

WORK ORDER

2016 193 010

(SMI)

A/C Reg	9M-LKE
A/C S/N	8297
A/C Type	AS350 B3 e

Schedule Maintenance Inspection /Retrofit/Role Change	Man Hours
1M Emergency Locator Transmitter Autotest. P/N: S1854501-01/704A45737078)	
KANNAD INTEGRA AP-H (ER)) i.a.w AMM 25-66-20, 5-1	
7D//10H para. 3 of the applicable ASB. Visual & periodic cond. chk of the elastomer of the TRH pitch change rod ball joint.	

Defect Rectification	Man Hours



WORK PACK INDEX

AIRCRAFT REGN: 9M-LKE

AIRCRAFT S/No : 8297

WORK ORDER No: 2016 193 010

AIRCRAFT TYPE : AS350B3 e

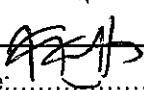

SHEET _____ OF _____

No.	Description	No. of Pages
1	WORK PACK CONTROL PAGE	1
2	CUSTOMER ORDER SHEET	1
3	WORK SHEETS	0

TOTAL	2
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PLANNING CLEARANCE

Remarks: _____

Signature: _____  

All work raised above has been accounted for and completed.



WORK PACK - CONTROL PAGE

1. WORK PACK COMPILED BY ...ZURAIDA...

2. Start date: (dd/mm/yy) <i>25/4/2017</i>		3. A/C Reg. No. 9M-LKE		4. Work Order No. / Customer Order No. 2016 193 010	
5. Customer's name KI HUAT TIMBER		6. Customer's Technical Log No. -		INTENTIONALLY LEFT BLANK	
8. A/C Type AS350 B3e		9. A/C S/N 8297		10. A/C TT <i>4.5</i>	11. Landings <i>28</i>
12. Engine Type: ARRIEL 2D	13. Engine S/N: Eng # 1 50942 Eng # 2 NA	14. Engine TT: <i>4.5</i> NA	15. Engine Ng Cycle: <i>70.00</i> NA	16. Engine Nf Cycle <i>8.00</i> NA	
17. Summary of customers order: Type of check (SMI) 1M Emergency Locator Transmitter Autotest. P/N: S1854501-01/704A45737078) KANNAD INTEGRA AP-H (ER)) i.a.w AMM 25-66-20, 5-1 7D//10H para. 3 of the applicable ASB. Visual & periodic cond. chk of the elastomer of the TRH pitch change rod ball joint.					
18. Number of Customer Order Sheets <i>1</i>		19. Number of Work Sheets <i>0</i>		20. Number of Part Used Sheets <i>0</i>	
21. Aircraft Maintenance Manual Rev. No. Rev 12 Dated 03 Nov 2016 <i>Rev 13 Dated 09 Jan 2017</i>		22. Engine Maintenance Manual Rev. No. Rev 12 Dated 30 May 2016 <i>Rev 13 Dated 30 Nov 2016</i>		23. Maintenance Schedule Ref. No. AHM/MS/AS350B3 Version, Issue no 1, Rev no 0, Date: 7 May 2014 <i>Rev 1, Date: 9 Jan 2017</i>	

[Signature]

[Signature]

e.i.e

24. Work Pack Closed		25. Technical Record Update	
Date (dd/mm/yy)	Signature & Stamp	Date (dd/mm/yy)	Signature & Stamp
<i>26/4/2017</i>	<i>[Signature]</i>	<i>27/4/17</i>	<i>[Signature]</i>



1.

Work Order No.:

2016 193 010

2. Item	3. Work Required	5. Performed by		6. Inspected by - CRS *		
		Sign	Auth. no.	Sign	Auth. no.	Date
0020	1M Emergency Locator Transmitter Autotest. P/N: S1854501-01/704A45737078) KANNAD INTEGRA AP-H (ER)) i.a.w AMM 25-66-20, 5-1 Remarks <u>CARRIED OUT, FOUND 'SATIS'</u>		328			25.04.17.
0030	7D//10H para. 3 of the applicable ASB. Visual & periodic cond. chk of the elastomer of the TRH pitch change rod ball joint. (References per EASB 05.00-86 Rev. 1) Remarks <u>C/O, SATIS.</u>					26/4/17

7. * The work recorded above has been carried out in accordance with the requirements of the ** Malaysian Civil Aviation Regulation / FAA Regulations / _____ for the time being in force and in that respect the aircraft /equipment is considered fit for Release to Service.
** Delete as required.



Emergency Airworthiness Directive

AD No.: 2017-0020-E

Issued: 07 February 2017

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) 216/2008 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EC) 216/2008, Article 14(4) exemption].

Design Approval Holder's Name:

AIRBUS HELICOPTERS

Type/Model designation(s):

AS 350 and AS 355 Helicopters

Effective Date: 09 February 2017

TCDS Number(s): EASA.R.008, EASA.R.146

Foreign AD: Not applicable

Supersedure: None

ATA 05 – Tail Rotor – Pitch Rod – Inspection

Manufacturer(s):

Airbus Helicopters (formerly Eurocopter, Eurocopter France, Aerospatiale).

Applicability:

AS 350 B, AS 350 BA, AS 350 BB, AS 350 B1, AS 350 B2, AS 350 B3, AS 355 E, AS 355 F, AS 355 F1, AS 355 F2, AS 355 N and AS 355 NP helicopters, all serial numbers, which embody modification (mod) 075601 or mod 076602.

Reason:

Several cases have been reported of finding significant damage of the elastomeric ball-joint of the tail rotor (TR) pitch rod (horn side) on AS 350 and AS 355 helicopters. Investigation is on-going to determine the cause for this damage.

This condition, if not detected and corrected, could lead to loss of control of the helicopter.

To address this unsafe condition, Airbus Helicopters (AH) issued Emergency Alert Safety Bulletin (ASB) No AS350-05.00.86 and ASB AS355-05.00.75, later revised (hereafter collectively referred to as 'the applicable ASB' in this AD) to reduce the interval of the visual inspection, as currently required by chapter 04-20-00 of the applicable Airworthiness Limitation Section (ALS), from 50 flight hours (FH) to 10 FH.



For the reasons described above, this AD requires repetitive inspections of the TR pitch rod, as specified in ALS chapter 04-20-00, but at reduced intervals.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

- (1) Before exceeding 50 FH since the last inspection per ALS chapter 04-20-00, or within 10 FH or 7 days, whichever occurs first after the effective date of this AD, and, thereafter, at intervals not to exceed 10 FH, visually inspect the TR pitch rod in accordance with the instructions of paragraph 3 of the applicable ASB.
- (2) If, during any inspection as required by paragraph (1) of this AD, any damage, as detailed in the instructions of the applicable ASB, is found on one or more layers, with a circumference of more than 90 degrees, before next flight, replace TR pitch change rod with a serviceable pitch rod in accordance with the instructions of the applicable ASB.
- (3) Replacement of a TR pitch rod on a helicopter, as required by paragraph (2) of this AD, does not constitute terminating action for the repetitive inspection as required by paragraph (1) of this AD for that helicopter.

Ref. Publications:

Airbus Helicopters Emergency ASB AS350-05.00.86 Revision 01 dated February 06, 2017.

Airbus Helicopters Emergency ASB AS355-05.00.75 Revision 01 dated February 06, 2017.

The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. The results of the safety assessment have indicated the need for immediate publication and notification, without the full consultation process.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters (EBSESB) – Aéroport de Marseille Provence, 13725 Marignane Cedex, France; Telephone: +33 (4) 12 85 97 97; Fax: +33 (4) 85 99 66; E-mail: Directive.technical-support@airbus.com.

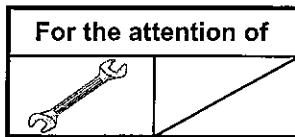




EMERGENCY ALERT SERVICE BULLETIN

SUBJECT: TIME LIMITS - MAINTENANCE CHECKS - Pitch change rod

Periodic check of the Tail Rotor Hub (TRH) pitch change rod



HELICOPTER(S) CONCERNED	NUMBER	Version(s)	
		Civil	Military
AS350	05.00.86	B, BA, BB, B1, B2, B3, D	L1
AS355	05.00.75	E, F, F1, F2, N, NP	
AS550	05.00.62		A2, C2, C3, U2
AS555	05.00.55		AF, AN, AP, SN, UF, UN

Revision No.	Date of issue
Revision 0	2017-01-23
Revision 1	2017-02-06

Summary:

Visual and periodic condition check of the elastomer of the TRH pitch change rod ball joint.

Reason for last Revision:

To add a periodic time limit which does not exceed 10 flying hours to paragraph 1.E.2. Compliance in service.

Compliance:

Airbus Helicopters renders compliance with this ALERT SERVICE BULLETIN mandatory.

1. PLANNING INFORMATION

1.A. EFFECTIVITY

1.A.1. Helicopters/installed equipment or parts

Helicopters which embody MOD 075601 or 076602.

NOTE 1

MODs 075601 and 076602 consist in replacing the tail rotor pitch change rod with an elastomer ball joint rod.

NOTE 2

*Modification 076602 is available through the following Service Bulletins:
AS350 No. 64.00.09
AS355 No. 64.00.07
AS550 No. 64.00.06
AS555 No. 64.00.05.*

NOTE 3

Refer to the aircraft individual inspection record (MOD record), the log card of the Tail gearbox or the aircraft log book to identify the actual configuration of the helicopter.

1.A.2. Non-installed equipment or parts

Not applicable.

1.B. ASSOCIATED REQUIREMENTS

Not applicable.

1.C. REASON

Revision 0:

Airbus Helicopters has been informed of several cases of significant damage to the elastomer ball-joint (rupture of an elastomer layer) of the pitch change rod attached on the tail rotor pitch horn side.

This damage was found during:

- the visual inspection required by the Airworthiness Limitations Section (ALS) every 50 flying hours,
- and during the detailed inspection required by the maintenance program (MSM) as part of each T inspection.

In order to improve flight safety and pending test results to confirm the current inspection intervals, Airbus Helicopters has decided to reduce the interval between two visual inspections **from 50 to 10 flying hours**. Airbus Helicopters renders compliance with this ALERT SERVICE BULLETIN mandatory.

Revision 1:

The purpose of Revision 1 is to add a periodic time limit which does not exceed 10 flying hours to paragraph 1.E.2.

Revision 1 does not affect compliance with this ALERT SERVICE BULLETIN.
Airbus Helicopters makes compliance with this ALERT SERVICE BULLETIN mandatory.

1.D. DESCRIPTION

Compliance with this ALERT SERVICE BULLETIN consists in:

- visually and periodically checking the condition of the elastomer,
- replacing the pitch change rod if damage beyond removal criteria is found.

1.E. COMPLIANCE**1.E.1. Compliance at H/C manufacturer level**

Not applicable.

1.E.2. Compliance in service

The work on the helicopter is to be performed by the operator.

Helicopters/installed equipment or parts:

Comply with paragraph 3. within 10 flying hours following receipt of this ALERT SERVICE BULLETIN, issued on January 23, 2017, without exceeding 50 flying hours since the latest inspection as per ALS chapter 04-20-00.

Then,

Comply with paragraph 3.B.1. at intervals which do not exceed 10 flying hours.

Non-installed equipment or parts:

Not applicable.

1.F. APPROVALApproval of modifications:

Not applicable.

Approval of this document:

The technical information contained in this ALERT SERVICE BULLETIN Revision 0 was approved on January 23, 2017 under the authority of EASA Design Organization Approval No. 21J.700 for civil version helicopters subject to an Airworthiness Certificate.

The technical information contained in this ALERT SERVICE BULLETIN Revision 0 was approved on January 23, 2017 under the prerogatives of the recognition of design capability FRA21J-002-DGA for French Government helicopters.

The technical information contained in this ALERT SERVICE BULLETIN Revision 0 was approved on January 23, 2017 by the Airbus Helicopters Airworthiness Department for export military versions.

The technical information contained in this ALERT SERVICE BULLETIN Revision 1 was approved on February 06, 2017 under the authority of EASA Design Organization Approval No. 21J.700 for civil version helicopters subject to an Airworthiness Certificate.

The technical information contained in this ALERT SERVICE BULLETIN Revision 1 was approved on February 06, 2017 under the prerogatives of the recognition of design capability FRA21J-002-DGA for French Government helicopters.

The technical information contained in this ALERT SERVICE BULLETIN Revision 1 was approved on February 06, 2017 by the Airbus Helicopters Airworthiness Department for export military versions.

1.G. MANPOWER



For compliance with this ALERT SERVICE BULLETIN, Airbus Helicopters recommends the following personnel qualifications:



Qualification: 1 Mechanical Engineering Technician or 1 Pilot.

The time for the operations is given for information purposes, for a standard configuration.

Time for the operations: (approximately) 5 minutes.

1.H. WEIGHT AND BALANCE

Not applicable.

1.I. POWER CONSUMPTION

Not applicable.

1.J. SOFTWARE UPGRADES/UPDATES

Not applicable.

1.K. REFERENCES

The documents required for compliance with this ALERT SERVICE BULLETIN are as follows:

Aircraft Maintenance Manual (AMM):

Task: 65-21-00, 4-12b: Removal / Installation - Pitch-Change Links POST MOD 075601 or POST MOD 076602

Maintenance Manual (MET):

Work Card: 64.10.00.402: Tail rotor pitch-change links (POST MOD 076602): Removal - Installation

1.L. OTHER AFFECTED PUBLICATIONS

Not applicable.

1.M. PART INTERCHANGEABILITY OR MIXABILITY

Not applicable.

2. EQUIPMENT OR PARTS INFORMATION
2.A. EQUIPMENT OR PARTS: PRICE - AVAILABILITY - PROCUREMENT

For any information concerning the kits and/or components or for assistance, contact the Airbus Helicopters Network Sales & Customer Relations Department.

Airbus Helicopters
 Etablissement de Marignane
 Direction Ventes et Relations Client
 13725 MARIIGNANE CEDEX
 France

NOTE 1

On the purchase order, please specify the mode of transport, the destination and the serial numbers of the helicopters to be modified.

NOTE 2

*For ALERT SERVICE BULLETINS, order by:
 Telex: HELICOP 410 969F
 Fax: +33 (0)4.42.85.99.96.*

2.B. LOGISTIC INFORMATION

Not applicable.

2.C. EQUIPMENT OR PARTS REQUIRED PER HELICOPTER/COMPONENT

Equipment or parts to be ordered separately:

Key Word	Qty	New P/N	Item	Former P/N →	Instruction
Elastomer pitch change rod	A/R	LB6-1231-3-1 (704A47137035)	1	/	/
Elastomer pitch change rod	A/R	LB6-1231-3-2 (704A47137036)	2	/	/

Consumables to be ordered separately:

As per Work Cards and Tasks mentioned in this ALERT SERVICE BULLETIN.

The consumables can be ordered separately from the INTERTURBINE AVIATION LOGISTICS company:

Website: <http://www.interturbine.com>
 Telephone: +49.41.91.809.300
 AOG: +49.41.91.809.444

2.D. EQUIPMENT OR PARTS TO BE RETURNED

Not applicable.

3. ACCOMPLISHMENT INSTRUCTIONS

3.A. GENERAL

Not applicable.

3.B. WORK STEPS

3.B.1. Visual check of the TRH pitch change rod (Figure 1)

- Manually induce a flapping movement in the tail rotor blade in order to make pitch change rod (a) rotate by more than 10°.
- Visually check both faces of the ball joint (only on the blade side) to ensure that the elastomer (Area A, Figure 1) is not damaged (debonding, extrusion or crack):
 - . If no damage is found, or if damage is found on one or several layers with a circumference of less than 90°, leave as is.
 - . If damage is found on one or several layers, with a circumference of more than 90°, replace pitch change rod (a) with a pitch change rod (1) or (2) as per AMM Task 65-21-00, 4-12b or MET Work Card 64.10.00.402.

3.C. COMPLIANCE CONFIRMATION

Compliance with this document:

Record first compliance with this ALERT SERVICE BULLETIN, with the revision number, on the Log Card of the Tail Gearbox.

3.D. OPERATING AND MAINTENANCE INSTRUCTIONS

Not applicable.

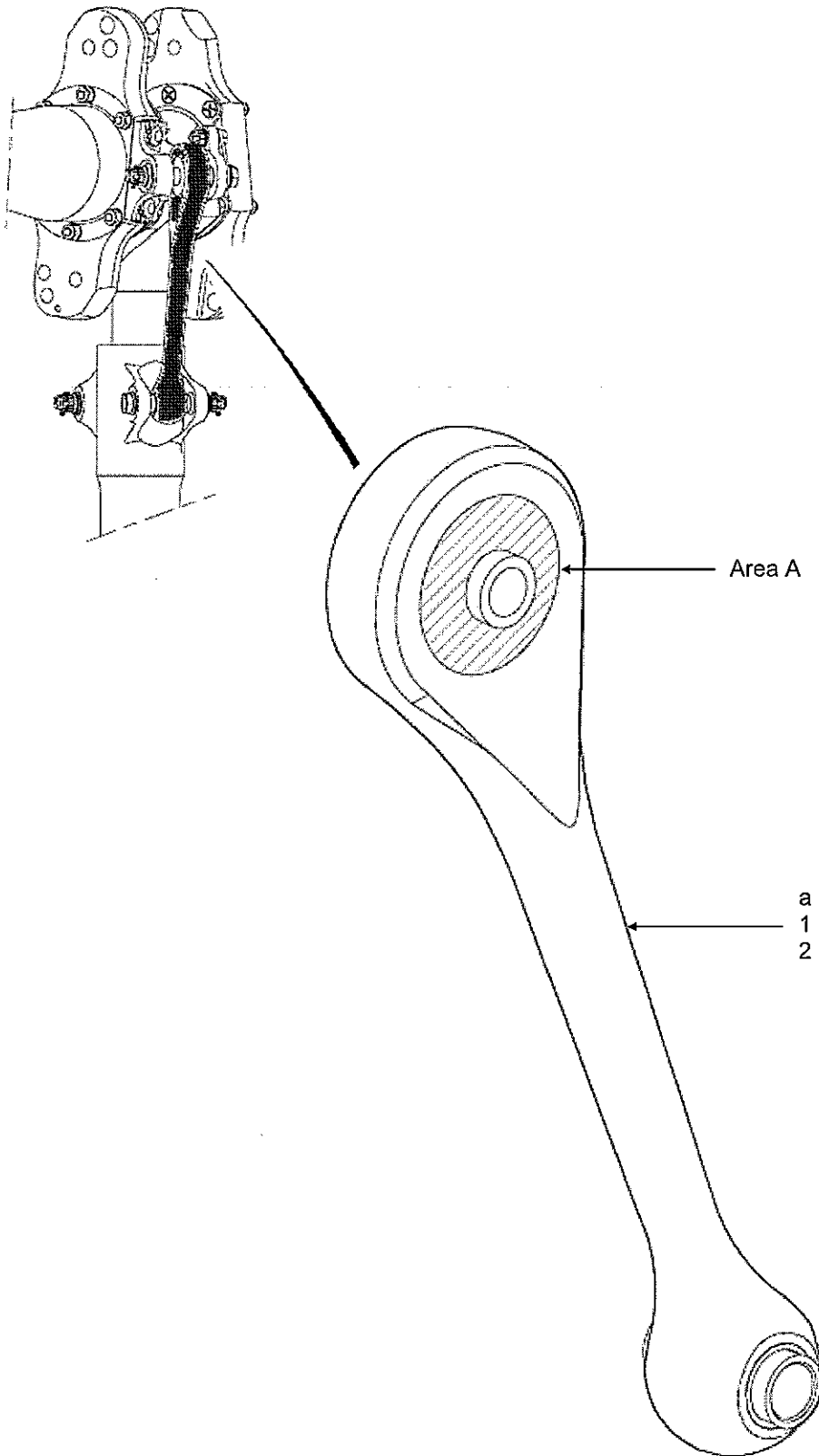


Figure 1: TRH pitch change rod POST MOD 076602 or 075601



MALAYSIA
JABATAN PENERBANGAN AWAM
DEPARTMENT OF CIVIL AVIATION

PERMIT UNTUK TERBANG <i>PERMIT TO FLY</i>		Nombor Sijil <i>Certificate Number</i> PTF/2016/044
Kenegaraan Dan Tanda Pendaftaran <i>Nationality And Registration Mark</i>	Pembuat dan Nama Kapal Udara diberi oleh Pembuat <i>Aircraft Manufacturer/Type Designation</i>	Nombor Siri Kapal Udara <i>Aircraft Serial Number</i>
9M-LKE	Airbus Helicopter AS350	8297

Tujuan Penerbangan <i>Purpose of flight</i>	Flying the aircraft for DCA acceptance
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Permit untuk terbang ini dikeluarkan menurut peraturan 29, Peraturan-peraturan Penerbangan Awam (CAR) 2016, dan mengesahkan bahawa kapal udara berkenaan berupaya untuk melakukan penerbangan dengan selamat untuk tujuan penerbangan tersebut dan dalam lingkungan syarat-syarat penerbangan yang telah ditetapkan.



Permit untuk terbang ini hanya memberi kuasa untuk penerbangan dalam Malaysia sahaja. Penerbangan merentasi Negara lain memerlukan kebenaran daripada pihak berkuasa penerbangan awam Negara tersebut dan adalah menjadi tanggungjawab pemilik/operator kapal udara untuk mendapatkan kebenaran tersebut.

This permit to fly is issued pursuant to regulation 29 of the Civil Aviation Regulations (CAR) 2016, and certifies that the aircraft is capable of safe flight for the purpose stated above and within the prescribed flight conditions.

This permit to fly only authorises flights within Malaysia. Flights over other countries require permission from the civil aviation authority of that country and it shall be the responsibility of the aircraft owner/operator to obtain such permission.

Permit untuk terbang ini sah untuk tempoh <i>This permit to fly is valid for the period</i>	Dari <i>From</i>	23 Dec 2016	Hingga <i>To</i>	22 Feb 2017
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Syarat-syarat penerbangan yang dirujuk adalah merupakan sebahagian daripada permit untuk terbang ini <i>The referenced flight conditions shall form part of this permit to fly</i>	Flight conditions approval ref. no: FC/2016/036 dated 23 December 2016
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Tarikh dikeluarkan: 23 December 2016 <i>Date of issue</i>	 HARIS FAZILLAH BIN CHEE b/p Ketua Pengarah Penerbangan Awam for Director General of Civil Aviation	
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JABATAN PENERBANGAN AWAM MALAYSIA
(DEPARTMENT OF CIVIL AVIATION MALAYSIA)
ARAS 1-4, BLOK PODIUM
NO. 27, PERSIARAN PERDANA, PRESINT 4
PUSAT PENTADBIRAN KERAJAAN PERSEKUTUAN
62618 PUTRAJAYA
MALAYSIA

Telefon : 6-03-88714000
Faks : 6-03-88901640
AFTN : WMKKYAYX
Teleks : PENAMA MA 30128
Kawat : CIVIL KUALA LUMPUR
Web : http://www.dca.gov.my





PERMIT TO FLY FLIGHT CONDITIONS

Flight conditions approval ref. no.	FC/2016/036
Aircraft nationality and registration mark(s)	9M-LKE
Aircraft manufacturer	Airbus Helicopter
Aircraft type/model	AS350
Aircraft serial number(s)	8297

The aircraft identified above shall be operated in accordance with the flight conditions/restrictions prescribed below:

1. Aircraft shall not fly for the purpose of commercial air transport operations.
2. No flight over congested or densely populated areas, except for take-off and landing.
3. Only minimum flight crew and required technical personnel on board.
4. Flight crew must have the appropriate licence and must be familiar with aircraft configuration and special operational procedures required under these flight conditions.
5. Flight shall be conducted in daylight under Visual Flight Rules (VFR) conditions.
6. Aircraft shall be maintained in accordance with specific continuing airworthiness arrangement including maintenance instructions and regime under which they will be performed.
7. The aircraft maintenance programme and related manuals remain applicable.
8. The flight test shall be conducted in accordance with **DCA Approved Airworthiness Flight Test Schedule ref no: AHM/AFTS/AS350B3e Rev 1 Dated 15 Sep 2015**
9. The basic Flight Manual and the relevant Supplements remain applicable.
10. The permit to fly and the prescribed flight conditions shall be carried on board and displayed in the aircraft in accordance with Airworthiness Notice 8305.

The flight conditions approval remains valid provided the declared configuration is applicable, the aircraft is maintained in accordance with defined instructions, and compliance with airworthiness directives is observed.

Date of issue: 23 December 2016	 HARIS FAZILLAH BIN CHEE Airworthiness Sector for Director General of Civil Aviation	
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MALAYSIA
JABATAN PENERBANGAN AWAM
DEPARTMENT OF CIVIL AVIATION

PERMIT UNTUK TERBANG <i>PERMIT TO FLY</i>		Nombor Sijil <i>Certificate Number</i> PTF/2016/043
Kenegaraan Dan Tanda Pendaftaran <i>Nationality And Registration Mark</i>	Pembuat dan Nama Kapal Udara diberi oleh Pembuat <i>Aircraft Manufacturer/Type Designation</i>	Nombor Siri Kapal Udara <i>Aircraft Serial Number</i>
9M-LKE	Airbus Helicopter AS350	8297

Tujuan Penerbangan <i>Purpose of flight</i>	To carry out Inspection Flight Test.
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Permit untuk terbang ini dikeluarkan menurut peraturan 29, Peraturan-peraturan Penerbangan Awam (CAR) 2016, dan mengesahkan bahawa kapal udara berkenaan berupaya untuk melakukan penerbangan dengan selamat untuk tujuan penerbangan tersebut dan dalam lingkungan syarat-syarat penerbangan yang telah ditetapkan.


Permit untuk terbang ini hanya memberi kuasa untuk penerbangan dalam Malaysia sahaja. Penerbangan merentasi Negara lain memerlukan kebenaran daripada pihak berkuasa penerbangan awam Negara tersebut dan adalah menjadi tanggungjawab pemilik/operator kapal udara untuk mendapatkan kebenaran tersebut.

This permit to fly is issued pursuant to regulation 29 of the Civil Aviation Regulations (CAR) 2016, and certifies that the aircraft is capable of safe flight for the purpose stated above and within the prescribed flight conditions.

This permit to fly only authorises flights within Malaysia. Flights over other countries require permission from the civil aviation authority of that country and it shall be the responsibility of the aircraft owner/operator to obtain such permission.

Permit untuk terbang ini sah untuk tempoh <i>This permit to fly is valid for the period</i>	Dari <i>From</i>	23 Dec 2016	Hingga <i>To</i>	22 Feb 2017
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Syarat-syarat penerbangan yang dirujuk adalah merupakan sebahagian daripada permit untuk terbang ini <i>The referenced flight conditions shall form part of this permit to fly</i>	Flight conditions approval ref. no: FC/2016/035 dated 23 December 2016
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Tarikh dikeluarkan: 23 December 2016 <i>Date of issue</i>	 HARIS FAZILLAH BIN CHEE b/p Ketua Pengarah Penerbangan Awam <i>for Director General of Civil Aviation</i>
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JABATAN PENERBANGAN AWAM MALAYSIA
(DEPARTMENT OF CIVIL AVIATION MALAYSIA)
ARAS 1-4, BLOK PODIUM
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AFTN : WMKKYAYX
Teleks : PENAMA MA 30128
Kawat : CIVIL KUALA LUMPUR
Web : http://www.dca.gov.my





PERMIT TO FLY FLIGHT CONDITIONS

Flight conditions approval ref. no.	FC/2016/035
Aircraft nationality and registration mark(s)	9M-LKE
Aircraft manufacturer	Airbus Helicopter
Aircraft type/model	AS350
Aircraft serial number(s)	8297

The aircraft identified above shall be operated in accordance with the flight conditions/restrictions prescribed below:

1. Aircraft shall not fly for the purpose of commercial air transport operations.
2. No flight over congested or densely populated areas, except for take-off and landing.
3. Only minimum flight crew and required technical personnel on board.
4. Flight crew must have the appropriate licence and must be familiar with aircraft configuration and special operational procedures required under these flight conditions.
5. Flight shall be conducted in daylight under Visual Flight Rules (VFR) conditions.
6. Aircraft shall be maintained in accordance with specific continuing airworthiness arrangement including maintenance instructions and regime under which they will be performed.
7. The aircraft maintenance programme and related manuals remain applicable.
8. The flight test shall be conducted in accordance with **DCA Approved Maintenance Flight Test Schedule ref no: MFTS/AS350B3e Rev.0 dated 23 Dec 2016.**
9. The basic Flight Manual and the relevant Supplements remain applicable.
10. The permit to fly and the prescribed flight conditions shall be carried on board and displayed in the aircraft in accordance with Airworthiness Notice 8305.

The flight conditions approval remains valid provided the declared configuration is applicable, the aircraft is maintained in accordance with defined instructions, and compliance with airworthiness directives is observed.

Date of issue: 23 December 2016	 HARIS FAZILLAH BIN CHEE Airworthiness Sector for Director General of Civil Aviation	
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