

RECORD OF TEMPORARY REVISIONS

ACTIVE TEMPORARY REVISIONS

The following Temporary Revisions are still active and must be retained in your printed manual.

For Temporary Revisions, changes from the basic issue are printed in red.

TR N°	Page	Approval	Note
2-3	2-6, 2-9	EASA Approval N° R.A. 01396 dated 22 October 2007	-
5-4	5-2, 5-2A, S17-i, S17-iA, S17-1, S17-1A	Approved with NDC-139G2350-012 dated 30 November 2022 under the authority of DOA ref. EASA.21J.005	_



INACTIVE TEMPORARY REVISIONS						
The following Temporary Revisions are inactive and must not be retained in your printed manual:						
TR N°	Page	Approval	Note			
5-1	S11-8A, S11-8B, S11-21, S11-22, S11-24, S11-25	Approved with NDC-139G0257-026 P. EASA 10072085	Introduced with RFM Issue 2 Revision 25.			
5-2	5-5A, 5-5, S52-iA, S52-i, S52-1A, S52-1, S52-6, S52-7	Approved with NDC-139G9350-010 dated 23 September 2020 under the authority of DOA ref. EASA.21J.005	Introduced with RFM Issue 2 Revision 27.			
5-3	5-5A, 5-5, S52-iA, S52-i, S52-1A, S52-1, S52-2, S52-3, S52-4, S52-4A, S52-4B, S52-4C	Approved with NDC-139G9350-012 dated 28 May 2021 under the authority of DOA ref. EASA.21J.005	Introduced with RFM Issue 2 Revision 28.			



Supplement No.	Name of equipment	P/N
7	VVIP Interior	3G2520F00311 3G2520F00312
8	Radio Equipment Primus HF 1050 Version with Cabin Controller	3G2310F00211 3G2310F00212
9	Ditching Configurations	-
10	Crash Position Indicator with Deployable E.L.T.	3G2560F00311 3G2560F00312 3G2560F00313
11	External Hoist Operations	3G2591F00111 or 3G2591F00112 or 3G2591F00113
12	Category A Operations	-
13	Cargo Hook Operations	3G2592F00111
14	Removed	N/A
15	Auxiliary Fuel Tank	3G281500F00111
16	Radio Equipment V/UHF FLEXCOMM II Second FLEXCOMM II	3G2310F00111 4G2310F01211 3G2310F00711 4G2310F01212
17	External and Internal Public Address System	3G2350F00211 4G2350F00511 4G2350F00611 4G2350F00111 4G2350F02111 4G2350F00512 4G2350F03011
18	Second Radar Altimeter RT300	3G3440F00111



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The information contained in this document supplements the information of the Basic Flight Manual. For limitations, procedures and performance data not contained in this Supplement, consult the basic Rotorcraft Flight Manual.

SUPPLEMENT 17

EXTERNAL AND INTERNAL PUBLIC ADDRESS SYSTEM

P/N 3G2350F00211 P/N 4G2350F00511 P/N 4G2350F00611 P/N 4G2350F00111 P/N 4G2350F02111 P/N 4G2350F00512 P/N 4G2350F03011



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GENERAL INFORMATION

The External and Internal Public Address System (PA) installation consists of an external loud speaker unit (mounted under the tail cone) and internally mounted speakers. The loud speakers are controlled by the NAT AA21 PA or alternatively the LSC22-033N control panel mounted in the interseat console (See Figure 1). The panel allows selection of internal (INT) or external (EXT) speakers, siren tone (WAIL/YELP), volume control and selection of PA or RADIO

re-broadcast (provision only). The panel LSC-033N has an additional AUX plug to permit external devices to be played and a record setting to record AUX or MIC. The PA control panel is powered continuously when the circuit breaker is in.

The control panel is interfaced with the aircraft ICS AV900 Audio panel (See Figure 3). Pushing the PA button the Pilot/Copilot microphone messages may be broadcast internally or externally.

The system gives an additional internal audio tone when the NO SMOKING and/or FASTEN SEAT BELT signs are activated using the ICS audio panel.

The direction of the loud speaker configuration, under the tail cone, for kit P/N 3G2350F00211, P/N 4G2350F00511, P/N 4G2350F00611, P/N 4G2350F00512 and P/N 4G2350F03011 is vertically downwards with respect to the aircraft BL.

However, for Kit P/N 4G2350F02111 the speakers are inclined 30° to the right (see Figure 4). This puts ground personnel in the right hand quadrant of the helicopter in the optimum position to hear the speaker communication and allows the pilot to visually see the ground personnel to whom he is addressing.

The PA panels are NVIS green compliant.



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