

AIRBUS

EC 155

MASTER SERVICING MANUAL (MSM)

EC 155 B1



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IMPORTANT NOTE

The practical value of this manual depends entirely upon it being updated correctly by the operator.

The successive revisions should be recorded on the relevant page of the manual.

LOEDU

1 Update

Refer to the "Update" Document Unit

2 "Type" Column Code

N-New, to be inserted

R-Revised, to be replaced

D-To be removed from the manual

- (No code) Unchanged with respect to the previous issue

D.U.	Type	Date	D.U.	Type	Date
TITLE	N	2022.02.01	05-10-00 ATA 63	R	2022.02.01
LOEDU	N	2022.02.01	05-10-00 ATA 64	-	2019.09.16
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TECHNICAL PUBLICATIONS	-	2018.09.10	05-10-00 ATA 71	-	2012.11.08
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UPDATE OF THE MSM	R	2022.02.01	05-10-00 ATA 93	-	2017.09.11
BREAKDOWN OF THE MSM	R	2022.02.01	05-20-00 ATA 05	-	2012.11.08
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SB INCORPORATED	R	2022.02.01	05-20-00 ATA 23	-	2012.11.08
05-10-00 ATA 23	-	2018.09.10	05-20-00 ATA 24	-	2019.09.16
05-10-00 ATA 24	-	2016.09.12	05-20-00 ATA 25	R	2022.02.01
05-10-00 ATA 25	R	2022.02.01	05-20-00 ATA 26	-	2019.09.16
05-10-00 ATA 26	-	2019.09.16	05-20-00 ATA 28	R	2022.02.01
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05-10-00 ATA 29	R	2022.02.01	05-20-00 ATA 30	R	2022.02.01
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05-10-00 ATA 45	R	2022.02.01	05-20-00 ATA 45	R	2022.02.01
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			05-20-00 ATA 53	R	2022.02.01
			05-20-00 ATA 55	-	2012.11.08
			05-20-00 ATA 56	-	2016.01.21

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05-20-00 ATA 60	-	2018.09.10	05-20-00 ATA 79	-	2019.09.16
05-20-00 ATA 62	-	2016.01.21	05-20-00 ATA 80	R	2022.02.01
05-20-00 ATA 63	R	2022.02.01	05-20-00 ATA 88	-	2012.11.08
05-20-00 ATA 64	-	2017.09.11	05-20-00 ATA 93	-	2017.09.11
05-20-00 ATA 65	-	2014.05.29	05-50-00 ATA 53	-	2013.09.12
05-20-00 ATA 67	R	2022.02.01	05-50-00 ATA 62	-	2012.11.08
05-20-00 ATA 71	-	2014.05.29	05-50-00 ATA 63	R	2022.02.01
05-20-00 ATA 73	-	2012.11.08	05-50-00 ATA 64	-	2021.01.22
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SB INCORPORATED

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HIGHLIGHTS OF THE REVISION

1 GENERAL

- Check that the content of the sections is in accordance with the List Of Effective Document Units (LOEDU).
- Return the acknowledgment card.

2 OUTLINE OF THE REVISION

The revision is codified as follows:

- Revision 011 2022.02.01

3 DETAILED DESCRIPTION

3.1 Section INTRO

- GLOSSARY modified
- UPDATE OF THE MSM Paragraph 5 modified
- BREAKDOWN OF THE MSM Paragraph 2 modified
- USE OF THE MSM Paragraph 5.2.2 modified

3.2 Section SB INCORPORATED

- SB INCORPORATED Modified

3.3 Section 05.10.00

25-63 OPT - ELECTRIC HOIST

- 25/63/02/403/000/100 10 Y // 1000 HC
Electric hoist 90m
Description modified

25-66 EMERGENCY LOCATOR TRANSMITTER

- 25/66/03/000/000/000 ED
Underwater locator beacon
Title modified
- 25/66/03/000/000/005 ED
Underwater locator beacon - Battery kit
Title modified
- 25/66/04/000/000/000 ED
Underwater locator beacon
Title modified
- 25/66/04/000/000/010 ED
Underwater locator beacon - Battery kit
Title and Description modified
- 25/66/04/000/000/020 ED
Underwater locator beacon - Battery kit
Title modified
- 25/66/18/510/000/000 ED
Underwater locator beacon - Battery kit
New task integrated

-	25/66/19/510/000/100 Underwater locator beacon - Battery kit New task integrated	ED
-	25/66/19/510/000/110 Underwater locator beacon New task integrated	ED
29-10 MAIN HYDRAULIC SYSTEM		
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-	45/11/01/510/000/100 SSCVFDR recorder New task integrated	OC
-	45/11/04/510/000/000 SSCVFDR recorder Title and Description modified	OC
63-10 ENGINE/MGB COUPLING		
-	63/10/00/510/000/000 Flexible coupling - Inputs New task integrated	OC
-	63/10/00/510/000/010 MGB - Tachymeter generator New task integrated	OC
-	63/10/00/510/000/020 Rotor brake New task integrated	OC
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-	63/20/00/000/000/138 MGB - Bearings New task integrated	3500 FH
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-	63/20/00/000/000/165 MGB - Tail rotor output pinion assembly New task integrated	1000 FH // 11 Y
-	63/20/00/000/000/170 MGB - Epicyclic gear train New task integrated	1000 FH // 11 Y
-	63/20/00/000/000/185 MGB - Dual lubrication pump assembly New task integrated	1000 FH // 11 Y
-	63/20/00/000/000/190 MGB - Fan output assembly New task integrated	1000 FH // 11 Y
-	63/20/00/000/000/200 MGB plate New task integrated	3840 FH // 19200 TC
-	63/20/00/000/000/205 Input bevel gear New task integrated	11529 FH
-	63/20/00/000/000/210 Double bevel gear New task integrated	40000 FH

- 63/20/00/000/000/215 6918 FH
Freewheel shaft
New task integrated
 - 63/20/00/000/000/220 115600 TC
Engine flange
New task integrated
 - 63/20/00/000/000/240 100 *
Rotor brake - Pads
New task integrated
- 80-10 CRANKING**
- 80/10/00/000/000/000 1200 FH // 5 Y
Starter generator
Description and limit modified

3.4 Section 05.20.00

25-66 EMERGENCY LOCATOR TRANSMITTER

- 25/66/03/000/000/010 1 Y
Underwater locator beacon
Title modified
- 25/66/03/000/000/100 2 Y
Underwater locator beacon
Title modified
- 25/66/04/000/000/005 1 Y
Underwater locator beacon
Title and Description modified
- 25/66/04/000/000/015 1 Y
Underwater locator beacon
Title and Description modified
- 25/66/04/000/000/025 1 Y
Underwater locator beacon - Battery kit
Title and Description modified
- 25/66/04/000/000/100 2 Y
Underwater locator beacon
Title and Description modified
- 25/66/04/000/000/200 2 Y
Underwater locator beacon
Title and Description modified
- 25/66/04/520/000/030 2 Y
Underwater locator beacon
New task integrated
- 25/66/18/520/000/000 2 Y
Underwater locator beacon
New task integrated
- 25/66/19/520/000/100 ED
Underwater locator beacon
New task integrated

28-20 FUEL DISTRIBUTION

-	28/20/00/211/000/010 Transfer pump Description modified Documentation modified	2400 FH
29-11 RIGHT MAIN HYDRAULIC SYSTEM		
-	29/11/00/612/000/000 Right hydraulic system Limit modified	2400 FH // 4 Y
29-12 LEFT MAIN HYDRAULIC SYSTEM		
-	29/12/00/612/000/000 Left hydraulic system Limit modified	2400 FH // 4 Y
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-	30/42/00/000/000/010 Windshield wiper blades Task deleted	1 Y
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-	33/50/03/000/000/050 HEEL system - Battery New task integrated	(*)
-	33/50/03/000/000/020 HEEL cabin roof - Battery Documentation modified	18 M
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45-11 OPT - CVFDR		
-	45/11/01/000/000/000 CVR installation Task deleted	6 M
-	45/11/01/000/000/010 FDR and CVR data control Task deleted	1 Y
-	45/11/01/000/000/015 Discrete signals Task deleted	1 Y
-	45/11/01/520/000/000 SSCVFDR recorder - Locator beacon New task integrated	2 Y
-	45/11/01/520/721/000 CVR recorder New task integrated	6 M
-	45/11/01/520/722/000 CVR recorder New task integrated	6 M
-	45/11/01/520/723/000 Immersion probe New task integrated	600 FH // 2 Y

-	45/11/01/520/724/000 CVR recorder New task integrated	1 Y
-	45/11/01/520/724/010 FDR recorder New task integrated	1 Y
-	45/11/01/723/000/000 Immersion probe Task deleted	600 FH // 2 Y
-	45/11/04/000/000/100 CVFDR recorder Task deleted	6 M
-	45/11/04/000/000/105 CVFDR recorder Task deleted	6 M
-	45/11/04/000/000/110 CVFDR recorder Task deleted	1 Y
-	45/11/04/000/000/115 CVFDR recorder Task deleted	1 Y
-	45/11/04/000/000/130 RIPS Task deleted	1 Y
-	45/11/04/000/000/140 Immersion probe Task deleted	600 FH // 2 Y
-	45/11/04/520/000/100 CVFDR recorder - Emergency locator transmitter Task deleted	2 Y
-	45/11/04/520/000/105 CVFDR recorder Task deleted	1 Y
-	45/11/04/520/000/200 SSCVFDR recorder - Locator beacon New task integrated	2 Y
-	45/11/04/520/724/200 CVR recorder New task integrated	6 M
-	45/11/04/520/725/200 SSCVFDR recorder New task integrated	600 FH // 2 Y
-	45/11/04/520/726/200 CVR recorder New task integrated	6 M
-	45/11/04/520/726/205 SSCVFDR recorder New task integrated	2 Y
-	45/11/04/520/726/210 SSCVFDR recorder New task integrated	2 Y

-	45/11/04/520/727/200 CVR recorder New task integrated	1 Y
-	45/11/04/520/727/210 FDR recorder New task integrated	1 Y
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-	53/60/00/520/226/005 Baggage compartment - External skin New task integrated	600 FH
63-10 ENGINE/MGB COUPLING		
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-	63/10/02/221/002/005 Flexible coupling attachment bolt on MGB side New task integrated	5 FH
-	63/10/62/221/000/040 Flexible coupling - Inputs and tail output New task integrated	5 FH
-	63/10/00/213/000/000 Flexible coupling - Inputs New task integrated	10 FH // 5 D
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-	63/10/00/221/000/010 Engine-MGB coupling Task deleted	600 FH
-	63/10/00/221/000/015 Engine-MGB coupling New task integrated	600 FH
-	63/10/00/221/000/020 Engine-MGB coupling New task integrated	600 FH
-	63/10/02/221/000/010 Flexible coupling Task deleted	600 FH
-	63/10/02/221/000/015 Flexible coupling New task integrated	600 FH

-	63/10/62/221/000/000 Flexible coupling - Inputs and tail output New task integrated	600 FH // 4 Y
-	63/10/62/221/000/010 Flexible coupling - Inputs and tail output New task integrated	600 FH // 2 Y
-	63/10/62/221/000/020 Flexible coupling - Inputs and tail output New task integrated	600 FH // 4 Y
-	63/10/62/221/000/030 Flexible coupling - Inputs and tail output New task integrated	600 FH // 2 Y
-	63/10/64/222/000/000 Engine/MGB coupling - Gimbals pins New task integrated	100 FH
63-20 MAIN GEARBOX		
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-	63/20/00/611/000/001 MGB New task integrated	45 FH
-	63/20/00/611/000/005 MGB New task integrated	50 FH
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-	63/20/60/216/000/000 MGB New task integrated	BFF
-	63/20/60/722/000/100 MGB - Actuator attachment fitting axis New task integrated	5 FH
-	63/20/60/722/000/110 MGB - Actuator attachment fitting axis New task integrated	75 FH
-	63/20/00/211/000/000 MGB Task deleted	100 FH // 1 Y
-	63/20/00/211/000/005 MGB New task integrated	100 FH // 1 Y
-	63/20/00/221/000/010 MGB Task deleted	600 FH
-	63/20/00/221/000/015 MGB New task integrated	600 FH

-	63/20/00/221/000/020 Attachment bolt MGB / servo-control Task deleted	4 Y
-	63/20/00/221/000/025 Attachment bolt MGB / servo-control New task integrated	4 Y
-	63/20/00/611/000/010 MGB - Mineral oil Task deleted	400 FH // 2 Y
-	63/20/00/611/000/015 MGB - Mineral oil New task integrated	400 FH // 2 Y
-	63/20/60/211/000/000 MGB New task integrated	100 FH
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-	63/20/60/211/000/090 MGB New task integrated	600 FH
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-	63/20/60/215/000/020 MGB New task integrated	600 FH
-	63/20/60/215/000/030 MGB New task integrated	600 FH
-	63/20/60/215/000/060 MGB New task integrated	600 FH
-	63/20/60/215/000/070 MGB New task integrated	600 FH
-	63/20/60/219/000/000 MGB New task integrated	10 FH // 5 D
-	63/20/60/219/000/020 MGB New task integrated	ALF
-	63/20/60/219/000/030 MGB - Plate New task integrated	5 FH

-	63/20/60/219/000/040 MGB - Main rotor actuator attachment New task integrated	ALF
-	63/20/60/219/000/050 MGB - Fan New task integrated	ALF
-	63/20/60/221/000/010 MGB - Input flange New task integrated	600 FH
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-	63/20/60/221/000/050 MGB - Tail rotor output pinion assembly New task integrated	600 FH
-	63/20/60/221/000/080 MGB New task integrated	600 FH
-	63/20/60/282/000/000 MGB New task integrated	100 FH // 3 M
-	63/20/60/611/000/000 MGB - Mineral oil New task integrated	460 FH // 2 Y
-	63/20/61/721/000/000 MGB - Rear output flange New task integrated	600 FH
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-	63/21/02/221/000/006 MGB New task integrated	3 FH
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-	63/21/02/221/000/001 Oil filter New task integrated	100 FH // 1 Y
-	63/21/62/221/000/000 MGB - Oil filter New task integrated	600 FH // 1 Y
63-22 MGB - Fan		
-	63/22/60/221/000/000 MGB - Fan New task integrated	1200 FH
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-	63/22/60/221/000/020 MGB - Fan New task integrated	1200 FH
63-23 ROTOR BRAKE		
-	63/23/00/000/000/000 Rotor brake Task deleted	ALF
-	63/23/00/000/000/005 Rotor brake New task integrated	ALF
-	63/23/00/211/000/000 Rotor brake Task deleted	100 FH // 1 Y
-	63/23/00/211/000/005 Rotor brake New task integrated	100 FH // 1 Y
-	63/23/00/221/000/000 Rotor brake Task deleted	600 FH
-	63/23/00/221/000/001 Rotor brake Task deleted	600 FH // 2 Y
-	63/23/00/221/000/005 Rotor brake New task integrated	600 FH
-	63/23/00/221/000/006 Rotor brake New task integrated	600 FH // 2 Y
-	63/23/60/221/000/000 Rotor brake New task integrated	600 FH
-	63/23/60/221/000/010 Rotor brake New task integrated	600 FH
63-24 TAIL ROTOR DRIVE POWER TAKE-OFF		
-	63/24/01/221/001/000 MGB tail drive-shaft flange Task deleted	100 FH
-	63/24/01/221/001/001 MGB tail drive-shaft flange New task integrated	100 FH
-	63/24/01/221/001/005 MGB tail drive-shaft flange Task deleted	100 FH
-	63/24/01/221/001/010 MGB tail drive-shaft flange New task integrated	100 FH
-	63/24/01/221/000/000 MGB tail drive-shaft flange Task deleted	300 FH

-	63/24/01/221/000/001 MGB tail drive-shaft flange New task integrated	300 FH
-	63/24/01/221/000/005 MGB tail drive-shaft flange Task deleted	300 FH
-	63/24/01/221/000/006 MGB tail drive-shaft flange New task integrated	300 FH
-	63/24/01/221/000/010 MGB tail drive-shaft flange Task deleted	600 FH
-	63/24/01/221/000/015 MGB tail drive-shaft flange New task integrated	600 FH
63-40 INDICATING		
-	63/40/02/721/000/000 MGB - Electrical magnetic plug Task deleted	600 FH
-	63/40/02/721/000/005 MGB - Electrical magnetic plug New task integrated	600 FH
-	63/40/03/721/000/000 Oil temperature warning system Task deleted	6 Y
-	63/40/03/721/000/005 Oil temperature warning system New task integrated	6 Y
-	63/40/61/221/000/000 MGB - Tachymeter generator New task integrated	600 FH
-	63/40/61/221/000/010 MGB - Tachymeter generator New task integrated	600 FH
67-10 MAIN ROTOR CONTROL		
-	67/10/00/221/000/025 Main rotor controls MOD Applicability modified	12 Y
80-10 CRANKING		
-	80/10/01/221/002/000 Starter generator Task modified	600 FH
-	80/10/00/211/000/000 Starter generator Limit modified	600 FH
-	80/10/00/221/000/000 Starter generator Limit modified	600 FH

3.5 Section 05.50.00

63-20 MAIN GEARBOX

- 63/20/00/000/000/017
MGB
New task integrated
- 63/20/00/000/000/025
MGB
New task integrated
- 63/20/00/000/000/035
MGB
New task integrated
- 63/20/00/000/000/040
MGB
New task integrated
- 63/20/00/000/000/065
MGB
New task integrated
- 63/20/00/000/000/070
MGB
New task integrated
- 63/20/00/000/000/080
MGB
New task integrated
- 63/20/00/000/000/090
MGB - Chip detector
New task integrated
- 63/20/00/000/000/100
MGB
New task integrated
- 63/20/00/000/000/110
MGB
New task integrated
- 63/20/00/000/000/120
MGB
New task integrated
- 63/20/00/000/000/130
MGB
New task integrated

63-40 INDICATING

- 63/40/02/211/000/005
MGB - Electrical magnetic plug
Task deleted
- 63/40/02/211/000/006
MGB - Electrical magnetic plug
New task integrated

- 63/40/02/211/000/010
MGB - Electrical magnetic plug
Task deleted
- 63/40/02/211/000/015
MGB - Electrical magnetic plug
New task integrated

End of the Document Unit

TECHNICAL PUBLICATIONS

1 THE DOCUMENTATION

The technical publications intended for the operators is mainly broken down into five groups.

2 OPERATING DOCUMENTS

- **The Master Servicing Manual (MSM)** is specifically intended for the persons in charge of maintenance. It defines the operating time limits and the maintenance intervals.
- **The Airworthiness Limitations Section (ALS)** is intended for the persons in charge of maintenance. It defines the mandatory limitations and checks.
- **The Flight Manual (FLM)** is intended for flight crews. It defines the limitations, the normal and emergency procedures, and the performance of the aircraft.
- **The Master Minimum Equipment List (MMEL)** defines the minimum list of equipment required to carry out flights.
- **The weight and balance log (PMC).**

3 MAINTENANCE DOCUMENTS

- **The Aircraft Maintenance Manual (AMM)** describes the maintenance procedures necessary to carry out the maintenance of the helicopter.
- **The Systems Description Section (SDS)** is Part 1 of the AMM and covers the operation of the systems.
- **The Structural Repair Manual (SRM)** allows operators to identify and protect the materials and to repair or replace structural components, mechanical assemblies or blades.
- **The Standard Practices Manual (MTC)** contains all the general information and know-how for the preparation, operation, maintenance, checks, standard repairs, technical and safety instructions as well as storage conditions.
- **The Wiring Diagrams Manual (WDM)** contains all the wiring diagrams of the aircraft.
- **The Index of Modifications (SIM)** gives a list of the applicable modifications with the corresponding SBs (if any).

4 IDENTIFICATION DOCUMENTS

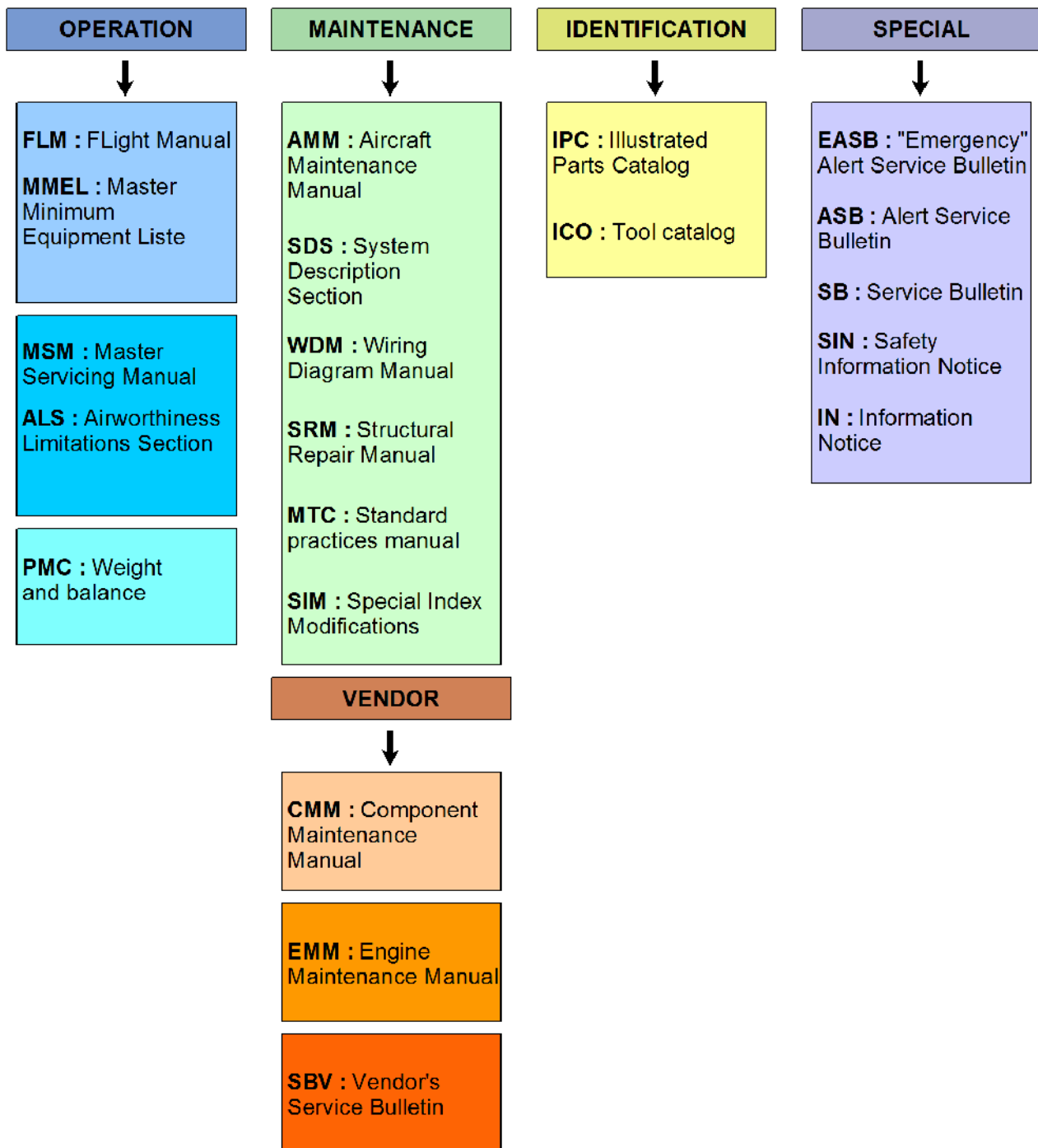
- **The Illustrated Parts Catalogue (IPC)** details all the spare parts necessary to maintain the helicopter.
- **The Illustrated Tools Catalogue (ICO)** details the tools necessary to maintain the helicopter.

5 SPECIAL DOCUMENTS

- **The Service Bulletins (SB), Alert Service Bulletins (ASB) and Emergency Alert Service Bulletins (EASB)** The purpose of these documents is to implement, with the operators, the directives defined further to a new event (incident, modification, etc.). They complete the aircraft publications into which they are incorporated through updates if necessary.
- **The Information Notices (IN), Safety Information Notices (SIN)** are information documents. Their purpose is to remind the existing maintenance procedures or to give information on changes validated by Airbus and in the process of being implemented.

6 VENDOR'S DOCUMENTS

- These publications include the following equipment manufacturer manuals : **the Vendor service bulletins (SBV), the Engine Maintenance Manual (EMM) and the Component Maintenance Manual (CMM).**



End of the Document Unit

GLOSSARY

(-) :	Not indicated
(P) :	Provisional limitation
& :	And
// :	Or
§:	Paragraph
A :	Basic technical inspection with a calendar limit
ALL MP/N :	All Manufacturer Part Numbers
ALF :	Check After the Last Flight of the day
ALS :	Airworthiness Limitations Section
ATA	Air Transport Association
BFF :	Check Before the First Flight of the day
CHK :	CHECK
D :	Day
DU:	Document Unit
EASA :	European Aviation Safety Agency
ED :	Expiry Date
EXC :	EXCEPT
FH :	Flight Hour
FLM :	FLIGHT MANUAL
FM :	Log card
GRT :	Group for staggered basic inspection
HC :	Hoist Cycle
HPC :	Harpoon Cycle
HT :	Hard Time maintenance
LOEDU :	List Of Effective Document Units
LC :	Landing Cycle
M :	Month
MGB :	Main Gear Box
MP/N :	Manufacturer Part/Number
MRH :	Main Rotor Head
MSM :	Master Servicing Manual (Maintenance Program)
OC :	On Condition
OPT :	OPTION
OTL :	Operating Time Limit
PN :	Part Number
PO :	Perform Once
POST :	After (POST MOD, POST SB...)
PRE :	Before (PRE MOD, PRE SB...)
RC :	Roping Cycle
SB :	Service Bulletin
SC :	Sling Cycle

SLL :	Service Life Limit
S/N :	Serial Number
T :	Basic Technical inspection with an hour-based time limit
TA :	Turn Around check
TBO :	Time Between Overhaul
TC :	Torque Cycle
TGB:	Tail Gear Box
TL :	Type Limit
TRH :	Tail Rotor Head
TSI :	Time Since Installation
TSM :	Time Since Manufacture
WC :	Work Card
Y :	Year

End of the Document Unit

STRUCTURE OF THE MSM

1 GENERAL STRUCTURE OF THE MSM

The Master Servicing Manual (MSM) is broken down into two parts.

The first part comprises general information which explains how the MSM operates and how it must be used.

The second part which defines the scheduled maintenance comprises the following sections:

- 05-10-00 Intervals and operating time limits

This section lists the components subject to an Operating Time Limit (OTL) or to a limited Time Between Overhauls (TBO) together with their corresponding limits.

This section also lists the components subject to On-Condition maintenance (OC).

- 05-20-00 Maintenance operations

This section lists all the maintenance operations to be carried out on the helicopter regardless of the operating conditions.

- 05-50-00 Special monitoring

This section contains the maintenance operations which must be carried out further to incidents or events.

These sections specify the maintenance operations which must be carried out on the helicopter, periodically or a limited number of times, whatever the helicopter operating conditions.

2 BREAKDOWN OF THE MSM

The various parts of the MSM are presented in the form of Document Units (DUs).

2.1 Identification of the DU

Each DU is unique and is identified by the following :

- the title of the manual at the top right-hand side of the page,
- the effectivity (helicopter type and version) marked at the bottom left-hand side of the page,
- the date code of the DU at the bottom left-hand side of the page below the effectivity (format: year.month.day),
- the number of the section or the title of the DU at the bottom right-hand side of the page,
- the page number at the bottom right-hand side of the page,
- the ATA number at the bottom center of the page only for Sections 05-10-00, 05-20-00 and 05-50-00.

The end of each DU is identified by the words "End of the Document Unit" on the last page.

2.2 Task codes

The DU in Sections 05-10-00, 05-20-00 and 05-50-00 comprise maintenance tasks. These tasks are identified by a unique code specific to Airbus, comprising a group of 15 characters.

Example :

62/20/00/000/000/180				
Rotor blade pin				
	AMM	62.10.00.061	300 FH	30 FH
		62.10.00.421.001	// 1 Y	36 D
Greasing.				

The first four digits (62/20/00/000/000/180) correspond to the ATA chapter/section breakdown.

The fifth and sixth digits (62/20/00/000/000/180) are used to subdivide an ATA chapter/section comprising several subjects into individual subjects.

The seventh to fifteenth digits (62/20/00/000/000/180) according to the section of the MSM are used to identify the tasks per ATA chapter/section/subject.

2.3 List Of Effective Document Units (LOEDU)

The LOEDU specifies the following information, for each DU, in a table:

- the Document Unit: this column specifies the number of the section and the ATA number or the title of the DU (example: 05-20-00 ATA 62).
- the type: this column is coded with the letter (R) if the DU is revised, (D) if the DU is deleted, (N) if the DU is new, or has no code (-) for DUs which are unchanged.
- the revision date: this column indicates the DU data code.

3 CLASSIFICATION OF INFORMATION

The DU in Sections 05-10-00, 05-20-00 and 05-50-00 are classified in increasing order according to ATA 100 numbering.

The tasks comprising the DUs in sections 05-20-00 are arranged in the following order: perform once tasks at the start of the DU, followed by periodic tasks.

End of the Document Unit

UPDATE OF THE MSM

1 REVISION OF THE MSM

The MSM is revised approximately every 6 months.

The MSM cover page bears the issue date code (first issue) and the current revision number taken in chronological order followed by the revision date code.

Example :

- issue date : 2012.01.30,
- revision 002 : 2012.03.28.

The last applicable revision is revision 002 with a date code of 2012.03.28.

The date code marked on the new or revised DU corresponds to the revision date code marked on the cover page.

2 UPDATE OF THE MSM

2.1 Update procedure

The user is responsible for updating his documentation each time he receives a revision or a new issue from the manufacturer. The MSM is updated using the new LOEDU which specifies which DU must be inserted (new DU), replaced (revised DU) or removed (deleted DU). A deleted DU is issued with the wording "DU TO BE REMOVED FROM THE MANUAL" in the revision concerned.

Once the MSM has been updated, the date code of each DU indicated at the bottom left-hand side of each page must correspond with the LOEDU.

The beginning of the MSM includes a "HIGHLIGHTS" page which explicitly summarizes the modifications made to each revision or new issue.

NOTE

After updating, the "RECORD OF REVISIONS" page at the beginning of the MSM must be completed (date, name and signature).

2.2 Identification of the modifications

Modifications are identified by a revision mark in the form of a vertical line in the right-hand margin of the page.

Revision marks are not used in the following DU :

- LOEDU,
- table of contents,
- highlights.

3 CUSTOMER COMPLEMENT (CC): ISSUED ON A GREEN BACKGROUND

The Customer Complement adds Document Units to the Basic Manual, which partially or totally modify the information it contains.

THE INFORMATION GIVEN ON THESE DOCUMENT UNITS TAKES PRECEDENCE OVER THE INFORMATION GIVEN IN THE BASIC MANUAL.

This information is specific to the customer and helicopter(s) identified at the bottom of the pages. The Customer Complement Document Units can be revised by reissue of the complete typesetting for a given customer. No Basic Manual Revision Document Unit is cancelled by a Customer Complement Revision. A supplement to the "LIST OF EFFECTIVE DOCUMENT UNITS" lists the Customer Complement Document Units.

The Basic Manual Revisions (on a white background) are completely separate from the Customer Complement Revisions (issued on a green background).

4 EFFECTIVITY

The operator must check the effectivity of the MSM used (correctly updated) for the helicopter concerned.

The effectivity (helicopter type and version) is indicated on the cover page and is repeated at the bottom of each page comprising the MSM DUs.

5 TECHNICAL QUERY

All technical queries to Airbus must be raised through the Technical Request Management tool, according to procedure MTC 20-08-05-107.

End of the Document Unit

BREAKDOWN OF THE MSM

1 PURPOSE OF THE MSM

The Master Servicing Manual (MSM) is drawn up by the helicopter manufacturer and gives all the maintenance operations to be performed by the operator.

It specifies all the maintenance operations and intervals recommended by the helicopter manufacturer and designed to ensure the continuous airworthiness, the operational availability and the performance of the helicopter (*), in all its varied operating missions.

For the maintenance operations and intervals assigned to the engine components, refer to the engine manufacturer's documentation.

(*) Helicopter (type/version) defined by Airbus.

2 CHANGES TO THE MSM

The MSM is modified by the manufacturer according to changes in helicopter definition and operational feedback.

The technical content of this document is approved under the authority of the DOA ref. EASA. 21J.700.

3 HELICOPTER MAINTENANCE

Helicopter maintenance consists in applying all the operations specified in the various sections of the MSM.

It must be performed by personnel with the required qualifications.

Maintenance includes the monitoring, servicing and reconditioning of a helicopter's components, as specified in the various manuals, taking into account the information that the manufacturer makes known to the operator via Alert Service Bulletins, Service Bulletins, Safety Information Notice, Information Notice etc.

4 TYPES OF MAINTENANCE

- Preventive maintenance

Preventive maintenance consists of all the scheduled, and generally repetitive, operations to check and maintain a given operating condition.

- Corrective maintenance

Corrective maintenance consists of all the operations that are performed after detection of a fault or failure in order to re-establish the initial operating condition.

- Remedial maintenance

Remedial maintenance consists of all the operations initiated to permanently correct a failure, and to partially or wholly eliminate the preventive and corrective maintenance.

5 DEFINITION OF THE MAINTENANCE MODES

There are three main maintenance modes :

- hard time maintenance,
- on-condition maintenance,
- condition monitoring maintenance.

The modes (defined below) differ mainly in how the replacement (or repair) of a component with a serviceable component is initiated. With the first two modes, the aim is to replace (or repair) the component before it fails, whereas in the third mode, to replace it after it fails.

5.1 Hard Time maintenance (HT)

A component subject to hard time maintenance must be removed at the latest when it reaches its specified limit. There are 3 types of hard time maintenance limits:

- **Service Life Limit (SLL)**

This is an Airworthiness Limitation.

Components which are essential for operating safety which are subject to undetectable damage due to the loads they withstand, are covered by a Service Life Limit. These components must be removed from service when the specified limit is reached.

These limits are listed in chapter 04.

- **Operating Time Limit (OTL)**

Components whose possible failure would have lesser consequences on flight safety than components with a Service Life Limit, are assigned an Operating Time Limit. These components must be removed from service when the specified limit is reached.

- **Time Between Overhauls (TBO)**

A TBO is assigned to a complete assembly. The interval corresponds to the operating time permitted before an overhaul is performed in a specialized workshop. The overhaul allows the assembly to be returned to service for a new period.

The interval corresponds to a period during which any increase in damage cannot be detected by routine maintenance (example: internal corrosion, fretting leading to the loosening of bolted assemblies, etc.).

NOTE

The time limits assigned to components guarantee flight safety with regard to the flight load fatigue alteration of components with no external deterioration. These time limits are not a commercial guarantee because a component may be removed due to wear, fretting corrosion or scoring, etc., before the expiry of the time limits.

The only warranty applicable to the helicopter and any component is the warranty included in the Sales Contract for the helicopter and the components.

5.2 On-Condition maintenance (OC)

A component subject to On-Condition maintenance must be inspected periodically to confirm:

- that there is no damage,
- or that the damage found is within the removal or maintenance criteria specified in the documentation.

In both cases, the component is kept in service until the next inspection. The component must be removed from service when it reaches the criterion for removal, or made serviceable again as per the instructions given in the documentation.

Consequently, the removal of such items cannot be programmed. It intervenes depending on the result of the inspections.

The visual inspection is aimed at the overall external appearance of the component (no distortion, failure, cracks, scratches, signs of heating or wear, etc.), which would modify its original condition.

STANDING INSTRUCTIONS RELATIVE TO CONDITION: Surface deterioration and deterioration to protection and paint on all helicopter components must be examined and treated without delay in accordance with the applicable instructions (criteria, then reworking).

5.3 Condition Monitoring (CM)

A component is subject to condition monitoring only after its failure (which has no impact on flight safety) has been detected. Such failures are detected during maintenance or in service. These components are not listed in the MSM.

NOTE

Some components can be subject to several maintenance modes, for example, an OTL during which a condition check (OC) is to be performed.

6 EFFECTIVITY - RESPONSIBILITY

The limits assigned to the component part numbers listed in the MSM are applicable to material:

- acquired directly from Airbus or through the Airbus subsidiaries or distribution network,
- purchased from the equipment vendors listed in our spare parts catalogues, either directly or through their own distribution network.

In all cases, the manufacturing source is specified in the "airworthiness document" (JAA form one, EASA Form One or equivalent).

Airbus will not carry out repairs on components which were procured other than through the Airbus distribution network.

WARNING

IT IS PROHIBITED TO REUSE PARTS, EQUIPMENT OR ASSEMBLIES, COMING FROM A HELICOPTER WHICH HAS INVOLVED IN AN ACCIDENT, WITHOUT FORMAL TECHNICAL APPROVAL FROM THE AIRBUS CUSTOMER TECHNICAL SUPPORT DEPARTMENT.

REMINDER: THE AIR ACCIDENT INVESTIGATION BOARD OF THE COUNTRY CONCERNED IS RESPONSIBLE FOR CLASSING AN EVENT AS AN ACCIDENT.

7 MAINTENANCE INTERVALS

7.1 Daily Checks

The daily checks are designed to ensure the operational availability of the helicopter. The following checks are defined:

- **15 Flight Hour Inspection (15 FH)**

The 15 FH inspection must be carried out at the latest :

- at 15 FH if the helicopter flies 15 hours in less than 7 days,

- at 7 days if the helicopter flies less than 15 hours in 7 days,
- before the flights are resumed if the helicopter is grounded for 7 days or more, in accordance with the conditions defined in the MTC and the EMM.

NOTE

On completion of a 15 FH inspection, the helicopter can fly again for a maximum period of 15 FH or 7 days.

For operational reasons, the 15 FH inspection can be brought forward in order to provide a new period of 15 FH or 7 days.

This inspection is designed to ensure the operational availability of the helicopter and it must be carried out by :

- by maintenance-qualified personnel,
- or
- a pilot or a flight mechanic qualified by the operator's maintenance organization.



**IN CASE OF DOUBT OR IF AN ANOMALY IS FOUND,
THE PILOT OR THE FLIGHT MECHANIC MUST CON-
TACT THE MAINTENANCE SUPERVISOR TO HAVE
THE MAINTENANCE OPERATIONS CARRIED OUT.**

In accordance with the criteria given in the publications, the mechanic shall decide :

- either to authorize the flights with monitoring of the faulty components. In this case, the mechanic shall inform the pilot of the criteria to be complied with to continue the flights of the day.
- or to carry out corrective actions before resuming the flights.

- **Acceptance of a Helicopter by the Pilot and/or Incorporation of a Modification on the Helicopter**

The mechanic or the maintenance manager must systematically inform the pilot in case of:

- embodiment of a design modification of the helicopter.
- any change in the maintenance intervals of the modified component.

- **Preflight check in cold weather and very cold weather**

Depending on the minimum temperatures reached, an additional schedule has been defined in the AMM.

- **Inspection of optional systems**

The detailed schedule is given in the Aircraft Maintenance Manual (AMM 05.21.00.214).

7.2 Supplementary check (S)

This supplementary check must be carried out every 100 FH without exceeding 1 year.

This check is focused on checking the condition of the components which are subject to short inspection intervals (less than that for the basic inspection).

7.3 Additional Check (F)

Some governing authorities require an inspection which complies with FAR 43-11 and to be carried out in certain circumstances (renewal of certificate of airworthiness, change of ownership, 100 FH, etc.).

- FAR or type F check (defined in AMM Work Card 05-25-00-212). The interval of the type F check is the same as that of the type S check. The operations of the type F check come as a complement to the operations of the type S check.

7.4 Basic Inspection (T or A)

The basic inspection is composed of checks related to:

- aircraft operation, to be carried out every 600 FH (type T inspection) or every 2 years (type A inspection).

The basic inspection includes :

- functional tests to monitor the behavior of components and systems,
- inspection of the condition of the components that have a direct influence on the airworthiness condition of the helicopter.

The purpose of the basic inspection is to check the general condition of the helicopter by means of detailed visual checks (for distortions, breaks, cracks, scores, corrosion, signs of overheating, wear, impacts, etc.) of the systems and equipment.

The basic inspection is supplemented:

- every 2nd time by a "2T or 2A" type inspection supplement,
- every 3rd time by a "3T or 3A" type inspection supplement,
- every 4th time by a "2T or 2A and 4T or 4A" type inspection supplement,
- and so on.

The visual examination covers the external appearance of the component and it is a way of ensuring that:

- there have been no alterations,
- or that the alterations found remain within the removal or repair criteria given in the technical publications.

NOTE

During each inspection, it is essential that the following be checked :

- no additional action has been requested recently by the manufacturer through issuing of an Alert Service Bulletin, Service Bulletin, etc.,

- no interval or limit has been modified through issuing of an Alert Service Bulletin, Service Bulletin, etc.,

- no interval or limit specified in the MSM has been reached. The limits (TBO, OTL, and SLL) specified in MSM Section 05-10-00 and ALS Section 04-10-00 must also be taken into account so that the replacement of certain components which will reach their due time before the aircraft is grounded again can be scheduled.

7.4.1 Principle basic inspections

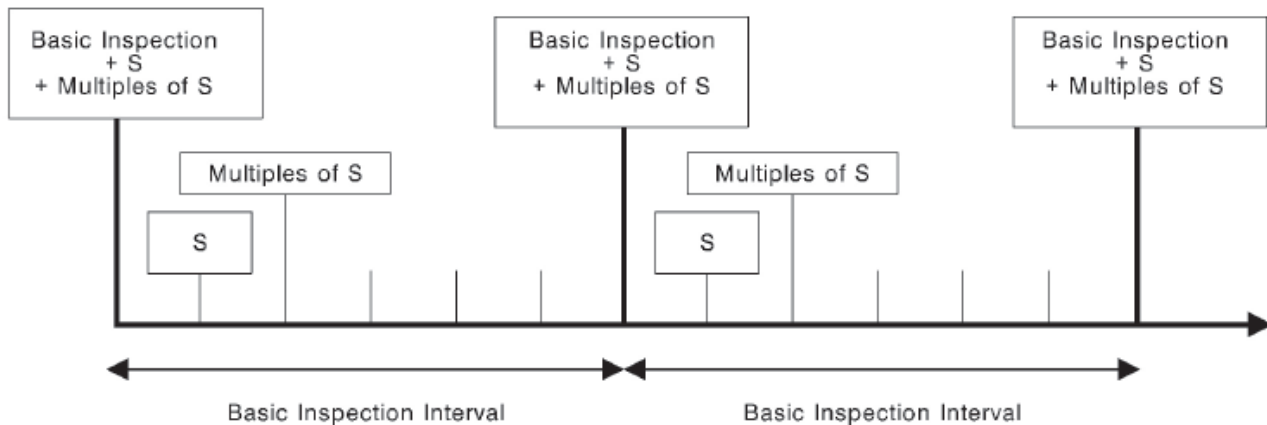
Airbus puts forward the following solutions with a view to minimizing the helicopter downtime for the basic inspection and in order to enable each operator to schedule and organize his maintenance in line with the operation of his helicopter:

- **Consolidated basic inspection:**

This option requires an extended helicopter downtime so that all the operations can be carried out.

All the operations that make up the basic inspection must be carried out by the specified due time at the latest.

Example : Consolidated basic inspection



- **Staggered basic inspection:**

If the operator wishes to replace the long downtime by a larger number of shorter downtimes, the basic inspection operations can be divided into groups. These groups of operations can be carried out at different times and staggered so that all the operations are completed by the time the basic inspection due time is reached.

Each operator can determine the number of groups he requires and define their content in keeping with his specific organization.

Airbus suggests an organization in six groups whereby the maintenance operations are optimized.

These groups have been determined according to helicopter zones and identified: GRT 01, GRT 02, GRT 03, GRT 04, GRT 05, GRT 06 in the Maintenance Program tasks in Section 05.20.00, "Documentation" column.

Group 1 (GRT 01) :

- Main Rotor Hub and Main Rotor Blades

Group 2 (GRT 02) :

- Tail Rotor Hub, Tail Rotor Blades, Tail Gearbox, Tail Boom, Horizontal Stabilizer, Outboard Fins.

Group 3 (GRT 03) :

- Main Gearbox, Landing Gear, Fuel System.

Group 4 (GRT 04) :

- Engines, Starter Generators, FADEC.

Group 5 (GRT 05) :

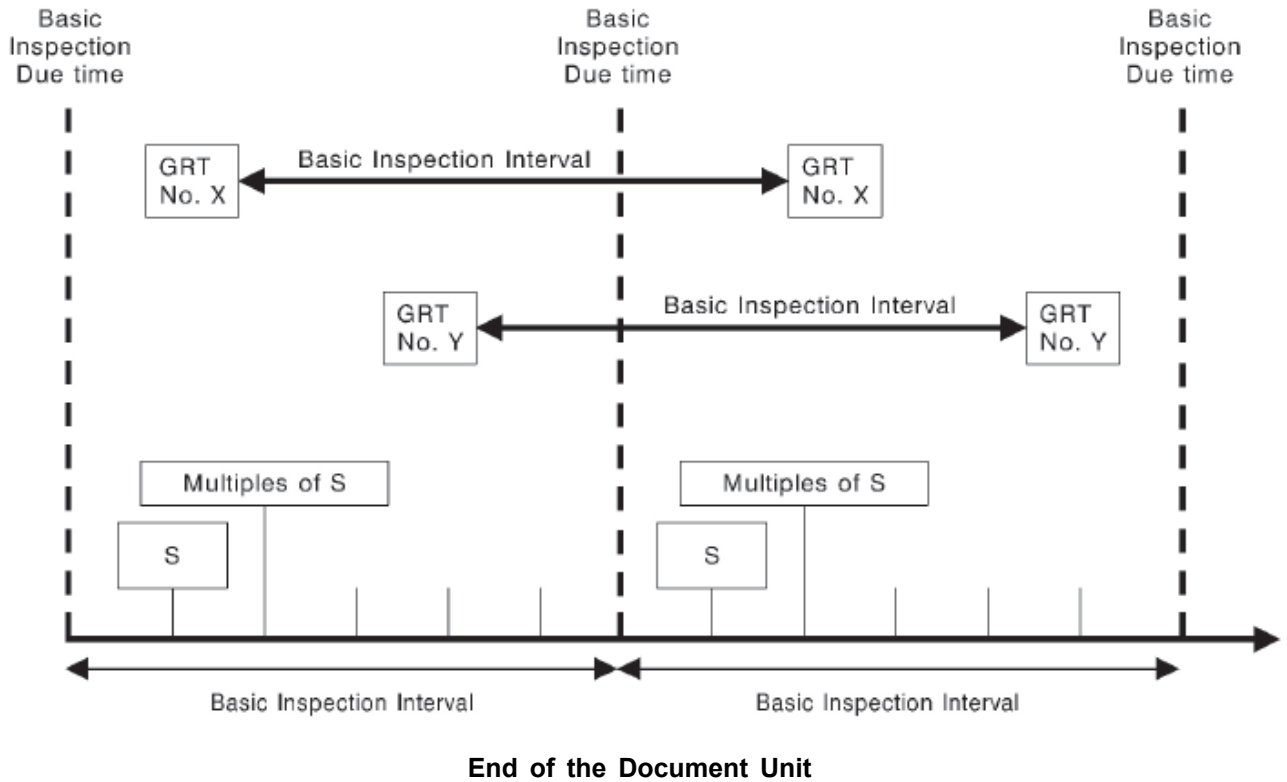
- Electrical Power System, Automatic Flight Control System, Optional Equipment, De-icing system.

Group 6 (GRT 06) :

- Structure (except for tail boom...), Air Conditioning System, Flight Controls.

These groups take into account the airworthiness operations scheduled at 660 FH.

Example : Staggered basic inspection



USE OF THE MSM

1 ADAPTATION OF THE MSM

The Master Servicing Manual (MSM) presents the limits for the maintenance operations recommended by the manufacturer.

It can be used as is or it can be adapted by each operator according to his operational needs on the condition he has the approval of his aviation authorities and that he complies with the interval for each maintenance operation.

2 INFORMATION FEEDBACK FROM THE OPERATOR TO THE MANUFACTURER

The operator must inform his local aviation authorities and the aircraft manufacturer of any significant failure or malfunction discovered in operation or in maintenance, particularly when the component in question is maintained by condition monitoring.

This feedback can be an essential factor for continuous airworthiness (refer to Chapter 04, DU: USE OF THE ALS: "OCCURRENCE REPORTING").

This is also applicable in case of an accident.

3 IN-SERVICE MONITORING

3.1 Airframe monitoring

In-service monitoring of the airframe must be recorded on the helicopter logbook.

3.2 Monitoring of components

All the components listed in Section 05-10-00 of the MSM are to be monitored in service, unless "not to be monitored in service" is specifically mentioned in the task.

The in-service monitoring of components consists in recording the date of installation, the number of flight hours and/or the calendar time and/or the cycles logged on the equipment log card (**FM**).

If the component does not have a Log Card (component monitored via the Log Card of its higher assembly, component without serial number, etc.), before installation, the operator must organize his documentation in order to comply with the inspection intervals and the limits relating to this component (refer to Standard Practices Manual Work Card 20.08.05.101).

These records must be kept if the component is transferred from one helicopter to another.

All the documentation must be organized to enable recording of all the parameters used to comply with the limits.

3.3 Particular case of elastomer hoses

The operator must calculate the aging of elastomer hoses for rejection in accordance with the instructions of Standard Practices Manual (MTC) Work Card 20.09.01.902.

The operator must issue a monitoring document (for example a Log Card) in order to comply with the operating time limits.

4 COMPONENT PART NUMBERS

4.1 Definition

- **Manufacturer Part Number (MP/N)** : this is the identification number of the component, marked on the identification plate or directly on the component.

- **Airbus Part Number (PN)** : this is Airbus part number which is associated with the Manufacturer Part Number (MP/N) for components which are not manufactured by Airbus.

The part numbers of components or assemblies which are given in the MSM are the Manufacturer Part Numbers (MP/N).

In the case of components which are not manufactured by Airbus, the MP/N is followed by the associated PN (in brackets).

Example :

28/20/00/404/000/040					
Booster pump					
	P94C16-606	(704A44510022)	TBO	2000 FH	0

In some cases the MP/N is not specified, in this case it is replaced by a hyphen.

4.2 Part number followed by a suffix

In some cases, the MP/N can be assigned a suffix (example: MP/N 365A33-6005-08M), used for spares, to indicate the installation of an optional component or a specific configuration.

These MP/Ns with a suffix generally do not appear in the MSM.

The limit applicable to an MP/N with a suffix is the same as that of the same MP/N without a suffix.

Example: the limit given for assembly MP/N 365A33-6005-08 is valid for assembly MP/N 365A33 6005 08 and assembly MP/N 365A33 6005 08M.

4.3 Applicability of the limits

The limit(s) specified for the first MP/N is (are) valid for all the other MP/Ns listed in the same task

Example :

29/10/00/402/000/020					
Hydraulic pump					
	C24999010 C24999010-1	(704A44320032) (704A44320038)	TBO	3000 FH	300 FH

4.4 Changes to part numbers

Unless otherwise specified, the service life limits and the inspections apply to the component whose Part Number is listed in the MSM as well as to its future changes when these are not already indicated in this chapter. In this case, contact the Airbus Technical Support Department to confirm the validity of installation of the dash number.

5 EXPRESSION OF THE LIMITS

A component or a maintenance operation can be subject to one or several limitations, expressed in :

- flight hours,
- calendar time,
- cycles,
- number of operations (for certain components).

5.1 Flight Hours

FH (Flight Hours) are counted from take-off to landing.

5.2 Calendar Time

Calendar time can be expressed :

- in years (Y),
- in months (M),
- in days (D),
- from the date written on the label of the equipment item (ED: Expiry Date).

5.2.1 Airframe initial aging date

The helicopter airframe initial aging date to be taken into account is the date marked on the helicopter identification plate.

5.2.2 Initial aging date for a component

Unless specified differently by mentions TSI (Time Since Installation) or TSM (Time Since Manufacture) in each MSM Task concerned, the start date for aging (T0) to be taken into account is the date of first installation or first depreservation to be recorded on the Log Card for:

- new equipments or mechanical assemblies (with OTL or TBO) and their constituents
- not new equipments or mechanical assemblies (with TBO) and their constituents after accomplishment of an overhaul

The storage time before installation is not to be taken into account in the aging calculation provided that storage was compliant with the applicable documentation, except for the special cases specified below.

Exception :

For some equipments and mechanical assemblies, the date to be taken into account for aging is the date of manufacture recorded on the Log Card. In this case, TSM (Time Since Manufacture) is specified in each MSM Task concerned.

Exemple:

26/22/00/000/000/000					
Hand fire extinguisher - Cylinder					
12085-01 H1-10AIR	(S262A10T1001) (704A32810008)	OTL	10 Y	TSM	0

Some equipments and mechanical assemblies are subject to both a limitation since installation and a limitation since manufacture. In this case, TSI (Time Since Installation) or TSM (Time Since Manufacture) is specified behind the respective limitation.

Exemple:

26/21/00/000/000/020					
Squib					
863080-00 863085-00	(704A38710040) (704A38710041)	OTL	//	5 Y TSI 7 Y TSM	0 0

NOTE

If there is no log card, when the component arrives in the stores, it is advisable to create a "Follow-up Sheet" and to fill out the helicopter logbook when the component is installed. When the component is installed on the helicopter, the applicable date is the date of manufacture marked on the component's identification plate.

NOTE

Dynamic components, which have been subject to work in a specialized workshop or long-term storage in accordance with the applicable instructions during this period, may be subject to an extension to this calendar time limit.

In this case the operator must make a request with the Airbus Technical Support Department and obtain approval from the manufacturer and the Airworthiness Authorities.

5.3 Cycles

Depending on the component, the cycles may be expressed in :

- **Landing Cycles (LC)**
1 LC = 1 take-off/landing.

- **Sling Cycles (SC)**
 - 1 load release on the ground = 1 SC (load on the ground)
 - 1 load release in flight = 3 SC (load not laid on the ground)

- **Roping Cycles (RC)**
1 RC = 1 rope release, or 4 commandos roping down without rope release.

- **Hoist Cycles (HC)**
1 HC =
 - in flight, 1 lowering plus one raising, whatever the unwound cable length and the load used.
 - on the ground, 1 lowering greater than or equal to 5 meters and one equivalent raising, whatever the load used.

- **OPerating Hours (OPH)**
The definition is specified in the task concerned.

5.4 Provisional limitations

The limits for which additional fatigue tests indicate the possibility of an extension, are identified by code "P" (Provisional). This code (P) follows the value of the limit in the tasks concerned.

The operator should therefore store the component in question after removal, pending a service life extension.

5.5 Periodic and perform once limits

The intervals given in Section 05-20-00 of the MSM are to be complied with :

- **Périodically** : the maintenance operation must be performed at the latest when the indicated interval is reached.

NOTE

For operational reasons, the maintenance operation can be performed before the limit is reached, however, the maximum interval between two maintenance operations must be complied with.

or

- **Once only** : the maintenance operation must be performed once only, when the indicated interval is reached.

It must not be performed before the interval is reached.

These maintenance operations are identified by a dotted line around the task and by code PO (Perform Once) in front of the limit value.

These operations can be performed either :

- after installation of a component removed from the same helicopter,
- after the introduction to service of a new, overhauled or repaired component,
- after installation of a component originating from another helicopter,
- each time after the component is installed.

The start of the maintenance operation is specified on each task concerned.

Example :

62/20/00/605/000/000

Mast and/or hub

AMM 62.20.00.22 PO 3 FH 7 FH

Check tightening torque loads.
Each time after the component is installed.

5.6 Performance interval

Some components and maintenance operations are assigned several intervals separated by the "/" or "&" sign.

In the case of intervals separated by the "/" sign, **the maintenance action must be carried out at the first interval reached.**

The next limit which will trigger the maintenance action will once again be the first limit reached.

In the case of intervals separated by the "&", **the maintenance action must be performed at each interval indicated.**

Example 1 : (Section 05-10-00)

67/30/00/000/000/015

Main rotor servo-control

SC8037 (704A44831149) TBO // 1000 FH 20 Y TSM 100 FH 180 D

TSM or overhaul.

A servo-control operates for 800 FH over a period of 20 Y (TSM). Therefore, the first interval reached by the servo-control is 20 Y, and it must be removed for Overhaul (TBO) before being released to service for a new period.

After Overhaul, this servo-control is installed on aircraft and then operates for 1000 FH in 10 Y. Therefore, the first interval reached by the servo-control is 1000 FH. It must therefore be removed for a new Overhaul (TBO).

Example 2 : (Section 05-20-00)

25/62/00/000/000/090

Float

CMM 25.69.35 PO 6 Y
& 9 Y
& 12 Y 180 D 180 D 180 D

216064-0 (-)
216166-0 (-)
216167-0 (-)
216080-0 (-)

Proof pressure test and detailed check.

The maintenance operation must be carried out once at 6 Y, once at 9 Y and once at 12 Y after manufacture. Then this maintenance operation must be carried out again after introduction into service of new, overhauled or repaired equipment.

5.7 Specific and severe atmospheric operating conditions

Certain specific and severe climatic conditions are considered as specific operating conditions. This concerns the use of helicopters in the following conditions:

- **Tropical and damp atmosphere**

Combinations of high ambient temperatures and high humidity levels (from +28°C (+82.4°F) and 75% relative humidity).

- **Salt-laden atmosphere**

Aircraft based on a ship or based more than 50% of the time inland at less than 1 km (.54 NM) from the coast or,

flying more than 50% of the time over the sea at low altitude (less than 1000 feet).

- **Sand- and/or dust-laden atmosphere**

Sand-laden wind, landing on sandy ground.

- **Cold weather**

As soon as snow or ice conditions with an outside air temperature above -15 °C (5 °F) to -25 °C (-13 °F) are noted or forecast.

- **Very cold weather**

temperatures from -25 °C (-13 °F) to -40 °C (-40 °F).

The scheduled maintenance relating to these operating conditions is covered with all the MSM tasks.

These operating conditions require not only the operations specified in the MSM and standard maintenance operations, but also the preventive measures recommended in AMM Work Card 05.50.09.211. Since the interval is not specified, the operator must reapply the measures according to his own experience in the corresponding conditions.

5.8 Maximum limit value

The maximum limit value represents the limit not to be exceeded. No tolerance is permitted on the maximum limit value.

The maximum limit value equals to a limit value + margin value.

5.8.1 Definition of the margin

To introduce "flexibility" into maintenance planning in order to compensate for unpredictable situations (e.g. unforeseeable increase in the helicopter utilization rate), Airbus defines a value, called "margin", to be added to a limit value.

This margin, added to the limit value to which it is applicable, results in the maximum limit value (limit value + margin value) to be taken into account for two consecutive inspections.

5.8.2 Application of the margin

The margin can be used repetitively for each limit value interval (refer to "Example of use of the margin" § 5.8.4).

However, to maintain certain "flexibility" in maintenance planning, Airbus recommends to plan the maintenance operation using the limit value without taking the margin into account.

5.8.3 Specification of the margin

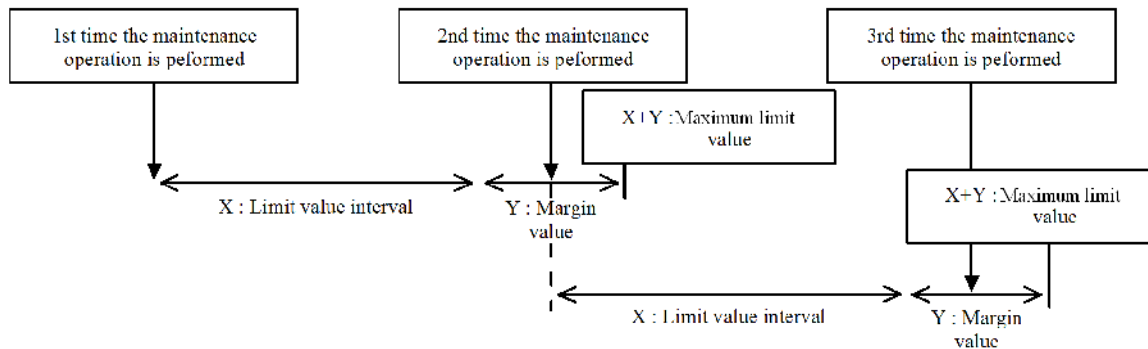
The value and the units of the margin are specified in the following example.

Example :

21/52/00/211/000/000			
Air conditioner			
AMM	21.52.00.211	100 FH // 1 Y	<div style="border: 1px solid black; border-radius: 50%; width: 40px; height: 40px; display: flex; flex-direction: column; align-items: center; justify-content: center; margin: 0 auto;"> 10 FH 36 D </div>
Inspection.			

There is no unit specified when the margin equal zero.

5.8.4 Example of use of the margin



X is the limit value for performing the maintenance operation and Y is the margin value added to the limit value.

After using the first margin value (partially or completely), the next maintenance operation is to be performed again within the X+Y interval, that is to say before the next maximum limit value.

6 TRANSFER OF A COMPONENT FROM ONE HELICOPTER VERSION TO ANOTHER HELICOPTER VERSION

6.1 Dauphin family helicopter versions concerned by the transfer

The versions concerned by the transfer of a component are identified in the following table :

Dauphin families	Civil versions	Military versions
SA 365	C, C1, C2, C3	
AS 365	N, N1, N2, N3	F, Fs (FC, FS), Fi, K, K2
EC 155	B, B1	
AS 565		MA, SA, MB, SB, UB, MBe

6.2 Components NOT subject to in-service monitoring (Condition Monitoring)

Transfer of components not subject to in-service monitoring is authorized between all versions specified in the table above after compliance with MTC Work Card 20.08.05.104.

6.3 Components subject to in-service monitoring

6.3.1 Components subject to a Service Life Limit (SLL)

The transfer rules are defined in the ALS section.

6.3.2 Components subject to a Time Between Overhauls (TBO), to an Operating Time Limit (OTL), to On-Condition Maintenance (OC)

6.3.2.1 Authorization to transfer

The transfer of a component (part or assembly) included in section 05-10-00, from helicopter version "x" to helicopter version "y", is authorized only if the following conditions are met simultaneously:

- both helicopter versions "x" and "y" must be listed in the table above. The transfer of components originating from any other version or any other type of helicopter is forbidden,
- the part number of the component to be transferred is listed in section 05-10-00 of helicopter version "x" and in section 05-10-00 of helicopter version "y",

- no temporary limitation for the component to be transferred is assigned in the MSM of version "x" or in the MSM of version "y".
- MTC Work Card 20.08.05.104 is to be complied with during the transfer process.

6.3.2.2 Application rules

If transfer is authorized, comply with the following application rules:

- **Components subject to Time Between Overhauls (TBO) and to Operating Time Limit (OTL)**

If the limits (flight hours and/or cycles and/or calendar time) are exactly the same on aircraft version "x" and "y", continue counting the flight hours and/or cycles and/or calendar time on aircraft version "y".

If the limits in flight hours and/or cycles and/or calendar time on aircraft version "x" and version "y" are different, convert the flight hours and/or cycles and/or calendar time already logged (since new or since last overhaul) in accordance with the calculation rule applicable to the life limit, defined in chapter 04 (DU: "USE OF THE ALS" - §: "Calculation rules"). Then continue counting the flight hours and/or cycles and/or calendar time and add them to the value obtained after applying the calculation rule.



WHEN THE FLIGHT TIME OF A COMPONENT SUBJECT TO A LIFE LIMIT (SLL) OR AN OPERATING TIME LIMIT (OTL) IS CONVERTED FOLLOWING A TRANSFER, THE OPERATOR MUST CHECK THAT THE LIFE LIMIT OR THE OPERATING TIME LIMIT OF THE COMPONENT ALLOWS THE HIGHER ASSEMBLY TO REACH ITS TBO.

IF THIS IS NOT THE CASE, THE ASSEMBLY MUST BE REMOVED WHEN THE LIFE LIMIT (SLL) OR THE OPERATING TIME LIMIT (OTL) OF THE COMPONENT IS REACHED.

- **Components subject to On-Condition maintenance (OC)**

If the inspection intervals (flight hours and/or torque cycles and/or calendar time) for the component are identical on version aircraft "x" and version aircraft "y", continue counting the flight hours and/or torque cycles and/or calendar time on version aircraft "y".

If the inspection intervals (flight hours and/or torque cycles and/or calendar time) are different on aircraft version "x" and aircraft version "y", perform the inspection when the component is transferred, then comply with the inspection interval specific to version aircraft "y".

6.4 Recording

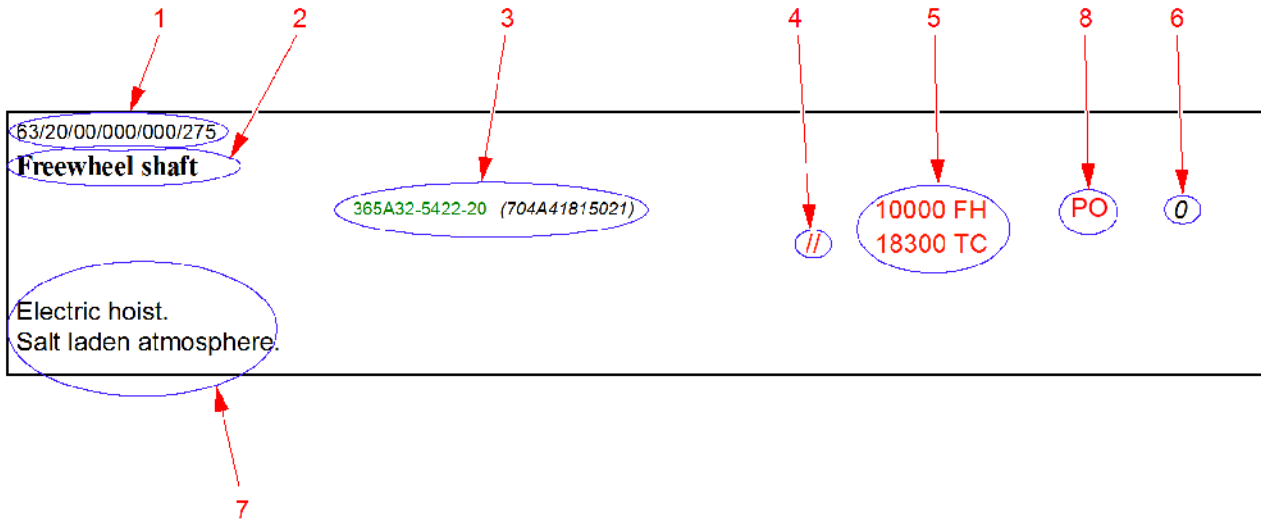
In all cases, the transfer information and the calculation must be recorded on the Log Card (FM) of the component concerned.

An individual log card must be drawn up if the component has been monitored via the log card of its higher assembly..

7 CONTENT OF A TASK

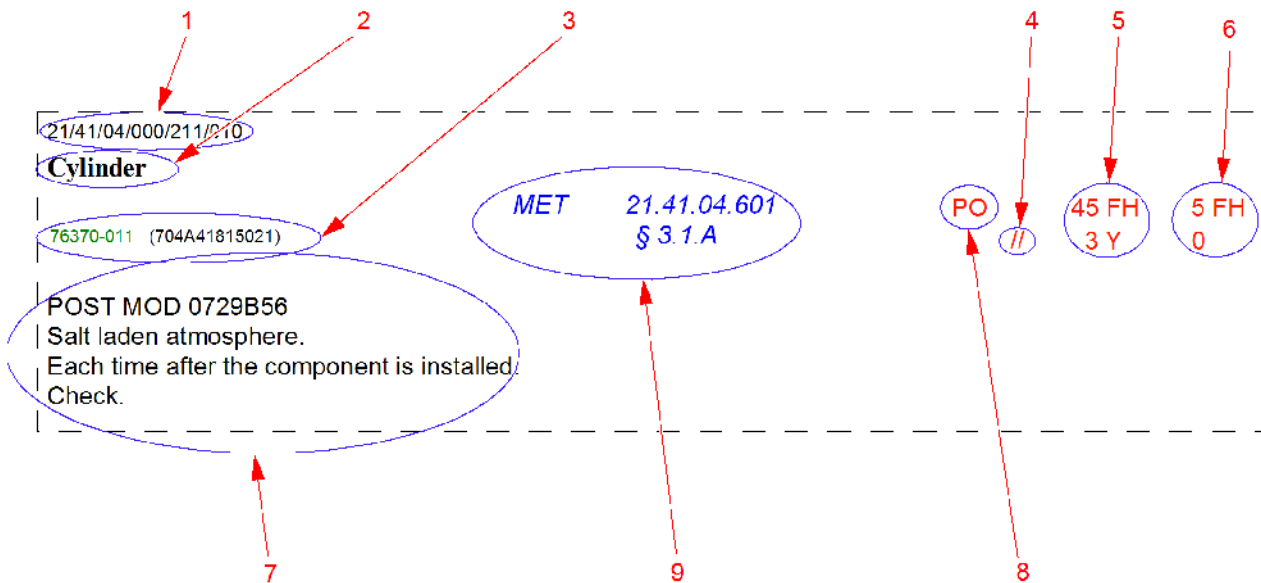
Section 05-10-00

Fictitious example.



Section 05-20-00

Fictitious example.



- 1 - Task number (refer to DU: "Structure of the MSM", § 2.2)
- 2 - Description of the component
- 3 - Component manufacturer Part Number and Airbus management number (refer to § 4)
- 4 - // (or), & (and) (refer to § 5.6)
- 5 - Value(s) and unit(s) of the limit (refer to § 5.1, 5.2 and 5.3)
- 6 - Value(s) and unit(s) of the margin (refer to § 5.8.3)
- 7 - Comments, if necessary
- 8 - PO (perform once): one-off operation (refer to § 5.5)
- 9 - Reference of the documentation: AMM, CMM, etc.

End of the Document Unit

OPTIONAL EQUIPMENT

1 COMPATIBILITY – INCOMPATIBILITY OF OPTIONAL EQUIPMENT

A system designed for specific operational requirements may sometimes impede other functions.

This incompatibility may be encountered at two different levels:

- a) in operation. The operating incompatibility of Optional Equipment is specified in the Flight Manual.
- b) On installation on the helicopter:
 - either provision is ensured on the helicopter for two optional systems but they cannot be installed at the same time: installation incompatibility,
 - or two optional systems can be installed at the same time provided that at least one of the systems is modified: compatibility subject to certain conditions.

The following table lists the approved Optional Equipment and specifies any incompatibility, capabilities or conditions and indicates when consequences on the helicopter operation are specified in a Flight Manual Supplement.

2 TABLE OF APPROVED OPTIONAL INSTALLATIONS

Key:

- A: Fixed parts for both these equipment items cannot be installed together.
- B: Simultaneous installation of the removable parts totally or partially incompatible.
- C: Can be installed together, but cannot be used at the same time.

ITEM	INSTALLATION	NATURE OF THE INCOMPATIBILITY		
		A	B	C
1	Air Conditioning System, Electrically powered	31		2
2	Windshield deicing system	31		1
3	AERAZUR life-raft external installation Life rafts in footstep containers	15; 19; 24		4; 6; 20
4	Non foldable seat on first passenger row (row facing aft or forward)			3
5	Wet Fit (Drain pan)	15; 24	6; 7; 32; 33	
6	Roping beam	15; 24; 26; 27; 28	5; 7; 32; 33	3; 13; 16; 17; 19; 20; 22; 40
7	12 Passenger seat layout	15; 24; 26; 27; 28	5; 6; 32; 33; 39	3; 16; 20
8	Auxiliary fuel tank in cargo hold (180 l - 47 US gal)			
10	HEEL System	15; 26; 27; 28		
11	Emergency flotation gear			
12	Sand filters			
13	Sling with electrically controlled mirror (1 600 kg - 3 527 lb)			6; 7; 16; 20; 24; 26; 27; 28; 32; 33
14	2nd extinguisher in cabin			

ITEM	INSTALLATION	NATURE OF THE INCOMPATIBILITY		
		A	B	C
15	LH and RH hinged rear cabin doors	3; 5; 6; 7; 10; 32; 33	16; 20	
16	AIR EQUIPEMENT Hoist (272 kg/90 m)	24; 26; 27; 28	15	6; 7; 13; 20; 32; 33
17	SPECTROLAB SX-16 Search-light	40		6; 20
18	High visibility markings of the main rotor blade tip caps			
19	External electrical hailer	3; 24		6; 20
20	Roping installation (Ropes are not included)	24; 26; 27; 28	15	3; 6; 7; 13; 16; 17; 19; 22; 32; 33; 40
21	Corrosion protection for SAR mission			
22	Pod-mounted FLIR installation on the RH side			6; 20
24	Retractable passenger footsteps instead of standard equipment (LH and RH sides)	3; 5; 6; 7; 16; 19; 20; 32; 33		13
25	Improved sound proofing (86 dB)	15; 26; 27; 28		
26	4/8-seat VIP installation with improved sound- proofing and carpeting	3; 6; 7; 10; 16; 20; 25; 32; 33; 36		13
27	VIP finish of the cockpit	3; 6; 7; 10; 16; 20; 25; 32; 33		13
28	Gold plated finish of the cabin metallized parts	3; 6; 7; 10; 16; 20; 25; 32; 33		13
29	Skis	35	30	
30	HELLAS			
31	Air Conditioning System Mechanically driven	1; 2		
32	9-Passengers seats lay out	15; 24; 26; 27; 28	5; 6; 7; 33	13; 16; 20
33	13-Passengers seats lay out	15; 24; 26; 27; 28	5; 6; 7; 32; 39	13; 16; 20
34	Quick loading luggage hold			
35	Anti-sticking installation	29		
36	Passengers address NAT AA20-431 with 6 loudspeakers	26		
37	Weather radar Telephonics RDR1400C	38		
38	Weather radar Honeywell RDR2000	37		
39	Hover mode		7; 33	
40	Search light TRAKKA	17		6; 20

End of the Document Unit

SB INCORPORATED

Airbus is required to issue Service Bulletins (ASB/SB) subsequent to new events (incidents, etc).

The information contained in the SBs is incorporated in the MSM if, and only if:

The SB contains new periodic checks, or modifies the existing MSM limitations.

The table below specifies for each SB incorporated:

- the number of the SB and its revision number,
- the subject of the SB,
- the ATA chapter/section concerned,
- the date on which the SB is incorporated in the MSM.

SB N°	Subject	ATA	AD number	Incorporation date
05A006 R1	Check and Replacement of Baggage Compartment Fire Extinguisher (ATA 26-23) and Engine Fire Extinguishers.	26-21		2005.01.20
05A007 R0	Check of the MGB Planet Gear Carrier.	63-20		2005.01.20
62A006 R0	Check for Correct Positioning of the Filler Wedge of the Tenon at the Main Rotor Blade Tip.	62-10		2005.01.20
64-003 R0	Tail rotor blade protection against sand erosion.	64-10		2005.01.20
32-005 R0	Brace Strut Actuator - Maintenance of the Various Components. (ATA 32-20)	32-10		2005.01.20
05A005 R1	Check of the Main Gearbox (MGB) Base Plate.	63-20		2005.06.09
05-008 R0	Visual Inspection of the RH and LH Engine Oil Return Hoses in the MGB Compartment. (ATA 29-12)	79-00		2005.12.22
05A011 R0	Tail Rotor Blade Check.	64-10		2006.11.14
53-023 R0	Visual Inspection of the Upper Flanges of the Forward Bottom Structure Longitudinal Beams.	53-10		2007.04.20
62A014 R0	MAIN ROTOR - Check of the Rotor Mast / Hub Nut.	62-20		2008.06.27
05-009 R1	TIME LIMITS - MAINTENANCE CHECKS - Check of the Hydraulic Hose Routing in the MGB Compartment.	29-10		2008.10.13
05A007 R2	Check of the MGB Planet Gear Carrier - This SB revision leads to no change the Maintenance Programme Section.	63-20		2010.04.08

SB N°	Subject	ATA	AD number	Incorporation date
25-092 R0	Container sealing improvement for reduction of quarter-hull life-raft maintenance (SR10 type) - This SB revision leads to no change the Maintenance Programme Section.	25-69		2010.07.29
56A003 R0	WINDOWS - Windows of hinged cabin doors.	56-30		2011.06.09
32-009 R0	LANDING GEAR - Skis installation	32-40		2013.09.12
24-019 R0	ELECTRICAL POWER - Electrical Bonding.	24-00		2013.09.12
24-019 R0	ELECTRICAL POWER - Electrical Bonding.	24-00		2013.09.12
05-024 R0	CHECKS - FLIGHT CONTROLS - Optimisation of flight controls maintenance	67-00		2014.05.29
05-025 R1	FUSELAGE Optimisation of maintenance effective for the structure of EC155 helicopters	53-10		2014.05.29
52-029 R0	Maintenance of VIP swing-doors POST-MOD OP53C80	52-12		2014.05.29
05-026 R0	Greasing the bolts of the reinforced scissors of the optional ski installation	32-40		2015.07.30
63A012 R1	Inspection of the Shur-Lok nut which retains the tail rotor drive flange of the Main Gearbox	63-24		2015.07.30
05-023 R1	EMERGENCY FLOTATION GEAR - SSCVFDR - Check of the immersion probe	25-62		2015.07.30
29-008 R0	Check of the Hydraulic Pipe Installation	63-20		2010.01.11
05-032 R1	Check for corrosion at the connection between the main landing gear legs and the upper part of the torque links	32-12		2018.09.10
05-035 R1	Reduction of the TBO of the deployment control unit of the CPI 503 beacon (HR SMITH)	25-66		2021.01.22
26-004 R0	FIRE PROTECTION - Obsolescence of cockpit fire extinguisher	26-22		2021.01.22
63A013 R1	MAIN ROTOR DRIVE - Tail rotor drive flange Installation of rear output stop	63-24		2021.01.22
05A033 R2	TIME LIMITS - MAINTENANCE CHECKS - Baggage compartment - Inspection of the baggage compartment skin	53-10		2022.02.01

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 23-COMMUNICATIONS

23-41 INTERCOMMUNICATION SYSTEM

23/41/00/000/000/000					
Wireless V.I.P. Phone - Battery					
	PE013179 PE014840	(706A45612031) (M442A00A1011)	OTL	3 Y	TSM 109 D
OXYTRONIC.					
Return the handset with its base to the manufacturer.					
3 years since manufacture or the last replacement of the battery.					

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 24-ELECTRICAL POWER

24-20 AC GENERATION SYSTEM

24/20/00/000/000/010 10-kVA alternator	ALT3577 (704A46220036)	TBO	3000 FH	300 FH
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24-30 DC GENERATION SYSTEM

24/30/00/000/000/035 Main battery	33490409000 40208-2 (704A46130049) (704A46130010)	OC		
24/30/00/000/000/040 Emergency battery	B1250000 EE0033 EE0417 (704A46130044) (704A46140000) (704A46130051)	OC		

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 25-EQUIPMENTS/FURNISHINGS

25-00 EQUIPMENT AND FURNISHINGS

25/00/00/000/000/000				
Harness - Pilot / copilot/passengers/emergency medical service				
	ALL MP/N (-)	OC		
It is advisable to record the follow up of this equipment in the aircraft log book. Pilot/copilot/passenger/operator.				

25-62 OPT - EMERGENCY FLOATATION GEAR

25/62/00/000/000/010				
Cylinder				
	216122-0 (704A42693007)	OTL	15 Y	TSM 0
	216122-1 (704A42693013)			
Return the equipment to an approved repair station for overhaul and replacement of the unequipped cylinder.				

25/62/00/000/000/070				
Float				
	216064-0 (NR101500123)	OTL	15 Y	TSM 180 D
	216166-0 (NR101500125)			
	216167-0 (NR101500124)			
	216280-0 (NR101500126)			
You can apply for an extension by returning the equipment in question to the manufacturer for examination.				

25-63 OPT - ELECTRIC HOIST

25/63/02/403/000/100				
Electric hoist 90m				
	76378-260 (704A41815065)	TBO	10 Y	180 D
	76378-260-D (704A41815084)		// 1000 HC	100 HC
TBO applicable since TSM or overhaul. GOODRICH FRANCE 600 LBS variable speed.				

25/63/02/403/000/120				
Electric hoist - Electronic control unit				
	61148-016 (704A41816044)	TBO	5 Y	180 D
GOODRICH FRANCE.				

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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25/63/02/403/000/130 Electric hoist - Cable 90m	710176 (NR100300010) 710176-100 (NR100370420)	OTL	2 Y TSI // 7 Y TSM // 500 HC	0 0 0
Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere GOODRICH FRANCE.				
25/63/02/403/000/140 Electric hoist - Cable 90m	710176 (NR100300010) 710176-100 (NR100370420)	OTL	4 Y TSI // 9 Y TSM // 500 HC	0 0 0
GOODRICH FRANCE.				
25/63/02/510/000/100 Electric hoist - Squib	20CF4D (N6714400100)	OTL	3 Y TSI // 5 Y TSM	0 0
GOODRICH FRANCE.				

25-66 EMERGENCY LOCATOR TRANSMITTER

25/66/00/000/000/001 Emergency locator transmitter - Battery	CMM 25.60.03 606-180-001 (NR017700738)	OTL	ED	0
ADELT CPT 609.				
25/66/00/000/000/002 Emergency locator transmitter - Deployment mechanism - Battery	CMM 25.60.02 00-23-1099 (704A46132002)	OTL	2 Y TSI // 5 Y TSM	0 0
ADELT CPT 609.				
25/66/00/000/000/004 Emergency locator transmitter - Squib	CMM 25.60.02 BC85-070 (NR663700001) BC85-070-1 (704A38710052)	OTL	1 Y TSI // 5 Y TSM	0 0
ADELT CPT 609.				

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
25/66/01/211/000/000 Emergency locator transmitter - Battery pack	JE2-1978-3 (N6742450122) JE2-1978-3NG (NR003100026)	OTL	ED // 1 OPH	0 0
JOLIET JE2/JE2NG. 1 OPH = 1 operating hour or operation of unknown duration.				
25/66/02/211/000/000 Emergency locator transmitter - Battery pack	452-0133-406 (-)	OTL	ED // 1 OPH	0 0
ARTEX 110 / 406HM. 1 OPH = Operation in continuous mode or of unknown duration.				
25/66/02/211/000/005 Emergency locator transmitter - Battery	452-0133 (-)	OTL	ED // 1 OPH	0 0
ARTEX C406-2HM. 1 OPH = 1 operating hour or operation of unknown duration.				
25/66/03/000/000/000 Underwater locator beacon		TBO	ED	0
DATASONIC ELP-362D With serial number starting with "S". Return the equipment to the manufacture or send it to an approved repair station.				
25/66/03/000/000/005 Underwater locator beacon - Battery kit	B362-06192-1 (-) B362-06192-2 (-)	OTL	ED	0
DATASONIC ELP-362D With serial number not starting with "S".				
25/66/04/000/000/000 Underwater locator beacon	DK100 (704A45737016)	TBO	ED	0
DK100. Emergency locator transmitter to be returned to the manufacturer for checking and replacement of the battery.				
25/66/04/000/000/010 Underwater locator beacon - Battery kit	810-2007/K (-) 810-2008/K (-)	OTL	ED	0
DK120. Beacon with "Code B" indicated on the label = Battery 810-2007/K. Beacon with "Code C" indicated on the label = Battery 810-2008/K.				

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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25/66/04/000/000/020 Underwater locator beacon - Battery kit 810-2042/K (-) OTL ED 0 DK120/90.				
25/66/04/510/000/100 Battery CMM 25.62.01 147-E5542-00 (-) OTL ED 0 DK120/90 ULB 90-Day L3COM MP/N 266-E5542-00				
25/66/06/000/000/001 Emergency locator transmitter - Battery pack CMM 25.63.05 S1820516-99 (-) OTL // ED 1 OPH 0 KANNAD 406-AF. KANNAD 406-AP. 1 OPH = 1 operating hour (excluding test) or operation of unknown duration.				
25/66/06/510/000/100 Emergency locator transmitter - Battery pack CMM 25.63.06 S1840510-01 (-) OTL // ED 1 OPH 0 KANNAD 406 - AP (ER) KANNAD AP INTEGRA (ER-N) 1 OPH = 1 operating hour (excluding test) or operation of unknown duration.				
25/66/07/000/000/000 Emergency locator transmitter - Battery pack CMM 25.56.97 A0696 (-) OTL // 5 Y TSM 1 OPH 0 HR SMITH 500-12,-27. 1OPH = 1 utilization.				
25/66/16/000/000/000 Emergency locator transmitter - Battery A01577 (-) OTL ED 0 CPI 503. Return the ELT and the attaching screw to the manufacturer to have the battery replaced. During the installation of the ELT put the new attaching screw.				
25/66/16/000/000/010 Emergency locator transmitter - Interface unit - Battery CMM 25.61.98 A01011 (NR033500002) OTL // 1 Y TSI ED 0 CPI 503. Battery to be replaced.				

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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25/66/16/000/000/020				
Emergency locator transmitter - Deployment control unit				
	503-41 (704A45737067)	TBO	7 Y // 4000 FH	TSM 0
CPI 503. Return the equipment to the manufacturer.				

25/66/16/000/000/060				
Emergency locator transmitter - Release unit				
	503-21 (704A45737059) 503-21-1 (704A45737075)	TBO	10 Y // 1 OPH	TSM 0
CPI 503. 1 OPH = 1 deployment. Return the equipment to the manufacturer.				

25/66/16/510/000/700				
Emergency locator transmitter				
	503-16 (704A45737056) 503-16-25 (704A45737080)	TBO	10 Y // 1 OPH	TSM 0
CPI 503. 1 OPH = 1 deployment or 1 transmission. Return the emergency locator transmitter to the manufacturer.				

25/66/16/510/000/800				
Emergency locator transmitter - Interface unit				
	503-42-6-G (704A45751330) 503-42-G (M256A30A1004)	TBO	10 Y	TSM 0
CPI 503. Return the equipment to the manufacturer.				

25/66/18/510/000/000				
Underwater locator beacon - Battery kit				
	810-2042/K (-) B362-06192-1 (-)	OTL	ED	0
Fitted ULB Honeywell 316-6004-009 (90 days) installed on SSCVFDR AR-602C & AR-204C.				

25/66/19/510/000/100				
Underwater locator beacon - Battery kit				
	147E0305-00 (-)	OTL	ED	0
L3-xxx. Fitted on 266-E0305-00 30 days underwater locator beacon, installed on SSCVFDR L3Com FA5001				

25/66/19/510/000/110				
Underwater locator beacon				
	147E5542-00 (-)	OTL	ED	0
L3-xxx. Fitted on 266-E5542-00 90 days underwater locator beacon, installed on SSCVFDR L3Com FA5001				

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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25-69 OPT - LIFE RAFTS

25/69/06/000/000/000 Life raft <i>CMM 25.64.47</i>	245431-0 245431-1 245434-0 245434-1	(704A42691043) (704A42691060) (704A42691059) (704A42691061)	OTL	12 Y TSM	0
Tropical and damp atmosphere Type SR10. You can apply for an extension by returning the assembly in question to the manufacturer for examination.					
25/69/06/000/000/010 Life raft <i>CMM 25.64.47</i>	245431-0 245431-1 245434-0 245434-1	(704A42691043) (704A42691060) (704A42691059) (704A42691061)	OTL	15 Y TSM	0
Type SR10. You can apply for an extension by returning the assembly in question to the manufacturer for examination.					
25/69/06/000/000/020 Life raft - Cylinder <i>CMM 25.64.47</i>	220097-0	(-)	OTL	15 Y TSM	180 D
Type SR10.					
25/69/06/100/000/000 Life raft - Cylinder <i>CMM 25.64.47</i>	217544-0	(-)	OC		
Type SR10.					
25/69/08/000/000/000 Life raft <i>CMM 25.59.07</i>	00051289	(-)	OC		
Type F10R					
25/69/08/100/000/000 Life raft - Cylinder <i>CMM 25.59.07</i>	44063002	(-)	OTL	15 Y TSM	0
Type F10R. Cylinder to be returned to the supplier to replace the body.					

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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25-83 OPT - ROPING

25/83/00/000/000/000 Roping beam SIREN.	AS29-05-02 (704A41815092)	TBO	5 Y // 1000 RC	180 D 100 RC
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25-91 OPT - LOADS UNDER HELICOPTER

25/91/01/000/000/010 Cargo hook SIREN. Storage limit: 5 years.	AS21-5-7 (704A41811029)	TBO	5 Y // 1500 SC	180 D 150 SC
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End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 26-FIRE PROTECTION

26-21 ENGINE FIRE EXTINGUISHING

26/21/00/000/000/020 Squib	863080-00 863085-00	(704A38710040) (704A38710041)	OTL	// 5 Y 7 Y	TSI TSM	0 0
26/21/00/401/000/007 Engine fire extinguisher - Cylinder	10230-01	(704A42820049)	OC			
26/21/00/401/000/012 Engine fire extinguisher - Cylinder	1023000	(704A42820043)	OC			

26-22 CABIN FIRE EXTINGUISHING

26/22/00/000/000/000 Hand fire extinguisher - Cylinder	12085-01 H1-10AIR	(S262A10T1001) (704A32810008)	OTL	10 Y	TSM	0
26/22/00/000/000/020 Hand fire extinguisher	P3APP003010A	(U262A50T1001)	OTL	7 Y	TSM	10 M
Charged extinguisher, which are removed from service due to end of lifetime or damage must be discharged and then discarded in accordance with the local regulations.						
26/22/01/000/000/000 Hand fire extinguisher	74-00	(800G6799980100)	OC			

26-23 BAGGAGE COMPARTMENT FIRE EXTINGUISHING

26/23/01/000/000/000 Squib	863080-00 863085-00	(704A38710040) (704A38710041)	OTL	// 5 Y 7 Y	TSI TSM	0 0
26/23/01/000/000/012 Fire extinguisher - Cylinder	864250-00 864250-01	(704A42820042) (704A42820048)	OC			

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 28-FUEL

28-10 STORAGE

28/10/00/000/000/000 Fuel tank <div style="display: flex; justify-content: space-around;"> ALL MP/N (-) OC </div>

28-20 FUEL DISTRIBUTION

28/20/00/404/000/030 Booster pump <div style="display: flex; justify-content: space-around;"> C11BC0080 C11BC0081 (-) (704A44510024) TBO 1800 FH 0 </div>
28/20/00/404/000/040 Booster pump <div style="display: flex; justify-content: space-around;"> P94C16-606 (704A44510022) TBO 2000 FH 0 </div>

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 29-HYDRAULIC POWER

29-10 MAIN HYDRAULIC SYSTEM

29/10/00/402/000/020					
Hydraulic pump	C24999010 C24999010-1 C24999012	(704A44320032) (704A44320038) (704A44320047)	TBO	3600 FH	<i>200 FH</i>

29-20 AUXILIARY HYDRAULIC SYSTEM

29/20/00/520/000/000					
Standby electric pump	A5029190	(704A44310025)	OC		

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 31-IND./RECORDING SYSTEMS

31-21 CLOCK

31/21/00/000/000/000				
Clock CH97-10 - Battery				
<i>CMM 31.21.97</i>		OTL	2Y	73D
Operating time limit that must not exceed the expiry date marked on the original packaging.				

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 32-LANDING GEAR

32-10 MAIN LANDING GEAR

32/10/00/000/000/000 Main landing gear	ALL MP/N	(-)	OC	
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32-20 NOSE LANDING GEAR

32/20/00/000/000/000 Nose landing gear	ALL MP/N	(-)	OC	
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32-40 WHEELS AND BRAKES

32/40/05/000/000/000 Anti-sinking shoes - Mechanical actuator	355090001	(704A41450036)	TBO	8Y	TSM	0
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End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 33-LIGHTS

33-43 OPT - SEARCHLIGHT

33/43/07/510/000/100 Searchlight - Gimbal arm	019058 (-) 019058-21 (-) 021715 (-)	OTL	7 Y	180 D
SPECTROLAB SX-16. Aluminium gimbal arm.				
33/43/07/510/000/200 Searchlight - Gimbal arm	022929 (-) 022929-11 (-) 022929-81 (-)	OTL	10 Y	180 D
SPECTROLAB SX-16. Steel gimbal arm.				
33/43/08/510/000/100 Searchlight	200005-0056321 (704A46822019)	OC		
TRAKKA A800.				
33/43/08/510/000/200 Searchlight - PCIU	011010-0020000 (704A46822018)	TBO	6000 FH	0
TRAKKA A800. Return the equipment to the manufacture.				

33-50 EMERGENCY LIGHTING

33/50/03/000/000/005 Sliding door HEEL system - Battery	8162-1 (706A36103001) 8162-2 (706A36103001)	OTL	2 Y	73 D
Before installation, make sure that time since manufacture is less than 1 year. Check the charge of the battery before installation.				
33/50/03/000/000/015 HEEL cabin roof - Battery	5776-1 (-)	OC		
33/50/03/510/000/100 HEEL cabin roof - Converter unit	6958-2 (706A36140000)	OC		

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 34-NAVIGATION

34-42 WEATHER RADAR

34/42/01/510/000/100				
Radar 1400 and 1600 antenna - Slaving unit				
4000504-0301	(N6707152800)	TBO	3000 FH	300 FH
4000504-0303	(N6707154000)		// 3 Y TSI	109 D
DA1203A.				
Return the equipment to an approved repair shop.				

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 45-CENTRAL MAINT SYSTEM

45-11 OPT - CVFDR

45/11/01/510/000/000 MFDAU+ - Lithium battery	CR2032RH (NR000705733)	OTL	5 Y TSI	180 D
OTL only applicable to aircraft fitted with HUMS, UMS or HOMP. Fitted on assemblies 360-00700-100/101. Return the equipment to Airbus for replacement of the battery.				
45/11/01/510/000/100 SSCVFDR recorder	980-6021-032 (704A45381025) 980-6021-066 (704A45381029)	OC		
SSCVFDR Honeywell AR-204C or AR-602C equipped of the underwater locator beacon: DATASONIC ELP-362D [30 days], or DUKANE DK100, or DUKANE DK120 [30 days], or DUKANE DK120/90 [90 days], or Honeywell 316-6004-009 [90 days].				
45/11/03/000/000/000 Recorder - Battery	M4T32-BR12SH6 (-)	OTL	// 10 Y TSI 20 Y TSM	180 D 180 D
QAR recorder. Operating time limit that must not exceed the time since manufacture. Fitted on assembly CTS200-E20-0000.				
45/11/04/510/000/000 SSCVFDR recorder	5001-6103-11 (704A45381032)	OC		
L3Com FA5001 Fitted with underwater locator beacon ULB 30-Day L3xx (266-E0305-00) or ULB 90-Day L3xx (266-E5542-00).				

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 62-ROTOR (S)

62-20 MAIN ROTOR HEAD

62/20/00/000/000/335 Pitch-change rod body Aging limit since MRH assembly.	T24497 (704A33633216)	OTL	2 Y	73 D
62/20/00/000/000/400 Swashplate bearing	121620 Y51BB10843S2M74 (704A33651157) (704A33651158)	OTL	5000 FH	0
62/20/00/000/000/420 Mast bearing <i>AMM 62.22.00.061</i>	Y45CB10700S7M74 (704A33651119)	OTL	2500 FH	0
62/20/00/605/000/010 MRH	ALL MP/N (-)	OC		

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 63-ROTOR DRIVE (S)

63-10 ENGINE/MGB COUPLING

63/10/00/401/000/000 Engine-MGB coupling	ALL MP/N	(-)	OC	
63/10/00/510/000/000 Flexible coupling - Inputs	365A32-6530-00 368A32-0730-00	(-) (-)	OC	
POST MOD 0763C88				
63/10/00/510/000/010 MGB - Tachymeter generator	365A32-0065-01	(-)	OC	
POST MOD 0763C88				
63/10/00/510/000/020 Rotor brake	368A32-0060-02	(-)	OC	
PRE MOD 0763C86				

63-20 MAIN GEARBOX

63/20/00/000/000/010 Right Accessory Gear Box (AGB)	366A32-1001-03	(-)	TBO	1000 FH // 11 Y	100 FH 1 Y
POST MOD 0763C89					
63/20/00/000/000/015 MGB	368A32-0005-01	(-)	TBO	1000 FH // 11 Y	100 FH 1 Y
POST MOD 0763C88					
63/20/00/000/000/020 AGB - bearing	PS-7015436-000	(-)	OTL	10500 FH	0
POST MOD 0763C88					
63/20/00/000/000/131 MGB - Bearings	RU13871S3 RU13871S4	(704A33652100) (704A33652101)	OTL	3500 FH	0
POST MOD 0763C88					

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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63/20/00/000/000/132 MGB - Bearings POST MOD 0763C88	RU13874S3 RU13874S4	(704A33652107) (704A33652104)	OTL	3500 FH	0
63/20/00/000/000/134 MGB - Bearings POST MOD 0763C88	XR13887S2	(704A33652096)	OTL	3500 FH	0
63/20/00/000/000/138 MGB - Bearings POST MOD 0763C88	AT13979S2	(704A33651271)	OTL	3500 FH	0
63/20/00/000/000/140 MGB - Bearings POST MOD 0763C88	RU13872S3 RU13872S4	(704A33652102) (704A33652103)	OTL	10500 FH	0
63/20/00/000/000/141 MGB - Bearings POST MOD 0763C88	TJ13873S2	(704A33651266)	OTL	10500 FH	0
63/20/00/000/000/142 MGB - Bearings POST MOD 0763C88	AB43627S1	(704A33651262)	OTL	10500 FH	0
63/20/00/000/000/144 MGB - Bearings POST MOD 0763C88	TJ13888S5 TJ13888S6	(704A33651264) (704A33651265)	OTL	10500 FH	0
63/20/00/000/000/145 MGB - Bearings POST MOD 0763C88	AB13875S2	(704A33651267)	OTL	10500 FH	0
63/20/00/000/000/148 MGB - Bearings POST MOD 0763C88	NU13876S2	(704A33652087)	OTL	10500 FH	0

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
63/20/00/000/000/150 MGB - Bearings POST MOD 0763C88	RU13877S3 (704A33652105) RU13877S4 (704A33652106)	OTL	7000 FH	0
63/20/00/000/000/153 MGB - Bearings POST MOD 0763C88	AT13709S1 (-)	OTL	7000 FH	0
63/20/00/000/000/165 MGB - Tail rotor output pinion assembly POST MOD 0763C88	368A32-0030-01 (-)	TBO	1000 FH // 11 Y	100 FH 1 Y
63/20/00/000/000/170 MGB - Epicyclic gear train POST MOD 0763C88	368A32-0040-01 (-)	TBO	1000 FH // 11 Y	100 FH 1 Y
63/20/00/000/000/185 MGB - Dual lubrication pump assembly POST MOD 0763C88	368A32-0050-01 (-)	TBO	1000 FH // 11 Y	100 FH 1 Y
63/20/00/000/000/190 MGB - Fan output assembly POST MOD 0763C88	368A32-0055-02 (-)	TBO	1000 FH // 11 Y	100 FH 1 Y
63/20/00/000/000/200 MGB plate POST MOD 0763C88	368A32-0163-00 (-)	OTL	3840 FH //19200 TC	0 0
63/20/00/000/000/205 Input bevel gear POST MOD 0763C88	368A32-0205-00 (-) 368A32-0205-01 (-) 368A32-0205-02 (-) 368A32-0205-03 (-) 368A32-0205-04 (-) 368A32-0205-05 (-) 368A32-0205-06 (-) 368A32-0205-07 (-)	OTL	11529 FH	0

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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63/20/00/000/000/210 Double bevel gear	368A32-0211-20 (-) 368A32-0211-21 (-) 368A32-0211-22 (-) 368A32-0211-23 (-)	OTL	40000 FH	0
POST MOD 0763C88				

63/20/00/000/000/215 Freewheel shaft	368A32-0230-20 (-)	OTL	6918 FH	0
POST MOD 0763C88				

63/20/00/000/000/220 Engine flange	368A32-0400-20 (-)	OTL	115600C	0
POST MOD 0763C88				

63/20/00/000/000/240 Rotor brake - Pads	439220X (704A33620025)	OTL	100 *	0
PRE MOD 0763C86				
* = braking cycle If the braking counting is not possible, the pads are changed every 50 FH with a conservative braking rate of 2 brake applications by FH.				

63/20/00/401/000/055 MGB	365A32-9001-00 (-)	TBO	1800 FH // 24 Y	0 0
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63-30 MOUNTS AND ATTACHMENTS

63/30/00/000/000/000 MGB supports and attachments - Suspension assy	ALL MP/N (-)	OC		
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End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 64-TAIL ROTOR

64-20 TAIL ROTOR HEAD

64/20/00/000/000/030 TRH 365A33-3501-03 (-) OC 365A33-3502-01 (-) 365A33-3502-02 (-)

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 65-TAIL ROTOR DRIVE

65-10 TAIL ROTOR DRIVE SHAFT

65/10/00/000/000/000				
Tail rotor drive shaft				
	ALL MP/N	(-)	OC	
65/10/00/000/000/010				
Bearing				
	593364	(704A33651180)	OTL	6000 FH
	83A904D-1C3	(704A33651142)		// 10 Y
	83A904DC3	(704A33651142)		0
				180 D
Bearings that cannot be dismantled. These bearings are not monitored in service, they will therefore be assigned the operating hours logged by the shaft element on which they are installed.				

65-20 TAIL GEARBOX

65/20/00/402/000/010				
TGB				
	365A33-6005-06	(-)	TBO	2400 FH
	365A33-6005-08	(-)		// 24 Y
	365A33-6005-09	(-)		240 FH
				180 D

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 67-ROTORS FLIGHT CONTROL

67-10 MAIN ROTOR CONTROL

67/10/00/000/000/010 Ball control	82CF9242 82CF9243	(704A34130197) (704A34130198)	OTL	2300 FH	230 FH
67/10/00/000/000/020 Ball control	82CF-10307 82CF-10308	(704A34130228) (704A34130229)	OTL	6000 FH	300 FH

67-20 TAIL ROTOR CONTROLS

67/20/00/000/000/010 Ball control	82CF9240 82CF9241	(704A34130196) (704A34130195)	OTL	2300 FH	230 FH
67/20/00/000/000/020 Ball control	82CF-10305 82CF-10306	(704A34130226) (704A34130227)	OTL	20000FH	300 FH

67-30 SERVO-CONTROLS

67/30/00/000/000/005 Tail rotor servo-control	SC7291	(704A44831153)	TBO	3000 FH // 20 Y	TSM 300 FH 180 D
TSM or overhaul.					
67/30/00/000/000/015 Main rotor servo-control	SC8037	(704A44831149)	TBO	1000 FH (P) // 20 Y	TSM 100 FH 180 D
TSM or overhaul.					
67/30/00/000/000/016 Main rotor servo-control	SC8037-1	(704A44831155)	TBO	3000 FH // 20 Y	TSM 300 FH 180 D
TSM or overhaul.					

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 71-POWER PLANT

71-00 POWER PLANT

71/00/00/000/000/000					
Engine bay fan	VCT1518D	(704A43691028)	TBO	1200 FH	120 FH
TECHNOFAN.					
Return the fan to the manufacturer for reconditioning.					

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 79-ENGINE OIL

79-21 OIL COOLING SYSTEM

79/21/00/000/000/000				
Engine/MGB oil cooler				
	34774	(704A33220037)	OC	
	53139	(704A33220038)		

End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 80-ENGINE STARTING

80-10 CRANKING

80/10/00/000/000/000 Starter generator TSM or overhaul.	524-031	(704A46101011)	TBO	1200 FH // 5 Y	20 FH 0
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End of the Document Unit

Task Number Description/Remarks	MP/N (PN)	Limit Type	Interval	Margin
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ATA 93-MONITORING SYSTEM

93-51 FLIR

93/51/04/510/000/100				
FLIR				
	350CF00029	(704A48410152)	TBO	12 Y
	350CF00036	(704A48410155)		180 D
	350CF00056	(704A48410159)		
FLIR UF350EP.				

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 05-TIME LIMITS/MTCE. CHKS

05-21 DAILY CHECKS

05/21/00/214/000/000 Inspection - Optional systems	AMM 05.21.00.214 05.21.00.214.002 05.21.00.214.003	TA	0
05/21/00/214/000/005 Inspection - Optional systems	AMM 05.21.00.214 05.21.00.214.001	BFF	0
05/21/00/214/000/010 Inspection - Optional systems	AMM 05.21.00.214 05.21.00.214.004	ALF	0
05/21/00/215/000/010 Preflight check in cold weather	AMM 05.21.00.215 05.21.00.215.001	BFF	0
05/21/00/215/000/015 Preflight check in cold weather	AMM 05.21.00.215 05.21.00.215.001	TA	0
05/21/00/215/000/020 Preflight check in cold weather	AMM 05.21.00.215 05.21.00.215.002	ALF	0
05/21/00/218/000/010 Preflight check in very cold weather	AMM 05.21.00.218 05.21.00.218.001	BFF	0
05/21/00/218/000/015 Preflight check in very cold weather	AMM 05.21.00.218 05.21.00.218.002	TA	0
05/21/00/218/000/020 Preflight check in very cold weather	AMM 05.21.00.218 05.21.00.218.003	ALF	0

Task Number Description/Remarks	Documentation	Interval	Margin
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05-22 FLIGHT RELATED CHECKS

05/22/00/213/000/000 15-hour check	AMM 05.22.00.213	// 15 FH 7 D	0 0
05/22/00/214/000/000 15-hour inspection of optional equipment	AMM 05.22.00.214	// 15 FH 7 D	0 0

05-25 MAINTENANCE CHECK

05/25/00/211/000/000 Additional check "S"	AMM 05.25.00.211	// 100 FH 1 Y	10 FH 36 D
05/25/00/212/000/000 Additional check "F"	AMM 05.25.00.212	// 100 FH 1 Y	10 FH 36 D

Check to be carried out only for helicopters covered by FAA requirements in addition to Check "S".

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 21-AIR CONDITIONING

21-00 AIR CONDITIONING

21/00/00/221/000/000 P2 air system Detailed inspection.	AMM 21.41.00.221	1200 FH	120 FH
21/00/00/221/000/010 Ventilation system Detailed inspection.	AMM 21.51.00.221	1200 FH	120 FH
21/00/00/221/000/015 Ventilation indication system Functional test.	AMM 21.51.00.722	6000 FH	300 FH

21-41 HEATING SYSTEM

21/41/00/721/000/000 Heating system Checking operation of electro-valve P2.	AMM 21.41.00.721	// 100 FH 1 Y	10 FH 36 D
21/41/04/211/000/000 Heating - FLEXIRAC union seals POST MOD OP21C01 Check.	AMM 21.41.04.211 GRT 06	600 FH	60 FH

21-52 OPT - AIR CONDITIONING

21/52/00/211/000/000 Air conditioner Inspection.	AMM 21.52.00.211	// 100 FH 1 Y	10 FH 36 D
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Task Number Description/Remarks	Documentation	Interval	Margin
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21/52/00/221/000/000 Air conditioner - Mechanical Detailed inspection.	AMM 21.52.00.221 21.52.00.221.001 GRT 06	600 FH	60 FH
21/52/00/221/000/005 Air conditioner - Electrical Detailed inspection.	AMM 21.52.00.221 21.52.00.221.001 GRT 06	600 FH	60 FH
21/52/00/221/000/020 Compressor, electrical air-conditioning unit - Belt Check of the belt tension.	AMM 21.52.00.067	1200 FH	120 FH
21/52/00/611/000/000 Air conditioner Maintenance.	AMM 21.52.00.611	6 Y	180 D
21/52/00/721/000/000 Air conditioning system Energizing the Air Conditioning System.	AMM 21.52.00.722	1 M	3 D

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 22-AUTO FLIGHT

22-10 AUTOMATIC PILOT

22/10/00/722/000/000 SAS indicator system Functional test.	<i>AMM 22.10.00.722</i>	<i>6000 FH</i>	<i>300 FH</i>
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22-11 AUTOPILOT

22/11/00/211/000/000 Ball end Check.	<i>AMM 22.11.00.211</i> <i>GRT 05</i>	<i>600 FH</i>	<i>60 FH</i>
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 23-COMMUNICATIONS

23-00 COMMUNICATIONS

<p>23/00/00/211/000/000 Radio-communication system Inspection.</p>	<p><i>AMM 23.00.00.211</i> <i>GRT 06</i></p>	<p><i>600 FH</i> <i>// 2 Y</i></p>	<p><i>60 FH</i> <i>73 D</i></p>
<p>23/00/01/221/000/000 External hailers Detailed inspection.</p>	<p><i>AMM 23.42.01.221</i> <i>AMM 23.42.03.221</i> <i>GRT 06</i></p>	<p><i>600 FH</i></p>	<p><i>60 FH</i></p>

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 24-ELECTRICAL POWER

24-00 ELECTRICAL POWER

24/00/00/211/000/000 Electrical power system Inspection.	AMM 24.00.00.211 GRT 05	600 FH	60 FH
24/00/00/761/000/000 Electrical power system Measurement - Electrical Bonding.	AMM 24.00.00.761	4 Y	146 D

24-20 AC GENERATION SYSTEM

24/20/00/000/000/070 10 kVA alternator ALT3577 (704A46220036) Splines check.	AMM 24.21.01.221	100 FH	10 FH
24/20/00/211/000/000 AC power system Inspection.	AMM 24.20.00.211 GRT 05	600 FH	60 FH

24-30 DC GENERATION SYSTEM

24/30/00/000/000/000 Main battery 40208-2 (704A46130010) Periodical check.	CMM 24.30.99.01 SECTION 5000	6 M	18 D
24/30/00/000/000/005 Main battery 33490409000 (704A46130049) Visual inspection and electrical check.	CMM 24.39.91 PARA. 5.2 PARA. 5.3	6 M	18 D

Task Number Description/Remarks	Documentation	Interval	Margin
24/30/00/000/000/010 Main battery 40208-2 (704A46130010) Regular check.	CMM 24.30.99.01 SECTION 5000	1 Y	36 D
24/30/00/000/000/015 Main battery 40208-2 (704A46130010) General overhaul.	CMM 24.30.99.01 SECTION 5000	2 Y	73 D
24/30/00/000/000/020 Main battery 33490409000 (704A46130049) Check.	CMM 24.39.91 PARA. 6	2 Y	73 D
24/30/00/000/000/025 Battery compartment Cleaning.	AMM 24.31.01.141	2 Y	73 D
24/30/00/000/000/045 Emergency battery B1250000 (704A46130044) Capacity check according to the § 2D chapter test.	CMM 24.39.38	2 Y	73 D
24/30/00/000/000/055 Emergency battery EE0033 (704A46140000) EE0417 (704A46130051) Charge check.	AMM 24.31.02.741	1 Y	36 D
24/30/00/722/000/000 Battery circuit Functional test.	AMM 24.30.00.722 24.30.00.722.001 GRT 05	600 FH	60 FH

Task Number Description/Remarks	Documentation	Interval	Margin
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24/30/00/722/000/010 Double power Check of the diodes in unit 37 ALPHA.	AMM 24.63.01.761 GRT 05	600 FH	60 FH
24/30/00/722/000/020 Generator circuits Functional test.	AMM 24.30.00.722 24.30.00.722.003 GRT 05	600 FH	60 FH

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 25-EQUIPMENTS/FURNISHINGS

25-00 EQUIPMENT AND FURNISHINGS

25/00/00/520/221/100			
Harness - Pilot / copilot / passengers / emergency medical service			
	<i>AMM 25.00.00.221</i>	1 Y	36 D
Detailed check. Operation also to be performed after each installation of the seat or the harness.			

25-62 OPT - EMERGENCY FLOATATION GEAR

25/62/00/000/000/090			
Floats			
	<i>CMM 25.69.35</i>	PO 6 Y	180 D
	<i>Section 500</i>	& 9 Y	180 D
		& 12 Y	180 D
216064-0	(NR101500123)		
216166-0	(NR101500125)		
216167-0	(NR101500124)		
216280-0	(NR101500126)		
Major inspection.			
	6 Y	TSM	
	9 Y	TSM	
	12 Y	TSM	

25/62/00/000/000/005			
Cylinder			
		3 Y	0
216122-0	(704A42693007)		
Retest to be carried out by the manufacturer or at an accredited repair facility.			
TSM			

25/62/00/000/000/007			
Cylinder			
		5 Y	0
216122-1	(704A42693013)		
Retest to be carried out by the manufacturer or at an accredited repair facility.			
TSM			

25/62/00/000/000/020			
Cylinder			
	<i>AMM 25.62.01.761</i>	1 Y	36 D
Check the inflation head frangible disc.			
TSM			

Task Number Description/Remarks	Documentation	Interval	Margin
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25/62/00/000/000/030 Cylinder Checking the gas cylinder charge. TSM	CMM 25.69.35 Section 500	1 Y	36 D
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25/62/00/000/000/210 Emergency floatation gear 216064-0 (NR101500123) 216166-0 (NR101500125) 216167-0 (NR101500124) 216280-0 (NR101500126) Minor inspection. TSM	CMM 25.69.35 Section 500	18 M	54 D
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25/62/00/000/000/220 Emergency floatation gear Check of the installation.	AMM 25.62.01.211 GRT 05	3 Y	109 D
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25/62/01/000/000/000 Immersion probe Functional test.	AMM 25.62.01.721 GRT 05	600 FH // 2 Y	60 FH 73 D
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25-63 OPT - ELECTRIC HOIST

25/63/02/520/211/100 Electric hoist Installation check.	AMM 25.63.01.211 GRT 05	600 FH // 2 Y	60 FH 73 D
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25/63/02/520/611/100 Electric hoist Salt-laden atmosphere GOODRICH FRANCE. Rinsing hoist and cable. Operation to be carried out if the hoist has not been used during that period.	AMM 12.30.00.611	7 D	0
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Task Number Description/Remarks	Documentation	Interval	Margin
25/63/02/520/722/100 Electric hoist GOODRICH FRANCE. Check the load release controls.	AMM 25.63.01.722 GRT 05	600 FH // 2 Y	60 FH 73 D
25/63/02/520/762/100 Electric hoist 76378-260 (704A41815065) 76378-260-D (704A41815084) Check the electrical bonding.	AMM 25.63.01.762 GRT 05	600 FH // 2 Y	60 FH 73 D
25/63/02/640/000/000 Electric hoist GOODRICH FRANCE. Greasing of the winder screw. Check the cable. Check the extract and winding cable. Check the hook. Check functioning idle(s) and ends of travel.	CMM 25.64.99 SECTION 500	6 M // 50 HC	0 0
25/63/02/640/000/010 Electric hoist GOODRICH FRANCE. Check the oil level in drum. Check of mechanical brake. Check of the motor electromagnetic brake. Check of the condition box of microswitches (except for amendment 23).	CMM 25.64.99 SECTION 500	1 Y // 500 HC	0 0
25/63/02/680/000/010 Electric hoist GOODRICH FRANCE. Drain the drum.	CMM 25.64.99 SECTION 500	2 Y	0

Task Number Description/Remarks	Documentation	Interval	Margin
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25-66 EMERGENCY LOCATOR TRANSMITTER

<p>25/66/00/000/000/006 Emergency locator transmitter - UHF/VHF transmitter ADELTA CPT 609. UHF/VHF transmitter test.</p>	<p><i>CMM 25.60.02</i> <i>PARA. 4.B</i></p>	<p>1 M</p>	<p>3 D</p>
<p>25/66/00/000/000/007 Emergency locator transmitter ADELTA CPT 609. Check the ramp system on the ground.</p>	<p><i>CMM 25.60.02</i> <i>PARA. 4.A</i></p>	<p>6 M</p>	<p>18 D</p>
<p>25/66/00/000/000/008 Emergency locator transmitter - Radar receiver ADELTA CPT 609. Radar receiver test.</p>	<p><i>CMM 25.60.02</i> <i>PARA. 4.C</i></p>	<p>1 Y</p>	<p>36 D</p>
<p>25/66/00/000/000/009 Emergency locator transmitter - Satellite UHF transmitter ADELTA CPT 609. Check. Test transmission with sat./test programmer.</p>	<p><i>CMM 25.60.03</i> <i>PAGE 104</i> <i>PARA. 3</i></p>	<p>1 Y</p>	<p>0</p>
<p>25/66/00/000/000/010 Emergency locator transmitter ADELTA CPT 609. Remove the equipment to test the deployment mechanism and the mount.</p>	<p><i>CMM 25.60.02</i> <i>PAGE 501</i> <i>PAGE 502</i></p>	<p>1 Y</p>	<p>36 D</p>
<p>25/66/00/211/000/000 Emergency locator transmitter ADELTA CPT 609. Check.</p>	<p><i>AMM 25.66.05.211</i> <i>GRT 05</i></p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>

Task Number Description/Remarks	Documentation	Interval	Margin
25/66/00/520/000/100 Emergency locator transmitter DK120/90 ULB 90-Day 266-E5542-00 Fitted on CVFDR L3COM. Cleaning and check.	CMM 25.62.01 § 6 § 3	2 Y	0
25/66/01/211/000/010 Emergency locator transmitter JOLIET JE2/JE2NG. Inspection.	AMM 25.66.01.211 GRT 05	2 Y	73 D
25/66/01/721/000/000 Emergency locator transmitter JOLIET JE2/JE2NG. Functional test.	AMM 25.66.01.721 25.66.01.721.002	6 M	18 D
25/66/02/211/000/010 Emergency locator transmitter ARTEX 110/406HM and C406-2HM. Inspection.	AMM 25.66.02.211 GRT 05	2 Y	73 D
25/66/02/721/000/000 Emergency locator transmitter ARTEX 110/406HM and C406-2HM. Functional test.	AMM 25.66.02.721	6 M	18 D
25/66/03/000/000/010 Underwater locator beacon DATASONIC ELP-362D. Not fitted on CVFDR HONEYWELL. Cleaning and inspection.	CMM 25.60.07	1 Y	36 D

Task Number Description/Remarks	Documentation	Interval	Margin
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25/66/03/000/000/100 Underwater locator beacon DATASONIC ELP-362D. Fitted on CVFDR HONEYWELL. Cleaning and check.	CMM 25.60.07	2 Y	0
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25/66/04/000/000/005 Underwater locator beacon DUKANE DK100. Not fitted on CVFDR HONEYWELL. Cleaning and inspection.	CMM 25.66.97 PARA 5.2 PARA 5.3	1 Y	36 D
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25/66/04/000/000/015 Underwater locator beacon DUKANE DK120. Not fitted on CVFDR HONEYWELL. Cleaning and inspection.	CMM 25.66.97 PARA 6.2 PARA 6.3	1 Y	36 D
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25/66/04/000/000/025 Underwater locator beacon - Battery kit DUKANE DK120/90. Not fitted on SSCVFDR HONEYWELL. Cleaning and check.	CMM 25.64.91 § 5.2 § 5.3	1 Y	0
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25/66/04/000/000/100 Underwater locator beacon DUKANE DK100. Fitted on CVFDR HONEYWELL. Cleaning and check.	CMM 25.66.97 PARA 5.2 PARA 5.3	2 Y	0
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Task Number Description/Remarks	Documentation	Interval	Margin
25/66/04/000/000/200 Underwater locator beacon DUKANE DK120. Fitted on CVFDR HONEYWELL. Cleaning and check.	CMM 25.66.97 PARA 6.2 PARA 6.3	2 Y	0
25/66/04/520/000/030 Underwater locator beacon DUKANE DK120/90. Fitted on CVFDR HONEYWELL. Cleaning and check.	CMM 25.66.91 PARA. 5.2 PARA. 5.3	2 Y	0
25/66/06/520/211/100 Emergency locator transmitter KANNAD 406. Check.	AMM 25.66.06.211 GRT 05	1 Y	36 D
25/66/06/520/721/100 Emergency locator transmitter KANNAD 406. Self-test of the emergency locator transmitter.	AMM 25.66.06.721	1 M	3 D
25/66/07/000/000/020 Emergency locator transmitter HR SMITH 500-27. Check. Checking of the integrated test.	AMM 25.66.17.741	6 M	18 D
25/66/07/000/000/100 Emergency locator transmitter HR SMITH 500-12. Self-test of the emergency locator transmitter.	CMM 25.56.97	1 Y	36 D

Task Number Description/Remarks	Documentation	Interval	Margin
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25/66/16/000/000/040 Emergency locator transmitter CPI 503. Self-test of the emergency locator transmitter.	AMM 25.66.07.741	6 M	18 D
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25/66/16/000/000/050 Emergency locator transmitter CPI 503. Check of the installation.	AMM 25.66.07.211	1200 FH // 2 Y	120 FH 73 D
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25/66/18/520/000/000 Underwater locator beacon 316-6004-009 (-) Fitted on SSCVFDR HONEYWELL. Cleaning and functional test.	CMM 23.70.48 § Cleaning § Testing of the ULB	2 Y	0
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25/66/19/520/000/100 Underwater locator beacon 266-E0305-00 (-) 266-E5542-00 (-) ULB L-3xx fitted to SSCVFDR L3Com. Cleaning and functional test.	CMM 25.62.01 § CLEANING § Tests for ULB	ED	0
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25-69 OPT - LIFE RAFTS

25/69/01/520/221/100 Life raft Type F10R. Type SR10. System inspection.	AMM 25.69.01.221	2 Y	73 D
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Task Number Description/Remarks	Documentation	Interval	Margin
25/69/06/200/000/010 Life raft PRE MOD 0745C43 // PRE SB 25-092 Type SR10. Detailed inspection. TSM	CMM 25.64.47 PAGE 100X EXC. PARA. 2.A EXC. PARA. 2.F PAGE 500X EXC. PARA. 4.A.3 EXC. PARA. 4.A.4	2 Y	73 D
25/69/06/200/000/015 Life raft POST MOD 0745C43 // POST SB 25-092 Type SR10. Detailed inspection. TSM	CMM 25.64.47 PAGE 100X EXC. PARA. 2.A EXC. PARA. 2.F PAGE 500X EXC. PARA. 4.A.3 EXC. PARA. 4.A.4	30 M	90 D
25/69/06/200/000/020 Life raft Type SR10. Overpressure test. TSM	CMM 25.64.47 PAGE 100X PARA. 2.A	5 Y	180 D
25/69/06/200/000/030 Life raft - Cylinder 217544-0 (-) 220097-0 (-) Type SR10. Retest to be carried out by the manufacturer or at an accredited repair facility. TSM		5 Y	0

Task Number Description/Remarks	Documentation	Interval	Margin
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25/69/06/221/000/000 Life raft - Cylinder 217544-0 (-) Type SR10. Corrosion inspection at the top part of the cylinder. TSM	AMM 25.69.01.221 25.69.01.221.003	1 Y	36 D
25/69/08/200/000/000 Life raft Type F10R. Detailed inspection. If Time since manufacture is more than 10 Y, the operation is to be carried out every 1 Y.	CMM 25.59.07 PAGE 100X PAGE 500X	2 Y	73 D
25/69/08/200/000/010 Life raft - Cylinder Type F10R. Retest to be carried out by the manufacturer or at an accredited repair facility. TSM		3 Y	0

25-83 OPT - ROPING

25/83/00/000/000/010 Roping installation Detailed check.	AMM 25.83.01.221 GRT 05	600 FH // 2 Y	60 FH 73 D
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25-91 OPT - LOADS UNDER HELICOPTER

25/91/00/000/000/000 Control mechanical release - Unit AS22-09-20 (NR081900027) Salt-laden atmosphere Check and greasing.	CMM 25.89.06	2 Y	73 D
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Task Number Description/Remarks	Documentation	Interval	Margin
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25/91/00/000/000/010 Control mechanical release - Unit AS22-09-20 (NR081900027) Check and greasing.	CMM 25.89.06	3 Y	109 D
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25/91/01/061/000/020 Sling system SIREN. Adjust the load indicator.	CMM 25.89.04 GRT 06	// 600 FH 2 Y	60 FH 73 D
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25/91/01/211/000/030 Sling system Inspection.	AMM 25.91.01.211	// 2 Y 500 SC	73 D 50 SC
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25/91/01/721/000/000 Cargo hook Functional test.	AMM 25.91.01.721	7 D	0
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25-95 FOOTSTEPS

25/95/01/000/000/000 VIP footsteps 2000 Inspection.	AMM 25.95.01.211	// 6000 FH 12 Y	300 FH 180 D
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 26-FIRE PROTECTION

26-21 ENGINE FIRE EXTINGUISHING

26/21/00/604/000/011			
Engine fire extinguisher - Cylinder	AMM 26.21.00.222	PO 1 Y	36 D
10230-01 (704A42820049)			
After the introduction to service of a new, overhauled or repaired component.			
Weighing.			

26/21/00/211/000/000			
Engine fire extinguishing system	AMM 26.21.00.211	100 FH // 1 Y	10 FH 36 D
Inspection.			

26/21/00/221/000/000			
Engine fire extinguishing system	AMM 26.21.00.721 26.21.00.721.002	2 Y	73 D
Functional test of the indication system.			

26/21/00/221/000/010			
Check valve	AMM 26.21.00.221	2 Y	73 D
Check with removal.			

26/21/00/401/000/011			
Engine fire extinguisher - Cylinder		5 Y	0
1023000 (704A42820043)			
Cylinder to be proof tested by the manufacturer or an approved repair shop.			
TSM			

26/21/00/401/000/013			
Engine fire extinguisher - Cylinder		7 Y	0
10230-01 (704A42820049)			
Cylinder to be proof tested by the manufacturer or an approved repair shop.			
TSM			

Task Number Description/Remarks	Documentation	Interval	Margin
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26/21/00/602/000/016 Fire extinguishing system - Squib	<i>AMM 26.21.00.761</i>	2 Y	<i>73 D</i>
863080-00 (704A38710040)			
863085-00 (704A38710041)			
Check.			

26/21/00/721/000/000 Engine fire extinguishing system	<i>AMM 26.21.00.721 26.21.00.721.001</i>	2 Y	<i>73 D</i>
10230-01 (704A42820049)			
Functional tests of the system control.			

26-22 CABIN FIRE EXTINGUISHING

26/22/00/000/000/010 Hand fire extinguisher - Cylinder	<i>CMM 26.24.04</i>	6 M	<i>18 D</i>
H1-10AIR (704A32810008)			
Weigh the cylinder.			

26/22/00/000/000/030 Hand fire extinguisher	<i>AMM 26.22.01.211</i>	1 M	<i>0</i>
P3APP003010A (U262A50T1001)			
Check.			

26/22/00/000/000/040 Hand fire extinguisher	<i>AMM 26.22.01.222</i>	24 M	<i>0</i>
P3APP003010A (U262A50T1001)			
Check.			

26/22/00/520/221/100 Hand fire extinguisher - Cylinder	<i>AMM 26.22.01.221</i>	1 Y	<i>36 D</i>
12085-01 (S262A10T1001)			
Weigh the cylinder.			

26/22/01/000/000/005 Hand fire extinguisher	<i>AMM 26.22.01.221</i>	2 Y	<i>73 D</i>
74-00 (800G6799980100)			
Detailed inspection.			

Task Number Description/Remarks	Documentation	Interval	Margin
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26/22/01/000/000/010 Hand fire extinguisher			
74-00 (800G6799980100)		5 Y	180 D
Retest to be carried out by the manufacturer or at an accredited repair facility.			
TSM			

26-23 BAGGAGE COMPARTMENT FIRE EXTINGUISHING

26/23/01/221/000/005 Fire extinguisher - Cylinder			
864250-01 (704A42820048)	AMM 26.23.01.221	PO 1 Y	36 D
After the introduction to service of a new, overhauled or repaired component.			
Weighing.			

26/23/01/000/000/015 Fire extinguisher - Cylinder			
864250-00 (704A42820042)		5 Y	0
Cylinder to be proof tested by the manufacturer or an approved repair shop.			
TSM			

26/23/01/000/000/020 Fire extinguisher - Cylinder			
864250-01 (704A42820048)		7 Y	0
Cylinder to be proof tested by the manufacturer or an approved repair shop.			
TSM			

26/23/01/761/000/000 Fire extinguisher system - Squib			
863080-00 (704A38710040)	AMM 26.23.01.761	2 Y	73 D
863085-00 (704A38710041)			
Check.			

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 28-FUEL

28-10 STORAGE

28/10/00/000/000/010 Fuel tank Tropical and damp atmosphere Check and maintenance.	MTC 20.08.06.401	3 M	9 D
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28-12 OPT - SUPPLEMENTARY FUEL TANKS

28/12/01/221/000/000 Supplementary fuel tank Examine.	AMM 28.12.01.221	1200 FH	120 FH
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28-14 TANK

28/14/00/721/000/000 Feeder tank - Bleed valve Test.	AMM 28.14.00.721	1200 FH	120 FH
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28-20 FUEL DISTRIBUTION

28/20/01/142/000/045 Tank Each time after the component is installed. Cleaning of the ejector.	AMM 28.20.01.141	PO 3 FH	7 FH
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28/20/00/211/000/000 Engine supply line Check.	AMM 28.20.00.211	1200 FH	120 FH
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28/20/00/211/000/010 Transfer pump Check according to the task 28-19-03-200-801-A01 of the CMM.	CMM 28.19.03	2400 FH	240 FH
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Task Number Description/Remarks	Documentation	Interval	Margin
28/20/00/721/000/000 Fuel transfer electric pump Functional test.	AMM 28.20.03.721 GRT 03	600 FH	60 FH
28/20/00/721/000/020 Fuel shut-off valves Functional test.	AMM 28.20.02.721	1200 FH	120 FH
28/20/01/141/000/000 Jet pump filter Clean.	AMM 28.20.01.141	6000 FH	300 FH

28-40 INDICATING

28/40/00/761/000/000 Diodes on fuel management panel Inspection.	AMM 28.40.00.761 GRT 03	600 FH	60 FH
28/40/01/721/000/000 Fuel quantity indicator Check.	AMM 28.41.01.821	6000 FH	300 FH
28/40/03/821/000/000 Feeder tank low-level sensor Do a functional test.	AMM 28.41.03.721	6 Y	180 D

28-42 OPT - FUEL FLOW CONTROL

28/42/00/211/000/000 Fuel flowmeter Check of the installation.	AMM 28.42.00.211	3000 FH	300 FH
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Task Number Description/Remarks	Documentation	Interval	Margin
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28-93 OPT - FUEL ANTI-ICING SYSTEM

28/93/01/211/000/000 Fuel anti-icing system Check.	AMM 28.93.01.211 GRT 03	600 FH	60 FH
28/93/01/721/000/000 Fuel anti-icing thermal switch Control of the thermal switches.	AMM 28.93.01.721	1200 FH	120 FH

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 29-HYDRAULIC POWER

29-00 HYDRAULIC

29/00/00/211/000/000 Hydraulic power system Inspection.	AMM 29.00.00.211	// 100 FH 1 Y	10 FH 36 D
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29-10 MAIN HYDRAULIC SYSTEM

29/10/00/000/000/030 Hydraulic reservoir and manifold assy - Filter Each time after the component is installed. Check the position of the clogging indicators. Operation to be performed after work on the hydraulic system.	AMM 29.00.00.211	PO 2 FH	3 FH
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29/10/00/211/000/000 MGB compartment hydraulic pipes PRE MOD 0729B68 Inspection. If a fault is found, carry out the inspection every 15 FH.	AMM 29.10.00.211 GRT 03	600 FH	60 FH
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29-11 RIGHT MAIN HYDRAULIC SYSTEM

29/11/00/612/000/000 Right hydraulic system Drain and fill the system.	AMM 29.11.00.612	// 2400 FH 4 Y	240 FH 146 D
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29/11/00/612/000/010 Right hydraulic system Salt-laden atmosphere Draining and filling.	AMM 29.11.00.612	2 Y	73 D
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29/11/00/640/000/000 Right hydraulic pump Greasing of the splines of the drive shaft.	AMM 29.11.00.062	1200 FH	120 FH
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Task Number Description/Remarks	Documentation	Interval	Margin
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29-12 LEFT MAIN HYDRAULIC SYSTEM

29/12/00/211/000/000 Left hydraulic system - Hose PRE MOD 0779B33 Check.	AMM 29.12.00.211	100 FH	10 FH
29/12/00/612/000/000 Left hydraulic system Drain and fill the system.	AMM 29.12.00.612	2400 FH // 4 Y	240 FH 146 D
29/12/00/612/000/010 Left hydraulic system Salt-laden atmosphere Draining and filling.	AMM 29.12.00.612	2 Y	73 D
29/12/00/640/000/000 Left hydraulic pump Greasing of the splines of the drive shaft.	AMM 29.12.00.062	1200 FH	120 FH

29-20 AUXILIARY HYDRAULIC SYSTEM

29/20/00/000/000/000 Standby electric pump Check of brushes and drive.	CMM 29.29.11	3000 FH	300 FH
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 30-ICE AND RAIN PROTECTION

30-42 WINDSHIELD WIPERS

30/42/00/711/000/005 Windshield wiper Do a functional test.	<i>AMM 30.42.00.721</i>	1200 FH	120 FH
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30-43 WINDSHIELD WASHER

30/43/00/211/000/005 Windshield washer Do an inspection of the system.	<i>AMM 30.43.01.211</i>	1200 FH	120 FH
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 31-IND./RECORDING SYSTEMS

31-00 INDICATING/RECORDING SYSTEMS

31/00/00/721/000/000 Central panel display system (CPDS) Functional test.	<i>AMM 31.60.01.721</i> <i>GRT 05</i>	600 FH	60 FH
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31-33 VIDEO RECORDING

31/33/00/520/211/100 Camera cockpit Vision 1000. Camera data recovery.	<i>AMM 31.33.00.725</i>	1 Y	36 D
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 32-LANDING GEAR

32-00 LANDING GEAR

32/00/00/000/000/000 Landing gear	<i>AMM 12.30.00.611</i>	7 D	0
Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere Rinse.			

32-10 MAIN LANDING GEAR

32/10/00/221/000/000 Main landing gear	<i>AMM 32.10.00.221</i>	1200 FH // 4 Y	120 FH 146 D
Inspection.			

32/10/00/221/000/001 Main landing gear	<i>AMM 32.10.00.221</i>	1200 FH // 2 Y	120 FH 73 D
Tropical and damp atmosphere Salt-laden atmosphere Inspection.			

32/10/00/221/000/010 Main landing gear	<i>AMM 32.10.00.613</i> <i>GRT 03</i>	600 FH // 1 Y	60 FH 36 D
Cleaning and greasing.			

32/10/00/611/001/000 Main landing gear	<i>AMM 32.10.00.611</i> <i>32.10.00.611.001</i> <i>GRT 03</i>	2 Y	73 D
Maintenance of microswitches.			

32/10/01/520/612/100 Main landing gear	<i>AMM 32.10.01.612</i>	1200 FH // 2 Y	120 FH 73 D
Shock absorber filling.			

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>32/10/01/781/000/000 Main landing gear - Shock absorbers Very cold weather Check inflation pressure, with the helicopter on jacks.</p>	<p><i>AMM 32.10.01.611</i> <i>32.10.01.611.001</i></p>	<p>50 FH</p>	<p>5 FH</p>
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<p>32/10/02/611/000/000 Actuating strut Check and lubricate swivel bearings.</p>	<p><i>AMM 32.10.02.611</i></p>	<p>1200 FH // 4 Y</p>	<p>120 FH 146 D</p>
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<p>32/10/02/611/000/005 Actuating strut Tropical and damp atmosphere Salt-laden atmosphere Check and lubricate swivel bearings.</p>	<p><i>AMM 32.10.02.611</i></p>	<p>1200 FH // 2 Y</p>	<p>120 FH 73 D</p>
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32-20 NOSE LANDING GEAR

<p>32/20/00/221/000/000 Nose landing gear Inspection.</p>	<p><i>AMM 32.20.00.221</i></p>	<p>1200 FH // 4 Y</p>	<p>120 FH 146 D</p>
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<p>32/20/00/221/000/001 Nose landing gear Tropical and damp atmosphere Salt-laden atmosphere Inspection.</p>	<p><i>AMM 32.20.00.221</i></p>	<p>1200 FH // 2 Y</p>	<p>120 FH 73 D</p>
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<p>32/20/00/221/000/010 Nose landing gear Cleaning and greasing.</p>	<p><i>AMM 32.20.00.613</i> <i>GRT 03</i></p>	<p>600 FH // 1 Y</p>	<p>60 FH 36 D</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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32/20/00/611/001/000 Nose landing gear Maintenance of microswitches.	AMM 32.20.00.611 32.20.00.611.001 GRT 03	2 Y	73 D
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32/20/01/520/612/100 Nose landing gear Shock absorber filling.	AMM 32.20.01.612	1200 FH // 2 Y	120 FH 73 D
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32/20/03/611/000/000 Actuating strut Check and lubricate swivel bearings.	AMM 32.20.03.611	1200 FH // 4 Y	120 FH 146 D
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32/20/03/611/000/005 Actuating strut Tropical and damp atmosphere Salt-laden atmosphere Check and lubricate swivel bearings.	AMM 32.20.03.611	1200 FH // 2 Y	120 FH 73 D
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32-30 EXTENTION AND RETRACTION

32/30/00/221/000/000 Extension/retraction control system Inspection.	AMM 32.30.00.221	1200 FH // 4 Y	120 FH 146 D
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32/30/00/221/000/005 Extension/retraction control system Tropical and damp atmosphere Salt-laden atmosphere Inspection.	AMM 32.30.00.221	1200 FH // 2 Y	120 FH 73 D
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Task Number Description/Remarks	Documentation	Interval	Margin
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<p>32/30/00/721/000/000 Landing gear Do a retraction and extension test.</p>	<p><i>AMM 32.30.00.721</i></p>	<p>1200 FH // 4 Y</p>	<p>120 FH 146 D</p>
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<p>32/30/00/721/000/005 Landing gear Tropical and damp atmosphere Salt-laden atmosphere Do a retraction and extension test.</p>	<p><i>AMM 32.30.00.721</i></p>	<p>1200 FH // 2 Y</p>	<p>120 FH 73 D</p>
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<p>32/30/00/722/000/000 Landing gear Do a test of retraction and extension safety systems.</p>	<p><i>AMM 32.30.00.722</i></p>	<p>1200 FH // 4 Y</p>	<p>120 FH 146 D</p>
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<p>32/30/00/722/000/005 Landing gear Tropical and damp atmosphere Salt-laden atmosphere Do a test of retraction and extension safety systems.</p>	<p><i>AMM 32.30.00.722</i></p>	<p>1200 FH // 2 Y</p>	<p>120 FH 73 D</p>
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32-40 WHEELS AND BRAKES

<p>32/40/00/211/000/000 Skis installation Each time after the component is installed. System inspection.</p>	<p><i>AMM 32.40.06.211</i></p>	<p>PO 45 FH</p>	<p>5 FH</p>
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<p>32/40/00/211/000/010 Wheel and tire Inspection.</p>	<p><i>AMM 32.40.00.211</i> <i>GRT 03</i></p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
32/40/00/211/000/015 Wheel and tire Tropical and damp atmosphere Salt-laden atmosphere Inspection.	AMM 32.40.00.211 GRT 03	600 FH // 1 Y	60 FH 36 D
32/40/00/212/000/000 Wheel Examine the half-rims.	AMM 32.40.00.212	1200 FH // 4 Y	120 FH 146 D
32/40/00/212/000/005 Wheel Tropical and damp atmosphere Salt-laden atmosphere Examine the half-rims.	AMM 32.40.00.212	1200 FH // 2 Y	120 FH 73 D
32/40/00/213/000/000 Brake system Inspection.	AMM 32.40.00.213 32.40.00.213.003 GRT 03	600 FH // 2 Y	60 FH 73 D
32/40/00/213/000/005 Brake system Tropical and damp atmosphere Salt-laden atmosphere Inspection.	AMM 32.40.00.213 32.40.00.213.003 GRT 03	600 FH // 1 Y	60 FH 36 D
32/40/00/520/211/000 Ski installation - Main landing gear PRE MOD 0732B47 Check with removal of the attachment bolt of the main landing gear lower scissor with skis equipped. This check to be performed at each removal of the installation.	AMM 32.40.06.211	25 FH	2 FH

Task Number Description/Remarks	Documentation	Interval	Margin
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32/40/03/721/000/000 Brake system Tests.	AMM 32.40.03.721	1200 FH // 4 Y	120 FH 146 D
32/40/03/721/000/005 Brake system Tropical and damp atmosphere Salt-laden atmosphere Tests.	AMM 32.40.03.721	1200 FH // 2 Y	120 FH 73 D
32/40/05/000/000/005 Shoe Anti-sinking shoes. Check.	AMM 32.40.05.211	100 FH // 1 Y	10 FH 36 D
32/40/06/520/611/100 Skis installation POST MOD OP32B31 & // POST MOD OP32B42 Greasing the bolts of the reinforced scissors of the optional ski installation.	AMM 32.40.06.611 GRT 03	600 FH // 1 Y	60 FH 36 D

32-60 POSITION INDICATING AND WARNINGS

32/60/00/721/000/000 Landing gear not extended audio warning device Functional test.	AMM 32.60.00.721 GRT 05	600 FH	60 FH
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32-70 LANDING GEAR PROTECTION

32/70/00/000/000/000 Tail guard Check of the tail guard support assembly.	AMM 32.70.01.221	2 Y	73 D
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Task Number Description/Remarks	Documentation	Interval	Margin
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32/70/00/000/000/005 Tail guard Check.	AMM 32.70.01.223	6000 FH	300 FH
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 33-LIGHTS

33-00 LIGHTS

<p>33/00/00/211/000/000 Lights Inspection.</p>	<p>AMM 33.00.00.211 GRT 06</p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>
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33-43 OPT - SEARCHLIGHT

<p>33/43/00/520/610/100 Retractable landing lights 4168748 (704A46820017) 4274479 (704A46820026) 4307405 (704A46820034) Check and cleaning.</p>	<p>AMM 33.43.00.610</p>	<p>1 Y</p>	<p>36 D</p>
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<p>33/43/07/520/221/100 Searchlight SPECTROLAB SX-16. Detailed inspection.</p>	<p>AMM 33.43.07.221 GRT 06</p>	<p>600 FH</p>	<p>60 FH</p>
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<p>33/43/07/520/222/100 Searchlight SPECTROLAB SX-16. Check the complete searchlight assembly.</p>	<p>AMM 33.43.07.222</p>	<p>300 FH</p>	<p>30 FH</p>
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<p>33/43/08/520/000/100 Searchlight TRAKKA A800. Check the hard stops.</p>	<p>CMM 33.45.93 PARA 8</p>	<p>100 FH</p>	<p>10 FH</p>
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<p>33/43/08/520/721/200 Searchlight - LSU TRAKKA A800. Check.</p>	<p>AMM 33.43.08.721 33.43.08.721.001 GRT 06</p>	<p>600 FH</p>	<p>60 FH</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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33-50 EMERGENCY LIGHTING

33/50/03/000/000/050

HEEL system - Battery

AMM 33.50.03.721 **PO** (*) 0

- (*) First installation on aircraft or after a replacement of PSU.
- (*) If the aircraft has not been used during more than 90 days in normal conditions, apply PSU battery maintenance before using again the aircraft.
- (*) If the aircraft has not been used during more than 40 days, and hot conditions occurred during this time, apply PSU battery maintenance before using again the aircraft.
- Functional test.

33/50/03/000/000/020

HEEL cabin roof - Battery

AMM 33.50.03.062 **18 M** 54 D

5776-1 (-)

Battery capacity check.

33/50/03/000/000/025

HEEL system

AMM 33.50.03.721 **18 M** 54 D

Functional test.

33/50/03/520/000/100

HEEL cabin roof - Converter unit

AMM 33.50.03.721 **18 M** 54 D

6958-2 (706A36140000)

Check.

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 34-NAVIGATION

34-00 NAVIGATION

<p>34/00/00/211/000/000 Navigation system Inspection.</p>	<p>AMM 34.00.00.211</p>	<p>// 100 FH 1 Y</p>	<p>10 FH 36 D</p>
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34-11 AIR DATA SYSTEM

<p>34/11/00/213/000/000 Air data system Detailed inspection.</p>	<p>AMM 34.11.00.221 GRT 05</p>	<p>600 FH</p>	<p>60 FH</p>
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<p>34/11/02/791/000/000 Air data system Leak test.</p>	<p>AMM 34.11.02.791</p>	<p>2 Y</p>	<p>73 D</p>
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34-42 WEATHER RADAR

<p>34/42/01/211/000/020 Radar 1400 Visual inspection of the antenna mechanism.</p>	<p>AMM 34.42.01.211 GRT 05</p>	<p>// 600 FH 2 Y</p>	<p>60 FH 73 D</p>
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<p>34/42/03/520/000/100 Radar 1600 Check.</p>	<p>AMM 34.42.03.211 GRT 05</p>	<p>// 1200 FH 3 Y</p>	<p>120 FH 109 D</p>
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34-70 DISPLAY

<p>34/70/01/721/000/000 Instrument control SMD - Functional test.</p>	<p>AMM 34.70.01.721</p>	<p>// 100 FH 1 Y</p>	<p>10 FH 36 D</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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34/70/01/721/000/010 Instrument control RCU - Functional test.	AMM 34.70.03.721	100 FH // 1 Y	10 FH 36 D
34/70/01/721/000/015 PELICAN rack fans Functional test.	AMM 34.70.05.721 GRT 05	600 FH	60 FH

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 45-CENTRAL MAINT SYSTEM

45-11 OPT - CVFDR

<p>45/11/01/520/000/000 SSCVFDR recorder - Locator beacon</p> <p>980-6021-032 (704A45381025) 980-6021-066 (704A45381029)</p> <p>SSCVFDR Honeywell AR-204C or AR-602C equipped of the underwater locator beacon: DATASONIC ELP-362D [30 days], or DUKANE DK100, or DUKANE DK120 [30 days], or DUKANE DK120/90 [90 days], or Honeywell 316-6004-009 [90 days]. Refer to the MSM chapter 05-20, ATA 25-66 for the fitted emergency locator beacon maintenance.</p>		<p>2 Y</p>	<p>0</p>
<p>45/11/01/520/721/000 CVR recorder</p> <p>980-6021-032 (704A45381025) 980-6021-066 (704A45381029)</p> <p>SSCVFDR AR-602C or AR-204C. Test of the audio signal acquisition. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 18 D)</p>	<p><i>AMM 45.11.01.721 45.11.01.721.002</i></p>	<p>6 M</p>	<p>0 *</p>
<p>45/11/01/520/722/000 CVR recorder</p> <p>980-6021-032 (704A45381025) 980-6021-066 (704A45381029)</p> <p>SSCVFDR AR-602C or AR-204C. Test of the erase function. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 18 D)</p>	<p><i>AMM 45.11.01.722</i></p>	<p>6 M</p>	<p>0 *</p>

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>45/11/01/520/723/000 Immersion probe 980-6021-032 (704A45381025) 980-6021-066 (704A45381029) SSCVFDR AR-602C or AR-204C. Functional test.</p>	<p>AMM 45.11.01.723 GRT 05</p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>
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<p>45/11/01/520/724/000 CVR recorder 980-6021-032 (704A45381025) 980-6021-066 (704A45381029) SSCVFDR AR-602C or AR-204C. Check and replay of the recording. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 36 D)</p>	<p>AMM 45.11.01.724 45.11.01.724.003</p>	<p>1 Y</p>	<p>0 *</p>
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<p>45/11/01/520/724/010 FDR recorder 980-6021-032 (704A45381025) 980-6021-066 (704A45381029) SSCVFDR AR-602C or AR-204C. Check and replay of the recording. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 36 D)</p>	<p>AMM 45.11.01.724 45.11.01.724.001</p>	<p>1 Y</p>	<p>0 *</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
45/11/04/520/000/200 SSCVFDR recorder - Locator beacon SSCVFDR L3Com FA5001 equipped of the underwater locator beacon: L-3xx 266-E0305-00 [30 days], or L-3xx 266-E5542-00 [90 days]. Refer to the MSM chapter 05-20, ATA 25-66 for the fitted emergency locator beacon maintenance.		2 Y	0
45/11/04/520/724/200 CVR recorder 5001-6103-11 (704A45381032) SSCVFDR FA5001. Test of the erase function. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 18 D)	AMM 45.11.04.724	6 M	0 *
45/11/04/520/725/200 SSCVFDR recorder 5001-6103-11 (704A45381032) Functional tests.	AMM 45.11.04.725 GRT 05	600 FH // 2 Y	60 FH 73 D
45/11/04/520/726/200 CVR recorder 5001-6103-11 (704A45381032) SSCVFDR FA5001. Test of the audio signal acquisition. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 18 D)	AMM 45.11.04.726 45.11.04.726.004	6 M	0 *
45/11/04/520/726/205 SSCVFDR recorder 5001-6103-11 (704A45381032) SSCVFDR FA5001. Check the power supply.	AMM 45.11.04.726 45.11.04.726.004	2 Y	73 D

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>45/11/04/520/726/210 SSCFDR recorder 5001-6103-11 (704A45381032) SSCVFDR FA5001. Check of indicators.</p>	<p>AMM 45.11.04.726 45.11.04.726.003</p>	<p>2 Y</p>	<p>73 D</p>
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<p>45/11/04/520/727/200 CVR recorder 5001-6103-11 (704A45381032) SSCVFDR FA5001. Check and replay of the recording. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 36 D)</p>	<p>AMM 45.11.04.727 45.11.04.727.004</p>	<p>1 Y</p>	<p>0 *</p>
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<p>45/11/04/520/727/210 FDR recorder 5001-6103-11 (704A45381032) SSCVFDR FA5001. Check and replay of the recording. (*) Margin = 0 or margin defined according to regulations of the regulatory authority (maximum allowable margin: 36 D)</p>	<p>AMM 45.11.04.727 45.11.04.727.001</p>	<p>1 Y</p>	<p>0 *</p>
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 52-DOORS

52-00 DOORS - GENERAL

52/00/00/211/000/000 Doors Inspection.	AMM 52.00.00.211	100 FH	10 FH
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52-11 CREW DOORS

52/11/00/221/000/000 Crew doors Detailed inspection.	AMM 52.11.00.221 GRT 06	// 600 FH 2 Y	60 FH 73 D
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52-12 PASSENGER DOORS

52/12/00/221/000/010 Sliding doors PRE MOD 52C23 Detailed inspection.	ASB 52A018		
52/12/00/520/211/100 Sliding doors POST MOD 52C23 Visual check of doors.	AMM 52.12.00.211 GRT 06	600 FH	60 FH
52/12/00/520/221/100 Sliding doors POST MOD 52C23 Detailed inspection.	AMM 52.12.00.221 GRT 06	// 600 FH 2 Y	60 FH 73 D
52/12/00/520/641/100 Sliding doors POST MOD 52C23 Greasing.	AMM 52.12.00.641 GRT 06	// 600 FH 2 Y	60 FH 73 D

Task Number Description/Remarks	Documentation	Interval	Margin
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52/12/01/221/000/000 Hinged doors System detailed inspection.	AMM 52.12.01.221 52.12.01.221.001	200 FH	20 FH
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52/12/01/520/000/100 VIP hinged door POST MOD OP53C80 Maintenance to be performed according to SB.	SB 52-029		
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52-30 Cargo door

52/30/00/520/221/000 Compartment and battery door locks POST MOD 0752C68 Check.	AMM 52.00.00.221 GRT 06	600 FH // 2 Y	60 FH 73 D
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 53-FUSELAGE

53-00 FUSELAGE

53/00/00/000/000/100 Structure Salt-laden atmosphere Cleaning.	<i>AMM 12.30.00.611</i>	7 D	<i>0</i>
53/00/00/000/000/105 Structure Salt-laden atmosphere Rinsing with fresh water. Jet of water with pressure not exceeding 10 bar. Dry areas where water may accumulate and bottom structure airframe (outside).	<i>MTC 20.04.01.403</i>	ALF	<i>0</i>
53/00/00/000/000/110 Structure - Bead Tropical and damp atmosphere Salt-laden atmosphere Check of sealant beads.	<i>AMM 53.00.00.211</i> <i>53.00.00.211.001</i>	100 FH // 6 M	<i>10 FH</i> <i>18 D</i>
53/00/00/211/000/000 Structure and cowlings Inspection.	<i>AMM 53.00.00.211</i>	100 FH // 1 Y	<i>10 FH</i> <i>36 D</i>

53-10 PRIMARY STRUCTURE

53/10/00/221/000/000 Forward bottom structure Detailed inspection.	<i>AMM 53.10.00.221</i> <i>GRT 06</i>	600 FH	<i>60 FH</i>
53/10/00/221/000/010 Forward bottom structure - Upper flanges of longitudinal beams Visual check.	<i>AMM 53.10.00.223</i>	2 Y	<i>73 D</i>

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>53/10/00/221/000/025 Transmission deck</p> <p>Detailed inspection.</p>	<p><i>AMM 53.40.00.221</i> <i>GRT 06</i></p>	<p>600 FH</p>	<p><i>60 FH</i></p>
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<p>53/10/00/520/222/100 Forward Bottom Structure</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.</p>	<p><i>AMM 53.10.00.222</i></p>	<p>10 Y // 14 Y</p>	<p><i>180 D</i> <i>180 D</i></p>
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<p>53/10/00/520/223/100 Engine deck</p> <p>Detailed inspection.</p>	<p><i>AMM 53.00.00.223</i></p>	<p>7200 FH</p>	<p><i>300 FH</i></p>
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53-20 CANOPY AND NOSE STRUCTURE

<p>53/20/00/221/000/121 Canopy and nose structure</p> <p>PRE MOD 0756B32 Detailed inspection.</p>	<p><i>AMM 53.20.00.221 (A)</i> <i>53.20.00.221.002</i> <i>GRT 06</i></p>	<p>600 FH</p>	<p><i>60 FH</i></p>
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<p>53/20/00/221/000/122 Canopy and nose structure</p> <p>POST MOD 0756B32 Detailed inspection.</p>	<p><i>AMM 53.20.00.221 (B)</i> <i>53.20.00.221.002</i> <i>GRT 06</i></p>	<p>600 FH</p>	<p><i>60 FH</i></p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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<p>53/20/00/520/223/101</p> <p>Cabin/carbon structure attachment</p> <p>PRE MOD 0756B32</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection of the canopy without removal.</p>	<p>AMM 53.20.00.223 (A) 53.20.00.223.001</p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>
<p>53/20/00/520/223/102</p> <p>Cabin/carbon structure attachment</p> <p>POST MOD 0756B32</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection of the canopy without removal.</p>	<p>AMM 53.20.00.223 (B) 53.20.00.223.001</p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>
<p>53/20/00/520/223/201</p> <p>25° frame</p> <p>PRE MOD 0756B32</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.</p>	<p>AMM 53.20.00.223 (A) 53.20.00.223.002</p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>
<p>53/20/00/520/223/202</p> <p>25° frame</p> <p>POST MOD 0756B32</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.</p>	<p>AMM 53.20.00.223 (B) 53.20.00.223.002</p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>53/20/00/520/223/301 12° frame</p> <p>PRE MOD 0756B32 Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.</p>	<p>AMM 53.20.00.223 (A) 53.20.00.223.003</p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>
<p>53/20/00/520/223/302 12° frame</p> <p>POST MOD 0756B32 Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.</p>	<p>AMM 53.20.00.223 (B) 53.20.00.223.003</p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>

53-30 CABIN STRUCTURE

<p>53/30/00/221/000/130 Cabin structure</p> <p>Detailed inspection.</p>	<p>AMM 53.30.00.221 GRT 06</p>	<p>600 FH</p>	<p>60 FH</p>
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53-40 TRANSMISSION DECK

<p>53/40/00/520/222/100 Transmission deck</p> <p>Torque check.</p>	<p>AMM 53.40.00.222 53.40.00.222.001</p>	<p>6000 FH</p>	<p>300 FH</p>
<p>53/40/00/520/222/200 Transmission deck</p> <p>Detailed inspection of the upper face and the lower faces.</p>	<p>AMM 53.40.00.222 53.40.00.222.002 53.40.00.222.003</p>	<p>7200 FH</p>	<p>300 FH</p>

Task Number Description/Remarks	Documentation	Interval	Margin
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53-50 COWLINGS AND FAIRINGS

53/50/00/221/000/150 Cowlings and fairings Detailed inspection.	AMM 53.50.00.221 GRT 06	600 FH // 2 Y	60 FH 73 D
53/50/00/221/000/155 MGB cowlings Inspection of the locks and hinges.	AMM 53.50.00.221 53.50.00.221.001	100 FH	10 FH
53/50/00/520/000/100 Cowling attachments Detailed inspection of the cowling attachments and hinge points.	AMM 53.50.00.222	6000 FH	300 FH

53-54 ENGINE COWLING

53/54/00/211/000/000 Aft removable cowling - Heat shields equipped or MI15 protection If defects are detected, perform this check during daily check.	AMM 53.54.00.211	100 FH	10 FH
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53-60 AFT STRUCTURE

53/60/00/221/000/160 Aft structure assembly Detailed inspection.	AMM 53.60.00.221 GRT 06	600 FH	60 FH
53/60/00/520/222/100 Frame X 4630 Detailed inspection of frame X 4630 on the cabin and baggage compartment side.	AMM 53.60.00.222 53.60.00.222.002	7200 FH	300 FH
53/60/00/520/222/200 9° frame Detailed inspection.	AMM 53.60.00.222 53.60.00.222.003	7200 FH	300 FH

Task Number Description/Remarks	Documentation	Interval	Margin
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53/60/00/520/222/300			
Feeder tank mounting plate assemblies			
	<i>AMM 53.60.00.222</i>	10 Y	180 D
	<i>53.60.00.222.004</i>	// 14 Y	180 D
Tropical and damp atmosphere			
Salt-laden atmosphere			
Sand-laden and/or dust-laden atmosphere			
Operation to be performed every 10 Y if use in climatic conditions specified above.			
Operation to be performed every 14 Y if use in other climatic conditions.			
Detailed inspection.			

53/60/00/520/222/400			
Frame X 4630			
	<i>AMM 53.60.00.222</i>	10 Y	180 D
	<i>53.60.00.222.001</i>	// 14 Y	180 D
Tropical and damp atmosphere			
Salt-laden atmosphere			
Sand-laden and/or dust-laden atmosphere			
Operation to be performed every 10 Y if use in climatic conditions specified above.			
Operation to be performed every 14 Y if use in other climatic conditions.			
Detailed inspection of the lower part.			

53/60/00/520/223/100			
Main landing gear leg fittings			
	<i>AMM 53.60.00.223</i>	10 Y	180 D
	<i>53.60.00.223.002</i>	// 14 Y	180 D
Tropical and damp atmosphere			
Salt-laden atmosphere			
Sand-laden and/or dust-laden atmosphere			
Operation to be performed every 10 Y if use in climatic conditions specified above.			
Operation to be performed every 14 Y if use in other climatic conditions.			
Detailed inspection.			

53/60/00/520/223/200			
Main landing gear brace strut actuator fittings			
	<i>AMM 53.60.00.223</i>	10 Y	180 D
	<i>53.60.00.223.001</i>	// 14 Y	180 D
Tropical and damp atmosphere			
Salt-laden atmosphere			
Sand-laden and/or dust-laden atmosphere			
Operation to be performed every 10 Y if use in climatic conditions specified above.			
Operation to be performed every 14 Y if use in other climatic conditions.			
Detailed inspection.			

Task Number Description/Remarks	Documentation	Interval	Margin
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53/60/00/520/225/100 Aft floatability cradles	<i>AMM 53.60.00.225</i>	8 Y // 14 Y	<i>180 D</i> <i>180 D</i>
Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere Operation to be performed every 8 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.			

53/60/00/520/226/000 Baggage compartment - External skin	<i>AMM 53.60.00.226</i> <i>53.60.00.226.001</i>	100 FH	<i>10 FH</i>
Visual inspection.			

53/60/00/520/226/005 Baggage compartment - External skin	<i>AMM 53.60.00.226</i> <i>53.60.00.226.002</i> <i>GRT 06</i>	600 FH	<i>60 FH</i>
Tapping inspection. If there is a new separation, proceed to 3 inspections by tapping at 100 FH (+10 FH) / 200 FH (+20FH) / 300 FH (+30 FH). If there is no evolution of the separation, continue the inspections according to the standard schedule.			

53-70 FLOORS

53/70/00/221/000/170 Cabin and baggage compartment floors	<i>AMM 53.70.00.221</i> <i>GRT 06</i>	600 FH	<i>60 FH</i>
Detailed inspection.			

53/70/00/520/222/100 Chords of Forward	<i>AMM 53.70.00.222a</i>	8 Y	<i>180 D</i>
PRE MOD 0753C71 Detailed inspection of the flanges.			

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>53/70/00/520/222/200</p> <p>Chords of Forward</p> <p>POST MOD 0753C71 Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere Operation to be performed every 10 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection of the flanges.</p>	<p><i>AMM 53.70.00.222b</i></p>	<p>10 Y // 14 Y</p>	<p>180 D 180 D</p>
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53-80 TAIL BOOM

<p>53/80/00/211/000/180</p> <p>Tail boom</p> <p>Detailed inspection.</p>	<p><i>AMM 53.80.00.221</i> <i>GRT 02</i></p>	<p>600 FH</p>	<p>60 FH</p>
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<p>53/80/00/520/211/100</p> <p>Structure to tail boom to fenestron junctions</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere Operation to be performed every 8 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection without removal.</p>	<p><i>AMM 53.80.00.211</i></p>	<p>8 Y // 14 Y</p>	<p>180 D 180 D</p>
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<p>53/80/00/520/222/100</p> <p>Structure to tail boom to fenestron junctions</p> <p>Detailed inspection of the liaisons.</p>	<p><i>AMM 53.80.00.222</i> <i>53.80.00.222.001</i></p>	<p>6000 FH</p>	<p>300 FH</p>
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<p>53/80/00/520/222/200</p> <p>Structure to tail boom to fenestron junctions</p> <p>External and internal detailed inspection of the tail boom.</p>	<p><i>AMM 53.80.00.222</i> <i>53.80.00.222.002</i> <i>53.80.00.222.003</i></p>	<p>7200 FH</p>	<p>300 FH</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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<p>53/80/00/520/222/300</p> <p>Tail boom</p> <p>Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere</p> <p>Operation to be performed every 8 Y if use in climatic conditions specified above. Operation to be performed every 14 Y if use in other climatic conditions. Detailed inspection.</p>	<p><i>AMM 53.80.00.222</i></p>	<p>8 Y // 14 Y</p>	<p>180 D 180 D</p>
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53-90 FENESTRON

<p>53/90/00/211/000/190</p> <p>Fenestron</p> <p>Detailed inspection.</p>	<p><i>AMM 53.90.00.221</i> <i>GRT 02</i></p>	<p>600 FH</p>	<p>60 FH</p>
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<p>53/90/00/520/222/100</p> <p>Shrouded fenestron</p> <p>Detailed inspection.</p>	<p><i>AMM 53.90.00.222</i></p>	<p>8 Y</p>	<p>180 D</p>
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53-92 VERTICAL FIN FENESTRON

<p>53/92/00/520/221/100</p> <p>Carbon tube - TGB mount</p> <p>Detailed inspection.</p>	<p><i>AMM 53.92.00.221</i></p>	<p>7200 FH</p>	<p>300 FH</p>
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53-93 TGB FLARED HOUSING

<p>53/93/00/520/221/100</p> <p>TGB flared coupling tube</p> <p>Detailed inspection of the seating faces.</p>	<p><i>AMM 53.93.00.221</i></p>	<p>7200 FH // 8 Y</p>	<p>300 FH 180 D</p>
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 55-STABILIZER

55-10 HORIZONTAL STABILIZER

<p>55/10/00/221/000/000 Horizontal stabilizer Detailed inspection.</p>	<p>AMM 55.10.00.221 GRT 02</p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>
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55-30 VERTICAL STABILIZER

<p>55/30/00/221/000/010 Out board fins Detailed inspection.</p>	<p>AMM 55.30.00.221 GRT 02</p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 56-WINDOWS

56-00 WINDOWS - GENERAL

56/00/00/520/000/100 Jettisonable windows Jettison check.	<i>AMM 56.00.00.721</i>	2 Y	73 D
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56/00/00/520/211/100 Sliding doors POST MOD 52C23 Check of window extraction ribbon.	<i>AMM 56.00.00.211</i> <i>56.00.00.211.003</i>	100 FH	10 FH
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 60-STD. PRACTS. PROP/ROTOR

60-00 ROTOR

<p>60/00/00/000/000/000</p> <p>MRH - Engine/MGB coupling - Tail rotor drive shaft - TRH</p> <p style="text-align: right;"><i>AMM 05.39.00.221</i></p> <p>Tropical and damp atmosphere Salt-laden atmosphere</p> <p>Check to be done from 8 Y (margin 180 D) after the first date of installation.</p>		<p>4 Y</p>	<p>146 D</p>
<p>60/00/00/000/000/010</p> <p>MRH - Engine/MGB coupling - Tail rotor drive shaft - TRH</p> <p style="text-align: right;"><i>AMM 05.39.00.221</i></p> <p>Normal climatic conditions</p> <p>Check to be done from 12 Y (margin 180 D) after the first date of installation.</p>		<p>6 Y</p>	<p>180 D</p>
<p>60/00/00/000/000/015</p> <p>MGB - TGB</p> <p style="text-align: right;"><i>AMM 05.39.00.221</i></p> <p>Tropical and damp atmosphere Salt-laden atmosphere</p> <p>Check to be done from 8 Y (margin 180 D) after the first date of installation or since overhaul.</p>		<p>4 Y</p>	<p>146 D</p>
<p>60/00/00/000/000/055</p> <p>MGB - TGB</p> <p style="text-align: right;"><i>AMM 05.39.00.221</i></p> <p>Normal climatic conditions</p> <p>Check to be done from 12 Y (margin 180 D) after the first date of installation or since overhaul.</p>		<p>6 Y</p>	<p>180 D</p>

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 62-ROTOR (S)

62-10 MAIN ROTOR BLADES

62/10/00/601/000/050

Main rotor blade

		AMM 62.10.00.221	PO	90 FH	10 FH
		62.10.00.221.002	&	180 FH	20 FH
			&	360 FH	40 FH

365A11-0080-00 (-)

365A11-0080-01 (-)

After the introduction to service of a new, overhauled or repaired component.

Remove the blade but not the tip cap.

Servicing.

Visual inspection.

Check the blade by tapping.

62/10/00/000/000/010

Main rotor blade

		AMM 62.10.00.211		15 FH	1 FH
		62.10.00.211.001			

365A11-0080-00 (-)

365A11-0080-01 (-)

Sand-laden and/or dust-laden atmosphere

Visual check for erosion.

62/10/00/211/016/000

Main rotor blade

		AMM 62.10.00.211		100 FH	10 FH
		62.10.00.211.002	//	1 Y	36 D

365A11-0080-00 (-)

365A11-0080-01 (-)

Visual check without blade removal.

62/10/00/221/000/000

Main rotor blade

		AMM 62.10.00.221		600 FH	60 FH
		GRT 01	//	2 Y	73 D

365A11-0080-00 (-)

365A11-0080-01 (-)

Remove the blade but not the tip cap.

If faults are found, do a tapping test, at 50, 100, 150 and 300 FH without blade removal.

If no changes are noted, resume the standard inspection cycle.

Task Number Description/Remarks	Documentation	Interval	Margin
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62/10/00/221/000/005 Main rotor blade - Tip cap	<i>AMM 62.10.00.221</i>	2 Y	73 D
365A11-0080-00 (-)			
365A11-0080-01 (-)			
Check after removing the blade tip cap. If faults are found, do a tapping test, at 50, 100, 150 and 300 FH without blade removal. If no changes are noted, resume the standard inspection cycle.			
62/10/00/301/000/000 Main rotor blade	<i>AMM 60.00.00.101</i>	50 FH	5 FH
365A11-0080-00 (-)			
365A11-0080-01 (-)			
Tropical and damp atmosphere Salt-laden atmosphere Maintenance.			

62-20 MAIN ROTOR HEAD

62/20/00/605/000/000 MRH or MRH component	<i>AMM 62.20.00.222</i>	PO	3 FH	7 FH
Each time after the component is installed. Readjust the tightening torque loads.				
62/20/00/605/000/005 MRH	<i>AMM 12.20.00.211</i>	PO	3 FH	7 FH
Each time after the component is installed. Check electrical magnetic plug.				

62/20/00/000/000/180 Rotor blade pin	<i>AMM 62.10.00.061</i> <i>62.10.00.421.001</i>	300 FH // 1 Y	30 FH 36 D
Greasing.			

Task Number Description/Remarks	Documentation	Interval	Margin
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62/20/00/211/000/020 MRH Inspection.	<i>AMM 62.20.00.211</i>	100 FH // 1 Y	10 FH 36 D
62/20/00/211/000/030 Frequency adapter Check axial play.	<i>AMM 62.20.00.211</i> <i>62.20.00.211.004</i>	50 FH	5 FH
62/20/00/211/000/040 MRH Detailed inspection.	<i>AMM 62.20.00.224</i>	1200 FH	120 FH
62/20/00/221/000/010 MRH Detailed inspection.	<i>AMM 62.20.00.221</i> <i>GRT 01</i>	600 FH	60 FH
62/20/00/223/000/000 MRH Detailed inspection.	<i>AMM 62.20.00.223</i>	3000 FH	300 FH

62-40 INDICATING

62/40/03/211/000/000 NR magnetic speed sensor Functional tests.	<i>AMM 62.40.01.721</i>	3000 FH	300 FH
62/40/03/211/000/010 Flared housing electrical magnetic plug Inspection.	<i>AMM 12.20.00.211</i>	100 FH // 1 Y	10 FH 36 D
62/40/03/221/000/010 Flared housing electrical magnetic plug Functional test.	<i>AMM 62.40.03.721</i> <i>GRT 01</i>	600 FH	60 FH

Task Number Description/Remarks	Documentation	Interval	Margin
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 63-ROTOR DRIVE (S)

63-10 ENGINE/MGB COUPLING

63/10/02/221/002/005			
Flexible coupling attachment bolt on MGB side			
	AMM 63.10.02.221 63.10.02.221.002	PO 5 FH	10 FH
PRE MOD 0763C88			
Check the torque load after installation of a new flexible coupling.			

63/10/62/221/000/040			
Flexible coupling - Inputs and tail output			
	AMM 63.10.62.221 63.10.62.221.001 AMM 65.10.00.222 65.10.00.222.001	PO 5 FH	10 FH
POST MOD 0763C88			
Check the tightening torque.			

63/10/00/213/000/000			
Flexible coupling - Inputs			
	AMM 05.22.00.213	// 10 FH 5 D	5 FH 2 D
POST MOD 0763C88			
Inspection.			

63/10/00/221/000/006			
Engine/MGB coupling - Gimbals pins			
	AMM 63.10.00.221 63.10.00.221.002	100 FH	10 FH
PRE MOD 0763C88			
Check and lubricate.			

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>63/10/00/221/000/015 Engine-MGB coupling PRE MOD 0763C88 Detailed inspection.</p>	<p>AMM 63.10.00.221 GRT 03</p>	<p>600 FH</p>	<p>60 FH</p>
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<p>63/10/00/221/000/020 Engine-MGB coupling POST MOD 0763C88 Detailed inspection.</p>	<p>AMM 63.10.60.221 63.10.60.221.001 GRT 03</p>	<p>600 FH</p>	<p>60 FH</p>
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<p>63/10/02/211/000/000 Flexible coupling (engine side) 365A32-2872-00 (-) 365A32-6530-00 (-) Detailed visual check.</p>	<p>AMM 63.10.02.211</p>	<p>50 FH</p>	<p>5 FH</p>
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<p>63/10/02/221/000/015 Flexible coupling PRE MOD 0763C88 Detailed inspection.</p>	<p>AMM 63.10.02.221 GRT 03</p>	<p>600 FH</p>	<p>60 FH</p>
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<p>63/10/62/221/000/000 Flexible coupling - Inputs and tail output POST MOD 0763C88 Detailed inspection.</p>	<p>AMM 63.10.62.221 63.10.62.221.002 AMM 65.10.00.222 65.10.00.222.002</p>	<p>600 FH // 4 Y</p>	<p>60 FH 4 M</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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63/10/62/221/000/010 Flexible coupling - Inputs and tail output	AMM 63.10.62.221 63.10.62.221.002 AMM 65.10.00.222 65.10.00.222.002	600 FH // 2 Y	60 FH 2 M
POST MOD 0763C88 Tropical and damp atmosphere Salt-laden atmosphere Detailed inspection.			

63/10/62/221/000/020 Flexible coupling - Inputs and tail output	AMM 63.10.62.221 63.10.62.221.001 AMM 65.10.00.222 65.10.00.222.001	600 FH // 4 Y	60 FH 4 M
POST MOD 0763C88 Check the tightening torque.			

63/10/62/221/000/030 Flexible coupling - Inputs and tail output	AMM 63.10.62.221 63.10.62.221.001 AMM 65.10.00.222 65.10.00.222.001	600 FH // 2 Y	60 FH 2 M
POST MOD 0763C88 Tropical and damp atmosphere Salt-laden atmosphere Check the tightening torque.			

63/10/64/222/000/000 Engine/MGB coupling - Gimbals pins	AMM 63.10.64.222 63.10.64.222.001	100 FH	10 FH
POST MOD 0763C88 Check.			

63-20 MAIN GEARBOX

63/20/00/611/000/001 MGB	AMM 63.20.00.611	PO 45 FH	5 FH
PRE MOD 0763C88 After the introduction to service of a new, overhauled or repaired component. Drain.			

Task Number Description/Remarks	Documentation	Interval	Margin
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63/20/00/611/000/005 MGB POST MOD 0763C88 Drain.	AMM 63.20.00.611	PO 50 FH	5 FH
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63/20/60/215/000/000 MGB POST MOD 0763C88 Each time after the component is installed.	AMM 63.20.60.213 63.20.60.215.001	PO	
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63/20/60/216/000/000 MGB POST MOD 0763C88 Search for seepage or leakage after intervention on MGB oil sealing.	AMM 63.20.60.216	PO BFF	0
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63/20/60/722/000/100 MGB - Actuator attachment fitting axis POST MOD 0763C88 Check the tightening torque load of the servo-control attachment bolts.	AMM 63.20.60.722	PO 5 FH	10 FH
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63/20/60/722/000/110 MGB - Actuator attachment fitting axis POST MOD 0763C88 Check the tightening torque load of the servo-control attachment bolts.	AMM 63.20.60.722	PO 75 FH	35 FH
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63/20/00/211/000/005 MGB PRE MOD 0763C88 Inspection.	AMM 63.20.00.211	100 FH // 1 Y	10 FH 36 D
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Task Number Description/Remarks	Documentation	Interval	Margin
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63/20/00/221/000/015 MGB PRE MOD 0763C88 Detailed inspection.	AMM 63.20.00.221 GRT 03	600 FH	60 FH
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63/20/00/221/000/025 Attachment bolt MGB / servo-control 360A32-1163-03 (-) PRE MOD 0763C88 Check with removal of the bolt equipped with fitting above.	AMM 63.20.00.222	4 Y	146 D
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63/20/00/611/000/015 MGB - Mineral oil PRE MOD 0763C88 Drain.	AMM 63.20.00.611	400 FH // 2 Y	40 FH 73 D
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63/20/60/211/000/000 MGB POST MOD 0763C88 Search for seepage or leakage.	AMM 63.20.60.211 63.20.60.211.001	100 FH	10 FH
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63/20/60/211/000/010 MGB - Link between housings POST MOD 0763C88 Visual inspection.	AMM 63.20.60.211 63.20.60.211.002	100 FH	10 FH
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63/20/60/211/000/090 MGB POST MOD 0763C95 Search for seepage or leakage at left hydraulic pump support interfaces. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.25.60.211 63.25.60.211.001	600 FH	60 FH
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Task Number Description/Remarks	Documentation	Interval	Margin
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<p>63/20/60/211/000/100</p> <p>MGB</p> <p>POST MOD 0763C95</p> <p>Search for seepage or leakage at right hydraulic pump support interfaces. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.</p>	<p><i>AMM 63.25.60.211</i> <i>63.25.60.211.002</i></p>	<p>600 FH</p>	<p>60 FH</p>
<p>63/20/60/215/000/005</p> <p>MGB</p> <p>POST MOD 0763C88</p> <p>Search for seepage or leakage on the main module. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.</p>	<p><i>AMM 63.20.60.215</i> <i>63.20.60.215.001</i></p>	<p>600 FH</p>	<p>60 FH</p>
<p>63/20/60/215/000/010</p> <p>MGB</p> <p>POST MOD 0763C88</p> <p>Search for seepage or leakage between main module and rear output. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.</p>	<p><i>AMM 63.20.60.215</i> <i>63.20.60.215.002</i></p>	<p>600 FH</p>	<p>60 FH</p>
<p>63/20/60/215/000/020</p> <p>MGB</p> <p>POST MOD 0763C88</p> <p>Search for seepage or leakage between main module and epicyclic module. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.</p>	<p><i>AMM 63.20.60.215</i> <i>63.20.60.215.003</i></p>	<p>600 FH</p>	<p>60 FH</p>
<p>63/20/60/215/000/030</p> <p>MGB</p> <p>POST MOD 0763C88</p> <p>Search for seepage or leakage between hydraulic pump support and input housing. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.</p>	<p><i>AMM 63.20.60.215</i> <i>63.20.60.215.004</i></p>	<p>600 FH</p>	<p>60 FH</p>
<p>63/20/60/215/000/060</p> <p>MGB</p> <p>POST MOD 0763C88</p> <p>Search for seepage or leakage on external elements. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.</p>	<p><i>AMM 63.20.60.215</i> <i>63.20.60.215.005</i></p>	<p>600 FH</p>	<p>60 FH</p>

Task Number Description/Remarks	Documentation	Interval	Margin
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63/20/60/215/000/070 MGB POST MOD 0763C88 Search for seepage or leakage between the main module and equipment. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.20.60.215 63.20.60.215.006	600 FH	60 FH
63/20/60/219/000/000 MGB POST MOD 0763C88 Visual check of oil level. Sampling every ALF according to SB Airbus Helicopters	AMM 63.20.60.219 63.20.60.219.006	// 10 FH 5 D	5 FH 2 D
63/20/60/219/000/020 MGB POST MOD 0763C88 Visual inspection.	AMM 63.20.60.219 63.20.60.219 .001	ALF	0
63/20/60/219/000/030 MGB - Plate POST MOD 0763C88 Visual inspection.	AMM 63.20.60.219 63.20.60.219.005	5 FH	10 FH
63/20/60/219/000/040 MGB - Main rotor actuator attachment POST MOD 0763C88 Visual inspection.	AMM 63.20.60.219 63.20.60.219.002	ALF	0
63/20/60/219/000/050 MGB - Fan POST MOD 0763C88 Check air inlet.	AMM 63.20.60.219 63.20.60.219.003	ALF	0

Task Number Description/Remarks	Documentation	Interval	Margin
63/20/60/221/000/010 MGB - Input flange POST MOD 0763C88 Check the tightening torque. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.20.60.221 63.20.60.221.001	600 FH	60 FH
63/20/60/221/000/040 MGB - input housing (Right and left) POST MOD 0763C88 Search for seepage or leakage on input module. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.20.60.221 63.20.60.221.002	600 FH	60 FH
63/20/60/221/000/050 MGB - Tail rotor output pinion assembly POST MOD 0763C88 Search for seepage or leakage on rear output module. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.20.60.221 63.20.60.221.003	600 FH	60 FH
63/20/60/221/000/080 MGB POST MOD 0763C88 Search for seepage or leakage at fan output. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.20.60.221 63.20.60.221.004	600 FH	60 FH
63/20/60/282/000/000 MGB POST MOD 0763C88 Oil spectral analysis.	AMM 63.20.60.282	100 FH // 3 M	10 FH 7 D
63/20/60/611/000/000 MGB - Mineral oil POST MOD 0763C88 Drain.	AMM 63.20.60.611	460 FH // 2 Y	40 FH 2 M

Task Number Description/Remarks	Documentation	Interval	Margin
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63/20/61/721/000/000 MGB - Rear output flange	<i>AMM 63.24.61.721</i>	600 FH	<i>60 FH</i>
POST MOD 0763C88 Check the tightening torque. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.			

63-21 OIL SYSTEM

63/21/02/221/000/006 MGB	<i>AMM 63.21.02.221</i>	PO 3 FH	<i>7 FH</i>
PRE MOD 0763C88 Each time after the component is installed. Check oil filter.			

63/21/02/221/000/001 Oil filter	<i>AMM 63.21.02.221</i>	100 FH // 1 Y	<i>10 FH</i> <i>36 D</i>
PRE MOD 0763C88 Inspection.			

63/21/62/221/000/000 MGB - Oil filter	<i>AMM 63.21.62.221</i>	600 FH // 1 Y	<i>60 FH</i> <i>30 D</i>
POST MOD 0763C88 Detailed inspection.			

63-22 MGB - Fan

63/22/60/221/000/000 MGB - Fan	<i>AMM 63.22.60.221</i> <i>63.22.60.221.001</i>	1200 FH	<i>120 FH</i>
POST MOD 0763C88 Check. Sampling every 600 FH (margin 60 FH) according to SB Airbus Helicopters.			

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>63/22/60/221/000/010 MGB - Fan POST MOD 0763C88 Check the tightening torque of the locking nut. Sampling every 600 FH (margin 60 FH) according to SB Airbus Helicopters.</p>	<p>AMM 63.22.60.221 63.22.60.221.002</p>	<p>1200 FH</p>	<p>120 FH</p>
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<p>63/22/60/221/000/020 MGB - Fan POST MOD 0763C88 Check the state of locking devices. Sampling every 600 FH (margin 60 FH) according to SB Airbus Helicopters.</p>	<p>AMM 63.22.60.221 63.22.60.221.003</p>	<p>1200 FH</p>	<p>120 FH</p>
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63-23 ROTOR BRAKE

<p>63/23/00/000/000/005 Rotor brake PRE MOD 0763C86 Check.</p>	<p>AMM 05.21.00.216</p>	<p>ALF</p>	<p>0</p>
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<p>63/23/00/211/000/005 Rotor brake PRE MOD 0763C86 Inspection.</p>	<p>AMM 63.23.00.211</p>	<p>100 FH // 1 Y</p>	<p>10 FH 36 D</p>
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<p>63/23/00/221/000/005 Rotor brake PRE MOD 0763C86 Detailed inspection.</p>	<p>AMM 63.23.00.221 GRT 03</p>	<p>600 FH</p>	<p>60 FH</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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63/23/00/221/000/006 Rotor brake PRE MOD 0763C86 Tropical and damp atmosphere Salt-laden atmosphere Detailed inspection.	AMM 63.23.00.221 GRT 03	600 FH // 2 Y	60 FH 73 D
63/23/60/221/000/000 Rotor brake POST MOD 0763C86 Detailed inspection. Sampling every 100 FH (margin 10 FH) according to SB Airbus Helicopters.	AMM 63.23.60.221 63.23.60.221.001	600 FH	60 FH
63/23/60/221/000/010 Rotor brake POST MOD 0763C86 Check the play between pads, Check the dimensions of the disk.	AMM 63.23.60.221 63.23.60.221.002	600 FH	60 FH

63-24 TAIL ROTOR DRIVE POWER TAKE-OFF

63/24/01/221/001/001 MGB tail drive-shaft flange PRE MOD 0763C81 // PRE MOD 0763C88 Each time after the component is installed. Check of the SHUR-LOK nut of the tail rotor drive-shaft.	AMM 63.24.01.221 63.24.01.221.001	PO 100 FH	10 FH
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Task Number Description/Remarks	Documentation	Interval	Margin
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<p>63/24/01/221/001/010</p> <p>MGB tail drive-shaft flange</p> <p>POST MOD 0763C81 // PRE MOD 0763C88</p> <p>Each time after the component is installed.</p> <p>Check the tail rotor drive-shaft flange play and reapply tightening torque of it's SHUR-LOK nut.</p>	<p>AMM 63.24.01.221 63.24.01.221.002</p>	<p>PO 100 FH</p>	<p>10 FH</p>
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<p>63/24/01/221/000/001</p> <p>MGB tail drive-shaft flange</p> <p>PRE MOD 0763C81 // PRE MOD 0763C88</p> <p>Check radial play of the flange.</p>	<p>AMM 63.24.01.221 63.24.01.221.001</p>	<p>300 FH</p>	<p>30 FH</p>
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<p>63/24/01/221/000/006</p> <p>MGB tail drive-shaft flange</p> <p>POST MOD 0763C81 // PRE MOD 0763C88</p> <p>Check axial and radial play of the flange.</p>	<p>AMM 63.24.01.221 63.24.01.281.001 63.24.01.281.003</p>	<p>300 FH</p>	<p>30 FH</p>
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<p>63/24/01/221/000/015</p> <p>MGB tail drive-shaft flange</p> <p>PRE MOD 0763C81 // PRE MOD 0763C88</p> <p>Check of the SHUR-LOK nut of the tail rotor drive-shaft flange.</p>	<p>AMM 63.24.01.221 63.24.01.221.001 GRT 03</p>	<p>600 FH</p>	<p>60 FH</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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63-30 MOUNTS AND ATTACHMENTS

63/30/00/211/000/010			
Support assembly			
	AMM 63.30.00.222 63.30.00.222.001 63.30.00.222.002	PO 3 FH	7 FH
Each time after the component is installed.			
Readjust the tightening torque loads:			
-MGB suspension bars,			
-attachment bolts of the fitting transmission deck,			
-bolts of the laminate pads.			

63/30/00/211/000/000			
Support assembly			
	AMM 63.30.00.211	100 FH // 1 Y	10 FH 36 D
Inspection.			

63/30/00/221/000/000			
Support assembly			
	AMM 63.30.00.221	1200 FH	120 FH
Detailed inspection with removal :			
- laminated pads,			
- MGB suspension bars,			
- attachment bolts.			

63-40 INDICATING

63/40/02/721/000/005			
MGB - Electrical magnetic plug			
	AMM 63.40.02.721 GRT 03	600 FH	60 FH
PRE MOD 0763C88			
Functional test.			

63/40/03/721/000/005			
Oil temperature warning system			
	AMM 63.40.03.721	6 Y	180 D
PRE MOD 0763C88			
Thermal switch functional test.			

Task Number Description/Remarks	Documentation	Interval	Margin
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63/40/61/221/000/000 MGB - Tachymeter generator POST MOD 0763C88 Check the state of fixation devices.	AMM 63.40.61.221 63.40.61.221.001	600 FH	60 FH
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63/40/61/221/000/010 MGB - Tachymeter generator POST MOD 0763C88 Check the drive condition.	AMM 63.40.61.221 63.40.61.221.002	600 FH	60 FH
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 64-TAIL ROTOR

64-10 ROTOR BLADES

64/10/00/211/001/000

Tail rotor blade

	AMM 64.10.00.211A	PO	5 FH	10 FH
	64.10.00.211.001	&	25 FH	5 FH
	64.10.00.211.002			
	AMM 64.20.00.222			
	64.20.00.222.001			
	AMM 64.20.00.224			

365A12-0060-01 (-)

After the introduction to service of a new, overhauled or repaired component.

Check the blades by tapping.

Check the safetying mark (on the blade horn).

Check the blade clearance.

Readjust the tightening torque load of the blade attachment bolt.

Blade on the helicopter.

TRH fairing removed from the helicopter.

64/10/00/211/001/003

Tail rotor blade

	AMM 64.10.00.211B	PO	5 FH	10 FH
	64.10.00.211.001	&	25 FH	5 FH
	64.10.00.211.002			
	AMM 64.20.00.222			
	64.20.00.222.001			
	AMM 64.20.00.224			

365A12-0070-00 (-)

365A12-0070-01 (-)

After the introduction to service of a new, overhauled or repaired component.

Check the blades by tapping.

Check the safetying mark (on the blade horn).

Check the blade clearance.

Readjust the tightening torque load of the blade attachment bolt.

Blade on the helicopter.

TRH fairing removed from the helicopter.

64/10/00/221/001/010

Tail rotor blade

	AMM 64.10.00.221A	PO	90 FH	10 FH
	64.10.00.221.001			
	64.10.00.221.008			

365A12-0060-01 (-)

After the introduction to service of a new, overhauled or repaired component.

Check the blades by tap testing.

Manually check that the crank pin is not loose.

Blade removed from the helicopter.

TRH fairing removed from the helicopter.

Task Number Description/Remarks	Documentation	Interval	Margin
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64/10/00/221/001/015 Tail rotor blade	AMM 64.10.00.221B 64.10.00.221.001 64.10.00.221.008	PO 90 FH	10 FH
365A12-0070-00 (-) 365A12-0070-01 (-) After the introduction to service of a new, overhauled or repaired component. Check the blades by tap testing. Manually check that the crank pin is not loose. Blade removed from the helicopter. TRH fairing removed from the helicopter.			
64/10/00/221/003/000 Tail rotor blade	AMM 64.10.00.211A 64.10.00.211.002 AMM 64.20.00.222 64.20.00.222.001	PO 5 FH	10 FH
365A12-0060-01 (-) After installation of a component removed from the same helicopter, After installation of a component originating from another helicopter. Check the safetying mark (on the blade horn). Readjust the tightening torque load of the blade attachment bolt. Blade on the helicopter. TRH fairing removed from the helicopter.			
64/10/00/221/003/005 Tail rotor blade	AMM 64.10.00.211B 64.10.00.211.002 AMM 64.20.00.222 64.20.00.222.001	PO 5 FH	10 FH
365A12-0070-00 (-) 365A12-0070-01 (-) After installation of a component removed from the same helicopter, After installation of a component originating from another helicopter. Check the safetying mark (on the blade horn). readjust the tightening torque load of the blade attachment bolt. Blad on the helicopter. TRH fairing removed from the helicopter.			

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>64/10/00/000/000/010</p> <p>Tail rotor blade - Protection against erosion</p> <p>365A12-0060-01 (-)</p> <p>365A12-0070-00 (-)</p> <p>365A12-0070-01 (-)</p> <p>POST OP61B33</p> <p>Replace polyurethane protective plates.</p>	<p>AMM 64.10.00.065</p>	<p>100 FH</p>	<p>10 FH</p>
<p>64/10/00/211/000/020</p> <p>Tail rotor blade</p> <p>365A12-0060-01 (-)</p> <p>Inspection. Check the safetying mark (on the blade horn). Blade on the helicopter. TRH fairing removed from the helicopter.</p>	<p>AMM 64.10.00.211A 64.10.00.211.002 64.10.00.211.004</p>	<p>100 FH</p>	<p>10 FH</p>
<p>64/10/00/211/000/025</p> <p>Tail rotor blade</p> <p>365A12-0070-00 (-)</p> <p>365A12-0070-01 (-)</p> <p>Visual inspection. Check the safetying mark (on the blade horn). Blade on the helicopter. TRH fairing removed from the helicopter.</p>	<p>AMM 64.10.00.211B 64.10.00.211.002 64.10.00.211.004</p>	<p>100 FH</p>	<p>10 FH</p>
<p>64/10/00/211/001/027</p> <p>Tail rotor blade</p> <p>Salt-laden atmosphere</p> <p>Clean blade. TRH fairing on the helicopter. Blade on the helicopter.</p>	<p>AMM 64.10.00.101</p>	<p>50 FH</p>	<p>5 FH</p>

Task Number Description/Remarks	Documentation	Interval	Margin
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64/10/00/221/001/020 Tail rotor blade 365A12-0060-01 (-) Sand-laden and/or dust-laden atmosphere Visual check for erosion. Blade on the helicopter. TRH fairing on the helicopter.	AMM 64.10.00.211A 64.10.00.211.003	15 FH	1 FH
64/10/00/221/001/025 Tail rotor blade 365A12-0070-00 (-) 365A12-0070-01 (-) Sand-laden and/or dust-laden atmosphere Visual check for erosion. Blade on the helicopter. TRH fairing on the helicopter.	AMM 64.10.00.211B 64.10.00.211.003	15 FH	1 FH
64/10/00/221/001/030 Tail rotor blade 365A12-0060-01 (-) Inspection. Blade removed from the helicopter. TRH fairing removed from the helicopter.	AMM 64.10.00.221A GRT 02	600 FH	60 FH
64/10/00/221/001/035 Tail rotor blade 365A12-0060-01 (-) Inspection and tapping test. Blade on the helicopter. TRH fairing removed from the helicopter.	AMM 64.10.00.211A 64.10.00.211.001	2 Y	73 D
64/10/00/221/001/040 Tail rotor blade 365A12-0070-00 (-) 365A12-0070-01 (-) Visual inspection. Blade removed from the helicopter. TRH fairing removed from the helicopter.	AMM 64.10.00.221B	1200 FH	120 FH

Task Number Description/Remarks	Documentation	Interval	Margin
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64/10/00/221/001/045 Tail rotor blade	AMM 64.10.00.211B 64.10.00.211.001	2 Y	73 D
365A12-0070-00 (-)			
365A12-0070-01 (-)			
Inspection and tapping test. Blade on the helicopter. TRH fairing removed from the helicopter.			

64-20 TAIL ROTOR HEAD

64/20/00/222/000/000 Tail rotor head	AMM 64.20.00.222 64.20.00.222.003	PO 5 FH	10 FH
Each time after the component is installed. Readjust the tightening torque load on the control shaft.			

64/20/00/211/000/000 Tail rotor head	AMM 64.20.00.211 AMM 64.20.00.224	100 FH	10 FH
Inspection. Check the blade clearance.			

64/20/00/225/000/000 Tail rotor head	AMM 64.20.00.225 64.20.00.225.001 64.20.00.225.002 64.20.00.225.003 64.20.00.225.004 64.20.00.225.005 64.20.00.225.006 64.20.00.225.007 64.20.00.225.009 64.20.00.225.010 64.20.00.225.011	1200 FH	120 FH
Detailed inspection.			

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 65-TAIL ROTOR DRIVE

65-10 TAIL ROTOR DRIVE SHAFT

65/10/00/000/000/015

Bearing

	AMM 65.10.00.211 65.10.00.211.003	PO	3 FH	7 FH
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Check each time after you grease it.

65/10/00/211/003/000

Tubes and bearings

	AMM 65.10.00.211 65.10.00.211.002 65.10.00.211.003		50 FH	5 FH
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Check the tubes, bearings and the sleeves.

65/10/00/222/000/000

Tail rotor drive shaft

	AMM 65.10.00.222 GRT 02		600 FH	60 FH
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Detailed inspection.

65/10/00/520/225/100

Bearing mounts of the tail rotor drive system

	AMM 65.10.00.225	//	8 Y 14 Y	180 D 180 D
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Tropical and damp atmosphere
Salt-laden atmosphere
Sand-laden and/or dust-laden atmosphere
Operation to be performed every 8 Y if use in climatic conditions specified above.
Operation to be performed every 14 Y if use in other climatic conditions.
Detailed inspection.

65/10/00/640/000/000

Tail rotor drive shaft

	AMM 65.10.00.641 GRT 02	//	600 FH 6 M	60 FH 18 D
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Lubricate the bearings.

65-20 TAIL GEARBOX

65/20/01/611/000/000

TGB

	AMM 65.20.01.611	PO	45 FH	5 FH
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After the introduction to service of a new, overhauled or repaired component.
Drain.

Task Number Description/Remarks	Documentation	Interval	Margin
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65/20/02/061/000/000

TGB

AMM 65.20.02.061 PO 3 FH 7 FH
65.20.02.421.001

Each time after the component is installed.
Check the tightening torque load on the attachment bolts.

65/20/00/211/000/000

TGB

AMM 65.20.00.211 // 100 FH 10 FH
1 Y 36 D

Inspection.

65/20/00/221/000/000

TGB

AMM 65.20.00.221 // 1200 FH 120 FH
65.20.00.221.002 // 4 Y 146 D

Check for absence of corrosion.

65/20/00/221/000/010

TGB

AMM 65.20.00.221 600 FH 60 FH
65.20.00.221.001
GRT 02

Detailed inspection.

65/20/01/611/000/010

TGB - Mineral oil

AMM 65.20.01.611 // 400 FH 40 FH
2 Y 73 D

Drain.

65-40 TAIL GEARBOX - INDICATING

65/40/01/211/000/000

TGB - Electrical magnetic plug

AMM 12.20.00.211 // 100 FH 10 FH
1 Y 36 D

Inspection.

65/40/01/721/000/000

TGB - Electrical magnetic plug

AMM 65.40.01.721 600 FH 60 FH
GRT 02

Functional test.

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 67-ROTORS FLIGHT CONTROL

67-00 ROTOR CONTROLS

67/00/00/221/000/000 Rotor controls	AMM 67.00.00.221 GRT 06	2 Y	73 D
Tropical and damp atmosphere Salt-laden atmosphere Inspection.			
67/00/00/520/222/100 Flight control systems	AMM 67.00.00.212 AMM 67.00.00.222	7200 FH // 10 Y // 14 Y	300 FH 180 D 180 D
Tropical and damp atmosphere Salt-laden atmosphere Sand-laden and/or dust-laden atmosphere Operation to be performed every 7200 FH // 10 Y if use in climatic conditions specified above. Operation to be performed every 7200 FH // 14 Y if use in other climatic conditions. Inspection.			

67-10 MAIN ROTOR CONTROL

67/10/00/221/000/025 Main rotor controls	AMM 67.10.00.223	PO 12 Y	180 D
PRE MOD 0767B43 Do a magnetism check on the cyclic sticks and collective levers.			

67/10/00/000/000/000 Main rotor controls	AMM 67.10.00.621	100 FH // 1 Y	10 FH 36 D
Tropical and damp atmosphere Salt-laden atmosphere Apply protection against corrosion.			
67/10/00/221/000/010 Main rotor controls	AMM 67.10.00.221 GRT 06	600 FH // 2 Y	60 FH 73 D
Inspection.			

Task Number Description/Remarks	Documentation	Interval	Margin
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67/10/00/224/000/000 Ball control Tropical and damp atmosphere Check.	AMM 67.10.00.224	6 M	18 D
67/10/00/224/000/003 Ball control Salt-laden atmosphere Check.	AMM 67.10.00.224	1 Y	36 D
67/10/00/224/000/005 Ball control Check.	AMM 67.10.00.224	2 Y	73 D

67-12 COLLECTIVE PITCH CONTROL

67/12/01/822/000/000 Collective lever Locking system check.	AMM 67.12.01.822 GRT 06	600 FH	60 FH
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67-20 TAIL ROTOR CONTROLS

67/20/00/000/000/000 Tail rotor controls Tropical and damp atmosphere Salt-laden atmosphere Apply protection against corrosion.	AMM 67.20.00.621	// 100 FH // 1 Y	10 FH 36 D
67/20/00/221/000/000 Tail rotor controls Inspection.	AMM 67.20.00.221 GRT 06	// 600 FH // 2 Y	60 FH 73 D

Task Number Description/Remarks	Documentation	Interval	Margin
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67/20/00/222/000/000 Ball control Tropical and damp atmosphere Check.	AMM 67.20.00.222	6 M	18 D
67/20/00/222/000/003 Ball control Salt-laden atmosphere Check.	AMM 67.20.00.222	1 Y	36 D
67/20/00/222/000/005 Ball control Check.	AMM 67.20.00.222	2 Y	73 D

67-30 SERVO-CONTROLS

67/30/01/221/000/010 Main rotor servo-control Each time after the component is installed. Readjust the tightening torque loads of the bolts.	AMM 67.30.01.061 MTC 20.02.05.404	PO 3 FH	7 FH
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67/30/00/000/000/006 Tail rotor servo-control - Input lever Check.	AMM 67.30.02.222 GRT 06	600 FH	60 FH
67/30/00/221/000/000 Servo-controls Inspection.	AMM 67.30.00.221	1200 FH	120 FH
67/30/01/221/000/000 Main rotor servo-control Check the tightening torque loads of the bolts.	AMM 67.30.01.061 GRT 06 MTC 20.02.05.404	// 600 FH 2 Y	60 FH 73 D

Task Number Description/Remarks	Documentation	Interval	Margin
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67/30/01/721/000/000 Hydraulic power assistance Test of the hydraulic power assistance given by the forward right servo control without hydraulic power.	AMM 67.30.01.721 GRT 06	600 FH	60 FH
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67/30/02/000/000/000 Servo-control - Test POST MOD 0739C30 & PRE MOD 0739C37 Cold weather Tests of servo-controls and of tail servo-control isolation.	AMM 67.30.00.711	// 15 FH 7 D	1 FH 0
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 71-POWER PLANT

71-00 POWER PLANT

<p>71/00/00/211/000/010 Engine bay fan</p> <p>Check.</p>	<p>AMM 71.00.00.211 71.00.00.211.002</p>	<p>50 FH</p>	<p>5 FH</p>
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<p>71/00/00/520/000/100 Power plant installation</p> <p>Detailed inspection.</p>	<p>AMM 71.00.00.211 GRT 04</p>	<p>600 FH // 2 Y</p>	<p>60 FH 73 D</p>
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71-20 ENGINE MOUNT

<p>71/20/00/520/221/100 Engine mounts</p> <p>Detailed inspection.</p>	<p>AMM 71.20.00.221</p>	<p>6000 FH</p>	<p>300 FH</p>
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71-30 FIRE WALL

<p>71/30/00/520/221/100 Fire walls</p> <p>Detailed inspection.</p>	<p>AMM 71.30.00.221</p>	<p>7200 FH</p>	<p>300 FH</p>
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71-60 AIR INTAKES

<p>71/60/20/221/000/000 Engine flushing system</p> <p>Detailed inspection.</p>	<p>AMM 71.60.20.221 GRT 04</p>	<p>600 FH</p>	<p>60 FH</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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71-61 SAND FILTER

71/61/01/221/000/000 Sand filter Inspection.	<i>AMM 71.61.01.221</i> <i>GRT 04</i>	<i>600 FH</i>	<i>60 FH</i>
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 73-ENG. FUEL & CONTROL

73-20 CONTROLLING

<p>73/20/00/761/000/000 FADEC unit Test the overspeed protection system.</p>	<p><i>AMM 73.20.00.761</i> <i>GRT 04</i></p>	<p>600 FH</p>	<p><i>60 FH</i></p>
<p>73/20/00/761/000/005 FADEC unit Check that "FADEC FAIL" message is displayed.</p>	<p><i>AMM 73.20.00.762</i> <i>73.20.00.762.002</i></p>	<p>100 FH // 1 Y</p>	<p><i>10 FH</i> <i>36 D</i></p>

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 77-ENGINE INDICATING

77-00 ENGINE INDICATING

77/00/00/520/000/100 Emergency control operation "EBCAU" Check.	<i>FLM Section 8.3</i> <i>§ 1.3</i> <i>GRT 04</i>	600 FH	<i>60 FH</i>
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77-30 ENGINE MONITORING - ANALYSER

77/30/00/721/000/000 Fuzz burner Functional test.	<i>AMM 77.30.00.721</i> <i>GRT 03</i>	600 FH <i>// 2 Y</i>	<i>60 FH</i> <i>73 D</i>
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 79-ENGINE OIL

79-00 OIL

<p>79/00/00/211/000/005</p> <p>Engine</p> <p>Each time after the component is installed.</p> <p>Check the magnetic plug.</p>	<p>AMM 79.00.00.211 79.00.00.211.002</p>	<p>PO 3 FH</p>	<p>7 FH</p>
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<p>79/00/00/211/000/002</p> <p>Engine / MGB oil cooling system</p> <p>PRE MOD 0779B33</p> <p>Visual check.</p> <p>For the left engine return line: Refer to criteria given in the "Replacement Criteria" column of the AMM Task 79.00.00.991.</p> <p>Criterion 1: Replace the line at the next type T inspection.</p> <p>Criterion 2: Replace the line w/within the next 100FH.</p>		<p>100 FH</p>	<p>10 FH</p>
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<p>79/00/00/211/000/003</p> <p>Engine / MGB oil cooling system</p> <p>PRE MOD 0750B32 // PRE MOD 0771C26</p> <p>Visual check.</p> <p>For the right engine return line: Refer to criteria given in the "Replacement Criteria" column of the AMM Task 79-00-00-991.</p> <p>Criterion 1: Replace the line at the next type T inspection.</p> <p>Criterion 2: Replace the line within the next 100 FH.</p>		<p>100 FH</p>	<p>10 FH</p>
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<p>79/00/00/211/000/020</p> <p>External oil system</p> <p>Inspection.</p>	<p>AMM 79.00.00.211B</p>	<p>6 Y</p>	<p>180 D</p>
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79-20 DISTRIBUTION

<p>79/20/00/721/000/005</p> <p>Electrical type magnetic plug of the engine oil tank</p> <p>Functional test.</p>	<p>AMM 79.20.00.721 GRT 04</p>	<p>600 FH</p>	<p>60 FH</p>
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Task Number Description/Remarks	Documentation	Interval	Margin
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79-30 INDICATING

<p>79/30/00/721/000/000 Engine oil thermostatic switch Functional check.</p>	<p><i>AMM 79.30.00.721</i></p>	<p>6000 FH // 6 Y</p>	<p>300 FH 180 D</p>
<p>79/30/00/722/000/000 Engine oil temperature probe Inspection and test after removal.</p>	<p><i>AMM 79.30.00.722</i></p>	<p>6 Y</p>	<p>180 D</p>

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 80-ENGINE STARTING

80-10 CRANKING

<p>80/10/00/211/000/000 Starter generator 524-031 (704A46101011) System inspection.</p>	<p>AMM 80.10.00.211 80.10.00.211.001 GRT 04</p>	<p>600 FH</p>	<p>50 FH</p>
<p>80/10/00/221/000/000 Starter generator 524-031 (704A46101011) Check the brushes with removal of the starter generator.</p>	<p>AMM 80.10.00.211 80.10.00.211.002 GRT 04</p>	<p>600 FH</p>	<p>50 FH</p>
<p>80/10/01/221/002/000 Starter generator 524-031 (704A46101011) Check the radial play of bearings.</p>	<p>AMM 80.10.01.212 GRT 04</p>	<p>600 FH</p>	<p>50 FH</p>

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 88-HARNESS

88-00 WIRING HARNESSES

88/00/00/211/000/000			
Connection and wiring	<i>AMM 88.00.00.211</i> <i>GRT 04</i>	<i>600 FH</i> <i>// 2 Y</i>	<i>60 FH</i> <i>73 D</i>
Inspection.			

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 93-MONITORING SYSTEM

93-51 FLIR

<p>93/51/01/221/000/000 FLIR installation</p> <p>WESCAM. Detailed inspection.</p>	<p><i>AMM 93.51.01.221</i> <i>GRT 05</i></p>	<p>600 FH</p>	<p><i>60 FH</i></p>
<p>93/51/02/211/000/000 FLIR installation</p> <p>U7000. Inspection.</p>	<p><i>AMM 93.51.02.211</i></p>	<p>100 FH // 1 Y</p>	<p><i>10 FH</i> <i>36 D</i></p>
<p>93/51/02/721/000/000 FLIR installation</p> <p>U7000. Functional tests.</p>	<p><i>AMM 93.51.02.721</i> <i>93.51.02.721.001</i></p>	<p>100 FH // 1 Y</p>	<p><i>10 FH</i> <i>36 D</i></p>
<p>93/51/04/520/000/100 FLIR installation</p> <p>FLIR UF350EP. Clean the turret and the lenses. Check the mechanism of the turret.</p>	<p><i>AMM 93.51.04.221</i> <i>93.51.04.221.002</i> <i>93.51.04.221.003</i></p>	<p>100 FH</p>	<p><i>10 FH</i></p>

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 53-FUSELAGE**53-54 ENGINE COWLING**

53/54/00/211/000/010

Aft removable cowling - Heat shields equipped or MI15 protection*AMM 53.54.00.211*

Check to be carried out after the last flight of the day until the replacement criteria are reached.

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 62-ROTOR (S)

62-20 MAIN ROTOR HEAD

<p>62/20/00/000/000/080</p> <p>Lower attach beam - Adapter plastic buches</p> <p align="center"><i>AMM 62.24.05.991</i> <i>62.24.05.991.002</i></p> <p>Check to be carried out every 15 FH until the replacement criteria are reached.</p>			
<p>62/20/00/000/000/102</p> <p>Upper attach beam - Adapter plastic buches</p> <p align="center"><i>AMM 62.24.05.991</i> <i>62.24.05.991.001</i></p> <p>Check to be carried out every 15 FH until the replacement criteria are reached.</p>			
<p>62/20/00/000/000/197</p> <p>Blade horn - Stop ring</p> <p align="center"><i>AMM 62.24.03.991</i> <i>62.24.03.991.002</i></p> <p>Check to be carried out every 15 FH until the replacement criteria are reached.</p>			
<p>62/20/00/000/000/325</p> <p>Pitch change rod - Swivel bearing</p> <p align="center"><i>AMM 62.26.01.991</i> <i>62.26.01.991.002</i></p> <p>Clearance to be checked every 15 FH until the replacement criteria are reached.</p>			

62-40 INDICATING

<p>62/40/03/211/000/015</p> <p>Rotor mast - Electrical magnetic plug</p> <p align="center"><i>AMM 12.20.00.211</i></p> <p>Check to be performed after each flight for 25 FH.</p> <ul style="list-style-type: none"> - After 25 FH if the criteria remain within tolerances, resume the normal checking cycle. - If the criteria are outside the tolerance limits, carry out the procedure given in AMM Task 05.50.02.211. 			
<p>62/40/03/211/000/017</p> <p>Rotor mast - Electrical magnetic plug</p> <p align="center"><i>AMM 12.20.00.211</i> <i>AMM 63.21.02.221</i></p> <p>After discovering particles on the rotor mast or MGB electrical magnetic plug, check filter every 5 FH for 25 FH.</p> <ul style="list-style-type: none"> - After 25 FH, if the criteria remain within the tolerance, resume the normal checking cycle. - If the criteria are outside the tolerance limits, carry out the procedure given in AMM Task 05.50.02.211. 			

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 63-ROTOR DRIVE (S)

63-20 MAIN GEARBOX

<p>63/20/00/000/000/017 MGB</p> <p style="text-align: center;"><i>AMM 05.50.09.223</i></p> <p>POST MOD 0763C88 Check in case of lightning on the main rotor or on the aircraft.</p>			
<p>63/20/00/000/000/025 MGB</p> <p style="text-align: center;"><i>AMM 05.50.00.224</i></p> <p>POST MOD 0763C88 Check in case of inflight collision (bird strike) on the main rotor or on the aircraft.</p>			
<p>63/20/00/000/000/035 MGB</p> <p style="text-align: center;"><i>AMM 05.50.05.221</i></p> <p>POST MOD 0763C88 Check after hard landing.</p>			
<p>63/20/00/000/000/040 MGB</p> <p style="text-align: center;"><i>AMM 05.50.00.222</i></p> <p>POST MOD 0763C88 Check after vibrations, resonance and abnormal dynamic phenomenon.</p>			
<p>63/20/00/000/000/065 MGB</p> <p style="text-align: center;"><i>AMM 05.50.02.213</i></p> <p>POST MOD 0763C88 Check after a rotor overspeed.</p>			
<p>63/20/00/000/000/070 MGB</p> <p style="text-align: center;"><i>AMM 05.50.02.212</i></p> <p>POST MOD 0763C88 Check after overtorque.</p>			
<p>63/20/00/000/000/080 MGB</p> <p style="text-align: center;"><i>AMM 05.50.08.221</i></p> <p>POST MOD 0763C88 Check after freewheel jamming.</p>			

Task Number Description/Remarks	Documentation	Interval	Margin
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63/20/00/000/000/090 MGB - Chip detector POST MOD 0763C88 Check after detection of particles.	AMM 05.50.01.211		
63/20/00/000/000/100 MGB POST MOD 0763C88 Check after over temperature.	AMM 05.50.01.213		
63/20/00/000/000/110 MGB POST MOD 0763C88 Check after oil pressure caution.	AMM 05.50.01.214		
63/20/00/000/000/120 MGB POST MOD 0763C88 Check after loss of oil.	AMM 05.50.01.215		
63/20/00/000/000/130 MGB POST MOD 0763C88 Check after dynamic braking above 120 RPM.	AMM 05.50.02.221		

63-30 MOUNTS AND ATTACHMENTS

63/30/00/221/003/015 MGB suspension bars - Bars: swivel bearings. Clearance to be checked every 100 FH or 1 Y until the replacement criteria are reached.	AMM 63.30.00.991 63.30.00.991.002		
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63-40 INDICATING

Task Number Description/Remarks	Documentation	Interval	Margin
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<p>63/40/02/211/000/006</p> <p>MGB - Electrical magnetic plug</p> <p>PRE MOD 0763C88</p> <p>Check to be performed after each flight for 25 FH.</p> <ul style="list-style-type: none"> - After 25 FH if the criteria remain within tolerances, resume the normal checking cycle. - If the criteria are outside the tolerance limits, carry out the procedure given in AMM Task 05.50.01.211. 	<p><i>AMM 12.20.00.211</i></p>		
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<p>63/40/02/211/000/015</p> <p>MGB - Electrical magnetic plug</p> <p>PRE MOD 0763C88</p> <p>After discovering particles on the rotor mast or MGB electrical magnetic plug, check filter every 5 FH for 25 FH.</p> <ul style="list-style-type: none"> - After 25 FH, if the criteria remain within the tolerance, resume the normal checking cycle. - If the criteria are outside the tolerance limits, carry out the procedure given in AMM Task 05.50.01.211. 	<p><i>AMM 12.20.00.211</i></p> <p><i>AMM 63.21.02.221</i></p>		
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End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 64-TAIL ROTOR

64-10 ROTOR BLADES

<p>64/10/00/221/001/032</p> <p>Tail rotor blade</p> <p style="text-align: right;"><i>AMM 64.10.00.221A</i> <i>64.10.00.221.003</i> <i>64.10.00.221.004</i></p> <p>365A12-0060-01 (-)</p> <p>Inspection. With blade removal. With TRH fairing removal. Note 1: If there is separation of blade shielding in zone A2 and A5: inspection every 100 FH or 1 Y, as long as the replacement criteria are not met (without blade removal). Note 2: If there is separation of blade shielding in zone C and E: inspection every 50 FH, 100 FH, 150 FH and 300 FH (without blade removal). If no change is noticed, resume the regular inspection cycle.</p>			
<p>64/10/00/221/001/042</p> <p>Tail rotor blade</p> <p style="text-align: right;"><i>AMM 64.10.00.221B</i> <i>64.10.00.221.003</i> <i>64.10.00.221.004</i></p> <p>365A12-0070-00 (-) 365A12-0070-01 (-)</p> <p>Visual inspection. With blade removal. With TRH fairing removal. Note 1: If there is separation of blade shielding in zone A2 and A5: inspection every 100 FH or 1 Y, as long as the replacement criteria are not met (without blade removal). Note 2: If there is separation of blade shielding in zone C and E: inspection every 50 FH, 100 FH, 150 FH and 300 FH (without blade removal). If no change is noticed, resume the regular inspection cycle.</p>			

End of the Document Unit

Task Number Description/Remarks	Documentation	Interval	Margin
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ATA 65-TAIL ROTOR DRIVE

65-40 TAIL GEARBOX - INDICATING

65/40/01/211/000/005

TGB - Electrical magnetic plug

AMM 12.20.00.211

Check to be performed after each flight for 25 FH.

- After 25 FH, if the criteria remain within tolerances, resume the normal checking cycle.
- If the criteria are outside the tolerance limits, carry out the procedure given in AMM Task 05.50.01.211.

End of the Document Unit

