second to	is occurrence was caused by wear of the selector control lever locking pins.									
	rise to a dangerous situat	ase, any untimely switching of an engine to idle can give ion for the helicopter and the passengers.									
	This AD has been revised paragraph 2 'Spares' of the	d to correct the part names and corresponding numbers in is directive.									
Effective Date:	01 September 2006.	1									
Compliance:	The following measures a	re mandatory as from the effective date of this AD:									
	applicable), replace the s	550 flying hours or 1 year (the first limit reached is selector switch guards and interlock on panel 12 ALPHA, in instructions specified in paragraph 2.B. of referenced lert Service Bulletin No. 76A001.									
	2. Spares:										
		ctor lifting interlocks and guard switch, listed below, is on the effective date of this AD:									
	- Selector lifting inte 801KU01C6AA or 801	rlocks P/No. 800KU01B6AA or 800KU01D6AA and KU01E6AA									
	- Guard switch P/No. 36	65A61-1807-00.									

Ref. Publications:	EUROCOPTER EC 155 Alert Service Bulletin No. 76A001 or later approved revisions.
Remarks:	 If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.
	 This AD was posted as PAD 06-157 for consultation on 23 June 2006 with a comment period until 16 July 2006. No comments were received during the consultation period.
	 Enquiries regarding this AD should be addressed to Mr. M. Capaccio, AD Focal Point, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu
	 For any questions concerning the technical content of the requirements in this AD, please contact: EUROCOPTER (STDI) - Aéroport de Marseille Provence 13725 Marignane Cedex, France. Telephone: +33 (0) 4 42 85 97 97; Facsimile: +33 (0) 4 42 85 99 66
	E-mail: Directive.technical-support@eurocopter.com