FEDERAL AVIATION ADMINISTRATION AIRWORTHINESS DIRECTIVES

SMALL AIRCRAFT, ROTORCRAFT, GLIDERS BALLOONS, AIRSHIPS, AND UAS

BIWEEKLY 2024-18

08/12/2024 - 08/25/2024



Federal Aviation Administration
Continued Operational Safety Policy Section, AIR-141
P.O. Box 25082
Oklahoma City, OK 73125-0460

	SMALL AIRCRAFT			
AD No.	Information	Manufacturer	Applicability	
	Info	ormation Key: E- Emergency; COR - Correction; R - Replac	res, A- Affects	
Biweekly 2024-01				
2023-26-03		WACO Classic Aircraft Corporation	2T-1A-2	
2024-01-52	Е	Hélicoptères Guimbal	CABRI G2	
Biweekly 2024-02				
2024-01-03	R 2023-01-07	GE Aviation Czech s.r.o.	H75-100, H75-200, H80, H80-100, H80-200, H85-100, H85-200	
2024-02-55	E	Bell Textron Canada Limited	505	
Biweekly 2024-03				
2024-01-11		Pacific Scientific Company Airbus Helicopters	Rotary Buckle Assembly	
2024-01-52	R 2023-24-51	Hélicoptères Guimbal	CABRI G2	
Biweekly 2024-04				
2024-02-01		Airbus Helicopters	EC225LP	
2024-02-04	R 2021-13-07	GE Aviation Czech s.r.o.	M601E-11, M601E-11A, M601E-11AS, M601E- 11S	
2024-04-51	Е	Pratt & Whitney Canada Corp.	PT6A-64, PT6A-66, PT6A-66A, PT6A-66B, PT6A-66D, PT6A-67, PT6A-67A, PT6A-67AF, PT6A-67AG, PT6A-67B, PT6A-67D, PT6A-67F, PT6A-67P, PT6A-67R, PT6A-67RM, PT6A-67T, PT6A-68, PT6A-68D, PT6E-66XT, PT6E-67XP	
Biweekly 2024-05 2024-02-55		Bell Textron Canada Limited	505	
2024-04-02		Robinson Helicopter Company	R22, R22 ALPHA, R22 BETA, R22 MARINER, R44, R44 II, R66	
2024-04-10		Airbus Helicopters Deutschland GmbH (AHD)	EC135P1, EC135P2, EC135P2+, EC135P3, EC135T1, EC135T2+/EC635T2+, EC135T3, EC635T2+, EC135T2	
2024-05-01		Austro Engine GmbH	E4, E4P	
2024-05-51	E	General Electric Company Delta Enterprise LLC Heliqwest International Inc. Pickering Aviation Inc. SIXTYHAWK TC LLC CAPITOL HELICOPTERS INC Central Copters Inc. Sikorsky Aircraft Corporation ACE Aeronautics LLC Billings Flying Service Inc. Blackhawk Mission Equipment Carson Helicopters Inc. High Performance Helicopters Corp.	CT7-2E1, CT7-2F1, CT7-8A, CT7-8E, CT7-8F5, EH-60A, HH-60L, S-70, S-70A, S-70C, S-70C(M), S-70C(M1), S-70M, UH-60A, CT7-8, CT7-2D, CT7-2D1	

	SMALL AIRCRAFT			
AD No.	Information	Manufacturer	Applicability	
	Info	ormation Key: E- Emergency; COR - Correction; R - Replaces Northwest Rotorcraft LLC PJ Helicopters Inc	CT7-2E1, CT7-2F1, CT7-8A, CT7-8E, CT7-8F5, EH-60A, HH-60L, S-70, S-70A, S-70C, S-70C(M),	
		Reeder Flying Service Inc. SKYDANCE BLACKHAWK OPERATIONS LLC Timberline Helicopters Inc. Unical Air Inc.	S-70C(M1), S-70M, UH-60A, CT7-8, CT7-2D, C CT7-2D1	
Biweekly 2024-06				
2024-03-05	A 2021-13-07 A 2022-13-16 A 2022-14-12 A2023-01-10	GE Aviation Czech s.r.o.	M601D-11, M601E-11, M601E-11A, M601E- 11AS, M601E-11S, M601F	
2024-04-01	112020 01 10	Airbus Helicopters Deutschland GmbH (AHD)	EC135P1, EC135P2, EC135P2+, EC135P3, EC135T1, EC135T2, EC135T2+, EC135T3, MBB-BK 117 C-2, MBB-BK 117 D-2, MBB-BK 117 D-3	
2024-04-05		Leonardo S.p.a.	AB412, AB412 EP	
2024-04-51		Pratt & Whitney Canada Corp.	PT6A-64, PT6A-66, PT6A-66A, PT6A-66B, PT6A-66D, PT6A-67, PT6A-67A, PT6A-67AF, PT6A-67AG, PT6A-67B, PT6A-67D, PT6A-67F, PT6A-67P, PT6A-67R, PT6A-67RM, PT6A-67T, PT6A-68D, PT6A-68, PT6E-67XP, PT6E-66XT	
2024-05-51		General Electric Company Delta Enterprise Heliqwest International Inc. Pickering Aviation Inc. SIXTYHAWK TC LLC CAPITOL HELICOPTERS INC Central Copters Inc. Sikorsky Aircraft Corporation ACE Aeronautics LLC Billings Flying Service Inc. Blackhawk Mission Equipment Carson Helicopters High Performance Helicopters Corp. Northwest Rotorcraft LLC PJ Helicopters Inc Reeder Flying Service Inc. SKYDANCE BLACKHAWK OPERATIONS LLC Timberline Helicopters Inc. Unical Air Inc.	CT7-2E1, CT7-2F1, CT7-8A, CT7-8E, CT7-8F5, EH-60A, HH-60L, S-70, S-70A, S-70C, S-70C(M), S-70C(M1), S-70M, UH-60A	
2024-06-51	E	General Electric Company Delta Enterprise Heliqwest International Inc. Pickering Aviation Inc. SIXTYHAWK TC LLC CAPITOL HELICOPTERS INC Central Copters Inc. Sikorsky Aircraft Corporation ACE Aeronautics LLC Billings Flying Service Inc. Blackhawk Mission Equipment Carson Helicopters High Performance Helicopters Corp. Northwest Rotorcraft LLC PJ Helicopters Inc Reeder Flying Service Inc.	CT7-2E1, CT7-2F1, CT7-8A, CT7-8E, CT7-8F5, EH-60A, HH-60L, S-70, S-70A, S-70C, S-70C(M), S-70C(M1), S-70M, UH-60A	

SMALL AIRCRAFT

AD No.	Information	Manufacturer	Applicability
L	Info	ormation Key: E- Emergency; COR - Correction; R - Re	eplaces, A- Affects
		SKYDANCE BLACKHAWK OPERATIONS Timberline Helicopters Inc. Unical Air Inc.	S LLC CT7-2E1, CT7-2F1, CT7-8A, CT7-8E, CT7-8F5, EH-60A, HH-60L, S-70, S-70A, S-70C, S-70C(M) S-70C(M1), S-70M, UH-60A
Biweekly 2024-07			
2024-06-02		GE Aviation Czech s.r.o.	M601D-11, M601E-11, M601E-11A, M601E- 11AS, M601E-11S, M601F
2024-07-51	Е	Bell Textron Canada Limited	429
Biweekly 2024-08			
2024-05-06		Leonardo S.p.a.	AW169
2024-05-07		Leonardo S.p.a.	AW189
2024-06-51	R 2024-05-51	General Electric Company	CT7-2E1, CT7-2F1, CT7-8A, CT7-8E, CT7-8F5
2024-07-03		Diamond Aircraft Industries Inc	DA 62
Biweekly 2024-09			
2024-06-13	R 2022-21-15	Diamond Aircraft Industries GmbH	DA 42, DA 42 NG, DA 42 M-NG
2024-07-01		Hamilton Sundstrand Corporation	14SF- 7, 14SF-15, 14SF-23
2024-07-07	R 2010-18-06	GA 8 Airvan (Pty) Ltd	GA8, GA8-TC320
2024-08-03		Britten-Norman Aircraft Ltd.	BN-2, BN-2A, BN-2A-2, BN-2A-3, BN-2A-6, BN-2A-8, BN-2A-9, BN-2A-20, BN-2A-21, BN-2A-26, BN-2A-27, BN-2B-20, BN-2B-21, BN-2B-26, BN-2B-27, BN-2T, BN-2T-4R, BN2T-4S, BN2A MK. III, BN2A MK. III-2, BN2A MK. III-3
2024-08-07	R 2023-12-17	Pilatus Aircraft Ltd.	PC-12, PC-12/45, PC-12/47, PC-12/47E
Biweekly 2024-10 No ADs			
Biweekly 2024-11			
2024-07-51		Bell Textron Canada Limited	429
2024-09-02		Leonardo S.p.a.	AW169
2024-10-04		Piper Aircraft Inc.	PA-28-181, PA-28R-201, PA-44-180, PA-34-220T (Seneca V)
Biweekly 2024-12			
2024-08-09		GA8 Airvan (Pty) Ltd	GA8, GA8-TC320
Discoolaler 2024-12			

Biweekly 2024-13

SMALL AIRCRAFT

AD No.	Information	Manufacturer Manufacturer	Applicability
	Info	rmation Key: E- Emergency; COR - Correction;	R - Replaces, A- Affects
2024-10-02		Leonardo S.p.a.	AW189
2024-10-10		Airbus Helicopters	SA-365N, SA-365N1, AS-365N2, AS-365N3
2024-13-03		Lindstrand Balloons Ltd.	42A, 56A, 60A, 69A, 77A, 90A, 105A, 120A, 150A, 180A, 210A, 240A, 260A, 310A, 69B, 77B, 90B, 105B, Drinks Can
Biweekly 2024-14			
2024-10-08		Leonardo S.p.a.	AW189
2024-10-13		Airbus Helicopters	AS332C, AS332C1, AS332L, AS332L1, AS332L2, EC225LP
Biweekly 2024-15			
2024-10-12		Bell Textron Canada Limited	407
2024-12-10		Centerpointe Aerospace Inc.	S-58BT, S-58DT, S-58ET, S-58FT, S-58HT, S-58JT
2024-14-03		Garmin Commander Aircraft Corporation DAHER AEROSPACE Mooney International Corporation Piper Aircraft Inc. Textron Aviation Inc.	GFC 500, 112B, 112TC, 112TCA, 114, 114A, 114B, 114TC, TB 20, TB 21, M20C, M20D, M20E, M20F, M20G, M20J, M20K, M20M, M20R, M20S, PA-24, PA-24-250, PA-24-260, PA-28-140, PA-28-150, PA-28-151, PA-28-160, PA-28-161, PA-28-180, PA-28-181, PA-28-201T, PA-28-235, PA-28-236, PA-28R-180, PA-28R-200, PA-28R-201T, PA-28R-201T, PA-28R-201T, PA-30, PA-39, PA-32-260, PA-32-300, PA-32-301, PA-32-301FT, PA-32-301T, PA-32-301XTC, PA-32R-300, PA-32RT-300, PA-32RT-300, PA-32RT-300T, PA-32R-301 (HP), PA-32R-301 (SP), PA-32R-301T, 19A, B19, M19A, A23A, A23-19, A23-24, B23, C23, A24, A24R, B24R, C24R, C35, D35, E35, F35, G35, 35-33, 35-A33, 35-B33, 35-C33, S5-C33A, 36, A36, A36TC, B36TC, E33, E33A, E33C, F33, F33A, F33C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 172D, 172E, 172F (USAF T-41A), 172G, 172H (USAF T-41A), 172I, 172K, 172L, 172M, 172N, 172P, 172Q, 172R, 172S, F172E, F172F, F172G, F172H, F172D, R172K, F172L, F172M, F172P, 172RG, P172D, R172K, FR172K, 177B, 177RG, F177RG, 182E, 182F, 182G, 182H, 182J, 182K, 182L, 182M, 182N, 182P, 182Q, 182R, 182S, 182T, F182P, F182Q, FR182, R182, T182, T182T, TR182, 206H, P206C, P206D, P206E, T206H, TP206C, TP206D, TP206E, TU206C, U206D, U206E, U206F, U206F, U206G, U206C, U206D, U206E, U206F, U206G, 210D, 210E, 210F, 210G, 210H, 210J, 210K, 210L, 210M, 210N, T210F, T210G, T210H, T210J, T210K, T210L, T210M, T210N

SMALL AIRCRAFT

AD No. Infe	nformation	Manufacturer	Applicability
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Information Key: E- Emergency; COR - Correction; R - Replaces, A- Affects

Biweekly 2024-16

No ADs

Biweekly 2024-17

2024-16-01 R 2000-18-09 Bell Textron Inc. 205A, 205A-1, 205B, 212, 412, 412CF, 412EP

2024-16-06 R 2023-15-07 Air Tractor Inc. AT-802, AT-802A

Biweekly 2024-18

2024-16-01 R 2000-18-09 Bell Textron Inc. 205A, 205A-1, 205B, 212, 412, 412CF, 412EP

2024-16-06 R 2023-15-07 Air Tractor Inc. AT-802, AT-802A

PART 39-AIRWORTHINESS DIRECTIVES

The authority citation for part 39 continues to read as follows:

[Amended]

The FAA amends §39.13 by:

Removing Airworthiness Directive 2000-18-09, Amendment 39-11894 (, September 13, 2000); and

Adding the following new airworthiness directive:

2024-16-01 Bell Textron Inc. (Type Certificate previously held by Bell Helicopter Textron, Inc.): Amendment 39-22807; Docket No. FAA-2024-2010; Project Identifier AD-2024-00366-R.

(a) Effective Date

This airworthiness directive (AD) is effective September 4, 2024.

(b) Affected ADs

This AD replaces AD 2000-18-09, Amendment 39-11894 (, September 13, 2000).

(c) Applicability

This AD applies to Bell Textron Inc. (type certificate previously held by Bell Helicopter Textron, Inc.), Model 205A, 205A-1, 205B, 212, 412, 412CF, and 412EP helicopters, certificated in any category, identified in paragraphs (c)(1) and (2) of this AD.

- (1) Model 205A, 205A-1, and 205B helicopters, with an upper left-hand cap angle (cap angle) part number (P/N) 205-030-207-005 installed.
- (2) Model 212, 412, 412CF, and 412EP helicopters, with a cap angle P/N 212-030-191-001 installed.

(d) Subject

Joint Aircraft System Component (JASC) Code: 5300, Fuselage structure.

(e) Unsafe Condition

This AD was prompted by reports of a fatigue crack in a tail boom attachment cap angle. The FAA is issuing this AD to prevent failure of a cap angle. The unsafe condition, if not addressed, could result in failure of the fuselage and bulkhead, and subsequent separation of the tail boom and loss control of the helicopter.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

- (1) For Model 205A, 205A-1, 205B, 212, 412CF, and 412 helicopters identified in paragraph (c) of this AD; and for Model 412EP helicopters identified in paragraph (c)(2) of this AD with serial numbers 36087 through 36693 inclusive, within 25 hours time-in-service (TIS) after the effective date of this AD, and thereafter at intervals not to exceed 25 hours TIS; and for Model 412EP helicopters identified in paragraph (c)(2) of this AD with serial numbers 36694 through 36999 inclusive and 37002 through 37999 inclusive, within 50 hour TIS after the effective date of this AD, and thereafter at intervals not to exceed 50 hours TIS, accomplish the actions required by paragraphs (g)(1)(i) and (ii) of this AD.
- (i) Visually inspect for the presence of sealant around the forward edge of the left-hand tail boom cap angle fitting in the "Inspection Area" depicted in Figure 1., of Bell Alert Service Bulletin (ASB) 205-24-122, Bell ASB 205B-24-77, Bell ASB 212-24-169, Bell ASB 412-24-197, or Bell ASB 412CF-24-77, all dated April 22, 2024, as applicable to your model helicopter. If there is any sealant, before further flight, remove any sealant by following the Accomplishment Instructions, paragraph 2., of Bell ASB 205-24-122, Bell ASB 205B-24-77, Bell ASB 212-24-169, Bell ASB 412-24-197, or Bell ASB 412CF-24-77, all dated April 22, 2024, as applicable to your model helicopter.
- (ii) Using a 10X or higher power magnifying glass and flashlight, visually inspect the cap angle and adjacent structure for a crack in the "Inspection Area" depicted in Figure 1., of Bell ASB 205-24-122, Bell ASB 205B-24-77, Bell ASB 212-24-169, Bell ASB 412-24-197, or Bell ASB 412CF-24-77, all dated April 22, 2024, as applicable to your model helicopter.
- (A) If there is any crack in the cap angle, before further flight, remove the cap angle from service and install an airworthy cap angle.
- (B) If there is any crack in the adjacent structure, before further flight, repair the structure in accordance with a method approved by the Manager, Central Certification Branch, FAA. For a repair method to be approved by the Manager, Central Certification Branch, FAA; as required by this paragraph, the Manager's approval letter must specifically refer to this AD.
- (2) If there is a crack as a result of any instance of an inspection required by paragraph (g)(1)(ii) of this AD, within 10 days after completing the inspection, report the information in Appendix 1 to this AD by email to .

(h) Special Flight Permit

Special flight permits may be issued in accordance with and to operate the helicopter to a location where the actions required by this AD can be performed, provided after takeoff, the flight is straight and level until landing, and avoids areas of known turbulence and provided no passengers are onboard.

(i) Alternative Methods of Compliance (AMOCs)

- (1) The Manager, Central Certification Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in . In accordance with , send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: .
- (2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(j) Related Information

For more information about this AD, contact Hung Nguyen, Aviation Safety Engineer, FAA, 1801 S Airport Road, Wichita, KS 67209; phone: (562) 627-5362; email: .

(k) Material Incorporated by Reference

- (1) The Director of the Federal Register approved the incorporation by reference (IBR) of the material listed in this paragraph under and .
- (2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.
- (i) Bell Alert Service Bulletin (ASB) 205-24-122, dated April 22, 2024.
- (ii) Bell ASB 205B-24-77, dated April 22, 2024.
- (iii) Bell ASB 212-24-169, dated April 22, 2024.
- (iv) Bell ASB 412-24-197, dated April 22, 2024.
- (v) Bell ASB 412CF-24-77, dated April 22, 2024.
- (3) For Bell material identified in this AD, contact Bell Textron, Inc., P.O. Box 482, Fort Worth, TX 76101; phone: (450) 437-2862 or (800) 363-8023; fax: (450) 433-0272; email: ; website: *bellflight.com/support/contact-support*.
- (4) You may view this material at the FAA, Office of Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222-5110.
- (5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit or email .

Appendix 1 to AD 2024-16-01

Report the following information for each crack by email to: .

In the subject line of the email, include the text "AD 2024-16-01".

- (1) Date of inspection that revealed a crack in the cap angle or in the adjacent structure.
- (2) Helicopter Model:
- (3) Date of previous inspection of the cap angle and adjacent structure:
- (4) Helicopter serial number:
- (5) Total hours time-in-service accumulated since new on the airframe:
- (6) Helicopter N-number:

- (7) Cap angle(s) part number:
- (8) Describe in detail any information and findings, including any previous maintenance or modification of the cracked area, any cracks in the surrounding areas such as the fitting or web, and, if possible, provide photos.

Issued on August 1, 2024.

Victor Wicklund,

Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[Filed 8-15-24; 11:15 am]

BILLING CODE 4910-13-P

PART 39-AIRWORTHINESS DIRECTIVES

The authority citation for part 39 continues to read as follows:

[Amended]

The FAA amends §39.13 by:

Removing Airworthiness Directive 2023-15-07, Amendment 39-22519 (, August 9, 2023); and

Adding the following new airworthiness directive:

2024-16-06 Air Tractor, Inc.: Amendment 39-22812; Docket No. FAA-2024-2013; Project Identifier AD-2024-00363-A.

(a) Effective Date

This airworthiness directive (AD) is effective September 4, 2024.

(b) Affected ADs

This AD replaces AD 2023-15-07, Amendment 39-22519 (, August 9, 2023) (AD 2023-15-07).

(c) Applicability

This AD applies to Air Tractor, Inc. Model AT-802 and AT-802A airplanes, all serial numbers, certificated in any category, that have Wipaire, Inc. Supplemental Type Certificate (STC) No. SA01795CH installed.

(d) Subject

Joint Aircraft System Component (JASC) Code 5510, Horizontal Stabilizer Structure; 5511 Horizontal stabilizer, Spar/Rib; 5514, Horizontal Stabilizer Miscellaneous Structure; 5530, Vertical Stabilizer Structure.

(e) Unsafe Condition

This AD was prompted by additional reports of cracks found in the horizontal stabilizer spars and the need to incorporate a new finlet attach design on the horizontal stabilizer spars to reduce the cracking. The FAA is issuing this AD to prevent structural failure of the horizontal stabilizer spars. The unsafe condition, if not addressed, could result in structural failure of the horizontal tail with consequent loss of control of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

- (1) Within 7 days after the effective date of this AD or before the airplane accumulates 110 hours time-inservice (TIS) after installation of STC No. SA01795CH, whichever occurs later, and thereafter at intervals not to exceed 110 hours TIS until the airplane is modified as required by paragraph (g)(3) of this AD: Inspect the left and right horizontal stabilizer spars for cracks in accordance with Steps 1 through 9 of the Work Instructions-Inspection, Method 1 in Wipaire, Inc. Service Letter 253, Revision D, dated July 3, 2024 (Wipaire SL 253D).
- (2) If any crack is found in a horizontal stabilizer spar during any inspection required by paragraph (g)(1) of this AD, or if any crack, elongated hole, or corrosion is found in a horizontal stabilizer spar during any inspection required by paragraph (g)(4) of this AD, before further flight, replace the horizontal stabilizer spar.
- (3) Within 300 hours TIS or 12 months after the effective date of this AD, whichever occurs first, install bathtub fittings (Service Kit 1012347-01 or 1012347-02) in accordance with Steps 1 through 10 of the Work Instructions-Install Bathtub Fittings in Wipaire SL 253D except where Step 2 specifies that to be eligible for reinstallation, finlet mount weldments must include the welded gussets shown in figure 8 of Wipaire SL 253D, that constraint is not required by this AD. If any spars were previously modified by installing 7D1-4399 Revision L or earlier, regardless of condition, those spars must be replaced at the same time the bathtub fittings kit is installed.
- (4) Within 110 hours TIS after installing the bathtub fittings, and thereafter at intervals not to exceed 110 hours TIS, inspect the horizontal stabilizer spars for cracks, elongated holes, and corrosion in accordance with Steps 1, 2, 4 through 6, 9, and 10 of the Work Instructions-Inspection, Method 2 in Wipaire SL 253D.
- (5) Within 5 days after each inspection required by paragraphs (g)(1) and (4) of this AD or within 5 days after the effective date of this AD, whichever occur later, report the following to the FAA at the address in paragraph (j)(1) of this AD. Report this information regardless of whether cracks are found.
- (i) Model, engine configuration (with horsepower limits), and propeller type;
- (ii) Serial number and N number;
- (iii) Total hours TIS on airframe;
- (iv) Total hours TIS operated with floats, if known;
- (v) STC configuration and total hours with STC installed;
- (vi) Crack location (right or left, upper/lower caps inboard/outboard hole);
- (vii) Crack size;
- (viii) Photos of cracks found, if available; and
- (ix) Any additional operator/mechanic comments.

(h) Credit for Previous Actions

You may take credit for the initial inspection required by paragraph (g)(1) of this AD if, before the effective date of this AD, you complied with Wipaire, Inc. Service Letter 253, Revision A, dated April 5, 2023; or Wipaire Service Letter 253, Revision B, dated July 27, 2023.

(i) Alternative Methods of Compliance (AMOCs)

- (1) The Manager, Central Certification Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in . In accordance with , send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the Certification Branch, send it to the attention of the person identified in paragraph (j)(1) of this AD and email to: .
- (2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(j) Related Information

- (1) For more information about this AD, contact Tim Eichor, Aviation Safety Engineer, Central Certification Branch, FAA, 1801 S. Airport Road, Wichita, KS 67209; phone: (847) 294-7141; email: .
- (2) Material identified in this AD that is not incorporated by reference is available at the address specified in paragraph (k)(3) of this AD.

(k) Material Incorporated by Reference

- (1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under and .
- (2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.
- (i) Wipaire, Inc. Service Letter 253, Revision D, dated July 3, 2024.
- (ii) [Reserved]
- (3) For Wipaire, Inc. material identified in this AD, contact Wipaire, Inc., 1700 Henry Avenue, Fleming Field (KSGS), South St. Paul, MN 55075; phone: (651) 451-1205; email: ; website: wipaire.com.
- (4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222-5110.
- (5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit or email.

Issued on August 14, 2024.

Victor Wicklund.

Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[Filed 8-15-24; 11:15 am]

BILLING CODE 4910-13-P