

GALAXY AEROSPACE (M) SDN. BHD. [1040262-D]

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GENERAL DETAILS
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Aircraft Type:	EC120 B	Fuel:			
Aircraft Fleet:	E120	Fuel Burn:	0.00 L/Hr	Engine PN:	0319008000
Registration:	9M-GAS	Empty Weight:	MAUW 0.00 Kg	Engine SN:	34077
Callsign:	9M-GAS	Weight:	3000.00 Kg		
Serial No:	1089	Company: (GALAXY AEROSPACE (M) SDN.		

MAINTENANCE

Date of Values: 04/08/2022

Airframe: 3572:47 hours 0.00 505 4561.00 landings

Engine 1: 4621:54 hours 3818.03 Np cycles 3760.60 Ng cycles

Time: 08:13:36

Torque Events: 0.00

Report Date: 05/08/2022

Verified by (Sign/Stamp):



maintenance.repair.overhaul Aircraft Registration: 9M-GAS 04/08/2022 Page No: 2 Date: Inspection MANDATORY REGULATION INSPECTION General AMP REVIEW 1 Years 01/03/2022 01/03/2023 209 Days General CERTIFICATE OF AIRWORTHINESS 1 Years 25/05/2022 24/05/2023 293 Days CERTIFICATE OF REGISTRATION General **CAAM AN 8301** 3 Years 30/11/2021 30/11/2024 849 Days General AIRCRAFT WEIGHING CAT.POL.MAB.100 4 Years 24/09/2020 23/09/2024 781 Days General AIRCRAFT INSURANCE 2 Years 07/01/2022 06/01/2023 155 Days General RADIO LICENCE 21619/21619/683287/1/1 001/2021 0 Years 10/01/2021 31/12/2024 880 Days NAVIGATION DATABASE UPDATE 28 Days 14/07/2022 11/08/2022 General 7 Days AIRFRAME SCHEDULED INSPECTIONS FLIGHT HOUR INTERVAL Reference Period **Last Done Next Due** Remaining Airframe Inspection 15H INSPECTION ALS 04-20-00 15 Airframe Hours 3503:34 0:00 0:00 Hours Limitation Note: INCLUDED IN 15FH//7D INSPECTION Scheduled Inspections 25 FH INSPECTION ALS 04-20-00 25 Airframe Hours 3563:58 3588:58 16:11 Hours MSM 05-21-01 3500:00 3600:00 100H INSPECTION 100 Airframe Hours 27:13 Hours Airframe Inspection Airframe Inspection 300H INSPECTION MSM 05-25-00 300 Airframe Hours 3439:30 3739:30 166:43 Hours Airframe Inspection 500 FH Inspection MSM 05-22-01 500 Not Applicable Airframe Inspection 600H INSPECTION (EC120B) MSM 05-25-00 600 Not Applicable Airframe Inspection 1000H INSPECTION AMM 64-21-00, 6-8 1000 Airframe Hours 2899:54 3899:54 327:07 Hours Airframe Inspection 1200H INSPECTION (EC120B) MSM 05-25-00 1200 Not Applicable AIRFRAME SCHEDULED INSPECTIONS CALENDAR INTERVAL Reference Period **Last Done Next Due** Remaining UnScheduled BFF / ALF INSPECTION 0 On Condition Limitation Note: DI General 7D INSPECTION MSM 05-25-00 7 Days 00/00/0000 00/00/0000 0 Day Limitation Note: DI Airframe Inspection 1M INSPECTION MSM 05-25-00 1 Months 13/05/2022 00/00/0000 0 Day Limitation Note: REFER MODOFOCATION TAB - KANNAD AF-H INTEGRA SELF-TEST General 3M INSPECTION MSM 05-25-00 3 Months 14/07/2022 14/10/2022 71 Days **6M INSPECTION** MSM 05-25-00 6 Months 15/04/2022 Airframe Inspection 15/10/2022 72 Days Airframe Inspection 12M INSPECTION MSM 05-25-00 12 Months 15/04/2022 15/04/2023 254 Days CMM 24.33.96 CHECK § 2.C AMM 25-67-00, 6-2 CMM 25.69.32 AMM 25-67-00, 6-3 AMM 26-22-00, 6-1 AMM 24-33-00, 5-2 AMM 53-50-00, 6-3 Airframe Inspection 18M INSPECTION MSM 05-25-00 18 Months 02/12/2021 26/05/2023 295 Days



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Engine Inspection	100H/12M ENGINE INSPECTION	EMM 05-20-00-200-801-A01	12 Mont	:hs	15/04/2022	15/04/2023	254 Days
Engine Inspections	150H/12M ENGINE INSPECTION	EMM 05-20-10-201-825-A01	150 Airframe Hou	ırs	3500:00	3650:00	77:13 Hours
Engine Inspections	150H/12M ENGINE INSPECTION	EMM 05-20-10-201-825-A01	12 Mont	ths	15/04/2022	15/04/2023	254 Days
Airframe Inspection	500H/24M ENGINE INSPECTION	EMM 05-20-10-201-846-A01	500 Airframe Hou	ırs	3500:00	4000:00	427:13 Hours
Airframe Inspection	500H/24M ENGINE INSPECTION	EMM 05-20-10-201-846-A01	24 Mont	:hs	15/04/2022	15/04/2024	620 Days
Engine Inspections	**600H/24M ENGINE INSPECTION	EMM 05-20-10-201-850-A01	600 Airframe Hou	ırs	3500:00	4100:00	527:13 Hours
Engine Inspections	**600H/24M ENGINE INSPECTION	EMM 05-20-10-201-850-A01	24 Mont	:hs	15/04/2022	15/04/2024	620 Days
General	**15Y ENGINE INSPECTION		15 Yea	ars	10/07/2015	06/07/2030	2893 Days
General	ENGINE CLEANING		100 Airframe Hou	ırs	3535:25	3635:25	62:37 Hours
UnScheduled	ENGINE RINSING	ARRIUS 2F MM 71-01-01-610-801	0 On Condition	on	-	-	-
AIRFRAME CONDITION	ONAL/ UNSCHEDULE INSPECTIONS	Reference	Perio	od	Last Done	Next Due	Remaining
UnScheduled	Servocontrol Each time after the component is	MSM 05-26-00	0 On Conditi	on	-	-	-
	installed.						
UnScheduled	TGB After the introduction to service of a new,	MSM 05-26-00	0 On Condition	on	-	-	-
	overhauled or repaired component.						
UnScheduled	Tail rotor Each time after the component is installed.	MSM 05-26-00	0 On Condition	on	-	-	-
UnScheduled	MGB After the introduction to service of a new,	MSM 05-26-00	0 On Condition	on	-	-	-
	overhauled or repaired component.						
UnScheduled	Main rotor drive Each time after the component is	MSM 05-26-00	0 On Conditi	on	-	-	-
	installed.						
UnScheduled	Mast - scissors drive coupling Each time after the component is installed.	MSM 05-26-00	0 On Condition	on	-	-	-
UnScheduled	Main rotor blade After the introduction to service of a	MSM 05-26-00	0 On Conditi	on	_	-	-
	new, overhauled or repaired component.						
UnScheduled	Main rotor Each time after the component is installed.	MSM 05-26-00	0 On Conditi	on	_	-	-
UnScheduled	Cabin antivibrator support Each time after the	MSM 05-26-00	0 On Condition		_	-	-
	component is installed.						
UnScheduled	Strainers Each time after the component is installed.	MSM 05-26-00	0 On Condition	on	-	-	-
UnScheduled	Starter generator After the introduction to service of a	MSM 05-26-00	0 On Condition	on	-	-	-
	new or overhauled component.						
PART E: Component	Inspection	Reference	Perio	od	Last Done	Next Due	Remaining
Scheduled Inspections	EC 120B dynamic comp. check	AMM 05-40-00	4 Yea	ars	00/00/0000	00/00/0000	0 Day
Scheduled Inspections	EC 120B dynamic comp. check	AMM 05-40-00	8 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	Tail Gear Box Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	4 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	Tail Gear Box Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	8 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	MGB Bars Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	4 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	MGB Bars Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	8 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	MGB Main Module Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	4 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	MGB Main Module Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	8 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	Epicyclic Module Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	4 Yea	ars	00/00/0000	00/00/0000	0 Day
Airframe Inspection	Epicyclic Module Dynamic Comp. Check	MSM 05-25-00 (AMM 05-40-00, 6-1)	8 Yea	ars	00/00/0000	00/00/0000	0 Day
Scheduled Inspections	Arm Assy. Dynamic Comp. Check	AMM 05-40-00	4 Yea	ars	00/00/0000	00/00/0000	0 Day
Scheduled Inspections	Arm Assy. Dynamic Comp. Check	AMM 05-40-00	8 Yea	ars	00/00/0000	00/00/0000	0 Day
Scheduled Inspection	Cabin Extinguisher Weighing	AMM 26-22-00.6-1 (MSM 05-21-02)	12 Mont	hs	00/00/0000	00/00/0000	0 Day
Scheduled Inspection	Starter Generator Brush Check	AMM 24-36-01,6-1	500 Airframe Hou	ırs	0:00	0:00	0:00 Hours
General	First Aid Kit	EPM 1-20	12 Mont	hs	00/00/0000	00/00/0000	0 Day



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General	Main Battery	Inspect i.a.w CMM 24.33.96 CHECK § 2.B		6 Monti	hs	00/00/0000	00/00/0000	0 Day
General	Main Battery	Inspect i.a.w CMM 24.33.96 CHECK § 2.C		12 Mont	hs	00/00/0000	00/00/0000	0 Day
General	Main Battery	Overhaul 96 CHECK § 2.D CHECK § 2.E CHECK § 2.F	i.a.w CMM 24-33-	24 Monti	hs	00/00/0000	00/00/0000	0 Day
Avionic Inspection		Reference		Perio	od	Last Done	Next Due	Remaining
UnScheduled	VERTICAL SPEED INDICATOR (VSI) PHYSICAL INPECTION	CAGM 8709 TIC-GEN-22-0036			0	0:00	0:00	0:00
General Inspection	MRB PERFORM ONCE INSPECTION AT 3600FH & 3700FH	MSM 05-26-00		100 Airframe Hou	rs	3500:00	3600:00	27:13 Hours
Airframe Inspection	10Y/2000H Inspection			2000 Airframe Hou	rs	0:00	0:00	0:00 Hours
Airframe Inspection	10Y/2000H Inspection			10 Yea	rs	00/00/0000	00/00/0000	0 Day
Airframe Inspection	10Y/800H Inspection (EC120B)			800 Airframe Hou	rs	0:00	0:00	0:00 Hours
Airframe Inspection	10Y/800H Inspection (EC120B)			10 Yea	rs	00/00/0000	00/00/0000	0 Day
Airframe Inspection	24M/600H Inspection (EC120B)			600 Airframe Hou	rs	0:00	0:00	0:00 Hours
Airframe Inspection	24M/600H Inspection (EC120B)			24 Mont	hs	00/00/0000	00/00/0000	0 Day
Airframe Lifed Compor	nents							
Part No.	Description	Serial No.	Life At Install	Perio	od	Last Done	Next Due	Remaining
7050A4673006 (SC5091-1) (OH)	SERVOCONTROL	1490	0:00	4000 Airframe Hou	rs	2323:12	6323:12	2750:25 Hours OH
7050A4673006 (SC5091-1) (OH)	SERVOCONTROL	1490	0:00	240 Month	hs	17/09/2021	26/05/2025	1026 Days OH
7050A4673006 (SC5091-1) (LL)	SERVOCONTROL	1490	1588:36	20000 Airframe Hou	rs	2323:12	20734:36	17161:49 Hours LL
7050A4673006 (SC5091-1) (OH)	SERVOCONTROL	272	0:00	4000 Airframe Hou	rs	3439:30	7439:30	3866:43 Hours OH
7050A4673006 (SC5091-1) (OH)	SERVOCONTROL	272	0:00	240 Montl	hs	17/09/2021	28/08/2040	6599 Days OH
7050A4673006 (SC5091-1) (LL)	SERVOCONTROL	272	0:00	20000 Airframe Hou	rs	1340:06	21340:06	17767:19 Hours LL
7050A4673006 (SC5091-1) (OH)	SERVOCONTROL	1845	0:00	240 Montl	hs	10/08/2015	25/04/2027	1725 Days OH
7050A4673006 (SC5091-1) (OH)	SERVOCONTROL	1845	0:00	4000 Airframe Hou	rs	2471:54	6471:54	2899:07 Hours OH
7050A4673006 (SC5091-1) (LL)	SERVOCONTROL	1845	553:18	20000 Airframe Hou	rs	2471:54	21918:36	18345:49 Hours LL
C623A2005102 (LL)	EQUIPPED SCISSOR LINKS	DA175	0:00	20000 Airframe Hou	rs	1827:48	21827:48	18255:01 Hours LL
C621A1006101 (LL)	MAIN ROTOR BLADE	259	2458:18	20000 Airframe Hou	rs	3165:00	20706:42	17133:55 Hours LL
C621A1006101 (LL)	MAIN ROTOR BLADE	211	2215:54	20000 Airframe Hou	rs	3165:00	20949:06	17376:19 Hours LL
C621A1006101 (LL)	MAIN ROTOR BLADE	210	2220:06	20000 Airframe Hou	rs	3165:00	20944:54	17372:07 Hours LL
S262A10T1001 (L'HOTELLIER) (LL)	PORTABLE FIRE EXTINGUISHER	ACW04827	0:00	120 Month	hs	17/09/2021	01/03/2030	2766 Days LL
ELT90A2560102001 (LL)	Battery Pack	KA0018134001	0:00	72 Mont	hs	17/09/2021	31/10/2023	453 Days LL
C652A1103601 (LL)	Bevel Gear	M93	0:00	20000 Airframe Hou	rs	0:00	20000:00	16427:13 Hours LL
C652A1102601 (LL)	Bevel Wheel	M105	0:00	20000 Airframe Hou	rs	0:00	20000:00	16427:13 Hours LL
C652A0101055 (OH)	Tail Gear Box	M192	0:00	288 Monti	hs	17/09/2021	19/12/2023	502 Days OH
C652A0101055 (OH)	Tail Gear Box	M192	0:00	5000 Airframe Hou	rs	0:00	5000:00	1427:13 Hours OH
C642A0300102 (LL)	Equipped Blade	M1176	0:00	8500 Airframe Hou	rs	0:00	8500:00	4927:13 Hours LL
C632A2114102 (LL)	Output housing	M276	0:00	20000 Airframe Hou	rs	202:30	20202:30	16629:43 Hours LL
C632A2110602 (LL)	Input Bevel Gear	M158	0:00	20000 Airframe Hou	rs	202:30	20202:30	16629:43 Hours LL
	Bi-bevel wheel	M176	0:00	20000 Airframe Hou		202:30	20202:30	16629:43 Hours LL



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C632A1111101 (LL)	Conical Box	M108	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C632A1103601 (LL)	Unequipped planet gear	M434	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C632A0201052 (OH)	MGB Main Module	M303	0:00	288 Mon	ths 17/09/2021	14/12/2024	863 Days OH
C632A0201052 (OH)	MGB Main Module	M303	0:00	3750 Airframe Ho	urs 202:30	3952:30	379:43 Hours OH
C622A2006601 (LL)	Blade Pin	M965	0:00	9800 Airframe Ho	urs 0:00	9800:00	6227:13 Hours LL
C622A2002102	Pitch Control Horn	DA170	0:00	11400 Airframe Ho	urs 0:00	11400:00	7827:13 Hours
C622A1003102 (LL)	Equipped scissors drive	DA12	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C622A1002103 (LL)	Equipped main hub	M111	0:00	6400 Airframe Ho	urs 0:00	6400:00	2827:13 Hours LL
C533C3204103 (LL)	MGB ATTACHMENT BAR FITTING	99097	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C533C3203103 (LL)	MGB ATTACHMENT BAR FITTING	99108	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
7050A4243050 (OH)	Starter Generator	S00024XL2	0:00	60 Mon	ths 17/09/2021	13/02/2024	558 Days OH
7050A4243050 (OH)	Starter Generator	S00024XL2	0:00	900 Airframe Ho	urs 2841:24	3741:24	168:37 Hours OH
7050A3651001 (LL)	BEARING,BALL	16-13623	0:00	3000 Airframe Ho	urs 2999:54	5999:54	2427:07 Hours LL
7050A3622059 (LL)	Damper	2606	0:00	20000 Airframe Ho	urs 3164:36	23164:36	19591:49 Hours LL
7050A3622059 (LL)	Damper	2615	0:00	20000 Airframe Ho	urs 3164:36	23164:36	19591:49 Hours LL
7050A3622059 (LL)	Damper	2631	0:00	20000 Airframe Ho	urs 3164:36	23164:36	19591:49 Hours LL
7050A3622024 (LL)	ADAPTATEUR END	F666	0:00	20000 Airframe Ho	urs 2899:54	22899:54	19327:07 Hours LL
7050A3622024 (LL)	ADAPTATEUR END	F817	0:00	20000 Airframe Ho	urs 2899:54	22899:54	19327:07 Hours LL
7050A3622024 (LL)	ADAPTATEUR END	F759	0:00	20000 Airframe Ho	urs 2899:54	22899:54	19327:07 Hours LL
7050A3622023 (LL)	ADAPTATEUR END	F275	0:00	20000 Airframe Ho	urs 2899:54	22899:54	19327:07 Hours LL
7050A3622023 (LL)	ADAPTATEUR END	F498	0:00	20000 Airframe Ho	urs 2899:54	22899:54	19327:07 Hours LL
7050A3622023 (LL)	ADAPTATEUR END	F230	0:00	20000 Airframe Ho	urs 2899:54	22899:54	19327:07 Hours LL
7050A3622022 (LL)	Spherical bearing	0235	0:00	7500 Airframe Ho	urs 0:00	7500:00	3927:13 Hours LL
7050A3632293 (LL)	Cooling Belt (Eurocopter Kit)	NIL	0:00	1500 Airframe Ho	urs 3133:18	4633:18	1060:31 Hours LL
7050A3632293 (LL)	Cooling Belt (Eurocopter Kit)	NIL	0:00	72 Mon	ths 01/03/2018	27/02/2024	572 Days LL
C533C3207101 (LL)	COUNTER-TORQUE FITTING	99113	0:00	82000 Airframe Landir		82000:00	77439:00 Landings
C533C3206101 (LL)	COUNTER-TORQUE FITTING	99113	0:00	82000 Airframe Landir	ngs 0:00	82000:00	77439:00 Landings
C533C3205102 (LL)	MGB ATTACHMENT BAR FITTING	99196	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C533C3205102 (LL)	MGB ATTACHMENT BAR FITTING	99201	0:00	20000 Airframe Ho		20000:00	16427:13 Hours LL
C622A2008101 (LL)	Upper Sleeve Flange	M115	0:00	11000 Airframe Ho		11000:00	7427:13 Hours LL
C622A2008101 (LL)	Upper Sleeve Flange	M123	0:00	11000 Airframe Ho		11000:00	7427:13 Hours LL
C622A2008101 (LL)	Upper Sleeve Flange	M142	0:00	11000 Airframe Ho		11000:00	7427:13 Hours LL
C622A2009101 (LL)	Lower Sleeve Flange	M109	0:00	11000 Airframe Ho		11000:00	7427:13 Hours LL
C622A2009101 (LL)	Lower Sleeve Flange	M123	0:00	11000 Airframe Ho		11000:00	7427:13 Hours LL
C622A2009101 (LL)	Lower Sleeve Flange	M126	0:00	11000 Airframe Ho		11000:00	7427:13 Hours LL
7050A3622036 (LL)	Spherical Bearing	0255	0:00	7500 Airframe Ho		7500:00	3927:13 Hours LL
7050A3622036 (LL)	Spherical Bearing	0284	0:00	7500 Airframe Ho		7500:00	3927:13 Hours LL
7050A3622037 (LL)	Spherical Bearing Bolt	350	0:00	9300 Airframe Ho		9300:00	5727:13 Hours LL
7050A3622037 (LL)	Spherical Bearing Bolt	580	0:00	9300 Airframe Ho		9300:00	5727:13 Hours LL
7050A3622037 (LL)	Spherical Bearing Bolt	606	0:00	9300 Airframe Ho		9300:00	5727:13 Hours LL
7050A3622037 (LL)	Spherical Bearing Bolt	712	0:00	9300 Airframe Ho		9300:00	5727:13 Hours LL
7050A3622037 (LL)	Spherical Bearing Bolt	749	0:00	9300 Airframe Ho		9300:00	5727:13 Hours LL
7050A3622037 (LL)	Spherical Bearing Bolt	750	0:00	9300 Airframe Ho		9300:00	5727:13 Hours LL
C622A1043201 (LL)	THRUST SCREW	4098	0:00	13900 Airframe Ho		17064:36	13491:49 Hours LL
		1000	0.00	10000 / Williams Ho	0107.00	17007.00	. 5 TO 1. TO T TOUTS EL



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C622A1043201 (LL)	THRUST SCREW	4373	0:00	13900 Airframe Ho	urs 3164:36	17064:36	13491:49 Hours LL
C622A1043201 (LL)	THRUST SCREW	4330	0:00	13900 Airframe Ho	urs 3164:36	17064:36	13491:49 Hours LL
C622A1043201 (LL)	THRUST SCREW	4094	0:00	13900 Airframe Ho	urs 3164:36	17064:36	13491:49 Hours LL
C622A1043201 (LL)	THRUST SCREW	4366	0:00	13900 Airframe Ho	urs 3164:36	17064:36	13491:49 Hours LL
C622A1043201 (LL)	THRUST SCREW	3915	0:00	13900 Airframe Ho	urs 3164:36	17064:36	13491:49 Hours LL
C623A2004102 (LL)	EQUIPPED DIAPASON SCISSOR LINKS	SD39	0:00	20000 Airframe Ho	urs 3162:36	23162:36	19589:49 Hours LL
C623A2004102 (LL)	EQUIPPED DIAPASON SCISSOR LINKS	SD40	0:00	20000 Airframe Ho	urs 3164:36	23164:36	19591:49 Hours LL
C623A2005102 (LL)	EQUIPPED SCISSOR LINKS	DA99	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C623A1001103 (LL)	Non Rotating Star	M129	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C623A1002102 (LL)	Rotating Star	M144	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
7050A3623019 (LL)	SNR Bearing	NR0627	0:00	5300 Airframe Ho	urs 1911:54	7211:54	3639:07 Hours LL
C623A3001102 (LL)	EQUIPPED BLADE ANGLE ROD	M397	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C623A3001102 (LL)	EQUIPPED BLADE ANGLE ROD	M398	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C623A3001102 (LL)	EQUIPPED BLADE ANGLE ROD	M399	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C631A1102601 (LL)	SHAFT CONNECTING	SP095	0:00	89000 Airframe Landir	gs 0:00	89000:00	84439:00 Hours LL
C631A1101101 (LL)	ARM ASSY	MAP195	0:00	20000 Airframe Ho	urs 988:30	20988:30	17415:43 Hours LL
C632A0101052 (OH)	EPICYCLIC MODULE	M192	0:00	5000 Airframe Ho	urs 0:00	5000:00	1427:13 Hours OH
C632A0101052 (OH)	EPICYCLIC MODULE	M192	0:00	24 Ye	ars 17/09/2021	24/11/2023	477 Days OH
C632A1110102 (LL)	Main Rotor Shaft	NIL	0:00	78000 Airframe Landir	gs 0:00	78000:00	73439:00 Landings LL
C632A1102602 (LL)	Ring Gear Fixed	M2	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C632A2106602 (LL)	Sun Gear	M147	0:00	20000 Airframe Ho	urs 202:30	20202:30	16629:43 Hours LL
C632A2107602 (LL)	Output Pinion	M245	0:00	8900 Airframe Ho	urs 202:30	9102:30	5529:43 Hours LL
C632A2108602 (LL)	OUTPUT WHEEL	M231	0:00	9500 Airframe Ho	urs 202:30	9702:30	6129:43 Hours LL
C632A2111601 (LL)	Free Wheel Shaft	M242	0:00	20000 Airframe Ho	urs 202:30	20202:30	16629:43 Hours LL
C632A2115102 (LL)	Lower housing	M401	0:00	70000 Airframe Landir	gs 248:00	70248:00	65687:00 Hours LL
C632A2116102 (LL)	Main housing	M310	0:00	70000 Airframe Landir	gs 248:00	70248:00	65687:00 Hours LL
C632A2112601 (LL)	Input flange	ML028	0:00	120000 Airframe Landir	gs -6314:00	113686:00	109125:00 Landings LL
C632A2158201 (LL)	FLANGE OUTPUT	PPT214	0:00	20000 Airframe Ho	urs 202:30	20202:30	16629:43 Hours LL
C632A2181102 (LL)	Input Box	NIL	0:00	20000 Airframe Ho	urs 1897:36	21897:36	18324:49 Hours LL
7050A3633016 (LL)	MGB Bar Fwd	193	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
7050A3633016 (LL)	MGB Bar Fwd	208	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
7050A3633015 (LL)	MGB BAR	200	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
7050A3633015 (LL)	MGB BAR	203	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C642A0100103 (LL)	TAIL ROTOR HUB	CUR00070	0:00	9500 Airframe Ho	urs 0:00	9500:00	5927:13 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	6378	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	5971	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	5393	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	909	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	6368	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	6367	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	5392	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
7050A3642044 (LL)	STOPPED PIN OF TORSION STAR	6067	0:00	20000 Airframe Ho	urs 2042:24	22042:24	18469:37 Hours LL
C651A3101102 (LL)	Front Shaft	PPT00866	0:00	20000 Airframe Ho	urs 2121:00	22121:00	18548:13 Hours LL
C651A1101201 (LL)	Splined Flange	PPT181	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL



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C651A3102052 (LL)	REAR SHAFT ASSY	PPT111	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C652A1005301 (LL)	Input Flange	PPT137	0:00	20000 Airframe Ho	urs -10275:48	9724:12	6151:25 Hours LL
C652A1108101 (LL)	Housing	M77	0:00	20000 Airframe Ho	urs -10275:48	9724:12	6151:25 Hours LL
C622A2002102 (LL)	Pitch Control Horn	DA177	0:00	11400 Airframe Ho	urs 0:00	11400:00	7827:13 Hours LL
C622A2002102 (LL)	Pitch Control Horn	DA222	0:00	11400 Airframe Ho	urs 0:00	11400:00	7827:13 Hours LL
C622A2006601 (LL)	Blade Pin	M974	0:00	9800 Airframe Ho	urs 0:00	9800:00	6227:13 Hours LL
C622A2006601 (LL)	Blade Pin	M978	0:00	9800 Airframe Ho	urs 0:00	9800:00	6227:13 Hours LL
C622A2006601 (LL)	Blade Pin	M982	0:00	9800 Airframe Ho	urs 0:00	9800:00	6227:13 Hours LL
C622A2006601 (LL)	Blade Pin	M1001	0:00	9800 Airframe Ho	urs 0:00	9800:00	6227:13 Hours LL
C622A2006601 (LL)	Blade Pin	M1042	0:00	9800 Airframe Ho	urs 0:00	9800:00	6227:13 Hours LL
7050A4632181 (LL)	OIL PUMP	1400	0:00	20000 Airframe Ho	urs 988:30	20988:30	17415:43 Hours LL
C632A1103601 (LL)	Unequipped planet gear	M440	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C632A1103601 (LL)	Unequipped planet gear	M450	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C632A1103601 (LL)	Unequipped planet gear	M458	0:00	20000 Airframe Ho	urs 0:00	20000:00	16427:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1178	0:00	8500 Airframe Ho	urs 0:00	8500:00	4927:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1183	0:00	8500 Airframe Ho	urs 0:00	8500:00	4927:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1192	0:00	8500 Airframe Ho		8500:00	4927:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1195	0:00	8500 Airframe Ho	urs 0:00	8500:00	4927:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1207	0:00	8500 Airframe Ho	urs 0:00	8500:00	4927:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1208	0:00	8500 Airframe Ho	urs 0:00	8500:00	4927:13 Hours LL
C642A0300102 (LL)	Equipped Blade	M1218	0:00	8500 Airframe Ho	urs 0:00	8500:00	4927:13 Hours LL
C622A1007105 (LL)	Fitting Adapter Upper	MAP1761	0:00	20000 Airframe Ho	urs 3500:00	23500:00	19927:13 Hours LL
C622A1007105 (LL)	Fitting Adapter Upper	MAP1843	0:00	20000 Airframe Ho	urs 3500:00	23500:00	19927:13 Hours LL
C622A1007105 (LL)	Fitting Adapter Upper	MAP1760	0:00	20000 Airframe Ho	urs 3500:00	23500:00	19927:13 Hours LL
C622A1006105 (LL)	Fitting Adapter Lower	MAP1612	0:00	20000 Airframe Ho	urs 3500:00	23500:00	19927:13 Hours LL
C622A1006105 (LL)	Fitting Adapter Lower	MAP1610	0:00	20000 Airframe Ho	urs 3500:00	23500:00	19927:13 Hours LL
C622A1006105 (LL)	Fitting Adapter Lower	MAP1618	0:00	20000 Airframe Ho	urs 3500:00	23500:00	19927:13 Hours LL
0141823J (LL)	ELT BATTERY	NIL	0:00	0 Mon	ths 30/05/2022	30/03/2024	N/A Days LL
SN3308-00-BR (OC)	HORIZONTAL SITUATION INDICATOR	1798	0:00	On Condit	ion N/A	On Condition	N/A OC
060-0015-00 (OC)	DIRECTIONAL GYRO	19254	0:00	On Condit	ion N/A	On Condition	N/A OC
504-0111-904 (OC)	ATTITUDE INDICATOR	504011190225532	0:00	On Condit		On Condition	N/A OC
7050A4634009 (OC)	SPEED MAGNETIC SENSOR	2298	0:00	On Condit		On Condition	N/A OC
78-8060-6086-5 (OC)	STORMSCOPE PROCESSOR, SERIES II	JFP07800466	0:00	On Condit	ion N/A	On Condition	N/A OC
	WEATHER MAPPING SYSTEM						
B19030FC05 (OC)	VEMD VERS. 2006 BASE	4724	0:00	On Condit	ion N/A	On Condition	N/A OC
Modification							
Туре	Maintenance Required	Reference		Peri	od Last Done	Next Due	Remaining
Modifications (Inspection)	INSTALLATION OF EXTERIOR PLACARD	GAM156/ICA		0 On Condit		-	
Modifications (Inspection)	DATAPLATE INSPECTION	GAM127/ICA		6 Mon		30/11/2022	118 Days
Modifications (Inspection)	ELT KANNAD AF-H INTEGRA INSPECTION	ICA 1230222		100 Airframe Ho		3620:14	47:26 Hours
Modifications (Inspection)	12M KANNAD AF-H INTEGRA	ICA 1230222		12 Mon		15/04/2023	254 Days
Modifications (Inspection)	KANNAD AF-H INTEGRA SELF-TEST	ICA 1230222		1 Mon		11/08/2022	7 Days
Modifications (Inspection)	OMNI AND LOW PROFILE ANTENNA SYSTEM	MM/2016/11 - ICA 1230141		0 On Condit			. Days
General	Bilingual Placard	MM/2015/30 - MYCAS		0 On Condit		_	-
	silingual Placard for illegibility, security and cleanliness			o on oonan			
Hours	VHF/GPS Comm 1 Garmin GNC 420 System.	00 Series Maintenance Manual 190-00140-05		28 Da	ays 00/00/0000	00/00/0000	0 Day
i iodi 3	VIII / OI O OOIIIIII I OAIIIIIII ONO 420 Systeiii.	00 Oches Maintenance Manual 130-00140-03		20 Da	193 00/00/0000	00/00/0000	U Day



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Limitation Note: REFER	NAVIGATION DATABASE UPDATE								
Modifications (Inspection)	VHF Comm 2 Bendix/King KY 196A System.	Bendix/King VHF Comm. Transceiver Number 006-00695-0003 ICA	s I 96A/ 197 A Manual		0 On Condit	ion	-	-	-
Modifications (Inspection)	GARMIN GMA-340	Maintenance Manual GMA-340			0 On Condit	ion	-	-	-
Modifications (Inspection)	ADF Bendix/King KR87 ADF with KI-227	KR 87 Installation Manual 006- 00184	-0006 ICA Para. 1.8		0 On Condit	ion	-	-	-
Modifications (Inspection)	Mode A Garmin GTX-320 Transponder & Ameriking AK 350 Encoder.	GTX320/320A Installation Manual 190	D-00133-0l		0 On Condit	ion	-	-	-
Modifications (Inspection)	Bendix/King KRA-10A Radar Altimeter System.	KRA-1 0A Installation Manual 006-00	152-0003		0 On Condit	ion	-	-	-
Modifications (Inspection)	Sandell SN 3308 EHSI	SN3308 Navigation Display Installatio	n Manual 90106-IM-J		0 On Condit	ion	-	-	-
Modifications (Inspection)	500H//24M GMX 200 Multi Function Display . INSPECTION	EC 120B/190/ICA PARA. 7. 1.			24 Mon	ths	02/09/2020	23/08/2022	19 Days
Modifications (Inspection)	500H//24M GMX 200 Multi Function Display . INSPECTION	EC 120B/190/ICA PARA. 7. 1.			500 Airframe Ho	urs	3439:30	3939:30	366:43 Hours
Modifications (Inspection)	12M Pioneer CDX-FM657 CD player. INSPECTION	Pioneer CDX-FM657 CD Player Instal	llation Manual		12 Mon	ths	15/04/2022	15/04/2023	254 Days
Modifications (Inspection)	500H EC120 53 58 03 - REPAIR BATTERY TRAY. INSPECTION	EC120 53 58			500 Airframe Ho	urs	3339:18	3839:18	266:31 Hours
Modifications (Inspection)	REPAIR BATTERY TRAY INSPECTION	EC120 53 58			100 Airframe Ho	urs	3520:14	3620:14	47:26 Hours
Modifications (Inspection)	500H EC120 53 68 03 - REPAIR TAIL BOOM. INSPECTION	EC120 53 68 03			500 Airframe Ho	urs	3339:18	3839:18	266:31 Hours
ADs									
Types	Subject	AD Number	Based On	Р	Warn	ing	Last Done	Next Due	Remaining
				е					
				ri					
				0					
				d					
Repetitive ADs	Main Rotor Head -Rotor Hub Inspection	2010-0026-E	Airframe Hours	1	7 Ho	urs	3572:43	3587:43	14:56
				5					
Limitation Note: REFER	WORKPACK 9M-GAS-8729-015.								
Repetitive ADs	ATA 63, 05, 71 - ROTOR DRIVE - ENGINE MOUNT	F-2003-325R1	Airframe Hours	2	14 Ho	urs	3563:58	3588:58	16:11
	AND ENGINE-TO-MAIN GEAR BOX (MGB)			5					
	COUPLING TUBE ASSEMBLY								
Limitation Note: COMPLI	ED WITH. REFER WORK PACK MYCAS-SBK-15-001-	03 ON 2.01.2015.							
REFER TIC-EC120B-21-0	104 DATED 14.12.2021.								
Repetitive ADs	SERVICEABILITY CHECK OF THE ELT	VAM AD NO 2004-001 (PART II)	Date (Months)	6	3 Mon	ths	30/05/2022	30/11/2022	118
Repetitive ADs	SERVICEABILITY CHECK OF THE ELT	VAM AD NO 2004-001 (PART I)	Date (Months)	4	2 Mon	ths	02/12/2021	00/00/0000	0
Limitation Note: REFER	1M ELT KANNAD SELF TEST								
Repetitive ADs	MRH INSPECTION ON ROTATING AND NON- ROTATING SCISSORS BOLTS	AD 2021-0046 R0 (15H//7D)	Date (Days)	7	3 Da	ays	03/08/2022	10/08/2022	5
	WORKPACK 9M-GAS-8729-015.								
Limitation Note: REFER		AD 2021-0046 R0 (15H//7D)	Airframe Hours	1	7 Ho	urs	3572:43	3587:43	14:56
Limitation Note: REFER \ Repetitive ADs	MRH INSPECTION ON ROTATING AND NON-	,							
	MRH INSPECTION ON ROTATING AND NON- ROTATING SCISSORS BOLTS	,		5					
Repetitive ADs		, ,		5					
Repetitive ADs	ROTATING SCISSORS BOLTS	AD 2021-0069 R0 (15FH)	Airframe Hours	5 1	7 Ho	urs	3572:43	3587:43	14:56



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Limitation Note: APPLICABLE, COMPLIANCE AT INTERVALS NOT TO EXCEED 15 FH, INSPECT EACH AFFECTED PART IN ACCORDANCE WITH THE INSTRUCTIONS OF SECTION 3.B.2 OF THE ASB.

REFER TIC-EC120B-21-0128 DATED 14.12.2021.

TAIL ROTOR HUB BODY - REPLACEMENT OF Repetitive ADs

BOLT. NUT AND WASHER

AD 2021-0069 R0 (1000FH)

Airframe Hours

50 Hours

0 Days

0

2899:54

3899:54

327:07

0 0 Λ

1

Limitation Note: AT INTERVALS NOT TO EXCEED 1 000 FH (SEE NOTE 1 OF THIS AD), ACCOMPLISH A DET OF THE INTERFACE BETWEEN THE AFFECTED PART AND THE TR SPLINED FLANGE IN ACCORDANCE WITH THE INSTRUCTIONS OF PARAGRAPH 1.E.2 OF THE ASB

REFER TIC-EC120B-21-0128 DATED 14.12.2021.

AD Number Remaining/Status Types Subject Based On Warning Last Done Next Due ri Airframe ADs Standard ADs SKID-TYPE LANDING GEAR INSPECTION 2022-0053 Date (Months) 3 Months 28/04/2024 633/TO BE

COMPLIED Limitation Note: APPLICABLE DUE TO COMPONENT FRONT CROSSBEAM (P/N C321A2104101), REAR CROSSBEAM (P/N C321A2105101), LEFT-HAND SKID ASSEMBLY (P/N C321A2107102) AND RIGHT-HAND SKID ASSEMBLY (P/N C321A2107102) AND RIGHT-HAND

C321A2108102) EQUIPPED ON AIRCRAFT, COMPLIANCE WITHIN 27 MONTHS INSPECT THE EXTERNAL AND INTERNAL TUBES OF THE AFFECTED PART.

ATA 65 - CANCELLED: TAIL ROTOR DRIVE -2007-0211 Date (Days) REAR DRIVE SHAFT FRICTION RING -

INSPECTION / REPLACEMENT

Limitation Note: AD CANCELED.

Standard ADs

REFER TIC-EC120B-21-0131 DATED 15.12.2021.

Standard ADs ATA 31 INDICATING / RECORDING SYSTEMS -2021-0168

CONTROL UNIT - INSPECTION

Limitation Note: NOT APPLICABLE. AIRCRAFT SN NOT AFFECTED IN ASB EC120-05A023

REFER TIC-EC120B-21-0130

Standard ADs ATA 64 - TAIL ROTOR - BLADES - INSPECTION 2021-0079 Date (Months) 3 Months 30/01/2023 179/TO BE

COMPLIED

NOT APPLICABLE

CANCELLED

Limitation Note: APPLICABLE DUE TO AIRCRAFT FITTED WITH TRB P/N C642A0300102 AND TO COMPLY WITHIN 550FH /24M AFTER THE EFFECTIVE DATE OF THIS AD.

REFER TIC-EC120B-21-0129 DATED 14.12.2021.

Standard ADs ATA 64 - TAIL ROTOR - BLADES - INSPECTION 2021-0079 Airframe Hours 50 Hours 0:00 3943:18 370:31/TO BE COMPLIED

Limitation Note: APPLICABLE DUE TO AIRCRAFT FITTED WITH TRB P/N C642A0300102 AND TO COMPLY WITHIN 550FH /24M AFTER THE EFFECTIVE DATE OF THIS AD.

REFER TIC-EC120B-21-0129 DATED 14.12.2021.

Standard ADs ATA 62 - MAIN ROTOR - HUB SCISSORS / 2021-0046 0 **COMPLIED WITH**

ATTACHMENT BOLTS - INSPECTION /

REPLACEMENT

Report Date: 05/08/2022 Time: 08:13:36



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Limitation Note: COM	IPLIED WITH.							
REFER TO HELISTAR	JOB CARD HRSB/2019/0395 & HRSB/2020/0377.							
Standard ADs	ATA 53 - FUSELAGE - TAIL BOOM - INSPECTION / MODIFICATION / REPAIR	2021-0015 (PARA 1 & 2)		-	0	-	- COMPLI	IED WITH
Limitation Note: COM	IPLIED WITH.							
REFER WORKPACK 9	9M-GAS-8729-020.							
Standard ADs	ATA 32 - LANDING GEAR - SKID TUBES - INSPECTION - ROTORCRAFT FLIGHT MANUAL - SECTION LIMITATIONS - AMENDMENT	2020-0247		-	0	-	- NOT APF	'LICABLI
Limitation Note: NOT	APPLICABLE. AIRCAFT SN NOT AFFECTED.							
REFER TIC-EC120B-0	125 DATED 14.12.2021.							
Standard ADs	EMERGENCY AIRWORTHINESS DIRECTIVES ATTITUDE INDICATOR (BENDIXKING MODEL, KI- 300 OR SANDIA MODEL SAI-340A)	FAA AD NO 2020-18-51, NEW ISSUE	Not Applicable	-	0	-	- NOT APF	'LICABLI
Limitation Note: NOT	APPLICABLE DUE TO BENDIXKING MODEL KI-300 OR S	ANDIA MODEL SAI-340A NOT INSTALLE	ED ON AIRCRAFT.					
Standard ADs	ATA 52 - DOORS - SLIDING DOOR STAR SUPPORT - MODIFICATION	2020-0095	Date (Days)	-	0 Days	22/06/2018	- COMPLI	IED WITH
Limitation Note: REFE	ER HELISTAR JOB CARD 4255.							
REFER TIC-EC120B-2	21-0124 DATED 14.12.2021.							
Standard ADs	ATA 25 - EQUIPMENT / FURNISHINGS - EMERGENCY FLOTATION SYSTEM / INFLATION HOSES - INSPECTION	2020-0064		-	0	-	- NOT APF	'LICABLI
Limitation Note: NOT	APPLICABLE. AIRCRAFT NOT FITTED WITH EFS.							
REFER TIC-EC120B-2	21-0123 DATED 14.12.2021.							
Standard ADs	ATA 65 - TAIL ROTOR - TAIL ROTOR HUB BODY - INSPECTION / REPLACEMENT	2019-0272R1		-	0	-	- SUPE	RSEDEI
	ERSEDED BY EASA AD 2021-0069							
Standard ADs	ATA 62 - MAIN ROTOR - HUB SCISSORS / ATTACHMENT BOLTS - INSPECTION / REPLACEMENT	2019-0139		-	0	-	- SUPE	ERSEDEI
Limitation Note: SUPE	ERSEDED BY EASA AD 2021-0046							
Standard ADs	ATA 62 - MAIN ROTOR - HUB SCISSORS ATTACHMENT BOLTS - INSPECTION / REPLACEMENT	2018-0186		-	0	-	- SUPE	ERSEDEI
Limitation Note: SUPE	ERSEDED BY EASA AD 2019-0139							
Standard ADs	ATA 64 - TAIL ROTOR - BLADE - LIFE LIMIT / RE- IDENTIFICATION	2018-0183		-	0	-	- NOT APP	LICABL



Galax	yAerospace							
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Limitation Note: NO	T APPLICABLE. PN INSTALLED ON AC IS PN: C642A0300°	102.						
REFER TIC-EC120B-	-21-0122 DATED 14.12.2021.							
Standard ADs	TAIL ROTOR DRIVE SYSTEM AIR CONDITIONER COMPRESSOR DRIVE PULLEY - TEST/ REPLACEMENT (EASA DECISION NOT TO ADOPT			-	0	-	- NOT APP	LICABLE
Limitation Note: NO	FAA AD 2017-06-11) T APPLICABLE. AIRCOMM NOT INSTALLED.							
	. ,							
	-21-0121 DATED 14.12.2021.							
Standard ADs	ATA 25 - EQUIPMENT & FURNISHINGS - EMERGENCY FLOATATION GEAR LIGHTING AND ANCILLARY CONTROL UNIT (LACU) PUSHBUTTON - IN-FLIGHT CHECK / REPLACEMENT	2016-0180	Date (Days)	-	0 Days	-	- NOT APP	PLICABLE
Limitation Note: NO	T APPLICABLE. PN NOT INSTALLED IN AC.							
REFER TIC-EC120B	-21-0120 DATED 14.12.2021.							
Standard ADs	ATA 65 - TAIL ROTOR DRIVE SYSTEM AIR CONDITIONER COMPRESSOR DRIVE PULLEY - TEST / REPLACEMENT	US-2015-24-51		-	0	-	- NOT APP	LICABLE
Limitation Note: NO	T APPLICABLE. AIRCOMM NOT INSTALLED.							
REFER TIC-EC120B-	-21-0119 DATED 14.12.2021.							
Standard ADs	ATA 52 - DOORS - SLIDING DOOR STAR SUPPORT - MODIFICATION	2015-0020	Date (Days)	-	0 Days	-	- SUPE	RSEDED
	PERSEDED BY EASA AD 2020-0095							
Standard ADs Limitation Note: NO	ATA 53 - FUSELAGE - TAIL BOOM - INSPECTION T APPLICABLE. VOR NOT INSTALLED.	2014-0167	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
REFER TIC-EC120B	-21-0118 DATED 14.12.2021.							
Standard ADs	ATA 52 - DOORS - SLIDING DOOR STAR SUPPORT - MODIFICATION	2013-0093	Date (Days)	-	0 Days	-	- SUPE	RSEDED
Limitation Note: SUI	PERSEDED BY EASA AD 2015-0020							
Standard ADs	ATA 25 - EQUIPMENT / FURNISHINGS - EMERGENCY FLOTATION GEAR - INSPECTION / MODIFICATION	2011-0185	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
Limitation Note: NO	T APPLICABLE. PN NOT INSTALLED IN AC.							
REFER TIC-EC120B	-21-0117 DATED 14.12.2021.							
Standard ADs	ATA 25 - EQUIPMENT / FURNISHINGS - LOAD RELEASE UNITS - INSPECTION / MODIFICATION / LIMITATION	2009-0122 R1	Date (Days)	-	0 Days	-	- NOT APP	PLICABLE
Limitation Note: NO	T APPLIOCABLE. CARGO HOOK NOT INSTALLED.							
REFER TIC-EC120B	-21-0116 DATED 14.12.2021.							



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Standard ADs	ATA 24 - ELECTRICAL POWER - EMERGENCY	2010-0078-E	Date (Days)	-	0 Days	-	- NOT APP	PLICABLI
Limitation Note: NOT	SWITCH (EMER SW) WIRING - MODIFICATION APPLICABLE. AC SN NOT AFFECTED.							
Zimitation Note: No	74 TEIGNEEL 76 GIVING F74 TEGTES.							
	21-0115 DATED 14.12.2021.							
Standard ADs	ATA 26 - FIRE PROTECTION - PORTABLE HALON	2009-0277 R1	Date (Years)	-	0 Years	-	- NOT APF	PLICABL
	1211 FIRE EXTINGUISHERS - IDENTIFICATION / REPLACEMENT							
Limitation Note: NOT	APPLICABLE DUE TO AIRCRAFT IS NOT FITTED WITH	AFFECTED P/N 863520-00.						
DEEED TIQ EQ100D	04 0440 DATED 44 40 0004							
Standard ADs	21-0113 DATED 14.12.2021. ATA 25 - EQUIPMENT / FURNISHINGS -	2009-0190	Date (Days)	_	0 Days	_	- SIIDE	ERSEDEI
Otalidald ADS	EMERGENCY FLOATATION GEAR - INSPECTION	2009-0190	Date (Days)	-	0 Days	-	- 301 L	INOLDLI
Limitation Note: SUP	PERSEDED BY EASA AD 2011-0185							
Standard ADs	ATA 76 - ENGINE CONTROLS - TWIST GRIP	2008-0185	Date (Days)	-	0 Days	08/05/2012	- COMPL	IED WITI
	DRIVE TUBE - REPLACEMENT / INSPECTION							
Limitation Note: CON	MPLIED WITH. REFER EUROCOPTER WORKSHEET NO.	2011 485 007 DATED 08.05.2012.						
REFER TIC-EC120B-	21-0112 DATED 14.12.2021.							
Standard ADs	ATA 25 - EQUIPMENT & FURNISHINGS -	2008-0177-E	Date (Days)	-	0 Days	-	- SUPE	ERSEDEI
	EMERGENCY FLOATATION GEAR LIGHTING AND	1						
	ANCILLARY CONTROL UNIT (LACU)							
	PUSHBUTTON - IN-FLIGHT CHECK / REPLACEMENT							
Limitation Note: SUP	PERSEDED BY EASA AD 2016-0180							
Standard ADs	ATA 25 - EQUIPMENT/FURNISHING - ENERGY	2007-0311	Date (Days)	=	0 Days	24/01/2008	- COMPL	IED WITI
	ABSORBING SEAT AND METALLISATION BRAID -							
	MODIFICATION							
Limitation Note: CON	MPLIED WITH. REFER EUROCOPTER WORKCARD NO. 2	008-006 DATED 24.01.2008.						
REFER TIC-EC120B-	21-0111 DATED 14.12.2021.							
Standard ADs	ATA 76 - ENGINE CONTROLS - TWIST GRIP	2006-0253	Date (Days)	-	0 Days	-	- SUPE	ERSEDEI
	ASSEMBLY							
Standard ADs	PERSEDED BY EASA AD 2008-0185 ATA 76 - ENGINE CONTROLS - TWIST GRIP	2006-0093			0		CLIDE	ERSEDEI
Statidatu ADS	ASSEMBLY	2000-0093		-	U	•	- 3076	LKSEDEI
Limitation Note: SUP	PERSEDED BY EASA AD 2006-0253							
Standard ADs	ATA 62 - GROUNDING OF SPHERICAL THRUST	F-2006-040	Date (Days)	-	0 Days	-	- NOT APF	PLICABL
	BEARINGS							
Limitation Note: NOT	APPLICABLE DUE TO FITTED SPHERICAL THRUST BE	ARING SN NOT AFFECTED.						
REFER TIC-EC120B-	21-0110 DATED 14.12.2021.							
Standard ADs	ATA 67 - ROTOR FLIGHT CONTROLS - THRUST	F-2005-175	Date (Days)	-	0 Days	-	- COMPL	IED WITI
	WASHERS FOR CYCLIC STICK FRICTION							

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ATA 04, 63 - ENGINE-TO-MAIN GEARBOX (MGB) - COUPLING TUBE	F-2000-058-003 R1	Date (Days)	-	0 Days	-	- NOT APF	'LICABL
APPLICABLE. AIRCRAFT IS FITTED WITH PN: C631A110	02601.						
1-0100 DATED 14.12.2021.							
ATA 04, 65 - TAIL GEARBOX APPLICABLE, TGB INSTALLED PART NUMBER NOT AFF	F-1999-151-001 R1 FECTED. AIRCRAFT FITTED WI	Date (Days) TH PN 652A0101055 SN M19	- 2.	0 Days	-	- NOT APF	'LICABLI
1-0099							
ATA 67 - ROTOR FLIGHT CONTROLS APPLICABLE, AIRCRAFT SN NOT AFFECTED IN THE RA	F-2001-373-008 ANGE.	Date (Days)	-	0 Days	-	- NOT APF	'LICABLI
1-0098							
		, , ,	-	0 Days	19/07/2001	- COMPLI	ED WITH
1-0097 DATED 14.12.2021.							
			- UTSIDE AIR TEMPERAT		-	- NOT APF	LICABL
1-0096							
ATA 72 - ENGINE - GAS GENERATOR FRONT	2007-0057R1		-	0	-	- NOT APF	PLICABL
	RAN MODIFICATION (MOD) Tf	2.					
1-0145							
ATA 79 - ENGINE OIL SYSTEM - ELECTRICAL MAGNETIC PLUG - INSPECTION / REPLACEMENT	2018-0012-E		-	0	-	- NOT APF	'LICABL
APPLICABLE DUE TO ELECTRICAL MAGNETIC PLUGS	HEAD FITTED IS P/N 95200115	96.					
1-0144							
ATA 73 - ENGINE FUEL & CONTROL - PIPE INJECTOR PREFERRED ASSEMBLY -	2017-0070		-	0	-	- NOT APF	'LICABL
	D ASSY P/N 0319730440 INSTA	LLED ON THE ENGINE.					
1-0143 DATED 15.12.2021.							
ATA 73 - ENGINE FUEL & CONTROL - PRIVILEGE INJECTOR - REPLACEMENT	2016-0138R1		-	0	-	- SUPE	ERSEDEI
RSEDED BY EASA AD 2017-0070	2012 2011			_			
ATA 73 - ENGINE FUEL & CONTROL - FUEL	2016-0041		-	0	-	- COMPLI	ED WITI
	ATA 04, 63 - ENGINE-TO-MAIN GEARBOX (MGB) - COUPLING TUBE APPLICABLE. AIRCRAFT IS FITTED WITH PN: C631A110 1-0100 DATED 14.12.2021. ATA 04, 65 - TAIL GEARBOX APPLICABLE, TGB INSTALLED PART NUMBER NOT AFF 1-0099 ATA 67 - ROTOR FLIGHT CONTROLS APPLICABLE, AIRCRAFT SN NOT AFFECTED IN THE RA 1-0098 ATA 05, 52 - CABIN SLIDING DOOR PLIED WITH ON 19.07.2001 AS PER VH-TRW AIRCRAFT 1-0097 DATED 14.12.2021. ATA 28 - FUEL ANTI-ICING ADDICTIVE APPLICABLE, AIRCRAFT OPERATING AT TROPICAL AN 1-0096 ATA 72 - ENGINE - GAS GENERATOR FRONT BEARING - MODIFICATION APPLICABLE DUE TO ENGINE WAS EMBODIED TO SAF 1-0145 ATA 79 - ENGINE OIL SYSTEM - ELECTRICAL MAGNETIC PLUG - INSPECTION / REPLACEMENT APPLICABLE DUE TO ELECTRICAL MAGNETIC PLUGS 1-0144 ATA 73 - ENGINE FUEL & CONTROL - PIPE INJECTOR PREFERRED ASSEMBLY - REPLACEMENT APPLICABLE DUE TO THE PIPE INJECTOR, PREFERRE 1-0143 DATED 15.12.2021. ATA 73 - ENGINE FUEL & CONTROL - PRIVILEGE INJECTOR - REPLACEMENT	ATA 04, 63 - ENGINE-TO-MAIN GEARBOX (MGB) - F-2000-058-003 R1 COUPLING TUBE APPLICABLE. AIRCRAFT IS FITTED WITH PN: C631A1102601. 1-0100 DATED 14.12.2021. ATA 04, 65 - TAIL GEARBOX APPLICABLE, TGB INSTALLED PART NUMBER NOT AFFECTED. AIRCRAFT FITTED WI 1-0099 ATA 67 - ROTOR FLIGHT CONTROLS F-2001-373-008 APPLICABLE, AIRCRAFT SN NOT AFFECTED IN THE RANGE. 1-0098 ATA 05, 52 - CABIN SLIDING DOOR F-2000-285-005 R2 PLIED WITH ON 19.07.2001 AS PER VH-TRW AIRCRAFT MAINTENANCE CERTIFICATION 1-0097 DATED 14.12.2021. ATA 28 - FUEL ANTI-ICING ADDICTIVE ATA 28 - FUEL ANTI-ICING ADDICTIVE F-2000-451-006 APPLICABLE, AIRCRAFT OPERATING AT TROPICAL AND DAMP ATMOSPHERE, NOT OF THE PROOF O	ATA 04, 63 - ENGINE-TO-MAIN GEARBOX (MGB) - F-2000-058-003 R1 Date (Days) COUPLING TUBE APPLICABLE. AIRCRAFT IS FITTED WITH PN: C631a1102601. -0100 DATED 14.12.2021. ATA 04, 65 - TAIL GEARBOX F-1999-151-001 R1 Date (Days) APPLICABLE, TGB INSTALLED PART NUMBER NOT AFFECTED. AIRCRAFT FITTED WITH PN 652A0101055 SN M19 PAIL AND A CONTROLS F-2001-373-008 Date (Days) ATA 67 - ROTOR FLIGHT CONTROLS F-2001-373-008 Date (Days) APPLICABLE, AIRCRAFT SN NOT AFFECTED IN THE RANGE. -0098 ATA 05, 52 - CABIN SLIDING DOOR F-2000-285-005 R2 Date (Days) PAPPLICABLE, AIRCRAFT SN NOT AFFECTED IN THE RANGE. -0097 DATED 14.12.2021 ATA 28 - FUEL ANTH-CING ADDICTIVE F-2000-451-006 Date (Days) APPLICABLE, AIRCRAFT OPERATING AT TROPICAL AND DAMP ATMOSPHERE, NOT OPERATING AT SUBZERO OP-0096 ATA 72 - ENGINE - GAS GENERATOR FRONT 2007-0057R1 BEARING - MODIFICATION APPLICABLE DUE TO ENGINE WAS EMBODIED TO SAFRAN MODIFICATION (MOD) TH12. -0145 ATA 79 - ENGINE OIL SYSTEM - ELECTRICAL 2018-0012-E MAGNETIC PLUG - INSPECTION / REPLACEMENT APPLICABLE DUE TO ELECTRICAL MAGNETIC PLUGS HEAD FITTED IS P/N 9520011596. -0144 ATA 73 - ENGINE FUEL & CONTROL - PIPE 2017-0070 INJECTOR PREFERRED ASSEMBLY - REPLACEMENT APPLICABLE DUE TO THE PIPE INJECTOR, PREFERRED ASSY P/N 0319730440 INSTALLED ON THE ENGINE. -0143 DATED 15.12.2021. ATA 73 - ENGINE FUEL & CONTROL - PRIVILEGE 2016-0138R1 INJECTOR - REPLACEMENT RSEDED BY EASA AD 2017-0070	ATA 04, 63 - ENGINE-TO-MAIN GEARBOX (MGB) - F-2000-058-003 R1 Date (Days) - COUPLING TUBE APPLICABLE, AIRCRAFT IS FITTED WITH PN: C631A1102601. -0100 DATED 14.12.2021. ATA 04, 65 - TAIL GEARBOX F-1999-151-001 R1 Date (Days) - APPLICABLE, TGB INSTALLED PART NUMBER NOT AFFECTED. AIRCRAFT FITTED WITH PN 652A0101055 SN M192. -0099 ATA 67 - ROTOR FLIGHT CONTROLS F-2001-373-008 Date (Days) - APPLICABLE, AIRCRAFT SN NOT AFFECTED IN THE RANGE. -0098 ATA 05, 52 - CABIN SLIDING DOOR F-2000-285-005 R2 Date (Days) - PLIED WITH ON 19.07.2001 AS PER VH-TRW AIRCRAFT MAINTENANCE CERTIFICATION LOG BOOK NO. 1. -0097 DATED 14.12.2021. ATA 28 - FUEL ANTH-ICING ADDICTIVE F-2000-451-006 Date (Days) - APPLICABLE, AIRCRAFT OPERATING AT TROPICAL AND DAMP ATMOSPHERE, NOT OPERATING AT SUBZERO OUTSIDE AIR TEMPERATIONS -0096 ATA 72 - ENGINE - GAS GENERATOR FRONT 2007-0057R1 - BEARING - MODIFICATION AND DAMP ATMOSPHERE, NOT OPERATING AT SUBZERO OUTSIDE AIR TEMPERATIONS -0145 ATA 79 - ENGINE OIL SYSTEM - ELECTRICAL 2018-0012-E - BATA 79 - ENGINE UNIS WAS EMBODIED TO SAFRAN MODIFICATION (MOD) TT12. -0145 ATA 79 - ENGINE FUEL & CONTROL - PIPE 2017-0070 - PREFERRED ASSEMBLY - REPLACEMENT	ATA 04, 63 : ENGINE-TO-MAIN GEARBOX (MGB) - F-2000-058-003 R1 Date (Days) - 0 Days COUPLING TUBE PPUICABLE. AIRCRAFT IS FITTED WITH PN: C631A1102601. -0100 DATED 14.12.2021. ATA 04, 65 - TAIL GEARBOX F-1999-151-001 R1 Date (Days) - 0 Days PPUICABLE, TOR INSTALLED PART NUMBER NOT AFFECTED. AIRCRAFT FITTED WITH PN 652A0101055 SN M192. -0099 -0099 -0099 -0099 -0099 -0099 -0099 -0098 -0099 -009	ATA 04, 63 - ENGINE-TO-MAIN GEARBOX (MGB) - F.2000-058-003 R1 Date (Days) - 0 Days - COUPLING TUBE PPEICABLE. ARRORAFT IS FITTED WITH PN: C631A1102601.	ATA 04, 63 - ENGINE-TO MAIN GEARBOX (McB) · F.2000-958-003 R1 Date (Days) · 0 Days · NOT APP COUPLING TUBE PURICABLE. AIRCRAFT IS FITTED WITH PN: C63141102001. -0100 DATED 14, 12, 2021. ATA 04, 65 - TAIL GEARBOX F. F.1999-151-001 R1 Date (Days) · 0 Days · NOT APP PURICABLE. RISK PART TUBE WITH PN: C63141102001. -0100 DATED 14, 12, 2021. ATA 04, 65 - TAIL GEARBOX F. F.1999-151-001 R1 Date (Days) · 0 Days · NOT APP PURICABLE. TOB INSTALLED PART NUMBER NOT AFFECTED. AIRCRAFT FITTED WITH PN 65240101055 SN M192. -0099 ATA 67 - ROTOR FLIGHT CONTROLS F. 2-201-373-009 Date (Days) · 0 Days · 1907/2001 ATA 67 - ROTOR FLIGHT CONTROLS F. 2-201-373-009 Date (Days) · 0 Days · 1907/2001 -0098 ATA 05, 52 - CABIN SLIDING DOOR F. 2-2000-285-005 R2 Date (Days) · 0 Days · 1907/2001 -0098 ATA 05, 52 - CABIN SLIDING DOOR F. 2-2000-285-005 R2 Date (Days) · 0 Days · 1907/2001 -0097 DATED 14, 12-2021. ATA 28 - FUEL ANTI-CING ADDICTIVE F. 2-2000-451-006 Date (Days) · 0 Days · 0 Days · NOT APP PURICABLE. AIRCRAFT OPERATING AT TROPICAL AND DAMP ATMOSPHERE, NOT OPERATING AT SUBZERO OUTSIDE AIR TEMPERATURE (OAT). -0096 ATA 72 - ENGINE - GAS GENERATOR FRONT 2007-0057R1 · 0 Days · NOT APP BEACHMENT PRUCABLE DUE TO ENGINE WAS EMBODIED TO SAFRAN MODIFICATION (MOD) TH2. -0146 ATA 79 - ENGINE OIL SYSTEM - ELECTRICAL 2018-0012-E · 0 · NOT APP MAGNETIC PLUG - INSPECTION / REPLACEMENT PRUCABLE DUE TO ELECTRICAL MAGNETIC PLUGS HEAD FITTED IS PN 9520011596. -0144 ATA 73 - ENGINE FUEL & CONTROL - PIPE 2017-0070 -0145 -0145 DATED 15, 12, 2021. -0147 -0147 -0148 ATA 73 - ENGINE FUEL & CONTROL - PIPE 2017-0070 -0 · NOT APP INJECTOR - PREFERRED ASSY PN 0319730440 INSTALLED ON THE ENGINE. -0149 DATED 15, 12, 2021. -0147 -0147 -0148 ATA 73 - ENGINE FUEL & CONTROL - PIPE 2017-0070

REFER TIC-EC120B-21-0142 DATED 15.12.2021.



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Standard ADs	ATA 79 - OIL - OIL PUMP DRIVE LINK /	2015-0049	-	0	-	- COMPLIE	ED WITH
turis New 200	LUBRICATING DEVICE - INSPECTION	AAATAAAAA AA DED MADIEIOATION AND DIII I E	TIN 0551/105 IN005505 ATES 540	- 111 05141 00 000	,		
imitation Note: CO	MPLIED WITH THROUGH THE ACCOMPLISHMENT OF SE	319794834 AS PER MODIFICATION AND BULLE	TIN SERVICE INCORPORATED PAG	E IN OEM LOG BOO	₹.		
REFER TIC-EC120-2	1-0141 DATED 15.12.2021.						
Standard ADs	ATA 79 - OIL - NOZZLE BONDING ON EJECTOR	2015-0057	-	0	-	- NOT APPI	LICABLI
	ASSEMBLY - INSPECTION						
imitation Note: NO	T APPLICABLE. LUBE DEVICE IS NOT LISTED IN SB.						
REFER TIC-EC120B-	-21-0140 DATED 15.12.2021.						
Standard ADs	ATA 79 - OIL - NOZZLE BONDING ON EJECTOR	2013-0243	-	0	-	- SUPER	RSEDE
	ASSEMBLY - INSPECTION						
imitation Note: SUF	PERSEDED BY EASA AD 2015-0057						
Standard ADs	ATA 73 - ENGINE FUEL & CONTROL - PRIVILEGE	2012-0150	-	0	-	- SUPER	RSEDE
	INJECTOR - REPLACEMENT						
imitation Note: SUF	PERSEDED BY EASA AD 2016-0138						
Standard ADs	ATA 73 - ENGINE FUEL & CONTROL - P3 AIR PIPE	2011-0182R1	-	0	-	- NOT APPI	LICABL
	- INSPECTION / MODIFICATION						
imitation Note: NO	T APPLICABLE. ENGINE IS POST TF 80.						
REFER TIC-EC120B-	-21-0139 DATED 15.12.2021.						
Standard ADs	ATA 73 - ENGINE FUEL & CONTROL - P3 AIR PIPE	2008-0134R1	-	0	-	- SUPER	RSEDE
	- INSPECTION / MODIFICATION						
imitation Note: SUF	PERSEDED BY EASA AD 2011-0182						
Standard ADs	ATA 73 - ENGINE FUEL & CONTROL -	2008-0170	-	0	-	- NOT APPI	LICABI
	LUBRICATION UNIT - MODIFICATION						
_imitation Note: NO	T APPLICABLE DUE TO MODIFICATION TF 075 HAVE BEI	EN EMBODIED AT THE TIME OF DELIVERY AS P	ER THE OEM ENGINE LOG BOOK.				
REFER TIC-EC120B-	-21-0138 DATED 15.12.2021.						
Standard ADs	ATA 73 - FUEL CONTROL UNIT (FCU) -	2006-0237	-	0	-	- COMPLIE	ED WIT
	REPLACEMENT OF THE CONSTANT DELTA						
	PRESSURE VALVE DIAPHRAGM:						
	INCORPORATION OF MODIFICATION TF55						
.imitation Note: NO	T APPLICABLE DUE TO TF-55 COMPLIANCE.						
REFER TIC-EC120B-	-21-0137 DATED 15.12.2021.						
Standard ADs	ATA 79 - OIL - REPLACEMENT OF CHECK -	2006-0141	-	0	-	- NOT APPI	LICABL
	VALVE PISTON O-RING						
imitation Note: NO	T APPLICABLE. ENGINE IS POST TF 75.						
	-21-0136						
REFER TIC-EC120B		F-2005-143 R1	_	0	-	- SUPF	RSEDE
	ATA 73 - FUEL CONTROL UNIT - REPLACEMENT			•		001 LI	JEDE
Standard ADs	ATA 73 - FUEL CONTROL UNIT - REPLACEMENT PERSEDED BY FASA AD 2006-0237						
	PERSEDED BY EASA AD 2006-0237	F-2005-122	-	0	-	- SUPFE	RSEDE
Standard ADs		F-2005-122	-	0	-	- SUPER	RSEDE



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Standard ADs	ATA 73 - FUEL SYSTEM AND CONTROLS - CHECK OF CORRECT POSITION OF ADJUSTED FCU FUEL FILTER	F-2005-088		-	0	-	-	COMPLIED WIT
Limitation Note: COM	PLIED WITH AT TMUSA AS PER FCU LOGCARD S/N 470	010.						
REFER TIC-EC120B-2	1-0135							
Standard ADs Limitation Note: NOT	ATA 72 - ENGINE - IMMEDIATE GROUNDING APPLICABLE. REF WO: 2007 183 024	F-2005-073R1		-	0	-	-	NOT APPLICABL
REFER TIC-EC120B-2	1-0134							
Standard ADs Limitation Note: NOT	ATA 72 - STOLEN ENGINE - S/N 34084 APPLICABLE DUE TO INSTALLED ENGINE NO. NOT AF	F-2003-219 FECTED.		-	0	-	-	NOT APPLICABL
	1-0133 DATED 15.12.2021.							
Standard ADs	ATA 73 - ENGINE FUEL & CONTROL - FCU MAX. FUEL FLOW STOP - REWORK	F-2000-482		-	0	-	-	NOT APPLICABL
Limitation Note: NOT FCU INSTALLED PN: (APPLICABLE BY FCU P/N AS LISTED IN SB 319 73 4808 031978010	L.						
REFER TIC-EC120B-2	1-0132							
Standard ADs	ATA 73 - REPLACEMENT OF THE FUEL INJECTION MANIFOLDS	F-1999-233		-	0	-	-	SUPERSEDE
Limitation Note: SUPI SBs	ERSEDED BY EASA AD 2012-0150							
Types	Subject	SB Number	Based On	Р	Warning	Last Done	Next Due	Remainir
				e ri				
				0				
Repetitive SBs	KANNAD ELT . CHECK OF THE ATTACHMENT STRAP	SIN 2740-S-25	Date (Months)	d 1	0.50 Months	11/07/2022	11/08/2022	
Repetitive SBs	COLLECTIVE PITCH CONTROL . CHECK OF THE DRIVE	SB 05-014	Date (Months)	2 6	3 Months	04/09/2020	25/08/2022	2
_imitation Note: COM	PLIED INITIAL PARA 3.B.I ON 04/05/2018 AT A/C 3133.33	BFH. REF.: MYCAS JC#4018.						
	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
REFER HELISTAR AIF	A RAINE SERVICE BULLETIN CONFLIANCE STATUS.			4	7 Hours	3572:43	3587:43	14:5
	MAIN ROTOR HUB . CHECK FOR CRACK	ASB 05A012	Airframe Hours	1 5	7 Hours			
Repetitive SBs			Airframe Hours	•	7 Hours			
Repetitive SBs Limitation Note: COM	MAIN ROTOR HUB . CHECK FOR CRACK		Airframe Hours	•	/ riouis			
Repetitive SBs Limitation Note: COM REFER HELISTAR AIF Repetitive SBs	MAIN ROTOR HUB . CHECK FOR CRACK PLIED INITIAL ON 01/07/2010 AT A/C 2054.20 FH. REFEI	R ECM WO# 2007 183 029. ASB 05A019 R1 (15H//7D)	Date (Days)	7	3 Days	03/08/2022	10/08/2022	



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Repetitive SBs	MRH INSPECTION ON ROTATING AND NON- ROTATING SCISSORS BOLTS	ASB 05A019 R1 (15H//7D)	Airframe Hours	1 5	7 Hours	3572:43	3587:43	14:50
Limitation Note: COM	IPLIED WITH 50H ASB 05A019 R1 PARA 3.B.2.B AT A/C	3480.60H ON 15/10/2020. REF. HRSB	JC#HRSB/2020/0377. RI	EPETITIVE 15H//7D I.A.W.	PARA 3.B.2.C I	NCLUDED IN DAILY INS	PECTION	
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Repetitive SBs	TAIL ROTOR HUB BODY - CHECK FOR CRACK	EASB 05A020 R2 (15FH)	Airframe Hours	1 5	7 Hours	3572:43	3587:43	14:56
Limitation Note: COM	MPLIED WITH PARA 3.B.1 & 3.B.2 ON 04/11/2019 AT A/C	3408.90 FH. REF. HRSB JC# 0510						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Repetitive SBs	TAIL ROTOR HUB BODY - REPLACEMENT OF BOLT, NUT AND WASHER	EASB 05A020 R2 (1000FH)	Airframe Hours	0	0 Hours	2899:54	3899:54	327:07
Limitation Note: COM	1PLIED WITH PARA 3.B.1 & 3.B.2 ON 04/11/2019 AT A/C	3408.90 FH. REF. HRSB JC# 0510		0				
REFER HELISTAR AIF	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Repetitive SBs	INSPECTION OF THE DRAIN HOLES ALIGNMENT OF TRB	ASB 64A003	Date (Months)	2 4	Months	09/02/2021	30/01/2023	179
Limitation Note: TO 0	COMPLY WITH ASB PARA 3 WITHIN 500H//24M FOR TRE	3 P/N C642A0300102						
	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Repetitive SBs	INSPECTION OF THE DRAIN HOLES ALIGNMENT OF TRB	ASB 64A003	Airframe Hours	5 5 0 0	0 Hours	3443:18	3943:18	370:3 ⁻
Limitation Note: TO 0	COMPLY WITH ASB PARA 3 WITHIN 500H//24M FOR TRE	3 P/N C642A0300102						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Repetitive SBs	IN 2853-I-52 (10H/7D)	Inspection of outside skin panels.	Date (Days)	7	90 Days	-	00/00/0000	(
Repetitive SBs	IN 2856-I-53 (15H/7D)	Corrosion of the Tail Boom Skin	Airframe Hours	1 20 5	0 Hours	0:00	0:00	0:00
Repetitive SBs	ENGINE-TO-MGB COUPLING TUBE ASSEMBLY - CHECK AND LIMITATION	05A003R3	Airframe Hours	2 1 5	4 Hours	3563:58	3588:58	16:17
	MPLIED WITH ON 07/04/2010 AT A.C CM WO# 2007 183 024.							
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Types	Subject	SB Number	Based On		Varning	Last Done	Next Due	Remaining/Status
				e ri				
				0				
				d				
Standard SBs	TAIL BOOM . SKIN CORROSION INSPECTION	IN 2856-I-53	Date (Months)	- 3	Months	19/08/2020	09/08/2022	FOR INFO ONLY
Standard SBs	REAR DRIVE SHAFT FRICTION RING . INSPECTION	ASB 05A006 R1	Airframe Hours	- 5	0 Hours	3433:12	3533:12	NOT APPLICABLE

Report Date: 05/08/2022 **Time:** 08:13:36



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Standard SBs	Flight Restriction for Engine-to-MGB Coupling Tube Assembly and Engine Support Fitting	04A002	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
Limitation Note: NOT	APPLICABLE DUE TO COUPLING PART NUMBER NOT	AFFECTED. INSTALLED ON A/C P	N C631A1101101.					
REFER HELISTAR AIF	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	SIREN Load Release Units: Load jammed on the cargo hook in the open position	04A004	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
Limitation Note: NOT	APPLICABLE DUE TO CARGO HOOK NOT INSTALLED.							
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Engine Mount	04A005	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
Limitation Note: NOT	APPLICABLE DUE TO COMPLIED WITH SB 71-003 ON 2	4/02/2003 AT A/C 988.50 FH.						
_	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Non-Airworthiness Declaration for Spherical Thrust Bearings,P/No. 7050A3622036	04A006	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
Limitation Note: NOT	APPLICABLE DUE TO PART NOT AFFECTED. PN 7050A	.3622022 SN 0235, 0255 & 0284 IN	STALLED.					
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Limitations - Emergency floatation gear control on the lighting and ancillary control unit	e 04A007R1	Date (Days)	-	0 Days	-	- NOT APP	LICABLE
Limitation Note: NOT	APPLICABLE DUE TO PART NOT FITTED. REF. MYCAS	JC 1409.						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs Limitation Note: N/A.	Monitoring of fuel system contamination AIRCRAFT SERIAL NUMBER 1089	05-002	Date (Days)	-	0 Days	-	- FOR INI	FO ONLY
DEEED AIDDIIS HELI	COPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS						
Standard SBs	Check of Horizontal Stabilizer Drain Holes	05-004	Date (Days)	_	0 Days	_	- COMPLI	ED WITH
	IPLIED WITH. REF.: VH-TRW NON RECURRING AD, SPE			OG BOOK ONE	o bayo		OOM! L	20 ******
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Calendar Check for Corrosion on Main Rotor Head Attach Beams	05-007	Date (Days)	-	0 Days	-	- FOR INI	FO ONLY
Standard SBs	Lubrication of the couplings of the non-rotating and	05-009	Date (Days)	-	0 Days	-	- FOR INI	FO ONLY
Standard SBs	rotating scissors Emergency floatation gear float cover	05-010R1	Date (Days)	<u>-</u>	0 Days	-	- FOR INI	FO ONLY
Standard SBs	SKURKA (ex APC) starter-generators (MP/N)	05-013	Date (Days)	-	0 Days	-		ED WITH
imitation Note: STA	160SG140Q1 and 160SG140Q1XL RTER GENERATOR LOGGED MORE 1000H. INITIAL 150I	H COMPLIED BY ECM ON 08 OCT	2013 AT 2370 6HRS REF	FR WIO# 2013-494-001 SI	LIBSEQUENTI Y BRU	ISH CONDITION CHECK	AT 300H AND OVE	RHALII
AT 900H.	VIEW GENERATION EGGGED MONE 1999 I. MITTINE 1990	TOOMI EIEB BT EOM ON 00 001	2010 X1 2070.01 IXO. IXE	ER WIG# 2010 434 001. 0	OBOL QUEITIET BITO	JOHN CONDITION CHECK	CAT SOUTHING OVE	IXI II KOL
REFER AIRBUS HELI	COPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.						
Standard SBs	Check of the engine-to-MGB coupling tube assembly			-	0	-		FO ONLY
Standard SBs	Cabin Sliding Door Adjustment	05A005R1	Date (Days)	-	0 Days	_	- COMPLI	ED WITH



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Limitation Note: CO	MPLIED WITH ON 19/07/2001 AT A/C 485.00FH. REFER A	USTRALIAN MODIFICATION BOOK						
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Tail Rotor Drive Shaft: Damper Friction Ring on the Rear Drive Shaft	05A006	Date (Days)	-	0 Days	-	- COMPL	LIED WITH
Limitation Note: CO	MPLIED WITH ON 25/03/2014 AT A/C 2396.80 FH. REFER	ECM WO# 2014 025 001						
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Crack in the Belt Buckle	05A008R2	Date (Days)	-	0 Days	=	- COMPL	LIED WITH
Limitation Note: RI:	COMPLIED WITH ON 15/11/2007 AT A/C 1897.60 FH. REF	ER ECM WO# 2007-533. BUCKLES R	EPLACED.					
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Emergency floatation gears: periodic check of the	05A011	Date (Days)	-	0 Days	-	- NOT AP	PLICABLE
	floats with intervals in hours							
Limitation Note: NO	T APPLICABLE DUE TO PART NOT INSTALLED							
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Safety improvement to TIRFOR locking unit on	07-001	Date (Days)	-	0 Days	=	- NOT AP	PLICABLE
	portable crane							
Limitation Note: N/A	A, PORTABLE CRANE NOT FITTED							
REFER AIRBUS HEL	LICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.						
Standard SBs	Installation of the Air-Conditioning Unit Optional	21-001		-	0	=	- (OPTIONAL
	Equipment							
Standard SBs	Improvement of cabin ventilation sealing	21-002		-	0	=	- (OPTIONAL
Standard SBs	Interference Suppression of the SIB 120	21-003		-	0	=	- (OPTIONAL
	Intercommunication System Audio Channels by							
	Replacing the Cabin Fan With an Interference-							
	Suppressed Fan							
Standard SBs	Preventing Damage to the Ventilation Duct due to	21-004		-	0	-	- (OPTIONAL
	Overheating Should the Air Conditioning Fan Fail							
Standard SBs	Provisions for Installation of Ballast in the Fenestron	53-001		-	0	-	- (OPTIONAL
Standard SBs	Reduction of the Noise Level of the Cabin Heating	21-005		-	0	-	- (OPTIONAL
Standard SBs	Ventilation duct insulation	21-006		-	0	-	- COMPL	LIED WITH
Limitation Note: CO	MPLIED WITH. REF.: VH-TRW NON-RECURRING AD, SPI	ECIAL INSPECTION AND MODIFICAT	ION CERTIFICATION L	OG BOOK ONE				
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Heating System For Very Cold Weather	21-008		-	0	-	- (OPTIONAL
Standard SBs	Elimination of the Outside Air Intakes at the Console	21-009		-	0	-	- (OPTIONAL
Standard SBs	Improved Draining of the Air Conditioning System	21-010		-	0	-	- COMPL	LIED WITH
Limitation Note: CO	MPLIED WITH ON 30 SEP 2003 AT A/F TT 1132.5 HRS. RI	EFER TO AUSTRALIAN MODIFICATIO	ON BOOK.					
REFER AIRBUS HEL	LICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.						
Standard SBs	Extreme cold weather heating option	21-011		<u>-</u>	0	-	- (OPTIONAL
Standard SBs	Reserved	21-012		_	0	-		OPTIONAL
Standard SBs	Rear/tail boom structure			-	0	-		NFO ONLY
					~		. 5.0	0 01121



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Standard SBs	Adding a water trap to the air conditioning system	21-013		-	-	0	-	-	OPTIONAL
Standard SBs	Modification of P2 valve control pin	21-014	Date (Days)	-	0 Day	/S	-	-	FOR INFO ONLY
Standard SBs	Modification of the Heating System Called Silencer System	21C007		-		0	-	-	OPTIONAL
Standard SBs	Replacement of the SIB 120 intercommunication	23-001		-		0	-	-	OPTIONAL
	system with the new model intercommunication system to prevent the risk of radio interference								
Standard SBs	External Connector for the ICS	23-002				0			OPTIONAL
Standard SBs	VHF installation KY196B for 8.33 kHz capability	23C003		-		0	-	_	OPTIONAL
Standard SBs	Relocating the battery aft	24-001		-		0	-	_	OPTIONAL
Standard SBs	Installation of the Ground Power Socket as Optional	24-001		-		0	-	_	OPTIONAL
	Equipment			-			-	_	
Standard SBs	To eliminate generation system breaking problem (following engine start-up with generator switch set to OFF) with a new electronic card Z100 replacing the	24-003		-		0	-	-	OPTIONAL
Ot OD-	former card inside the electrical master box	04.004				0			OPTIONAL
Standard SBs	Installation of the Battery Heater Optional Equipment			-		0	-	-	OPTIONAL
Standard SBs	Replacement of the Electrical Connector of the Tail Boom Position Light with a Connector Requiring no Tool for its Disconnection	24-005		-		0	-	-	OPTIONAL
Standard SBs	SUBJECT: ELECTRICAL POWER THALES Starter Generator Installation	24-006		-		0	-	-	OPTIONAL
Standard SBs	Electrical master box and SEFEE breaker panel installation	24-007		-		0	-	-	OPTIONAL
Standard SBs	Modification to Electrical Master Box	24-008		-		0	-	-	OPTIONAL
Standard SBs	Inhibition of the Flight/Ground Signal	24-009		-		0	-	-	OPTIONAL
Standard SBs	APC Starter-generator: Fan Attachment Nut	24-010	Date (Days)	-	0 Day	/S	-	-	FOR INFO ONLY
Standard SBs	Replacement of SEFEE SE07111 Electrical Master Box to Provide Power to the Map Readers When the	24-011	Date (Days)	-	0 Day	/s	-	-	FOR INFO ONLY
	BAT Switch is Set to OFF								
Standard SBs	Possible short-circuit	24-013		-		0	-	-	OPTIONAL
Standard SBs	Modification of the SEFEE electrical master box (P/N) SE07113	24-014	Date (Days)	-	0 Day	/S	-	-	FOR INFO ONLY
Standard SBs	Separation of the Direct Battery distribution from the electrical master box	24-015		-		0	-	-	FOR INFO ONLY
Standard SBs	Electrical Power - Starter-generator Improved SKURKA starter-generator	24-016R	Date (Days)	-	0 Day	/S	-	-	FOR INFO ONLY
Standard SBs	EMERgency SWitch (EMER SW) wiring	24A012	Date (Days)	-	0 Da	/S	-	_	NOT APPLICABLE
	T APPLICABLE DUE TO SERIAL NUMBER NOT AFFECTED		(= =, =,			, -			
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Equipping The Helicopter with the Optional Emergency Flotation Gear (Fixed parts)	25-001		-		0	-	-	OPTIONAL
Standard SBs	Cargo sling fixed part installation	25-002		-		0	-	_	OPTIONAL
Standard SBs	Installation of an electric rear-view mirror	25-003		-		0	-	-	OPTIONAL



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Standard SBs	Emergency Flotation Gear Đ Reinforcement of Drag	25-004		Date (Days)	-	0 Days	-	- NOT	APPLICABLE
	Patch Attachment								
Limitation Note: N/A	, EMERGENCY FLOTATION NOT INSTALLED								
REFER AIRBUS HEL	ICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.							
Standard SBs	Installation of the Optional "SOCATA" Emergency Locator Transmitter	25-005			-	0	-	-	OPTIONAL
Standard SBs	Installation of the optional SKI assembly	25-006			-	0	-	-	OPTIONAL
Standard SBs	Enhanced Sound-Proofing	25-007			-	0	-	-	OPTIONAL
Standard SBs	Installation of the Cabin Access Handles	25-008			-	0	-	-	OPTIONAL
Standard SBs	Installation of a filter on SOCATA emergency beacon	25-009			-	0	-	-	OPTIONAL
Standard SBs	Routing of the Removable Part of the Cargo Sling	25-012		Date (Days)	-	0 Days	-	- NOT	APPLICABLE
	Installation Mechanical Emergency Load Release								
	Unit Control								
Limitation Note: N/A	, CARGO SLING NOT INSTALLED								
REFER AIRBUS HEL	ICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.							
Standard SBs	Installation of the Wire Strike Protection System	25-013			-	0	-	-	OPTIONAL
Standard SBs	Rear Passenger Seat Bench: Corrosion-Preventive	25-014			-	0	-	-	OPTIONAL
	Treatment								
Standard SBs	Passenger seat: Longer Harness Strap	25-015			-	0	-	-	OPTIONAL
Standard SBs	Folding Stretcher Option	25-016			-	0	-	-	OPTIONAL
Standard SBs	KANNAD 406 AF-H Emergency Location Transmitter	r 25-017			-	0	-	-	OPTIONAL
	(ELT) installation								
Standard SBs	Checking the rear LH cable of the sling support	25-018R	R1	Date (Days)	-	0 Days	-	- FO	R INFO ONLY
Standard SBs	Optional KANNAD 121AF-H E.L.T. installation	25-019			-	0	-	-	OPTIONAL
Standard SBs	Reserved	25-020			-	0	-	- FO	R INFO ONLY
Standard SBs	First-Aid Kit	25-021			-	0	-	- FO	R INFO ONLY
Standard SBs	Reserved	25-022		Not Applicable	-	0	-	- FO	R INFO ONLY
Standard SBs	Emergency Floatation Gear - Floatation gear cylinder	r 25-024		• • •	-	0	-	-	OPTIONAL
	identification label on the rear RH bottom fairing								
Standard SBs	Sling: Under-estimation of the weight of the carried	25-025		Date (Days)	-	0 Days	-	- FO	R INFO ONLY
	load			(-, -,		,.			
Standard SBs	Protection of tri-band antenna MP/N 21-41	25-028		Date (Days)	-	0 Days	-	- FO	R INFO ONLY
Standard SBs	Electrical bonding of energy- absorbing seats	25A023		Date (Days)	-	0 Days	-	- CO	MPLIED WITH
	MPLIED WITH ON 24/01/2008 AT A/C 1911.90 FH. REFER			(1,7 1,		.,,			
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Protector installation on the rear bracket and the	25A026		Date (Days)	-	0 Days	-	- NOT	APPLICABLE
Clarida d ODO	supply couplings	20/1020		Dato (Dayo)		o Dayo			711 1 EIO/IBEE
Limitation Note: NO	T APPLICABLE DUE TO PART NOT INSTALLED								
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Cabin Comfort Interior	25C010			_	0	_	_	OPTIONAL
Standard SBs Standard SBs		25C010 25C011			-	0	-	-	OPTIONAL
Statitualu SBS	Retrofitting S/N 1115, 1140, 1150 Aircraft	250011			-	U	-	-	OF HUNAL



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Standard SBs	Installation of three-frequency Emergency Locator Transmitter (ELT) (JAR OPS 3)	25C027		-	0	-	-	FOR INFO ON
Standard SBs	Fire Protection Replacement of fire extinguisher further to obsolescence	26-001	Date (Days)	- () Days	02/09/2020	-	COMPLIED W
Limitation Note: CO	MPLIED WITH PARA 4.A. & 4.B. ON 02/09/2020 AT A/C 34	39.500 FH.	REF.: HRSB JC# 0319					
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Installation of New Fuel Quantity Probes	28-001		-	0	-	-	OPTION
Standard SBs	Optional fuel flowmeter	28-006		-	0	-	-	OPTION
Standard SBs	Use of Replacement Fuel Types (JP4 - JET B -TS1 -	28-007		-	0	-	-	NOT APPLICAE
	RT)							
Limitation Note: N/A	A DUE POST MOD TF 17 AND TF 39. FUEL TYPES NOT AV	/AILABLE I	N MALAYSIA. REFER W/O 2011 078 002					
REFER AIRBUS HEL	LICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.						
Standard SBs	Electrical bonding of upper fuel tank plate	28-008	Date (Days)	- () Days	-	-	RECOMMEND
Limitation Note: NO	T CARRIED OUT. RECOMMENDED SB							
REFER AIRBUS HEL	LICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.						
Standard SBs	Protection of engine fuel system without antifreeze additive down to -15¡C	28-009		-	0	-	-	OPTION
Standard SBs	Check of the integrity of the lower fuel tank (P/No. 501101-X) retaining studs	28-010	Date (Days)	- () Days	24/02/2008	-	COMPLIED W
Limitation Note: CO	MPLIED WITH ON 04/02/2008 AT A/C 1911.90FH. REF.: EC	CM W0#200	8-0060					
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Fuel evetem modification further to engine fuel	28-011		-	0	-	_	OPTION
	Fuel system modification further to engine fuel control unit evolution (FCU)	20 0						
	_	28-012	Date (Days)	- () Days	-	-	FOR INFO ON
Standard SBs	control unit evolution (FCU)		Date (Days)	- (Days 0	-	-	FOR INFO ON
Standard SBs Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit	28-012	Date (Days)	- - -		- - -	-	
Standard SBs Standard SBs Standard SBs Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure	28-012 28C002 28C003	Date (Days)	-	0	- - - 12/03/2001	-	OPTION
Standard SBs Standard SBs Standard SBs Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine	28-012 28C002 28C003 28C004	Date (Days)	-	0 0	- - - 12/03/2001		OPTION FOR INFO ON
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination	28-012 28C002 28C003 28C004	Date (Days)	-	0 0	- - - 12/03/2001	-	OPTION FOR INFO ON COMPLIED W
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL	28-012 28C002 28C003 28C004	Date (Days)	-	0 0 Days	- - - 12/03/2001	-	OPTION FOR INFO ON
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL	28-012 28C002 28C003 28C004	Date (Days)	-	0 0 0 Days	- - 12/03/2001 - -		OPTION FOR INFO ON COMPLIED W
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic	28-012 28C002 28C003 28C004 LIAN MODIF	Date (Days)	-	0 0 Days	- - 12/03/2001 - -		OPTION FOR INFO ON COMPLIED W FOR INFO ON
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic Power Pack	28-012 28C002 28C003 28C004 LIAN MODIF 28C005 29-001	Date (Days)	-	0 0 0 Days 0 0	- - 12/03/2001 - - -		OPTION FOR INFO ON COMPLIED W FOR INFO ON OPTION
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic Power Pack New Hydraulic Power Pack Pre-Clogging Indicator	28-012 28C002 28C003 28C004 LIAN MODIF 28C005 29-001	Date (Days)	-	0 0 0 Days	- - 12/03/2001 - - -		OPTION FOR INFO ON COMPLIED W FOR INFO ON OPTION
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic Power Pack New Hydraulic Power Pack Pre-Clogging Indicator GHC100 Compact Hydraulic Power Unit	28-012 28C002 28C003 28C004 LIAN MODIF 28C005 29-001 29-002 29-003	Date (Days)	-	0 0 0 Days 0 0 0	- - 12/03/2001 - - - -		OPTION FOR INFO ON COMPLIED W FOR INFO ON OPTION OPTION
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.1OH . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic Power Pack New Hydraulic Power Pack Pre-Clogging Indicator GHC100 Compact Hydraulic Power Unit Optional windscreen wiper installation	28-012 28C002 28C003 28C004 LIAN MODIF 28C005 29-001 29-002 29-003 30-001	Date (Days)	-	0 0 0 Days 0 0 0	- - 12/03/2001 - - - - - -		OPTION FOR INFO ON COMPLIED W FOR INFO ON OPTION OPTION OPTION
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO REFER HELISTAR A Standard SBs	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.10H . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic Power Pack New Hydraulic Power Pack Pre-Clogging Indicator GHC100 Compact Hydraulic Power Unit Optional windscreen wiper installation Deflectors for optional windshield wiper	28-012 28C002 28C003 28C004 LIAN MODIF 28C005 29-001 29-002 29-003 30-001 30-002	Date (Days)	-	0 0 Days	- - 12/03/2001 - - - - - - -		OPTION FOR INFO ON COMPLIED W FOR INFO ON OPTION OPTION OPTION OPTION OPTION
Standard SBs Standard SBs Standard SBs Standard SBs Limitation Note: CO	control unit evolution (FCU) Integration of a relay in the fuel pump control circuit Fuel Tank Cleaning Procedure Reserved Modification of the Fuel System to Protect the Engine From Contamination MPLIED ON 12/03/2001 AT A/C 296.10H . REF.: AUSTRAL AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reserved Improvement of the Electrovalve on the Hydraulic Power Pack New Hydraulic Power Pack Pre-Clogging Indicator GHC100 Compact Hydraulic Power Unit Optional windscreen wiper installation Deflectors for optional windshield wiper Reserved	28-012 28C002 28C003 28C004 LIAN MODIF 28C005 29-001 29-002 29-003 30-001 30-002 31-001	Date (Days)	-	0 0 Days	- - - 12/03/2001		OPTION FOR INFO ON COMPLIED W FOR INFO ON OPTION OPTION OPTION OPTION OPTION OPTION



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Standard SBs	LACU's SELA power supply unit retrofit	31-005		-		0	-	-	OPTIONAL
Standard SBs	Check of the warning panel	31-006R1	Date (Days)	-	0 D	ays	-	-	FOR INFO ONLY
Standard SBs	Check of the Twist Grip Drive Tube and Control	76A006		-		0	-	-	COMPLIED WITH
	Pinion Attachment								
Limitation Note: CA	RRIED OUT IAW PARA 2.B.4 ON 08/05/2012 AT A/C 2133.1	10 FH. REFER EC	WO #2011 485 007						
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Check of Control warning panel	31-007	Date (Days)	-	0 D	ays	=	-	FOR INFO ONLY
Standard SBs	Replacement of the ASU card No. 1	31-009	Date (Days)	-	0 D	ays	=	-	FOR INFO ONLY
Standard SBs	Doors -Reinforcement of the control rods of the standard cabin doors	52-019	Not Applicable	-		0	-	-	FOR INFO ONLY
Standard SBs	Indicating and Recording Systems Replacement of push-button FLOAT ARM of SMS unit	31A008	Date (Days)	-	0 D	ays	-	-	NOT APPLICABLE
Limitation Note: NO	T APPLICABLE DUE TO PART NOT FITIED. REF. MYCAS	JC 1409							
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Installation of end plates on the landing gear skids	32-001		-		0	-	_	OPTIONAL
Standard SBs	Preventing or Repairing Loose Rivets in the Landing		Date (Days)	-	0 D	ays	30/07/2002	-	COMPLIED WITH
	Gear Forward Cross-Tube / Skid Attachments		. , ,			•			
	MDUED ON 00/07/0000 DEE . VIU TOW NON DECURDING	AD CDECIAL IN	DECTION AND MODIFICATION CERTIF		ONE DO 00				
Limitation Note: CO	MPLIED ON 30/07/2002. REF.: VH-TRW NON-RECURRING	AD, SPECIAL IN	FECTION AND MODII ICATION CERTII	ICATION LOG BOOK	ONE PG.02				
	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.	S AD, SPECIAL IN	FECTION AND MODIFICATION CERTIF	ICATION LOG BOOK	ONE PG.02				
		32-003	FECTION AND MODIFICATION CENTIL	-	ONE PG.02	0	-	-	OPTIONAL
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.		ECTION AND MODIFICATION CENTIL	CATION LOG BOOK - -	ONE PG.02	0 0	-	-	OPTIONAL OPTIONAL
REFER HELISTAR A Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear /	32-003	ECTION AND MODIFICATION CENTIL	CATION LOG BOOK	ONE PG.02		- -	-	
REFER HELISTAR A Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection	32-003 32-004	Date (Days)	- - - -	0 D	0	- - -	- - -	OPTIONAL
REFER HELISTAR A Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment	32-003 32-004 32-005		- - - -		0	- - - -		OPTIONAL
REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp	32-003 32-004 32-005 32-006R1		CATION LOG BOOK		0 0 ays	- - - -	-	OPTIONAL FOR INFO ONLY
REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved	32-003 32-004 32-005 32-006R1 32-007		CATION LOG BOOK		0 0 ays	- - - - -		OPTIONAL FOR INFO ONLY FOR INFO ONLY
REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light	32-003 32-004 32-005 32-006R1 32-007 32C008		CATION LOG BOOK		0 0 ays 0 0	- - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY
REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0	- - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL
REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0	- - - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL FOR INFO ONLY
REFER HELISTAR A Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing Installation of fixed landing light indicator light Position and anticollision LED lights	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0 ays	- - - - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL OPTIONAL
REFER HELISTAR A Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing Installation of fixed landing light indicator light	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002 33-003 33-004	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0 ays 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	- - - - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL OPTIONAL OPTIONAL
REFER HELISTAR A Standard SBs Limitation Note: NO	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing Installation of fixed landing light indicator light Position and anticollision LED lights Cabin adaptation for night VF	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002 33-003 33-004 34-001	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0 ays 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	- - - - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL OPTIONAL OPTIONAL
REFER HELISTAR A Standard SBs Limitation Note: NO	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing Installation of fixed landing light indicator light Position and anticollision LED lights Cabin adaptation for night VF	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002 33-003 33-004 34-001	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0 ays 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	- - - - - - - - -		FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL OPTIONAL OPTIONAL
REFER HELISTAR A Standard SBs Limitation Note: NO	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing Installation of fixed landing light indicator light Position and anticollision LED lights Cabin adaptation for night VF T COMPLIED. REFER W/O 2011 078 002	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002 33-003 33-004 34-001 STATUS. 34-002	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0 ays 0 0			OPTIONAL FOR INFO ONLY FOR INFO ONLY OPTIONAL OPTIONAL OPTIONAL OPTIONAL
REFER HELISTAR A Standard SBs Limitation Note: NO REFER AIRBUS HEL Standard SBs	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS. Reinforced Wear Plates on Landing Gear Improvement of the Skid-Type Landing Gear / Access Footstep Connection Reserved Replacement of the landing gear rear attachment clamp Reserved Not translated Installation of the optional swiveling landing light Check of the Position Light and Anti-Collision Light Electrical Harness Routing on the Fin Top Fairing Installation of fixed landing light indicator light Position and anticollision LED lights Cabin adaptation for night VF T COMPLIED. REFER W/O 2011 078 002 ICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE OAT Measurement Calibration To Eliminate the Continuous Vibrations Affecting the	32-003 32-004 32-005 32-006R1 32-007 32C008 33-001 33-002 33-003 33-004 34-001 STATUS. 34-002	Date (Days)	CATION LOG BOOK	0 D	0 0 ays 0 0 0 0 ays 0 0			OPTIONAL FOR INFO ONLY FOR INFO ONLY FOR INFO ONLY OPTIONAL OPTIONAL OPTIONAL



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maintena	ance . repair . overhaul	Aircraft Registration:		9M-GAS	Date:	04/08/2022	Page No:	26
Limitation Note: COM	IPLIED WITH ON 15/12/2008 AT A/C 2019.60FH. REF.: EC	M WO 2008-183-005						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Reserved	34-007		-	0	-	- FC	OR INFO ONLY
Standard SBs	Radio Altimeter: AHV 16 (ERT160) Transceiver;	34-008		=	0	-	-	OPTIONAL
	improved reliability of the radio altimeter height signal transmitted for autopilot coupling and display of the							
	radio altimeter indication							
Standard SBs	Horizontal Situation Indicator KI 525A	34A006	Date (Days)	-	0 Days	-	- NOT	T APPLICABLE
Limitation Note: NOT	APPLICABLE DUE TO KI 525A NOT INSTALLED							
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Not translated	34C009		-	0	-	- FC	OR INFO ONLY
Standard SBs	Reserved	52-001		-	0	-	- FC	OR INFO ONLY
Standard SBs	To Allow Key Locking of Cargo Doors	52-002		-	0	-	-	OPTIONAL
Standard SBs	To prevent or repair loose attachment rivets of lower fitting of sliding door	52-003		-	0	-	-	OPTIONAL
Standard SBs	Improvement of the Door working	52-005		-	0	-	-	OPTIONAL
Standard SBs	Improvement of the Cabin Door Ergonomics	52-006		-	0	-	-	OPTIONAL
Standard SBs	New Door Seals	52-007	Date (Months)	-	0 Months	08/10/2013	- CC	OMPLIED WITH
Limitation Note: COM	IPLIED ON 08/10/2013 BY ECM AT A/C 2370.60FH. REF.:	WO 2013 494 001						
	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Modification to the Lower Receptacle of the Left Forward Cabin Door	52-008		-	0	-	-	OPTIONAL
Standard SBs	RH Fixed Panel Adjustment	52-009		-	0	-	-	OPTIONAL
Standard SBs	New Sliding Door Fittings	52-010		-	0	-	-	OPTIONAL
Standard SBs	New Cabin Door Handles	52-011		-	0	-	-	OPTIONAL
Standard SBs	Improvement of Cabin Sliding Door	52-012		-	0	-	-	OPTIONAL
Standard SBs	Doors protection	52-013		-	0	-	- FC	OR INFO ONLY
Standard SBs	Reserved	52-016		-	0	-	- FC	OR INFO ONLY
Standard SBs	Reinforcement of the control rods of the standard cabin doors	52-019 R1	Date (Days)	-	0 Days	-	- FC	OR INFO ONLY
Standard SBs	Installation of a Stop on the Upper Front Rail of the Cabin Sliding Door	52A004R1	Date (Days)	-	0 Days	-	- CC	OMPLIED WITH
Limitation Note: COM	IPLIED WITH ON 19/07/2001 AT A/C 485.00FH. REFER AU	JSTRALIAN MODIFICATION BOOK						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Attachment of the sliding door star support	52A014R2	Date (Days)	-	0 Days	-	- CC	OMPLIED WITH
Limitation Note: COM	1PLIED WITH AD 2013-0093 ON 14/03/2014 BY AHM. REF	. WO 2014 025 001.						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Check of the installation of the crew door actuators	52A017	Date (Days)	-	0 Days	-	- NOT	T APPLICABLE
Limitation Note: NOT	APPLICABLE DUE TO AIRCRAFT NOT FITTED WITH FLO	DATATION GEAR						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.							



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Standard SBs	Doors -Reinforcement of the door locking/unlocking	52A018R1	Date (Days)	-	0 Days	-	- CC	MPLIED WITH
Limitation Note: CO	mechanism MPLIED WITH ON 22/06/2018 AT A/C 3163.70 FH. REFER	MYCAS IC# 4255						
Limitation Note: 00	INIT EILD WITH ON 22/00/2010 AT A/C 3103.70 FH. KEI EK	WITOAG 30# 4200						
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Cabin doors standard upgrade	52C015		-	0	-	-	OPTIONAL
Standard SBs	Protection of the Battery and the Rear Compartment	53-002		-	0	-	-	OPTIONA
	Area Against Grease Expelled From the Tail Rotor							
	Drive Shaft Bearing							
Standard SBs	Reserved	53-003		-	0	=	- FC	OR INFO ONLY
Standard SBs	Reduced Instrument Panel Vibrations	53-004		-	0	=	-	OPTIONAL
Standard SBs	Increased possibilities for adjusting the center-of-	53-005		-	0	-	-	OPTIONAL
0: 1 100	gravity	50.007			2			ODTIONAL
Standard SBs	Improved Tuning of Z- and Y-Axis Anti-Vibration Units	53-007		-	0	-	-	OPTIONAL
Standard SBs	Easy-to-Adjust Cabin Antivibrators With Calibrated	53-008		-	0	-	-	OPTIONAL
	Blades				· ·			0
Standard SBs	Fenestron Ballast Increased to 19Êkg	53-009		-	0	<u>-</u>	-	OPTIONAL
Standard SBs	Reinforcement of X1710 frame for lateral Y vibration	53-010R1	Date (Days)	-	0 Days	20/09/2006	- CC	MPLIED WITH
	damper support fitting attachment		, ,		·			
Limitation Note: RO	: COMPLIED WITH ON 20/09/2006 AT A/C 1709.00FH. REF	:: ECM WO 2006-173						
	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Trimmed Fenestron Keel	53-011		-	0	-	-	OPTIONAL
Standard SBs	Reserved	53-012		-	0	-		OR INFO ONLY
Standard SBs	Cable anti-retention device on the canopy center pos		Date (Days)	-	0 Days	-		OR INFO ONLY
Standard SBs	Modification of the bolt length on the junction frame	53-014	Date (Days)	-	0 Days	-	- FC	OR INFO ONLY
Standard SBs	Repair of the tail boom at the VOR antenna attachments	53-016	On Condition	-	0	-	-	CANCELLED
Standard SBs	Corrosion on the tail boom	53A015R1	Date (Days)	-	0 Days	=	- NO	T APPLICABLE
Limitation Note: NO	T APPLICABLE. REF. : ECM WO# 2014 449 00							
DEEED HELISTAD A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Evaluation of Self-Adjusting Antivibrators in	53C006		_	0	_	_	OPTIONAL
Ciandara OBO	Op?rational Environment				ŭ			01 1101010
Standard SBs	Repair and/or Reinforcement of the Horizontal	55-001		-	0	<u>-</u>	-	OPTIONAL
0.0	Stabilizer Attachment	33 33.			ŭ			0
Standard SBs	Protection of Transparent Panel Bonding Areas	56-002	Airframe Hours	-	0 Hours	-	-	OPTIONAL
	Against UV Radiation							
Standard SBs	Strengthening RH and LH side windshield panels	56-003		-	0	-	-	OPTIONAL
Standard SBs	RH and LH side new windshields panels	56-004	Date (Years)	-	0 Years	14/03/2014	- CC	MPLIED WITH
	MPLIED WITH ON 14/03/2014 AT A/C 2396.80 FH. REFER		(
BEEEB 115	VDED 4.45 OF D.W. C. D							
	AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS.	500004			0			OPTION
Standard SBs	Reinforcement of the Pilot's and Copilot's Sliding	56C001		-	0	-	-	OPTIONAL
	Windows							



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	ance . repair . overhaul		Aircraft Registration:		9M-GAS	Date:	04/08/2022	Page I	No: 28
Standard SBs	Visualization of the Rotor Disk	62-001		-		0	-	-	OPTIONAL
Standard SBs	To Prevent Incorrect Greasing of Anti-vibrator	62-002		-		0	-	-	OPTIONAL
	Bearings								
Standard SBs	To Ease Installation of Stationary and Rotating Scissors Assemblies	62-003		-		0	-	-	OPTIONAL
Standard SBs	Improvement of the balance weight attachment fittings on the MRH	62-004	Date	e (Years) -	0 Ye	ears	26/01/2010	-	COMPLIED WITH
_imitation Note: CON	MPLIED WITH ON 26/01/2010 AT A/C 2121.00FH. REF.: EC	CM WO 20	11 485 002						
REFER HELISTAR AI	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Improvement of the MRH Electrical Bonding	62-005		-		0	-	-	OPTIONAL
Standard SBs	No. Abandoned	62-006		-		0	-	-	FOR INFO ONLY
Standard SBs	New Swashplate Adjustement Shim	62-007	Date	e (Months) -	0 Moi	nths	24/02/2003	-	COMPLIED WITH
Limitation Note: CON	MPLIED ON 24/02/2003. REF.: VH-TRW NON-RECURRING	AD, SPE	CIAL INSPECTION AND MODIFICATIO	N CERTIFICATION	LOG BOOK ONE PG.2				
REFER HELISTAR AI	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Check of the Scissors Drive Adapter Attachment	62-008	Date	e (Days) -	0 0	ays	-	-	FOR INFO ONLY
Standard SBs	Increased Clearance Between the Main Rotor Blades and the Tail Boom	62-009		-		0	-	-	OPTIONAL
Standard SBs	Increasing the blade clearance from tail boom	62-010		-		0	-	-	OPTIONAL
Standard SBs	Improved sealing of main blade balancing weight support	62-011		-		0	-	-	OPTIONAL
Standard SBs	New protection baffle plates for swashplate bearing	62-012		-		0	-	-	OPTIONAL
Standard SBs	Main blade sand protection	62-013		-		0	-	-	OPTIONAL
Standard SBs	Reserved	62-014		-		0	-	-	FOR INFO ONLY
Standard SBs	Improved wear resistance of the stationary and rotary scissors hinge bushes	y 62-015		-		0	-	-	OPTIONAL
Standard SBs	Modification of swashplate baffle plates	62-016	Date	e (Days) -	0 0	ays	-	-	FOR INFO ONLY
Standard SBs	Not translated	62C017		-		0	<u>=</u>	-	FOR INFO ONLY
Standard SBs	Installation of a Reinforced Engine-to-MGB Coupling Tube P/No. C631A1101101 and a New Engine Mount Fitting Assembly	63-001	Date	e (Days) -	0 0	ays	-	-	FOR INFO ONLY
Standard SBs	Reserved	63-002		-		0	-	-	FOR INFO ONLY
Standard SBs	MGB Input Flange Blanking Plug	63-003	Date	e (Days) -	0 0	ays	24/02/2003	-	COMPLIED WITH
Limitation Note: CON	MPLIED ON 24/02/2003. REF.: VH-TRW NON-RECURRING	AD, SPE			LOG BOOK ONE PG .2	•			
REFER HELISTAR AI	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Improving the Rotor Brake Handle Unit	63-004	Date	e (Years) -	0 Ye	ears	15/12/2008	-	COMPLIED WITH
Limitation Note: COM	MPLIED WITH ON 15/12/2008 AT A/C 2019.60FH. REF.: EC	CM WO 20	08 183 005						
REFER HELISTAR AI	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Improved Operation of The Rotor Brake Unit	63-005R	2 Date	e (Days) -	0 0	ays	-	-	FOR INFO ONLY
Standard SBs	Corrosion on Main Rotor Shaft	63-006R	.1 Date	e (Days) -	0 0	ays	-	-	COMPLIED WITH
Limitation Note: R1:	COMPLIED WITH ON 05 FEB 08 AT AIRFRAME TT 1911.9	HRS. RE							
DEEED AIDDUIG		0747110							

REFER AIRBUS HELICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE STATUS.



Galax	y Aci ospace								
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Standard SBs	New MGB Input Spiroid Worm Gear	63-007R1	Date (Days)	-	0 D	ays	-	-	FOR INFO ONLY
Standard SBs	Replacement of a Resistor in the NR Indicating System	63-009		-		0	-	-	OPTIONAL
Standard SBs	Installation of the New MGB Input Flange Plug	63-010	Date (Months)	-	0 Mor	iths	13/12/2005	-	COMPLIED WITH
Limitation Note: CO	MPLIED WITH ON 13/12/2005 AT A/C 1470.00FH. REF.: EC	CM WO 2005-026							
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Rotor brake disk spline wear suppression	63-011	Date (Days)	-	0 D	ays	28/01/2006	-	COMPLIED WITH
Limitation Note: CO	MPLIED WITH ON 28/01/2006 AT A/C 1525.20FH. REF.: EC	CM WO 2006-024. ROTOR BRAKE DISK	CPN-C635A1103201 8	TORLON SPLIT	RING PN-C365	A1111201			
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Check of MGB Main Module Tapered Pin Attaching Cones	63-012	Date (Days)	-	0 D	ays	-	-	FOR INFO ONLY
Standard SBs	Check for Absence of Corrosion Inside the Rotor Shaft	63-013	Date (Days)	-	0 D	ays	-	-	FOR INFO ONLY
Standard SBs	MGB housing corrosion protection improvement	63-014		-		0	-	-	OPTIONAL
Standard SBs	Follow-up of the Main Gear Box Input Housing	63-015	Date (Days)	-	0 D	ays	15/11/2017	-	COMPLIED WITH
Limitation Note: CO	MPLIED WITH ON 15/11/2007 A/C 1897.60FH. REF.: ECM	WO 2007-533. PN-C632A2181102							
	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Replacement of a BJE (leak-tight connector terminal)	63-016	Date (Days)	-	0 D	ays	21/01/2011	-	COMPLIED WITH
	module in the NR indicator circuit								
Limitation Note: CO	MPLIED WITH ON 21/01/2011 AT A/C 2062 .10FH. REF.: E	CM WO 2007 183 038							
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Change in NR/NF indicator	63-019		-		0	<u>=</u>	-	OPTIONAL
Standard SBs	Monitoring of Oil Filter Differential Pressure	63C008		-		0	-	-	OPTIONAL
Standard SBs	Not translated	63C017		-		0	-	-	FOR INFO ONLY
Standard SBs	Adaptation of N Rotor line	63C018		-		0	-	-	OPTIONAL
Standard SBs	Not translated	63C020		-		0	-	-	FOR INFO ONLY
Standard SBs	Longer rings on TRH	64-001		-		0	-	-	OPTIONAL
Standard SBs	Tail rotor hub (TRH) behavior monitoring in operation			-		0	-	-	OPTIONAL
Standard SBs	Replacement of the Existing Tail Gear Box Friction Bushes With Torlon Bushes	65-001	Date (Days)	-	0 D	ays	-	-	COMPLIED WITH
Limitation Note: FAC									
REFER HELISTAR A	IRFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Checking Two Tail Gearbox Casing Attachment Lugs	65-002	Date (Days)	-	0 D	ays	=	-	NOT APPLICABLE
Limitation Note: N/A	, TGB S/N M192 INSTALLED.								
REFER AIRBUS HEL	LICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.							
Standard SBs	Increase of the Centering Length on the Output Gear	65-003		-		0	-	-	NOT APPLICABLE
	of the Tail Rotor Gear Box								
Limitation Note: N/A	BY TGB P/N C652A0101055 INSTALLED								
REFER AIRBUS HEL	ICOPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.							



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Standard SBs	Installation of new TGB pitch control rod lubrification	65-005		-		0	-	-	OPTIONAL
	cap								
Standard SBs	TAIL ROTOR DRIVE - Tail rotor drive shaft. Installation of a blanking plate on the drive tube on	65-006	Date (Days)	-	0 D	ays	-	-	OPTIONAL
	the TGB side								
Standard SBs	Tail Rotor Drive Shaft Damper	65A004	Date (Days)	-	0 D	ays	-	-	COMPLIED WITH
Limitation Note: COM	MPLIED WITH ON 13/01/2004 AT A/C 1182.70 FH. REFER /	AUSTRAL	AN MODIFICATION BOOK						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Replacement of Pilot and Copilot Pedal Unit Links (RH Side)	67-001		-		0	-	-	OPTIONAL
Standard SBs	Improved Resistance of the Pedal Torque Tube to Hammering	67-002		-		0	-	-	OPTIONAL
Standard SBs	Provision for Quick Removal of LH Controls	67-004		-		0	-	-	OPTIONAL
Standard SBs	Increased Stability of Pedal Unit Adjustment	67-006		-		0	-	-	OPTIONAL
Standard SBs	Replacing the Rivets on the Hinge Housing of the LH Cyclic Stick	67-012	Date (Days)	-	0 D	ays	-	-	NOT APPLICABLE
Limitation Note: N/A	,								
REFER AIRBUS HELI	COPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.							
Standard SBs	Catching up play in collective levers, proposal for rework	67-015		-		0	-	-	OPTIONAL
Standard SBs	Reserved	67-016		_		0	-	_	FOR INFO ONLY
Standard SBs	Replacement of the RH Collective Lever End on the	67-017	Date (Days)	_	0 D	ays	05/05/2017	_	COMPLIED WITH
	Collective Torque Tube Assembly IPLIED ON 05/05/2017 BY MYCOPTER AT A/C 2913.60FH					,-			
Limitation Note: Colv	II EIED ON 03/03/2017 BT WITCOT TEN AT A/O 2913.00111	. IXLI IVI	UNU 30#2002						
	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Dual mechanical locking of flying control stops	67-018		-		0	-	-	OPTIONAL
Standard SBs	Nose-up stop modification after incorporation of a ring into the droop restrainer ring	67-019		-		0	-	-	OPTIONAL
Standard SBs	Fixed length anticipator link	67-020		-		0	-	-	OPTIONAL
Standard SBs	Collective Levers: Collective Lever Balancing	67-021		-		0	-	-	OPTIONAL
Standard SBs	RH pedal unit: Play in the torque tube-to-yaw ball-	67-022	Date (Days)	-	0 D	ays	18/01/2020	-	COMPLIED WITH
	type control lever assembly								
Limitation Note: COM	1PLIED ON 18/01/2020 AT A/C 2042.40FH. REF.: ECM WO	2008 183	023						
REFER HELISTAR AII	RFRAME SERVICE BULLETIN COMPLIANCE STATUS.								
Standard SBs	Reserved	67-023		-		0	-	-	FOR INFO ONLY
Standard SBs	Check for free movement and lubrication of the swivel bearing of the aft yaw ball-type control	67-024	Date (Days)	-	0 D	ays	-	-	NOT APPLICABLE
Limitation Note: N/A	BY AFT YAW MP/N: 76CF8921 (P/N: 7050A3672002.) REF	ER W/O 2	011 485 007 FOR DETAILS.						
REFER AIRBUS HELI	COPTER AIRFRAME SERVICE BULLETIN COMPLIANCE	STATUS.							
Standard SBs		67-025R							



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Standard SBs	Separation of the rear wall and air intake grid.	319 71 4025	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M01 module (reductor gearbox). Addition of a sealing ensured by preformed packing on the output gear.	319 72 4026	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Gas generator module. Gas generator rear bearing with increasing min. axial clearance.	319 72 4027	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	FCU assembly. Preformed packings made of 61D6.29 Apr 2020	319 73 4028	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02 module (gas generator). Air intake casing with built-in attachement flange for Tf 25A air intake grids.	319 72 4029	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02 module (gas generator). Borescope port blanking with renforced flange.	319 72 4030	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs Limitation Note: TF31	Module M01 (reduction gearbox).Reduction gear front casing with large link radius at the mating face of the torquemeter box cover. A TO BE EMBODIED UPON CUSTOMER REQUEST.	319 72 4031	Not Applicable	-	0	-	-	OPTIONAL
REFER AIRRUS HELIC	COPTER ENGINE SERVICE BULLETIN COMPLIANCE STA	ATUS						
Standard SBs	M01. Reduction gear front casing with large radius pitting at the bending radius of the torquemeter box cover mating face	319 72 4032	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Extension of fire detection harness	319 26 4033	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Oil system. BM 1030 electrical magnetic plug for the drain unit.	319 79 4034	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Oil system. BR1091 oil drain pump.	319 79 4035	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Oil system. BM 1030 rear bearing electrical magnetic plug.	319 79 4036	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Check valve seat assembly attached by bondings and with screws.	319 73 4037		-	0	-	-	CANCELLED
Standard SBs	Fuel Control Unit (FCU). Improved HP pump drain sealing ring.	319 72 4038	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	Fuel control unit (FCU). Adjustment of maximum flow for JP4 fuel.	319 73 4039	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02 module (gas generateor). Linkage between the rear bearing / bearing support / power turbine (PT) flange using underheard free-cut screws.	319 72 4040	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02 module (gas generator). Linkage between rear bearing/bearing support/power turbine (PT) flange using underhead free-cut screws and crimped nuts.	319 72 4041	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02 module (gas generator). Rear bear oil ducts with toleranced ends.	319 72 4043	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02 module (gas generator). Rear bearing C with increased diameter	319 72 4044	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02. Non-silver-plated mating face of power turbine nuts	319 72 4045	Not Applicable	-	0	-	-	FOR INFO ONLY
Standard SBs	M02. Simplified turbine casing	319 72 4046	Not Applicable	-	0	-	-	FOR INFO ONLY



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Standard SBs	Fuel Control Unit (FCU). NTL flyweight shaft with "plasma coating".	319 73 4050	Not Applicable	-	0	-	- FOR INFO	O ONLY
Standard SBs	Fuel Control Unit (FCU). Acceleration slope adjustment support held in place by a cover.	319 73 4051	Not Applicable	-	0	-	- FOR INFO	O ONLY
Standard SBs	M02. Laser drilled flame tube	319 72 4052	Not Applicable	-	0	-	- FOR INFO	O ONLY
Standard SBs	M02. Flame tube with prevaporizing tubes protected by chromium/aluminium-plating and HP turbine cooling air scallops	319 72 4053	Not Applicable	-	0	-	- FOR INFO	O ONLY
Standard SBs	Adjusted valve assembly. Start electro-valve with improved internal sealing.	319 73 4054	Not Applicable	-	0	-	- FOR INFO	O ONLY
Standard SBs	Fuel Control Unit (FCU). To ensure proper installation of constant delta P valve diaphragm.	319 73 4055	Not Applicable	-	0	-	- NOT APPL	ICABLE
.imitation Note: NC	OT APPLICABLE DUE TO FCU INSTALLED POST TF55. REF	FER SAFRAN ENGINE LOG BO	OOK					
REFER HELISTAR E	ENGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Installation of a load bearing magnetic seal on the starter drive.	319 72 4056	Not Applicable	-	0	-	- FOR INFO	O ONLY
tandard SBs	Module M02 (gas generator). Centrifugal diffuser assembled by brazing on half-through valves.	319 72 4057	Not Applicable	-	0	-	- COMPLIE	D WITH
.imitation Note: CC	OMPLIED WITH BY TURBOMECA ON 30 MAR 2007 AT ENG	INE TT 1828 HRS. REFER TUF	RBOMECA ENGINE LOG BOOK	K, SECTION E, W/O MD618	00			
REFER AIRBUS HE	LICOPTER ENGINE SERVICE BULLETIN COMPLIANCE ST	ATUS.						
Standard SBs	Module M01 (reduction gearbox).Installation of a magnetic seal with a lift face on the breather outlet.	319 72 4058		-	0	-	- OP	TIONAL
imitation Note: TF	58 TO BE EMBODIED UPON CUSTOMER'S REQUEST							
REFER AIRBUS HE	LICOPTER ENGINE SERVICE BULLETIN COMPLIANCE ST	ATUS.						
Standard SBs	Air system. Air intake, half grids with opening to allow entry of borescope duct.	319 71 4059	Not Applicable	-	0	-	- FOR INFO	O ONLY
Standard SBs	Module M01 (reduction gearbox)Attaching screw of the fuel control unit adaptor.	319 72 4060	Not Applicable	-	0	-	- COMPLIE	D WITH
.imitation Note: CC	MPLIED WITH BY TURBOMECA ON 30 MAR 2007 AT ENG	INE TT 1828 HRS. REFER TUP	RBOMECA ENGINE LOG BOOK	X, SECTION E, W/O MD618	00			
DEEED AIDDIIG HE	LICOPTER ENGINE SERVICE BULLETIN COMPLIANCE ST	ATUS.						
KEFEK AIKBUS HE					0		EOD INE	
	Module M01 (reduction gearbox). Addition of a steel torquemeter stop.	319 72 4061	Not Applicable	-	O	- -	- FOR INFO	OONLT
Standard SBs	- · · · · · · · · · · · · · · · · · · ·	319 72 4061 319 72 4062	Not Applicable	-	0	-	- FOR INFO	
Standard SBs Standard SBs	torquemeter stop. M02. Flame tube with prevaporizing tubes made of			-		-		O ONLY
Standard SBs	torquemeter stop. M02. Flame tube with prevaporizing tubes made of HS31 (KC25NW) and HP turbine cooling air scallops Changes to the pressure-reducing valve cover	319 72 4062	Not Applicable	-	0	- - -	- FOR INFO	O ONLY



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Standard SBs	Module M02 (gas generator). Reinforced rings to hold turbine casing/rear bearing support/High Pressure Turbine (HPT) ring. Stopping of incorporation / Elimination of modification Tf 71A	319 72 4071	Not Applicable	-	0	-	- FOR INF	FO ONLY
Standard SBs Limitation Note: NOT BOOK		319 79 4075 AN ENGINE LOG		-	0	-	- NOT APP	LICABLE
REFER HELISTAR EI	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Module M02 (gas generator). Centrifugal diffuser assembled by brazing on through double vanes. 6A TO BE EMBODIED WHEN REPLACEMENT OF M02 OR 6	319 72 4076	Not Applicable	-	0	-	- OF	PTIONAL
	ovio de embodied vineviver e de meivr d'impe divi	oenna oone bii i ooen oo						
	ICOPTER ENGINE SERVICE BULLETIN COMPLIANCE STA				0		NOT ADD	
Standard SBs	delta P valve.	319 73 4077		-	0	-	- NOT APP	LICABLE
Limitation Note: NOT	T APPLICABLE DUE TO FCU POST TF77 INSTALLED. REFE	ER SAFRAN ENGINE LOG BOO	K					
REFER HELISTAR E	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Module M01 (reduction gearbox). Output gear without damping ring.	319 72 4078	Not Applicable	-	0	-	- OF	PTIONAL
Limitation Note: TF7	8 TO BE EMBODIED WHEN REPLACEMENT OF M01 IS CA	RRY OUT.						
REFER AIRBUS HEL	ICOPTER ENGINE SERVICE BULLETIN COMPLIANCE STA	TUS.						
Standard SBs	Module M01 (reduction gearbox).Intermediate gear	319 72 4079	Not Applicable	-	0	-	- OF	PTIONAL
Limitation Note: TF 7	without damping ring. 79 TO BE EMBODIED WHEN REPLACEMENT OF M01 IS CA	ARRIED OUT.						
REFER EUROCOPTE	ER ENGINE SERVICE BULLETIN COMPLIANCE STATUS							
Standard SBs	interference between the right rear half-wall and the	319 71 4080		-	0	-	- OF	PTIONAL
Standard SBs	P3 pipe (first section). If post TF25 Adjusted valve assembly cover.Interference between the adjusted valve assembly cover and the fuel	319 71 4081		-	0	-	- CAN	NCELLED
Standard SBs	pipes. Flame tube with change of material and addition of a ceramic coating at the bottom of the chamber	319 72 4082		-	0	-	- OF	PTIONAL
Limitation Note: TO	BE CALL OUT AT NEXT INSPECTION							
REFER AIRBUS HELL	ICOPTER ENGINE SERVICE BULLETIN COMPLIANCE STA	TUS						
Standard SBs	Fuel system. Part Numbers created for the right-hand and left-hand injector half-manifolds and the privilege injector pipe, which are specific to the ARRIUS 2F variant.			-	0	-	- Of	PTIONAL



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Standard SBs	Gas generator module. Improved squeeze film for the	319 72 4084		-	0	-	- NOT APF	LICABL
	gas generator front bearing. Application of							
	modification Tf84							
imitation Note: NOT	APPLICABLE DUE TO ENGINE POST TF 12. REFER SAF	FRAN ENGINE LOG BOOK						
REFER HELISTAR EI	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
Standard SBs	Introduction of insulating seals below the injector	319 73 4085		-	0	-	- CAI	NCELLE
	flange of both the lefthand and righthand injector							
	halfmanifold assemblies and the privilege injector							
	pipe assembly. Incorporation of modification Tf85.							
tandard SBs	To improve attachement of fire detection harness to	319 26 4801		-	0	-	- COMPL	IED WI
	fire detectors. Addition of RHODORSIL CAF 730							
	sealant							
imitation Note: CON	MPLIED WITH BY TURBOMECA ON 30 MOR 2007 AT ENG	INE TT 1828 HRS. REFER TURBOMEC	CA ENGINE LOG BOOK	K, SECTION E, W/O# MD61	1800.			
EFER EUROCOPTE	R ENGINE SERVICE BULLETIN COMPLIANCE STATUS							
tandard SBs	Lubrication unit. Replacement of check valve piston	A319 79 4802	Not Applicable	-	0	-	- NOT APF	PLICAB
	preformed packing.Comments: If pre Tf 75							
imitation Note: NOT	APPLICABLE DUE TO ENGINE POST TF75. REFER SAF	RAN ENGINE LOG BOOK						
EFER HELISTAR EI	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4803		-	0	-	- RECOM	IMEND
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4804		-	0	-	- RECOM	IMEND
tandard SBs	Replacement procedure of the cooling ring	319 72 4805		-	0	-	- O	PTION
tandard SBs	Superseded by SB No. 4826	319 72 4806		-	0	-	- RECOM	IMEND
tandard SBs	Procedure to replace the power turbine ring	319 72 4807		-	0	-	- O	PTION
standard SBs	Fuel Control Unit. Increase of the max fuel flow "on	A319 73 4808	Not Applicable	-	0	-	- NOT APF	PLICAE
	site".							
imitation Note: NOT	APPLICABLE DUE TO FCU PN 0319878010 SN 47010 INS	STALLED						
EFER HELISTAR EI	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
tandard SBs	Superseded by SB No. 319 72 4826	319 73 4809		-	0	-	- RECOM	IMEND
tandard SBs	Inspection of clearance between the P3 air pipe (first	319 75 4810		-	50	-	- NOT APF	PLICAB
	section) and the RH rear halfwall							
imitation Note: NOT	APPLICABLE DUE TO ENGINE POST TF80. REFER SAFE	RAN ENG INE LOG BOOK						
EFER HELISTAR EI	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.							
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4811		-	0	-	- RECOM	IMEND
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4812		-	0	-	- RECOM	
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4813		_	0	-	- RECOM	
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4814		_	0	-	- RECOM	
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4815		-	0	-	- RECOM	
andard SBs	Superseded by SB No. 319 72 4826	319 72 4816		-	0	-	- RECOM	
andard SBs	Superseded by SB No. 319 72 4826	319 72 4817		-	0	-	- RECOM	
	·				0			IMEND
tandard SBs	Superseded by SB No. 319 72 4826	319 72 4818		-	U	-	- KEUUW	



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Standard SBs	Superseded by SB No. 319 72 4826	319 72 4820		-	0	-	- RECOMMENDE
Standard SBs	Superseded by SB No. 319 72 4826	319 72 4821		-	0	-	- RECOMMENDE
Standard SBs	Centrifugal diffuser vane check.	319 72 4822	Not Applicable	-	0	-	 NOT APPLICABL
Limitation Note: N/A	A. MO2 S/N: 322 POST TF57A						
REFER AIRBUS HEL	LICOPTER ENGINE SERVICE BULLETIN COMPLIANCE S	STATUS.					
Standard SBs	Check of correct position of adjusted FCU fuel filter	. A319 73 4823		-	0	-	 COMPLIED WIT
Limitation Note: CO	MPLIED WITH ENGINE 1470.0 FH . REFER ECM WO # 2	005-026					
REFER HELISTAR E	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.						
Standard SBs	Superseded by SB No. 319 72 4826	319 73 4824		-	0	-	- RECOMMENDE
Standard SBs	Fuel control unit. Check and replacement of the	A319 73 4825	Not Applicable	-	0	-	 NOT APPLICABL
	constant delta P device diaphragm.						
Limitation Note: NO	T APPLICABLE DUE TO FCU PN 0319878010 SN 47010 I	NSTALLED					
REFER HELISTAR E	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.						
Standard SBs	Deep Maintenance technical Instructions organized	319 72 4826	Not Applicable	-	0	-	 FOR INFO ONL
	Into 2 (two) books. Comments:Book X 319 L6 351 2	2					
	Book X 319 L6 352 2						
Standard SBs	Check of flow rate from fuel injection manifolds at a	319 73 4829	Not Applicable	-	0	-	 FOR INFO ONL
	Service Station. Comments: Expired						
Standard SBs	Module M02 (gas generator). Inspection of the oil je			=	0	-	- CANCELLE
	support of the power turbine rear bearing assembly						
Standard SBs	Lubrication unit. Inspection of jet bonding on ejector assembly	r 319 79 4831		-	0	-	- COMPLIED WIT
Limitation Note: CO	MPLIED WITH ON 06 AUG 10 AT ENGINE TT 2054.2 HRS	B BY TURBOMECA PERSONN	NEL. REFER ENG LOG BOOK PG. E	23.			
REFER AIRBUS HEI	LICOPTER ENGINE SERVICE BULLETIN COMPLIANCE S	STATUS					
Standard SBs	Replacement of post TF 77 regulator. Comments:S		Not Applicable	-	0	-	- FOR INFO ONL
Olandara ODO	expired after end of replacement and/or control of	.5 0.0.0 10.002	11017 (pp.11042)10		· ·		
	FCU. Restricted						
Standard SBs	Visual inspection of the drive-link splines of the	319 79 4834		-	0	-	- COMPLIED WIT
Limitation Note: CO	lubricating device MPLIED DURING ENGINE RETURN TO REPAIR CENTRI	= ON 08/05/2015 REFER SAF	FRAN ENGINE LOG BOOK				
Zimilation Noto: 00	III ELED BOMING ENGINE METORIA TO NEL 7 III GERTAIN	2 011 00/00/2010. 1121 211 0/11	TWIN ENGINE 200 BOOK				
REFER HELISTAR E	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.						
Standard SBs	Lubrication device. Check of the nozzle bonding on	319 79 4835		-	0	-	 NOT APPLICABL
Limitation Note: NO	the ejector assembly. T APPLICABLE DUE TO COMPLIED WITH MSB 319 79 4	931 VED A ON 06/ 09/2010 P	EEED MVCAS SBK 15 004				
Liiiitation Note. NO	1741 LIOADLE DOE 10 COMPLIED WITH MICE 319 79 44	001 VEINA ON 00/ 00/2010. R	E. E. MII ONO-ODIV-10-004				
	NGINE#1 SERVICE BULLETIN COMPLIANCE STATUS.						
Standard SBs	Updating of the engine log book to extend the TBO	of 319 73 4837		-	0	-	- OPTIONA
	the ARRIUS 2F Pump and Metering Valve						
	Assembly, Adjusted						



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Standard SBs	Updating of the engine log book to extend the TBO of the ARRIUS 2F Pump and Metering Valve Assembly, Adjusted	319 72 4838	Not Applicable	-	0	-	- FOR	INFO ONLY
Standard SBs	Return to service for civil operation of an ARRIUS 2 F or ARRIUS 2 R engine originating from an operator not controlled by a Civil Authority military, paramilitary, etc.)	319 73 4839	Date (Months)	- 1	0 Months	-	-	OPTIONAL
Standard SBs	Creation of a configuration applicable to ARRIUS 2B1, 2B1A, 2B2, 2F, 2K1 and 2K2 turboshaft engines during overhaul	319 73 4846		-	0	-	- FOR	INFO ONLY
Standard SBs	GTN 650 and GTN 750 Notification of Mod Level 5. This service bulletin provides instructions to inspect and return affected GTN 650/750 units to Garmin for a Mod Level change. This service bulletin also provides optional instructions for marking Mod Level 5 on the nameplate (serial tag) of unaffected GTN 650/750 units. Mandatory for units listed in Table 2. This Service Bulletin must be accomplished by October 31st, 2016	MSB No. 1559		-	0	-	- FOR	INFO ONLY