

## Attachment 4

CLIENT	OWNER: JABATAN BOMBA DAN		SERIAL NO.	HOURS	LDG/CYCLE	WORKSHEET	NO: 12762-001		
PENYELAMAT MALAYSIA AIRCRAFT TYPE, AW139 REGISTRATION: 9M-BOC BASE/FACILITY: MIRT DATE IN: WORKPACK OUT: WORKPACK		AIRCRAFT 31289		REFER W	ORKPACK	WORK/INSP/D	12102-001	200 FH INSPECTION	
		#1 ENGINE: #2 ENGINE:	PCE-KB0661 PCE-KB0659	REFER WORKPACK		WORKPACK REF: 9M-BOC-12762			
				REFER W	ORKPACK	AJL REF NO.:	REFER W	ORKPACK	
					NG/N1 NF/N2	SHEET:	1 OF	1	
200 FH IN JPDATE	n for raising: NSPECTION TO BE CARRIED OUT I.A.W. ID 2022-07-20 TO THE ALL-LATEST ISSUE MO/AMP/AW139 ISSUE 2 REV. 1, INSPECTION			28	Raised by and date: CAMO; Mohd Haffiz Ismail 14/09/2022	Other requirer	nents/information	1:	
Item		Description					* Eng. CRS	Date	
1	MIG4-05 TAIL ROTOR DAMPER (PART NUMBER 336420V00851) REFERENCE (DMC): 39-C-64-21-00-00A-31AB-A DETAILED INSPECTION OF ROD END FOR CRACKS IN THE ELASTOMER REMARKS: NOT APPLICABLE DUE TO DEFFERENT PART NUMBER					Technician	Ang. On the	-2.9.2	
2	CM30-01 ICE PROTECTION SYSTEM (FIPS) REFERENCE (DMC): 39-A-30-60-00-00A-340B-K OC TO VERIFY THE CORRECT FUNCTIONALITY OF THE ICE DETECTOR BOX 1 (IDB) FUNCTIONS REMARKS: NOT APPLICABLE DUE TO AC NOT EQUIPPED WITH FIPS					-	Ann .	22.9.2	
3	CM30-04 ICE PROTECTION SYSTEM (FIPS) REFERENCE (DMC): 39-A-30-60-00-00A-340B-K OC TO VERIFY THE CORRECT FUNCTIONALITY OF THE ICE DETECTOR BOX 2 (IDB) FUNCTIONS					-	Juin .	22.9.2	
	REMARKS: NOT APPLICABLE DUE TO AC NOT EQUIPPED WITH FIPS						ANOBO I		
y.	MI53-10 TAIL STRUCTURE ASSEMBLY (TAIL/REAR FUSELAGE ATTACHMENT FITTINGS) DETAILED INSPECTION FOR CRACKS (NOTE 13) NOTE 13: THE 200 FH INTERVAL IS EXTENDED TO 8350 LANDINGS FOR THE FOLLOWING CONFIGURATIONS: - TAIL ASSEMBLY PIN 3G5350A00136 NA - TAIL ASSEMBLY PIN 3G5350A00136 NA - TAIL ASSEMBLY PIN 3G5350A00136 NA - REAR FUSELAGE ASSEMBLY PIN 3P5340A00132 OR PIN 3G5340A01131 - REAR FUSELAGE ASSEMBLY PIN 3P5340A00132 OR PIN 3G5340A00133 PLUS RETROMODIFICATION PIN 3G5306P38811 (OPTIONAL BT 139-418) - TAIL ASSEMBLY PIN 3G5350A00134 OR PIN 3G5350A00133 OR PIN 3G5350A00132 PLUS THE FOLLOWING RETROMODIFICATIONS: 1 PIN 3G5306P02711 (PART I OF OPTIONAL BT 139-419) N/A 2 PIN 3G5306P02711 (PART I OF OPTIONAL BT 139-419) N/A DMC REFERENCE:39-A-53-40-00-00A-31AG-A REMARK: C / O FOUND SATIS PRCTORY					Alw]. 8320	Jun (1997)	22.9.2	
the air Certif and in	fies that the work specified above, excepter and a specified above, excepter and the work specified above, excepter to that work the aircraft / aircraft where APPLICABLE	I ready for relea of as otherwise s aft component i	se to service. pecified, was carr	ried out in a ly for releas	accordance with	a Requirements ar			

Technician column was found annotated with "-" which contradicted with MOE Part 2.13.3 Para 4.